

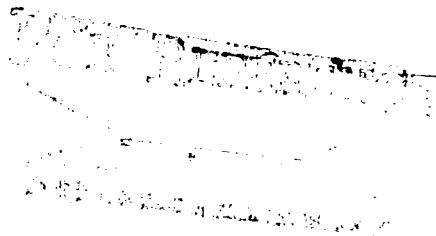
NASA Contractor Report 3918

NASA-CR-3918 19850025821

# Design and Testing of an Energy-Absorbing Crewseat for the F/FB-111 Aircraft

## *Volume III—Data From Seat Testing*

S. Joseph Shane



CONTRACT NAS1-17387  
AUGUST 1985

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HAMPTON, VIRGINIA

**NASA**



NASA Contractor Report 3918

# Design and Testing of an Energy-Absorbing Crewseat for the F/FB-111 Aircraft

*Volume III—Data From Seat Testing*

S. Joseph Shane  
*Simula Inc.*  
*Phoenix, Arizona*

Prepared for  
Langley Research Center  
under Contract NAS1-17387



National Aeronautics  
and Space Administration

Scientific and Technical  
Information Branch

1985



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## 1.0 INTRODUCTION

This volume presents data from the testing of an energy-absorbing crewseat in an F/FB-111 crew module. A complete description of the program is contained in Volume I: Final Report. Data from the dynamic test series on the seat is contained in Volume II.

Dynamic testing of the energy-absorbing seat and the operational F/FB-111 seat in an F/FB-111 crew module was conducted at NASA Langley Research Center. The testing is described in Section 4.0 of Volume I, and the resulting data are presented in Sections 2.0 through 14.0 of this volume.

Table 1 lists the data recorded. The four axes recorded on the seat pan include a measurement along the seat back tangent line as well as the three major axes. The seat deflection was measured along the seat back tangent line, but the dynamic response of the linear string potentiometers could not accurately show the seat stroke. Final seat stroke was measured and is recorded in Volume I.

DRI is presented using the same three acceleration measurements used in the seat testing: seat pan z-axis, seat pan along the back tangent line, and dummy pelvis z-axis.

The test conditions are listed at the beginning of each chapter. To be positive, the sign convention for the crew module orientation is for the following earth reference conditions:

- Pitch - nose up
- Roll - right wing down
- Yaw - nose right.

---

TABLE 1. RECORDED DATA FROM CREW MODULE TESTING

---

<u>Description of Data (Number of Channels)</u>	<u>Filter Class*</u>
Acceleration	
Left Bulkhead (3)	Class 60
Right Bulkhead (3)	Class 60
Left Seat Pan (4)	Class 180
Capsule Center of Gravity (C.G.)	Class 60
Left Dummy Pelvis	Class 180
Right Dummy Pelvis	Class 180
Right Dummy Chest	Class 180
Left Dummy Head	666 Hz
Right Dummy Head	666 Hz
Forces	
Left Seat Pan (3)	Class 60
Right Seat Pan (3)	Class 60
Exit Port Pressures (6)	Class 60
Seat Deflection (2)	Class 60
Dynamic Response Index (3)	-

---

\* Class 60 and Class 180 filters per SAE Recommended Practice J211b.

---



## 2.0 NASA TEST NO. 1

HORIZONTAL VELOCITY: 34 ft/sec

VERTICAL VELOCITY: 32 ft/sec

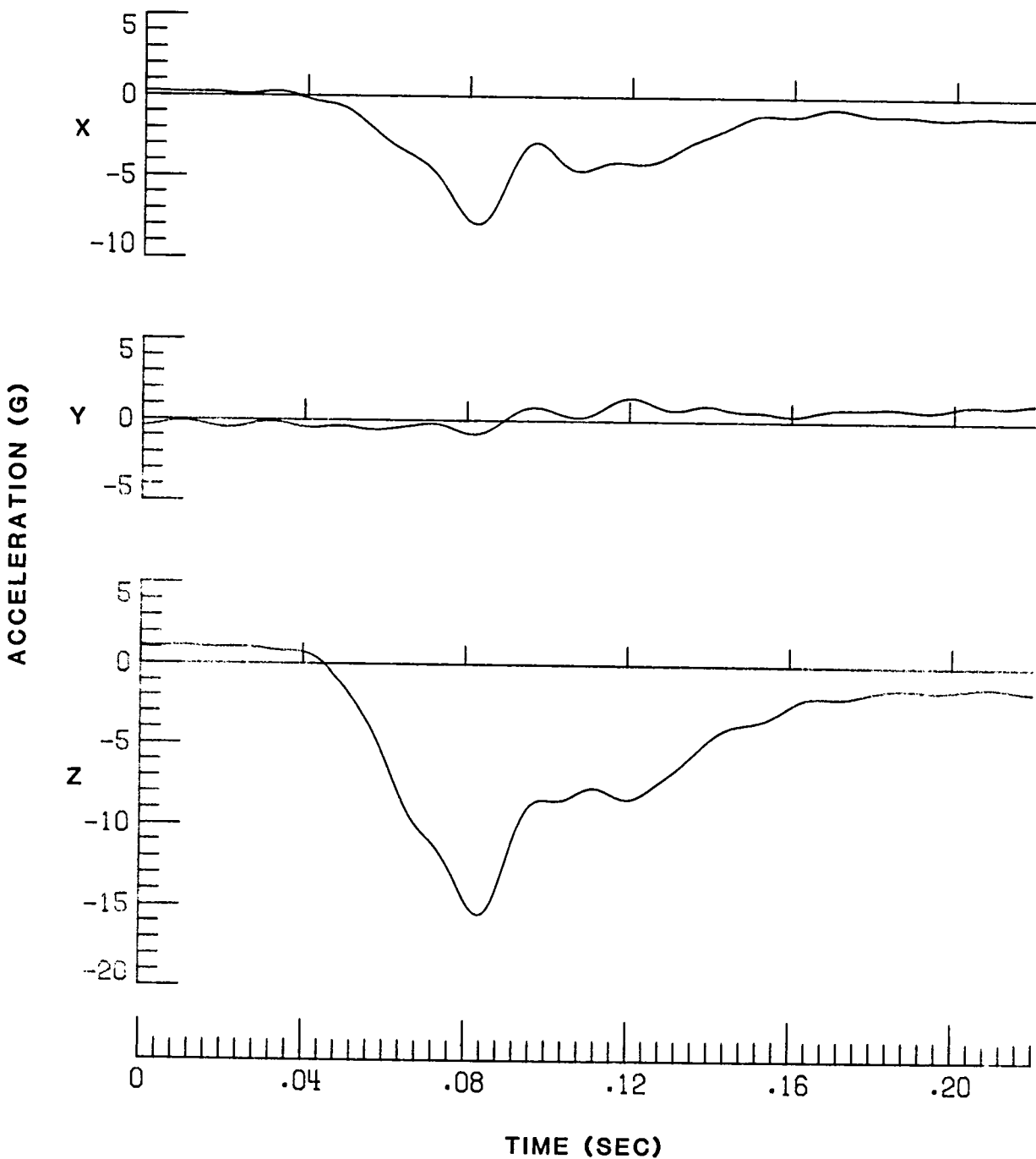
CREW MODULE PITCH:  $-13^{\circ}$

CREW MODULE ROLL:  $0^{\circ}$

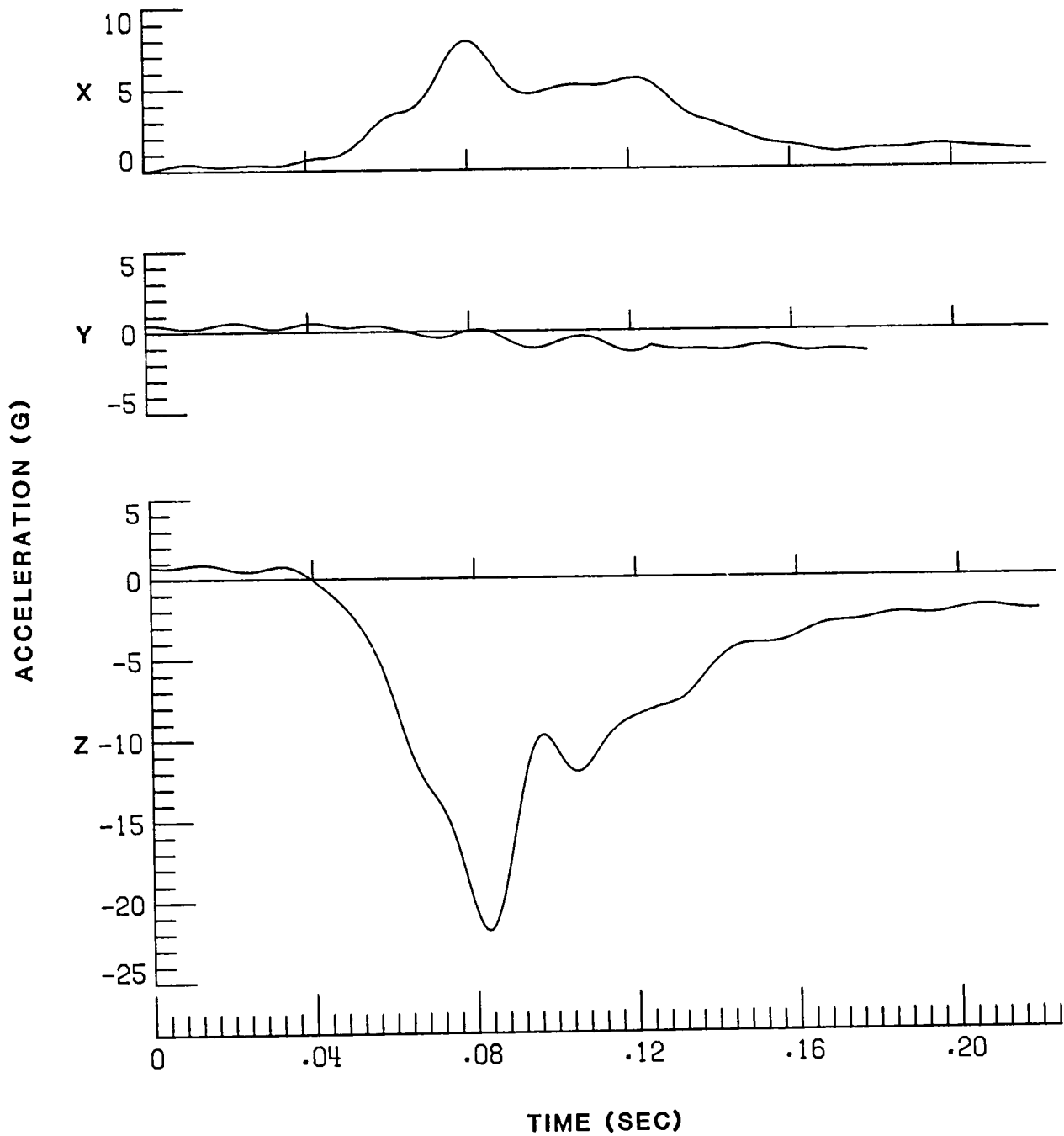
CREW MODULE YAW:  $45^{\circ}$

LEFT SEAT: Energy Absorbing

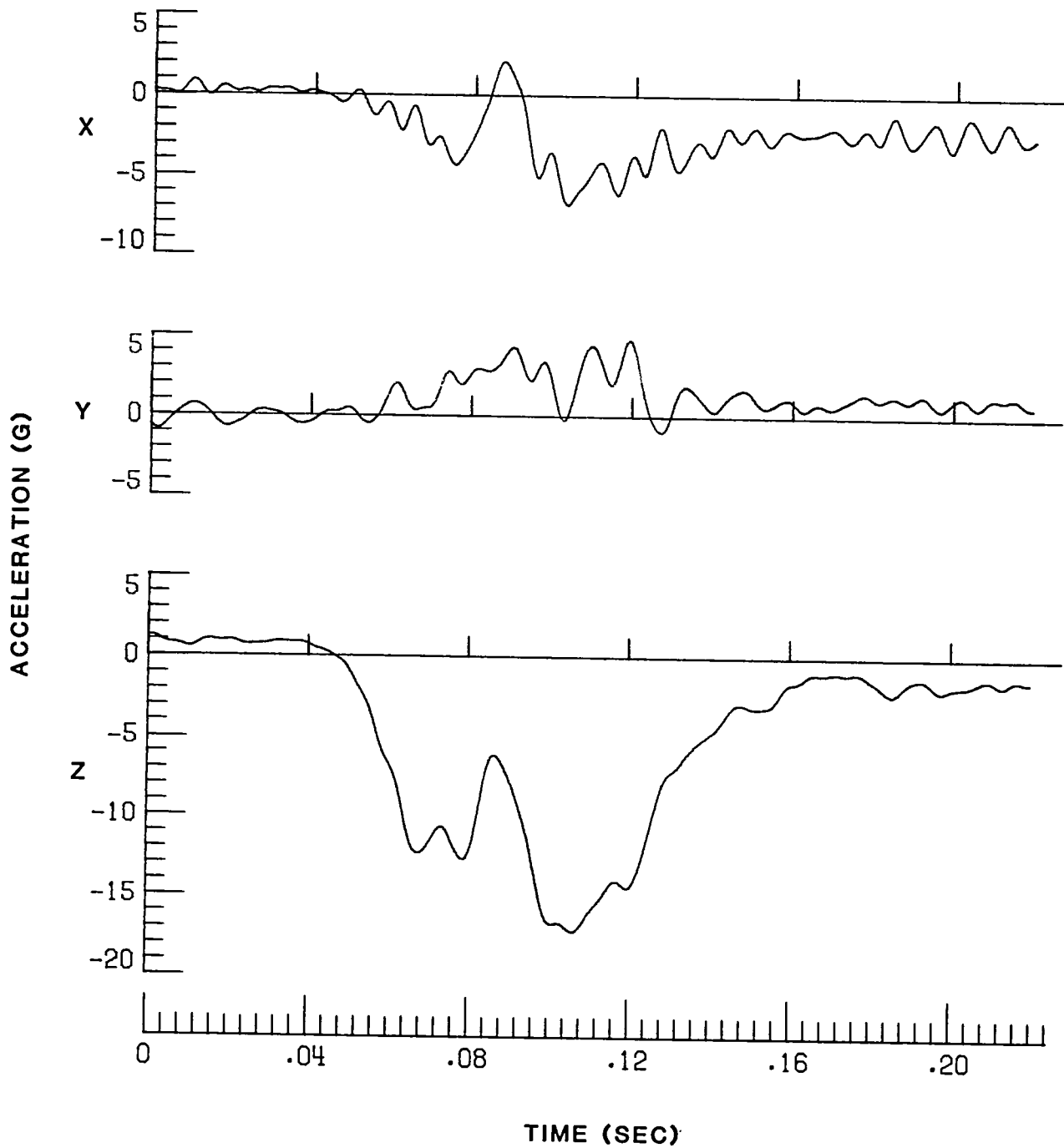
RIGHT SEAT: Energy Absorbing



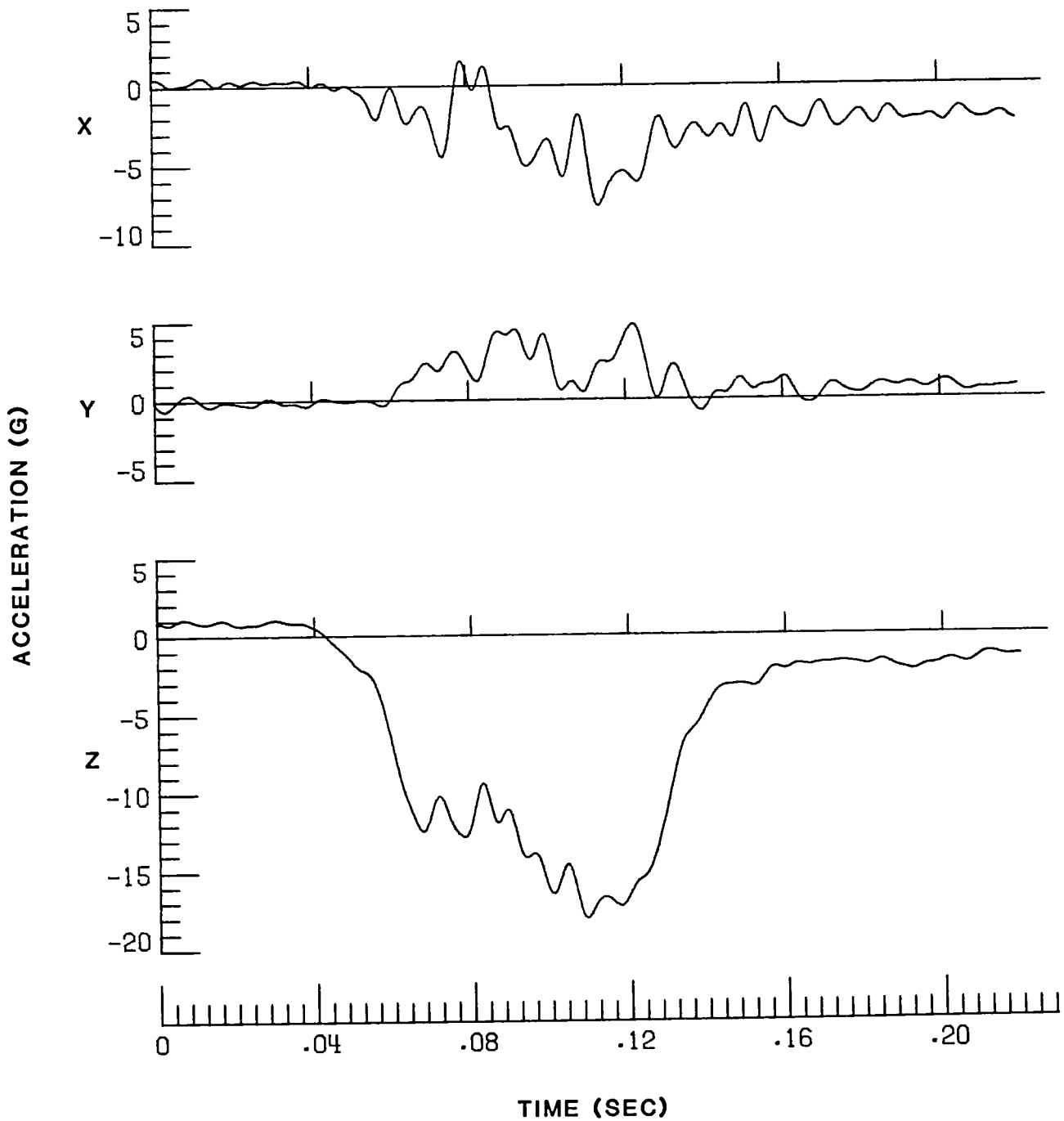
LEFT BULKHEAD  
NASA TEST NO. 1



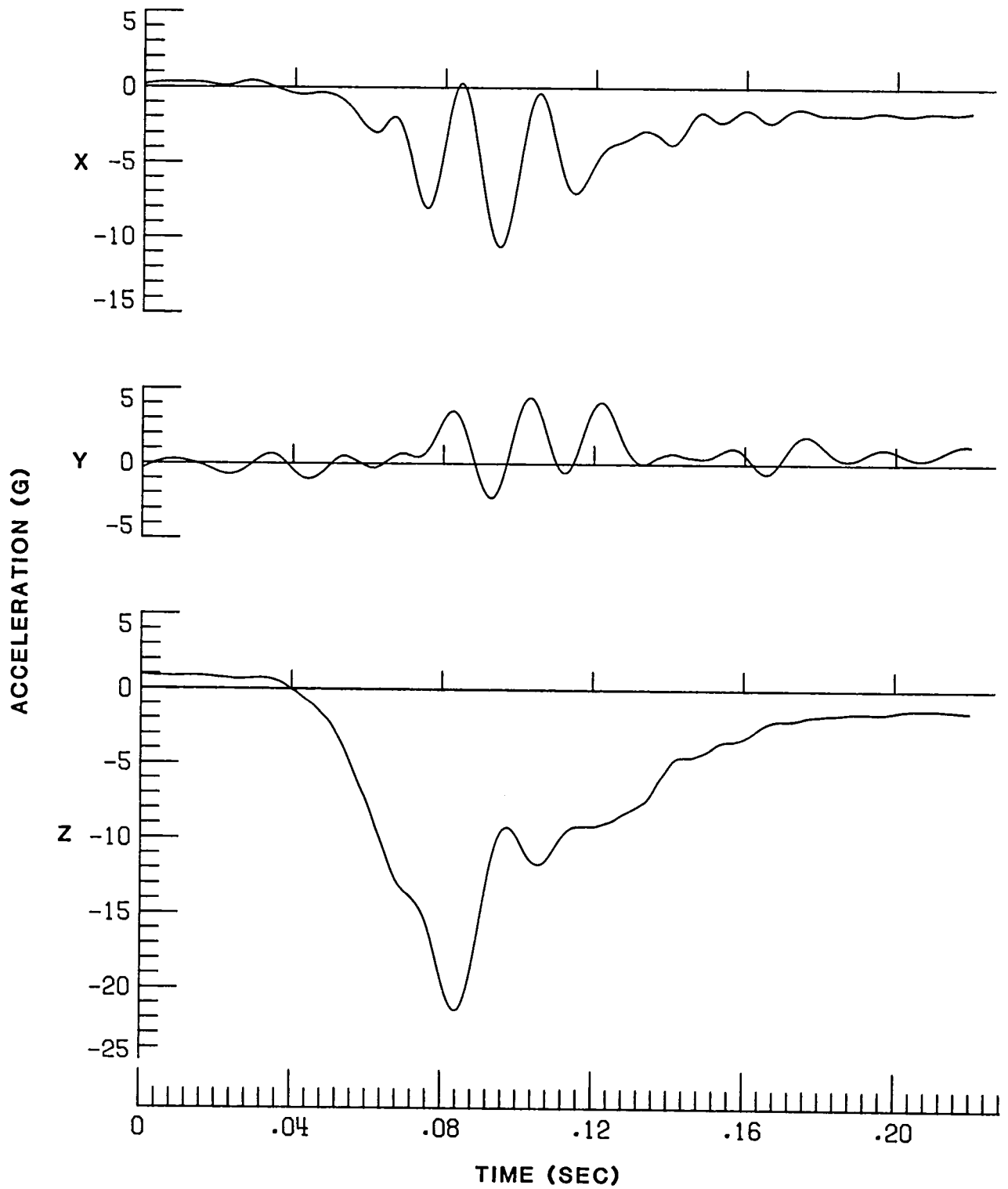
RIGHT BULKHEAD  
NASA TEST NO. 1



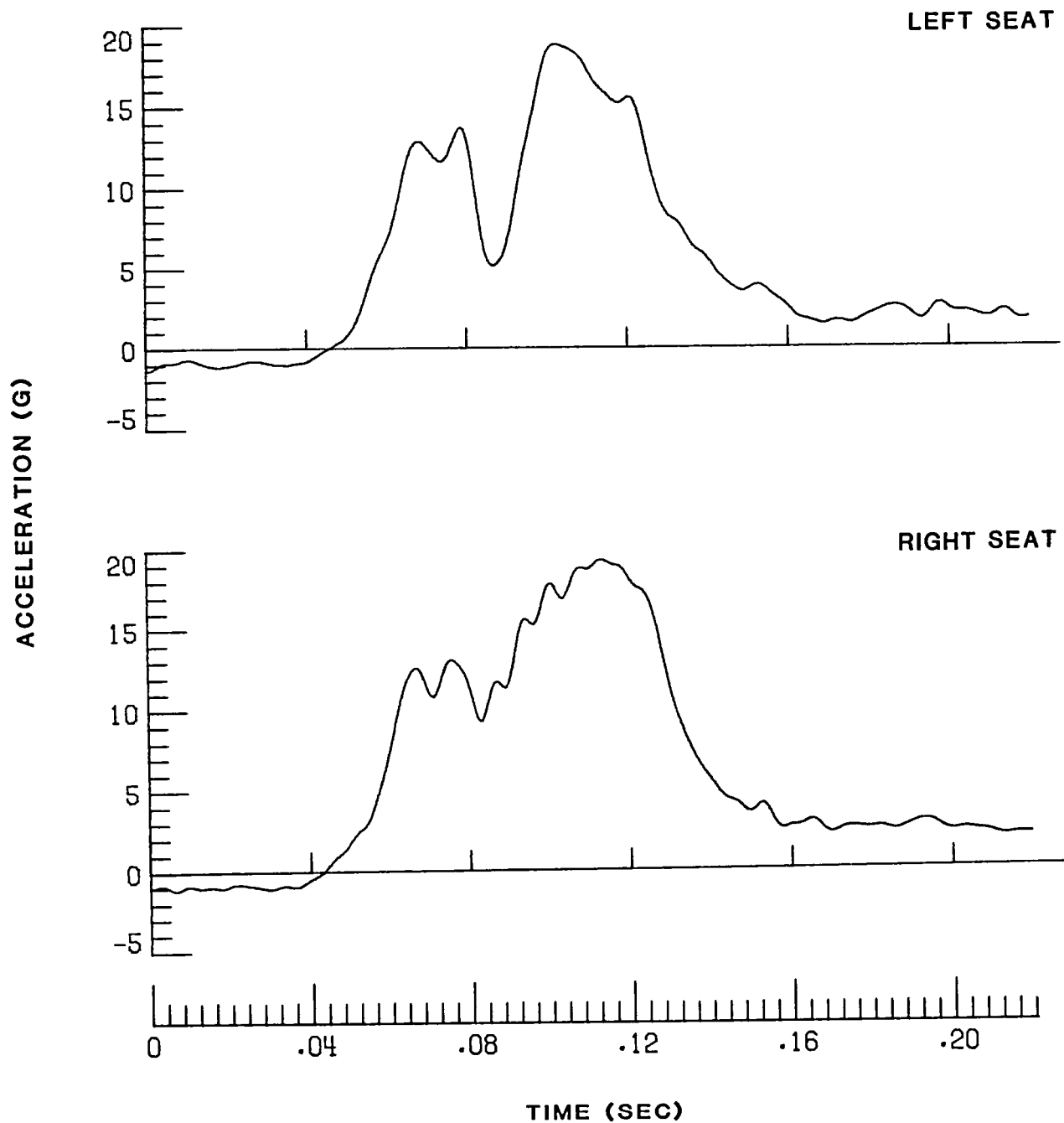
LEFT SEAT PAN  
NASA TEST NO. 1



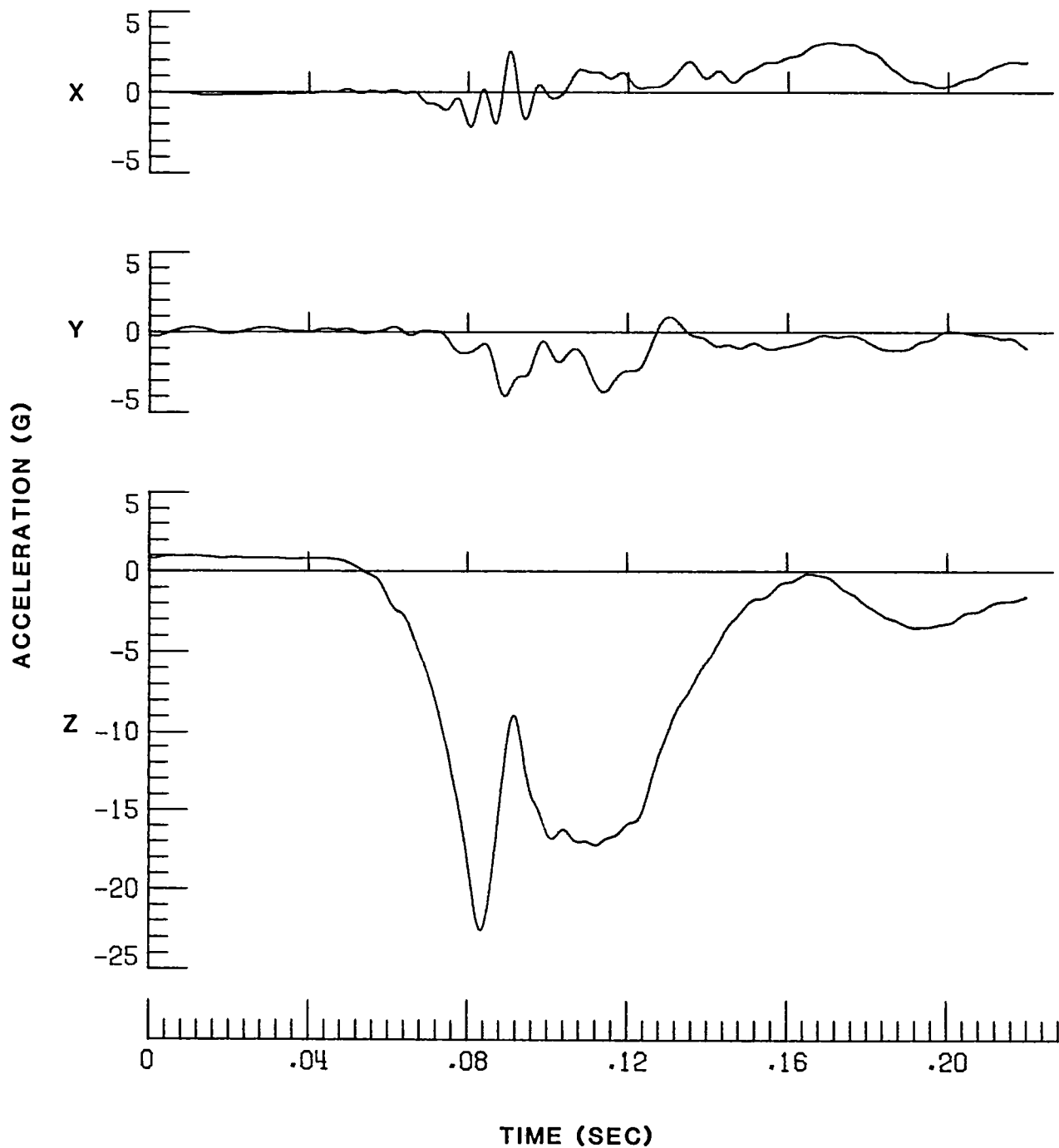
RIGHT SEAT PAN  
NASA TEST NO. 1



CAPSULE C.G.  
NASA TEST NO. 1

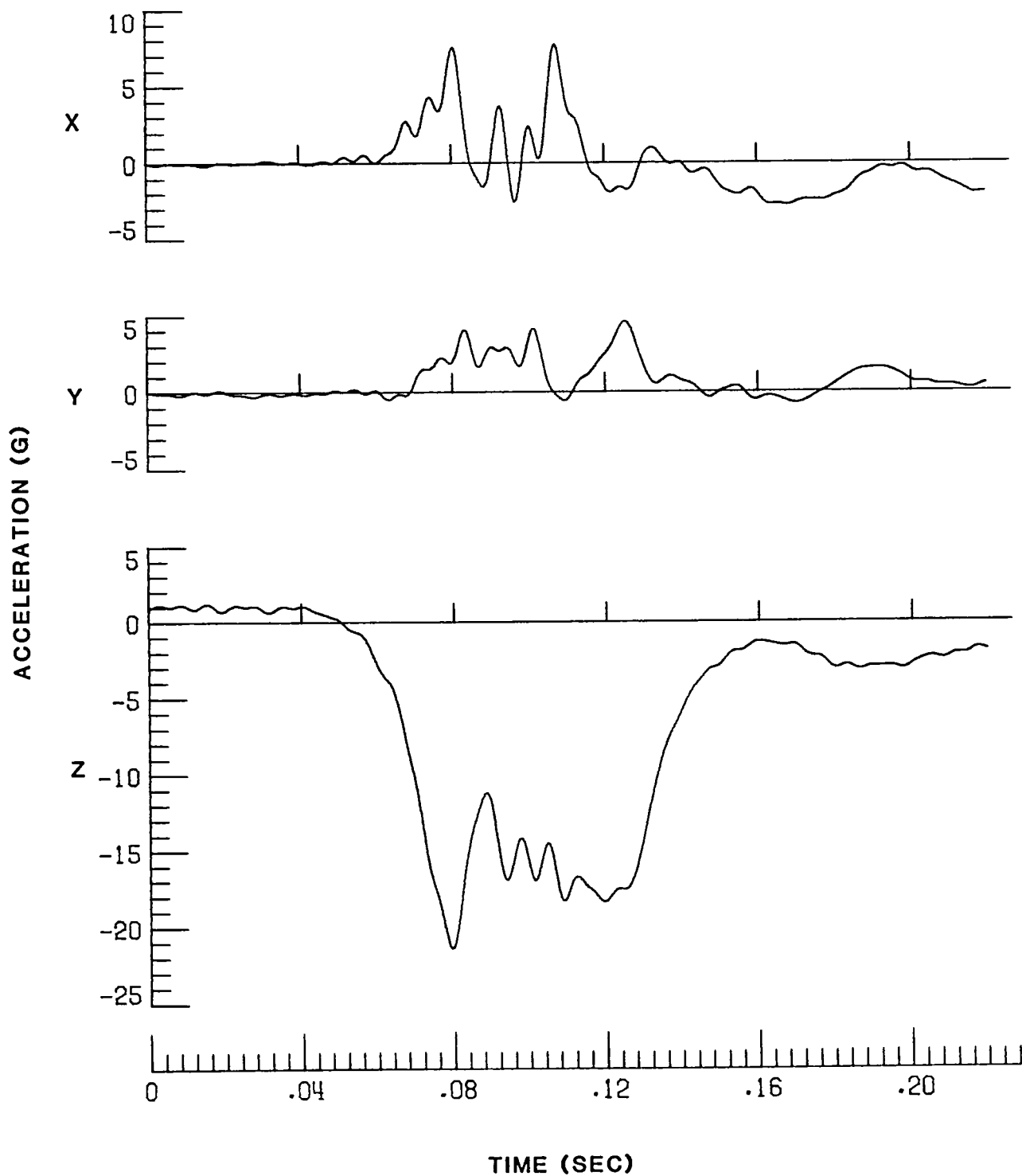


**SEAT PAN ALONG BACK TANGENT LINE  
NASA TEST NO. 1**

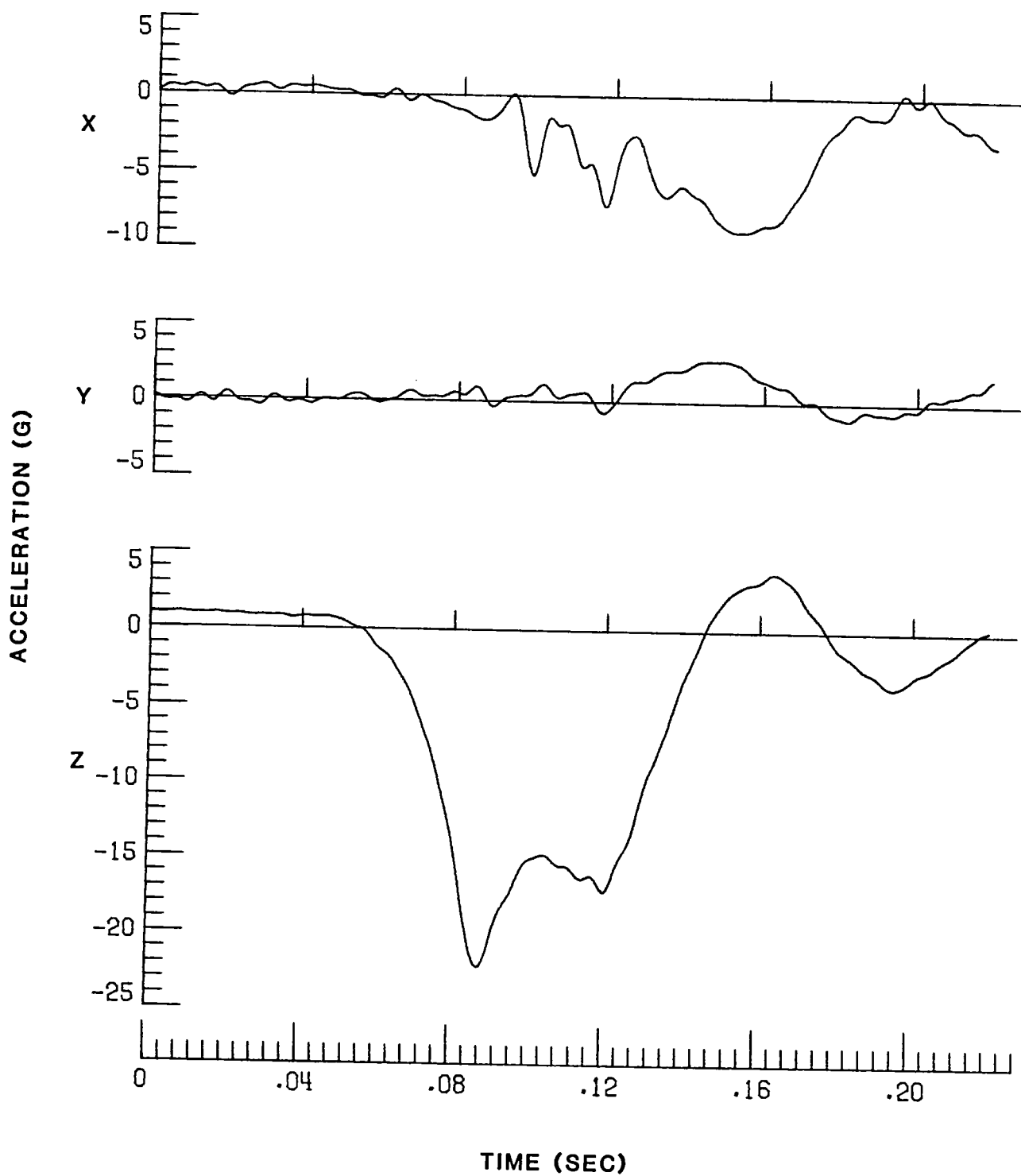


LEFT DUMMY PELVIS  
NASA TEST NO. 1

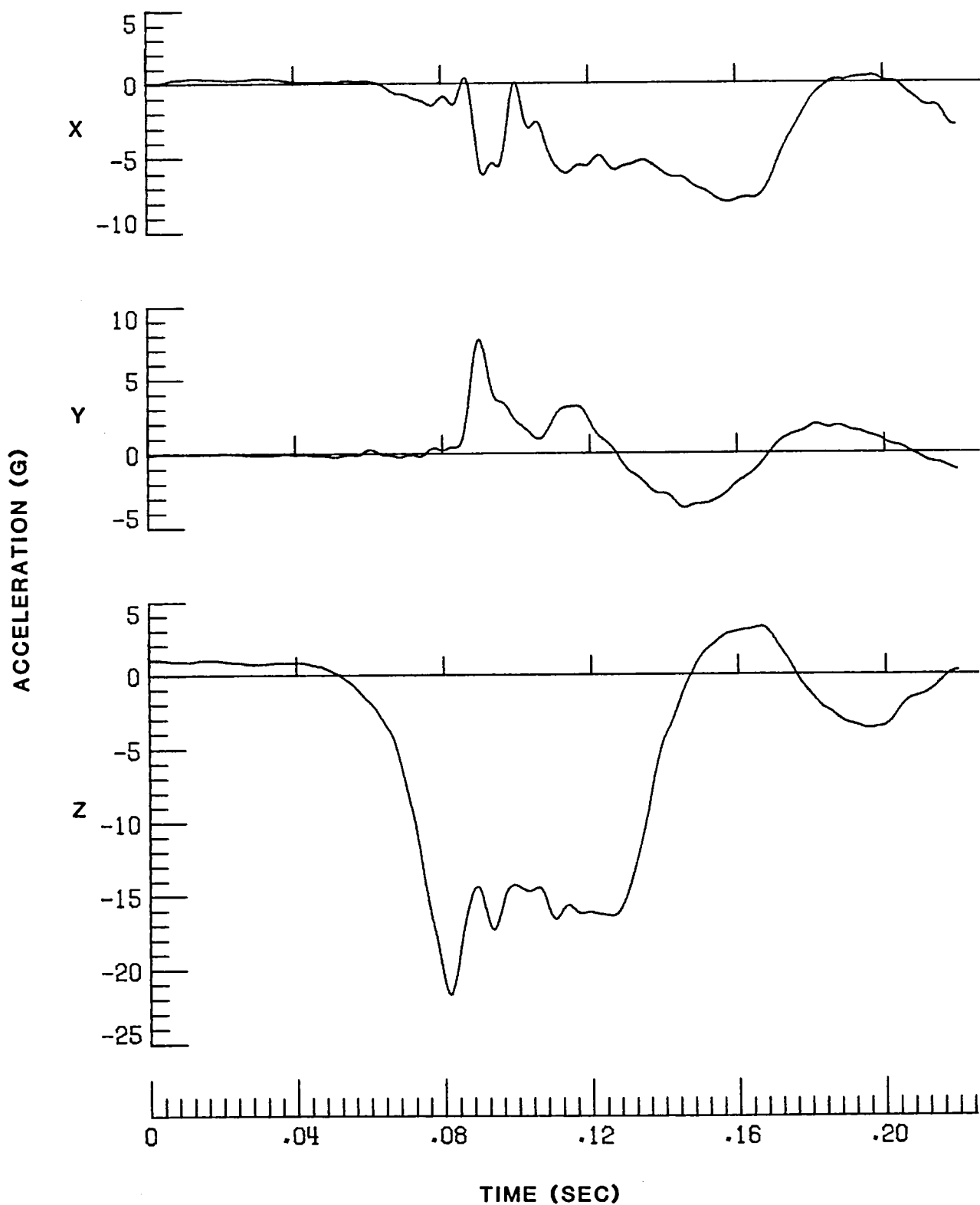




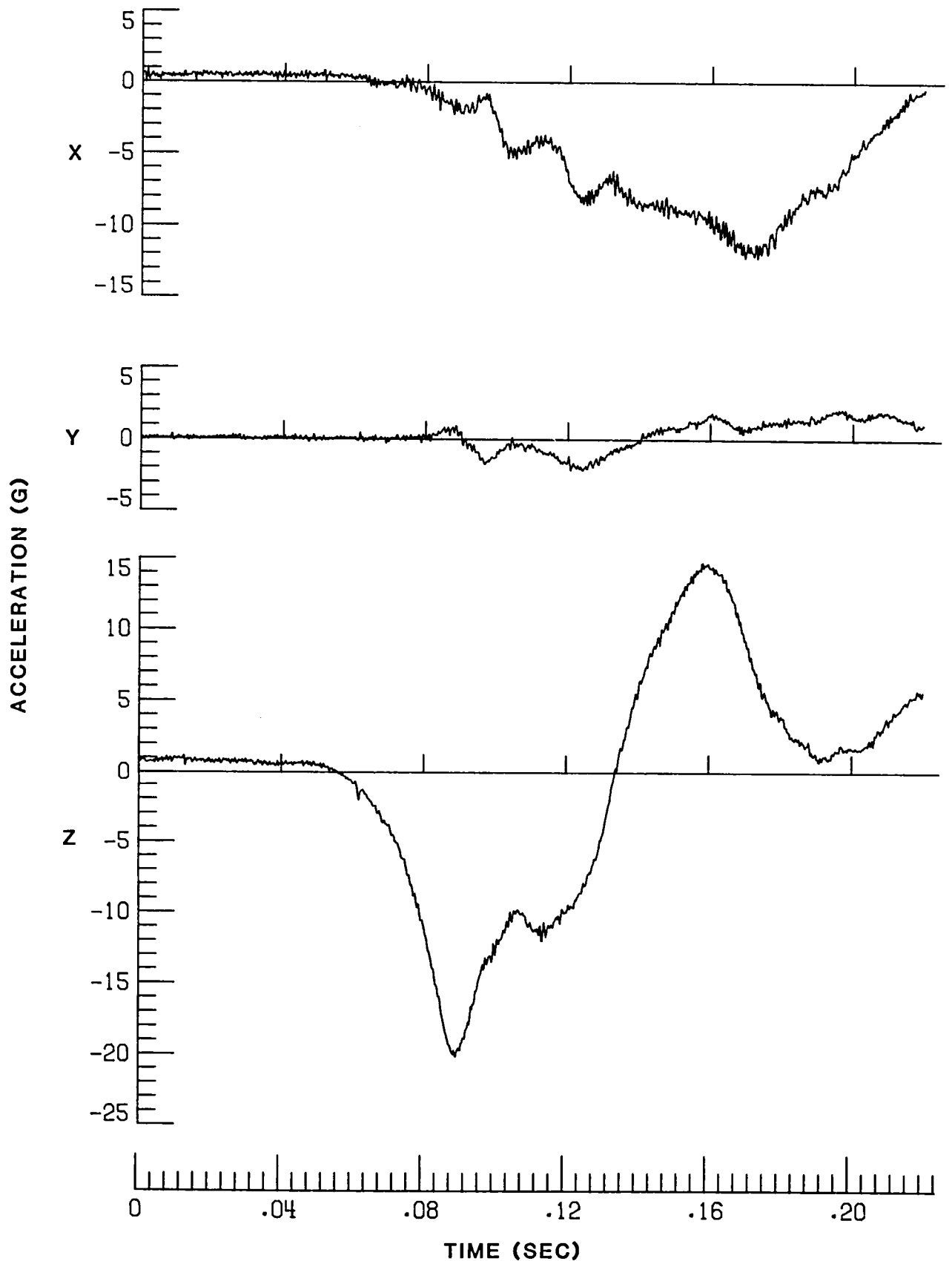
RIGHT DUMMY PELVIS  
NASA TEST NO. 1



LEFT DUMMY CHEST  
NASA TEST NO. 1

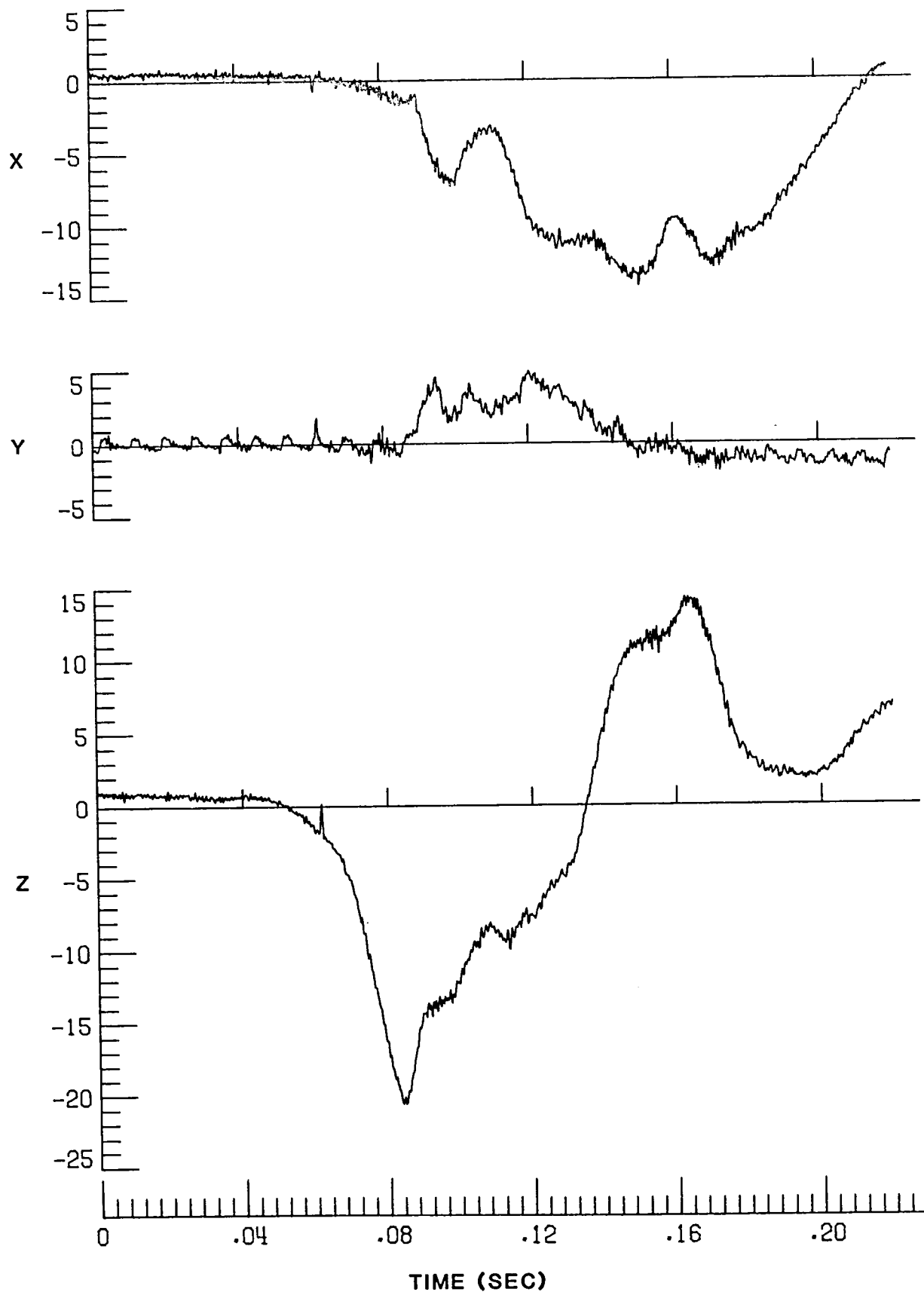


RIGHT DUMMY CHEST  
NASA TEST NO. 1

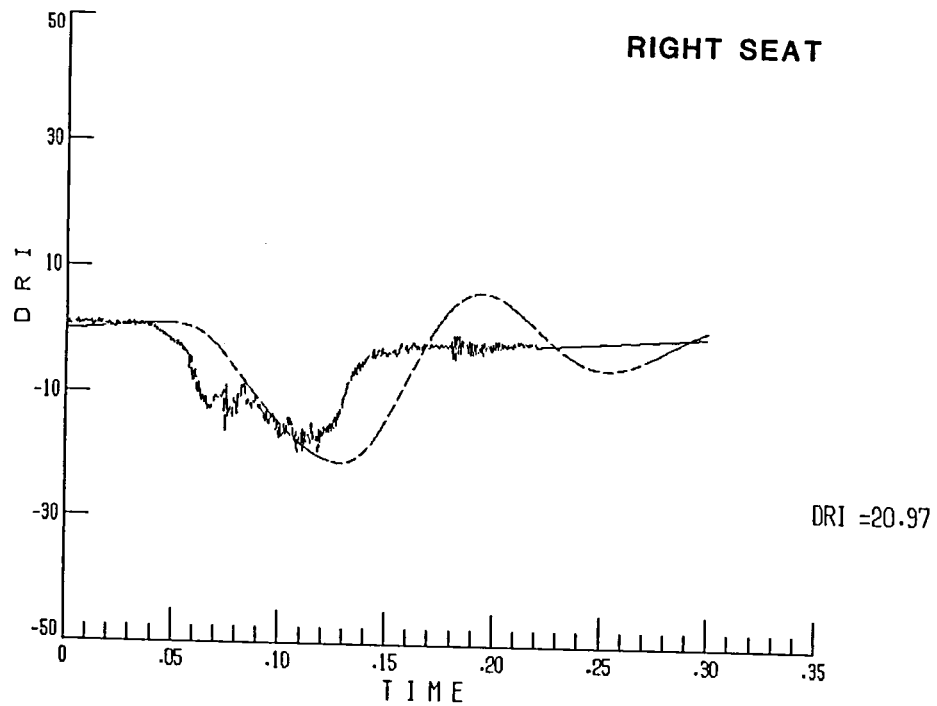
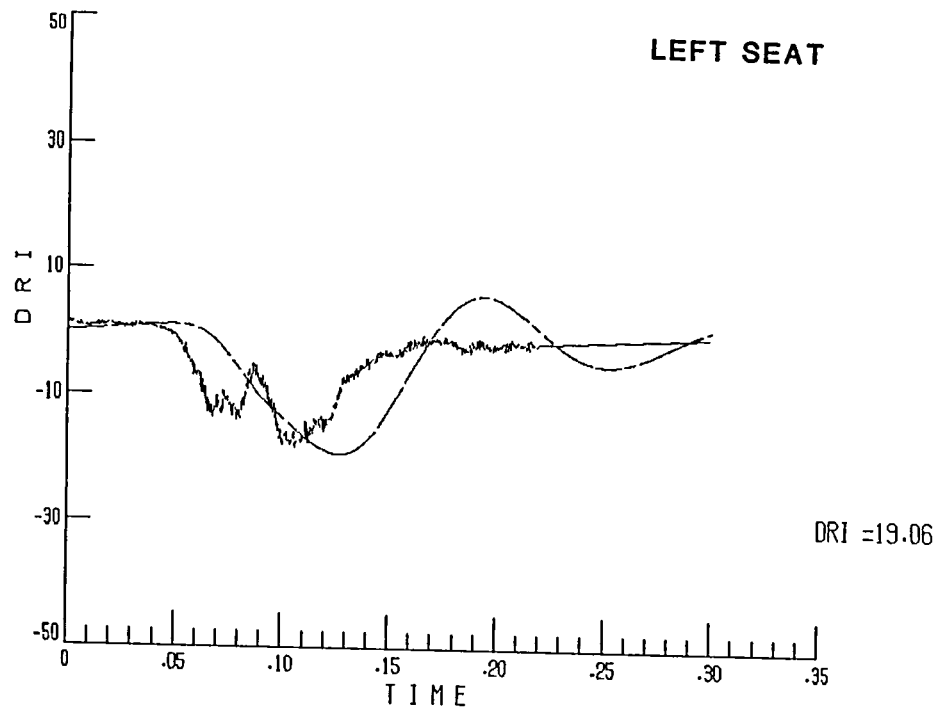


LEFT DUMMY HEAD  
NASA TEST NO. 1

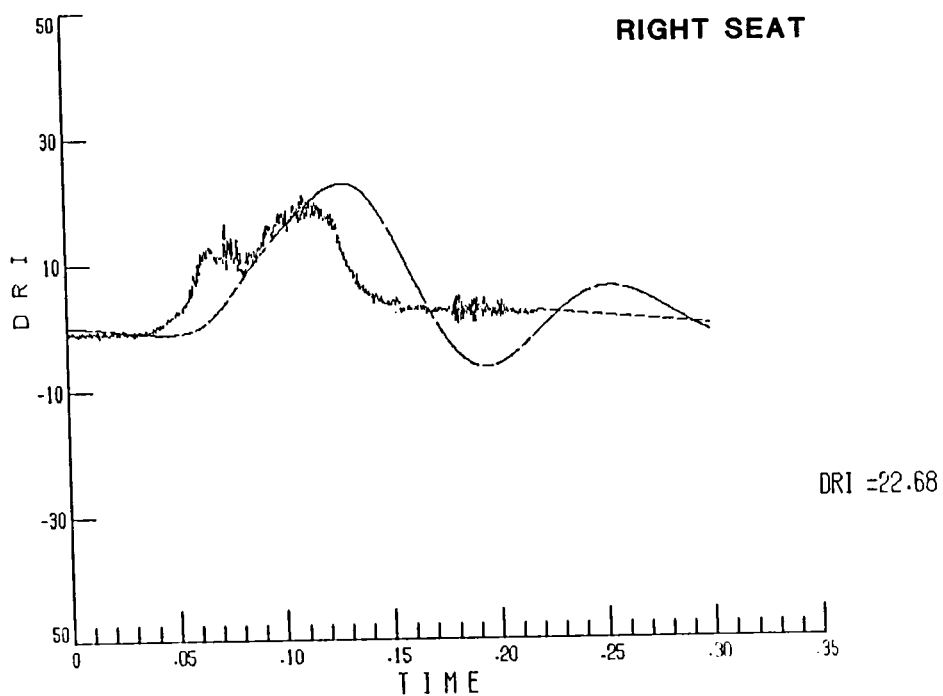
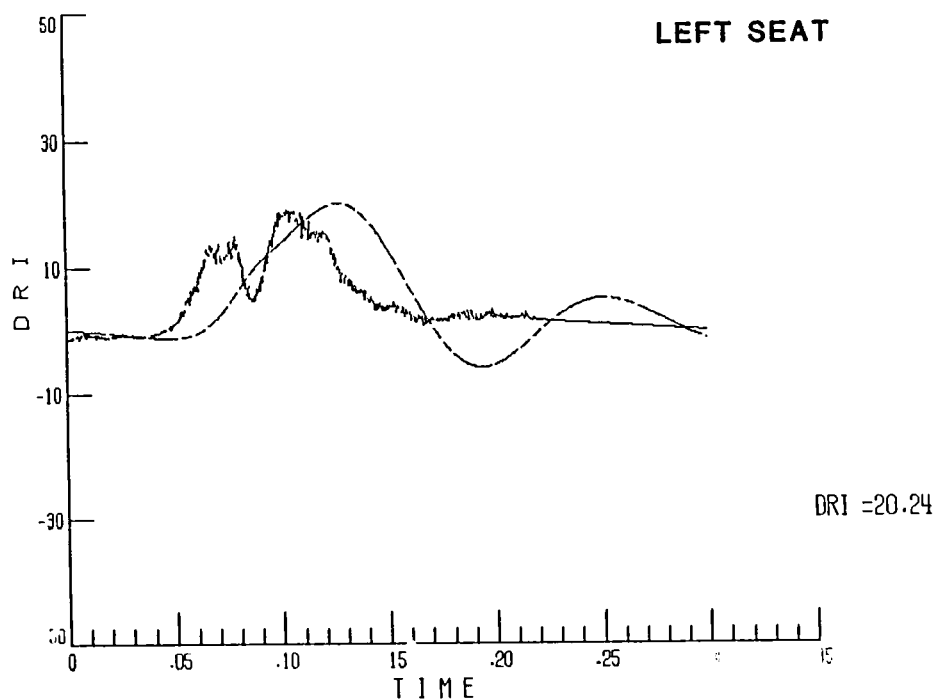
ACCELERATION (G)



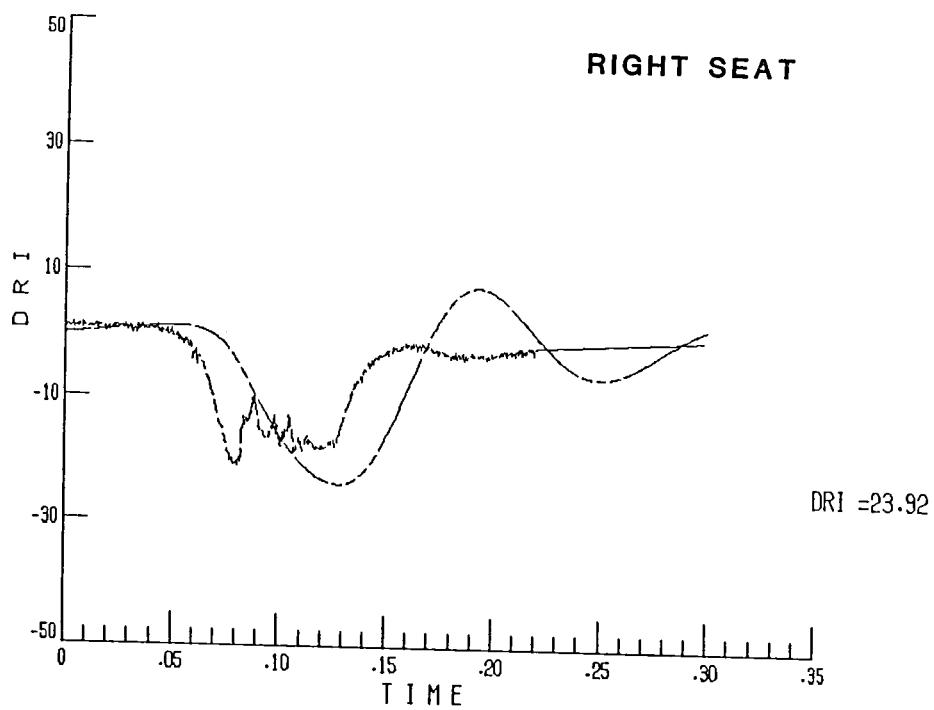
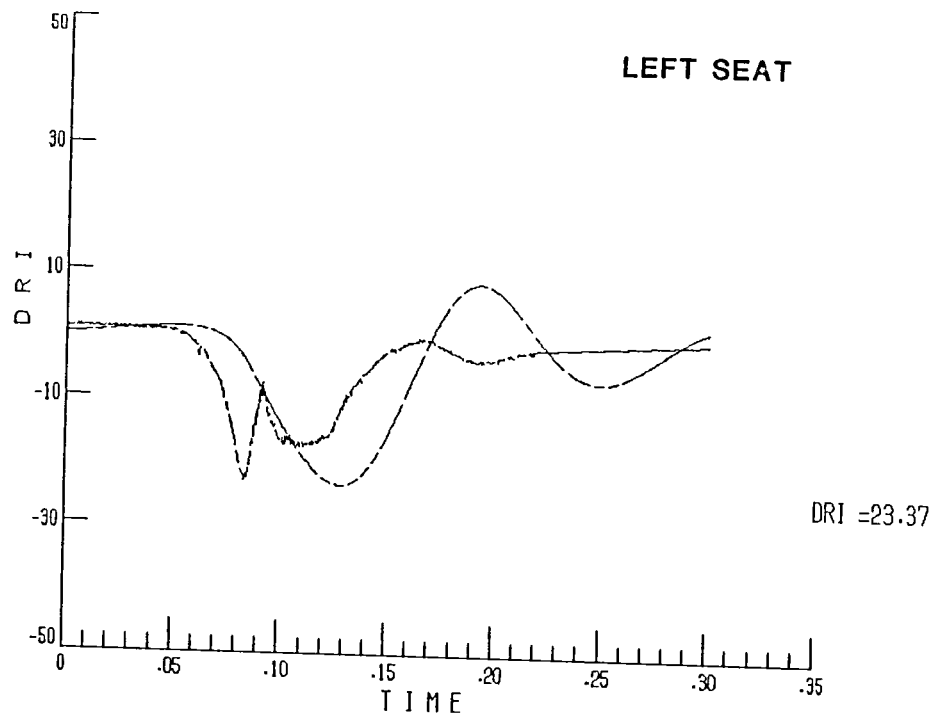
RIGHT DUMMY HEAD  
NASA TEST NO. 1



**DYNAMIC RESPONSE INDEX  
SEAT PAN Z-AXIS  
NASA TEST NO. 1**



**DYNAMIC RESPONSE INDEX  
BACK TANGENT LINE  
NASA TEST NO. 1**



**DYNAMIC RESPONSE INDEX  
DUMMY PELVIS Z-AXIS  
NASA TEST NO. 1**



### 3.0 NASA TEST NO. 2

HORIZONTAL VELOCITY: 34 ft/sec

VERTICAL VELOCITY: 32 ft/sec

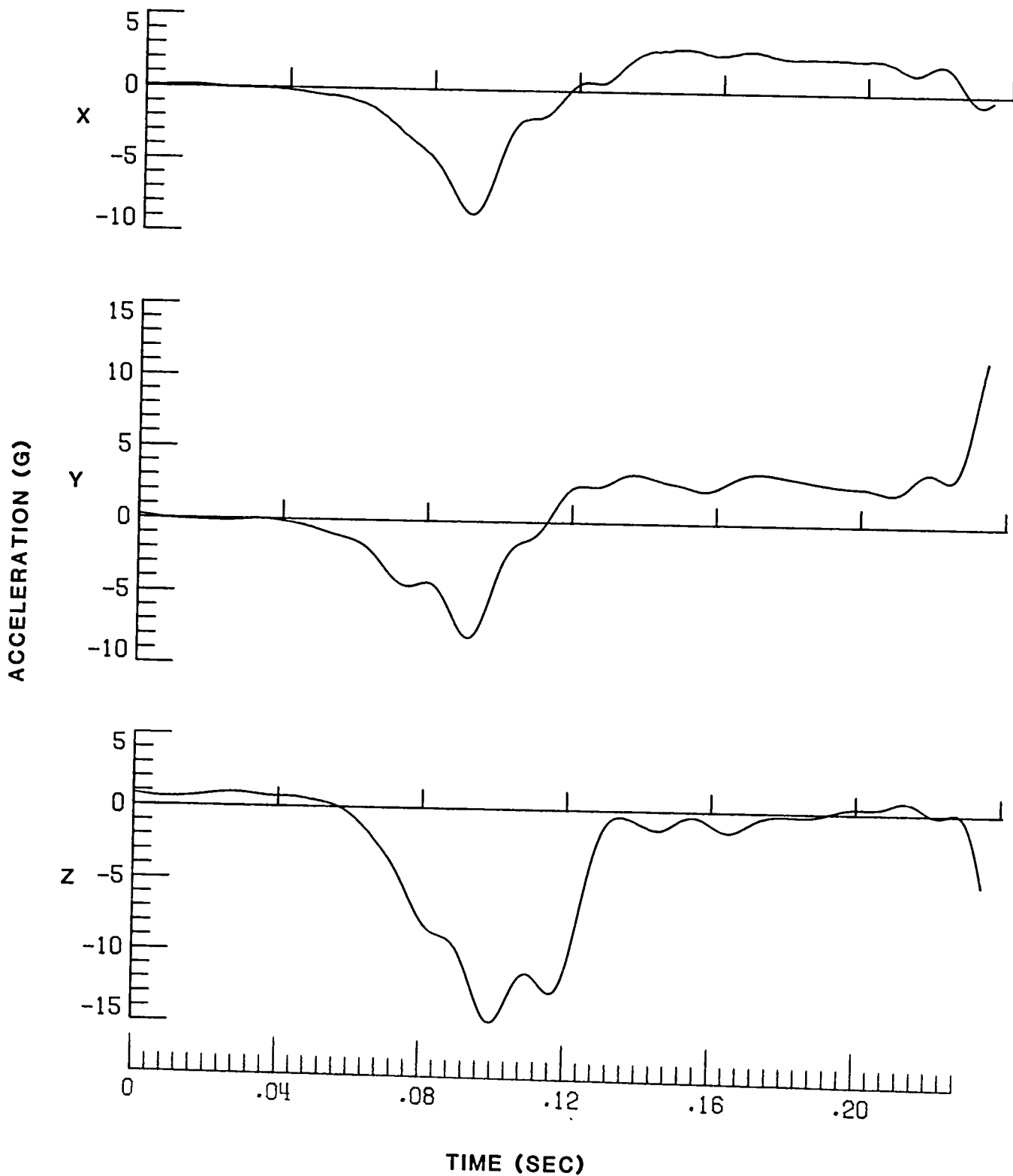
CREW MODULE PITCH:  $0^{\circ}$

CREW MODULE ROLL:  $15^{\circ}$

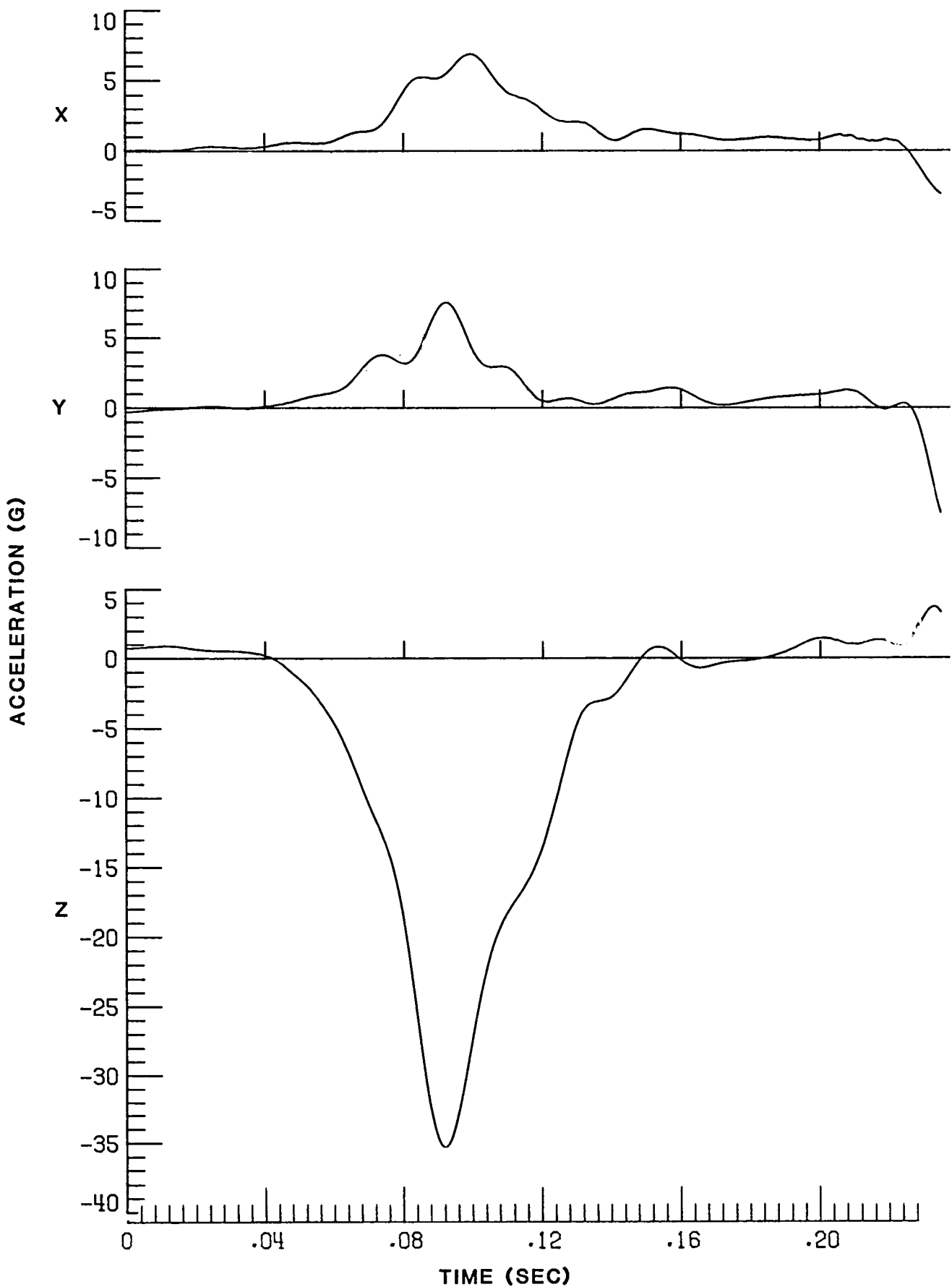
CREW MODULE YAW:  $90^{\circ}$

LEFT SEAT: Energy Absorbing

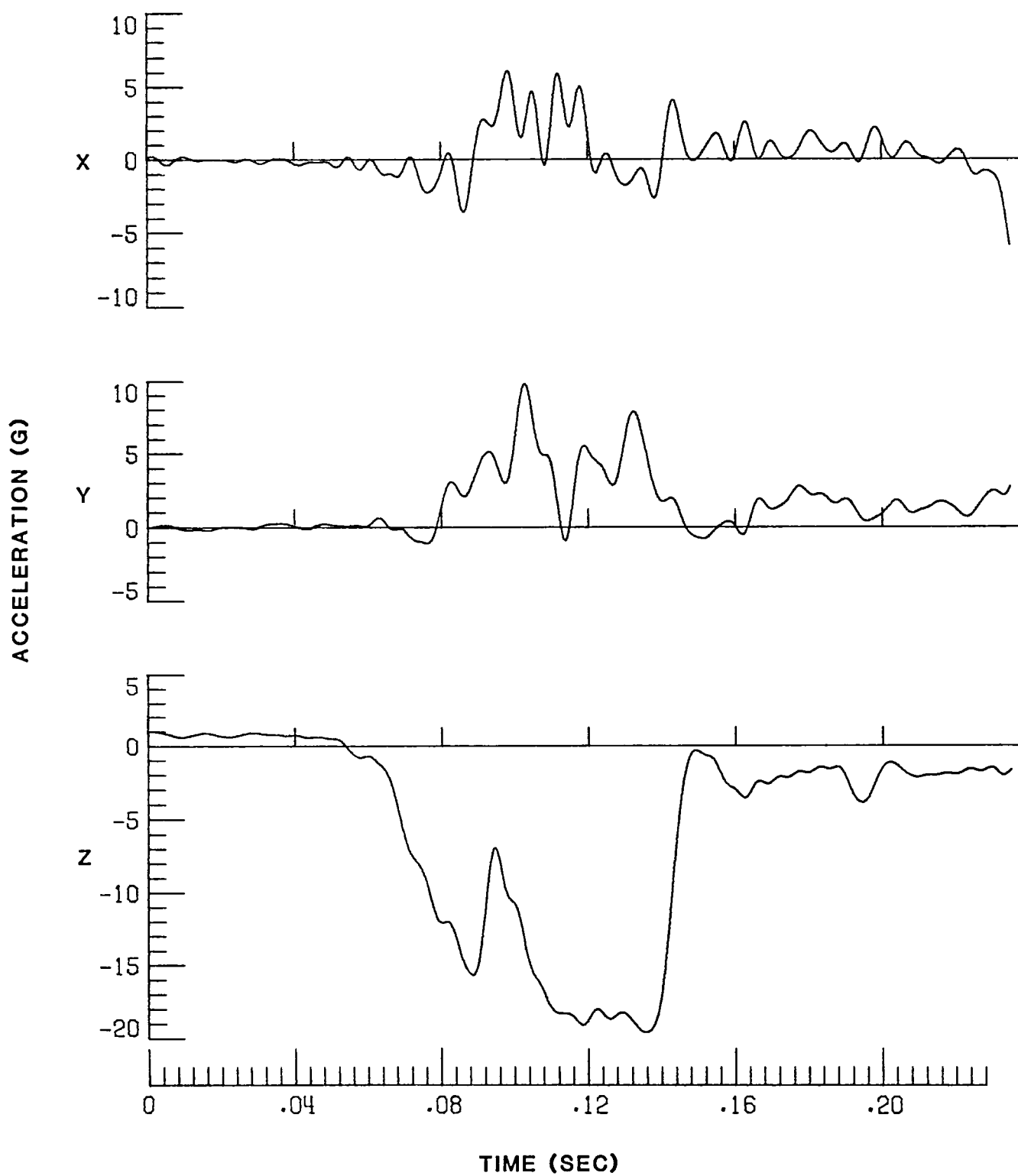
RIGHT SEAT: Energy Absorbing



LEFT BULKHEAD  
NASA TEST NO. 2

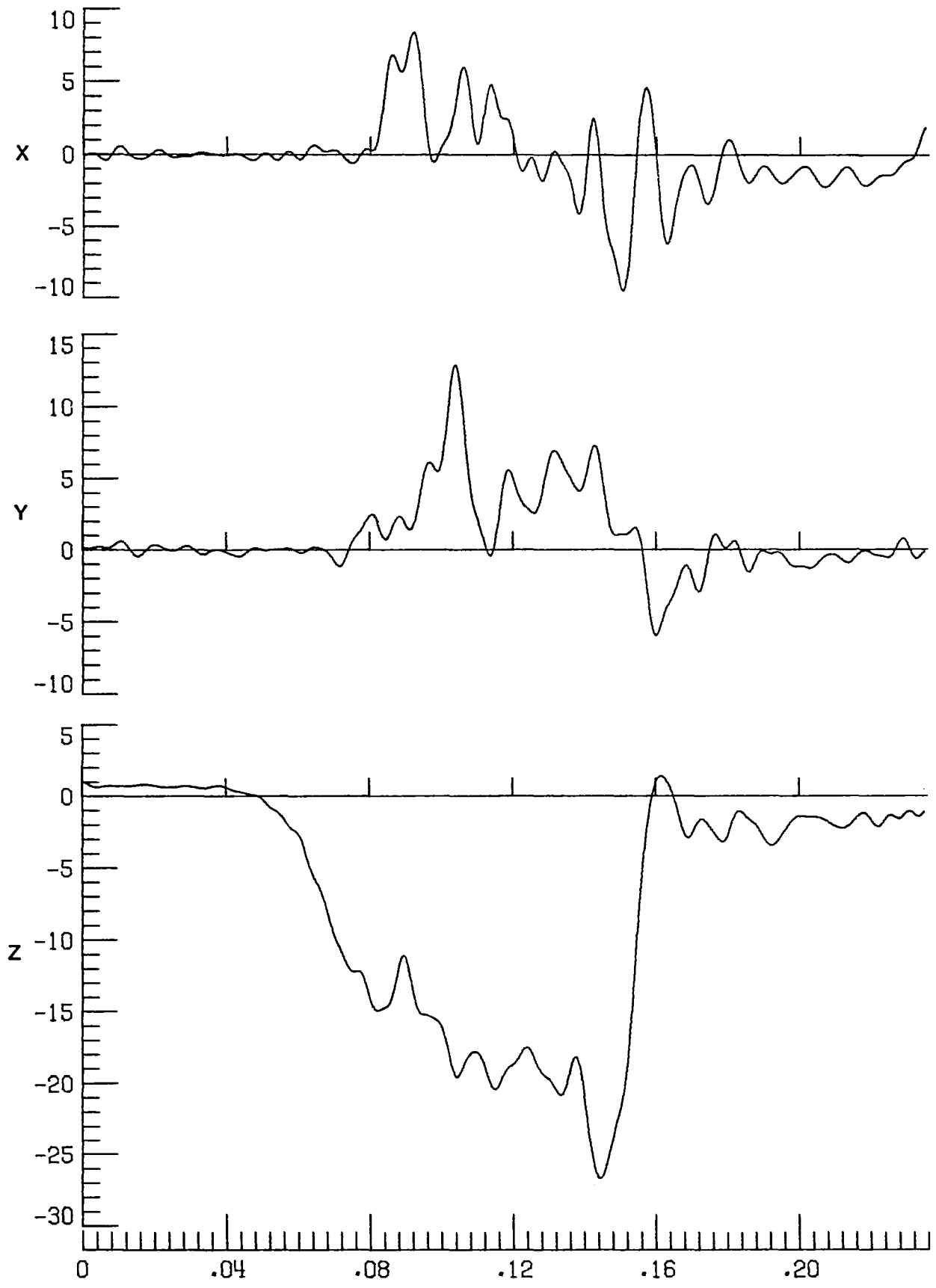


RIGHT BULKHEAD  
NASA TEST NO. 2

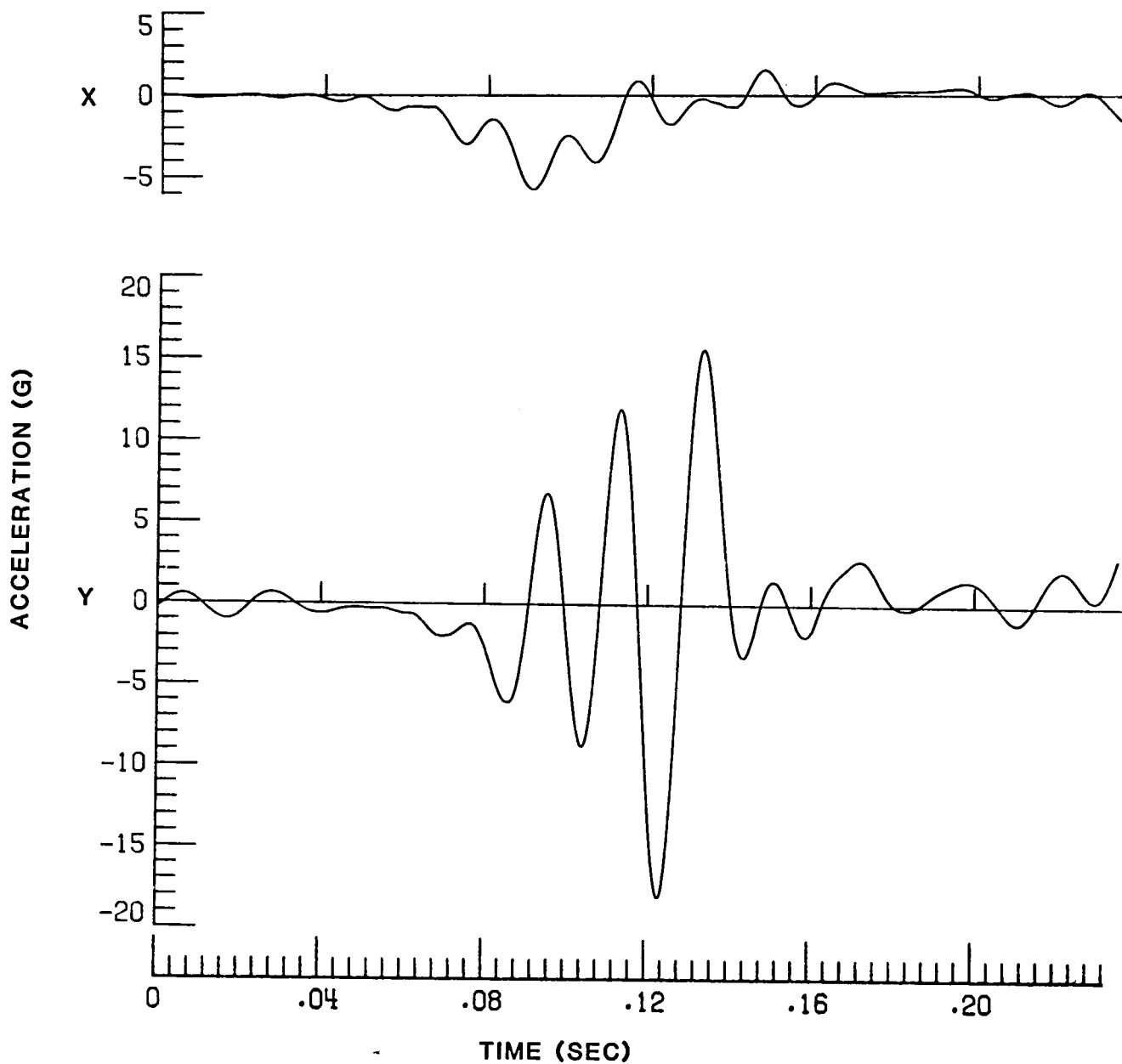


LEFT SEAT PAN  
NASA TEST NO. 2

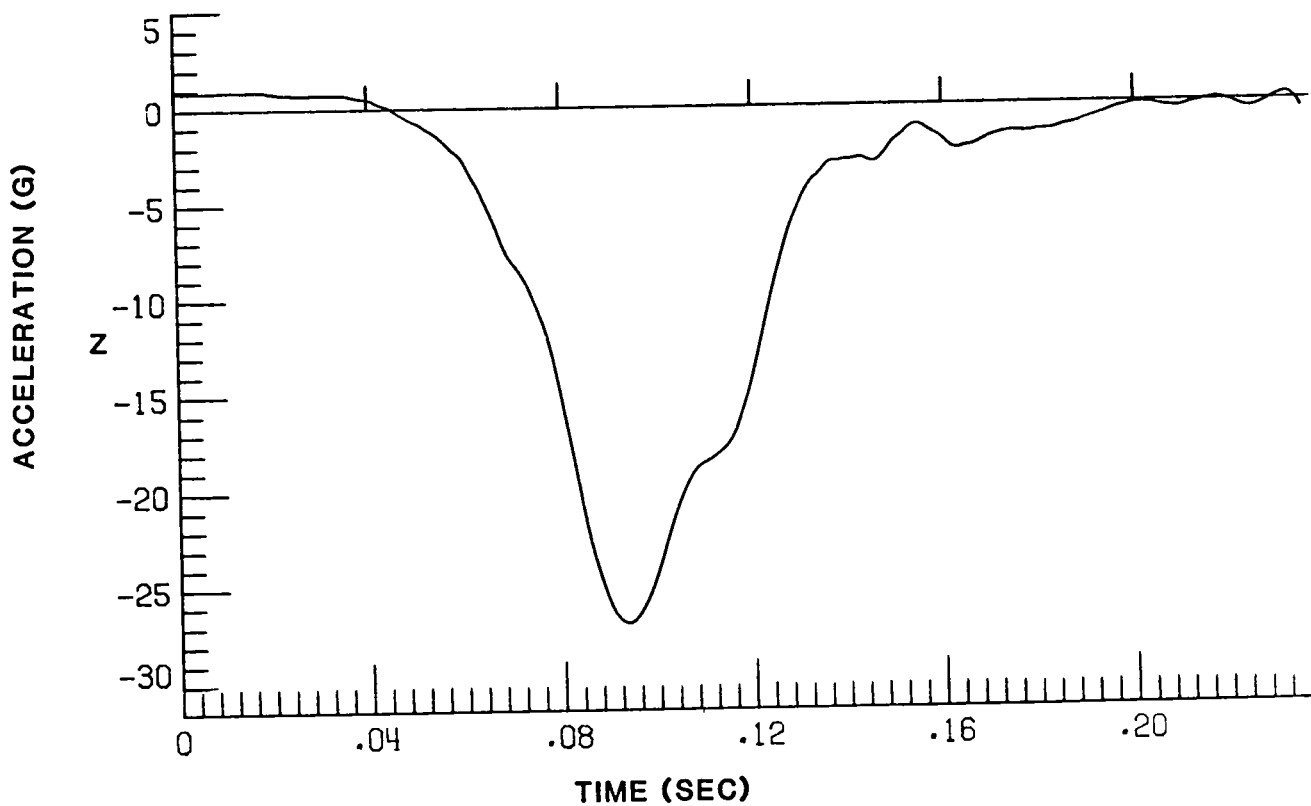
ACCELERATION (G)



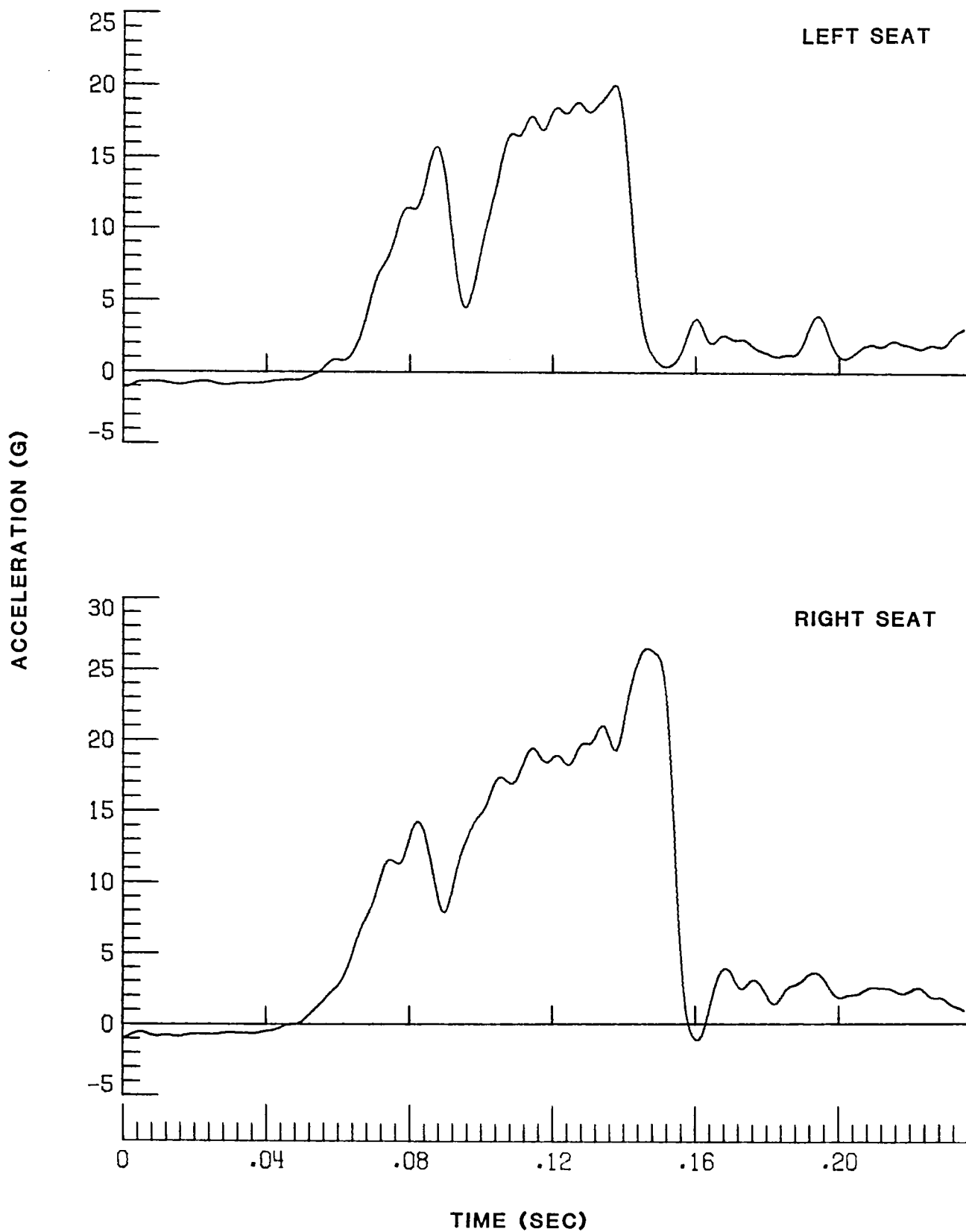
RIGHT SEAT PAN  
NASA TEST NO. 2



CAPSULE C.G.  
NASA TEST NO. 2

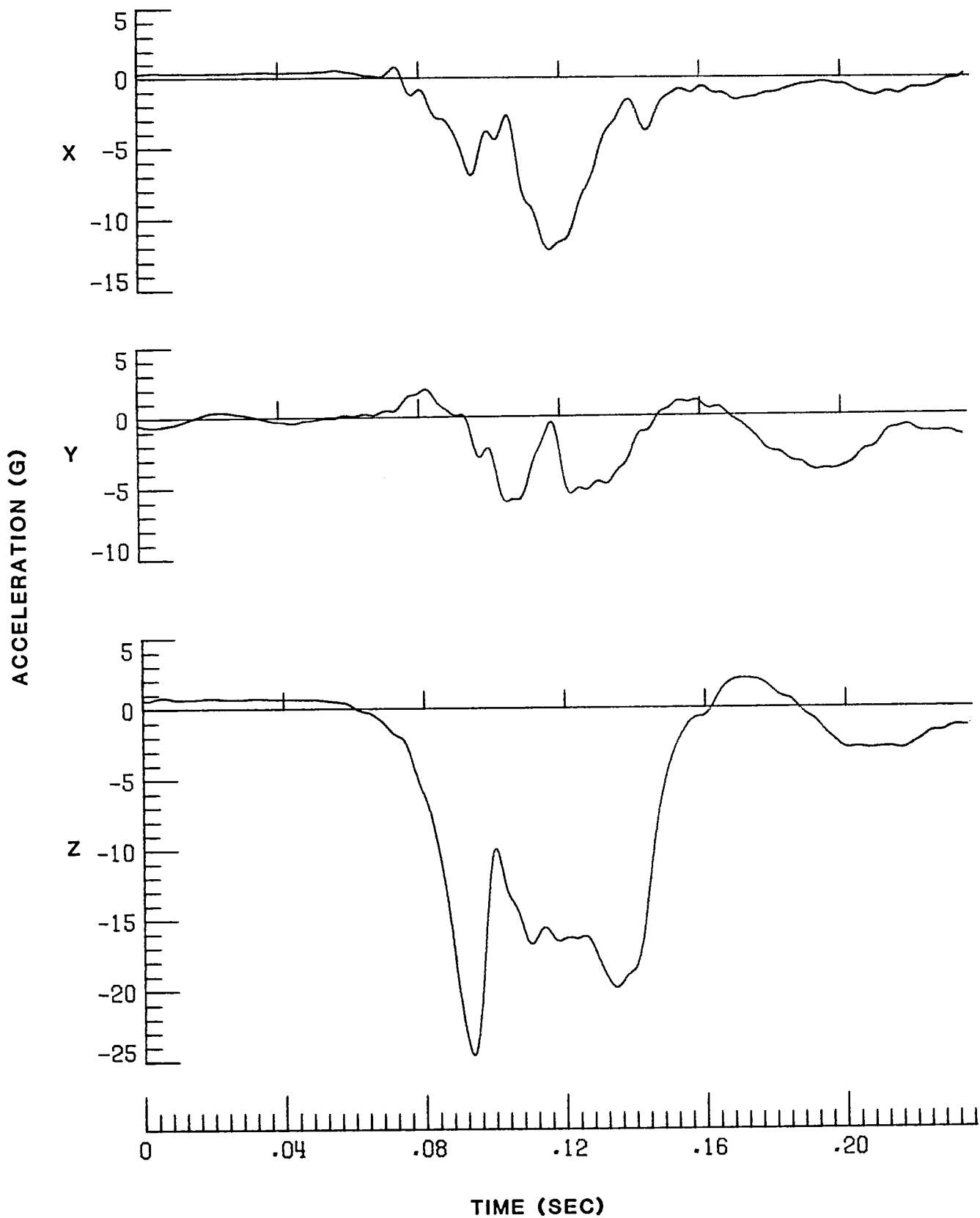


CAPSULE C.G.  
NASA TEST NO. 2

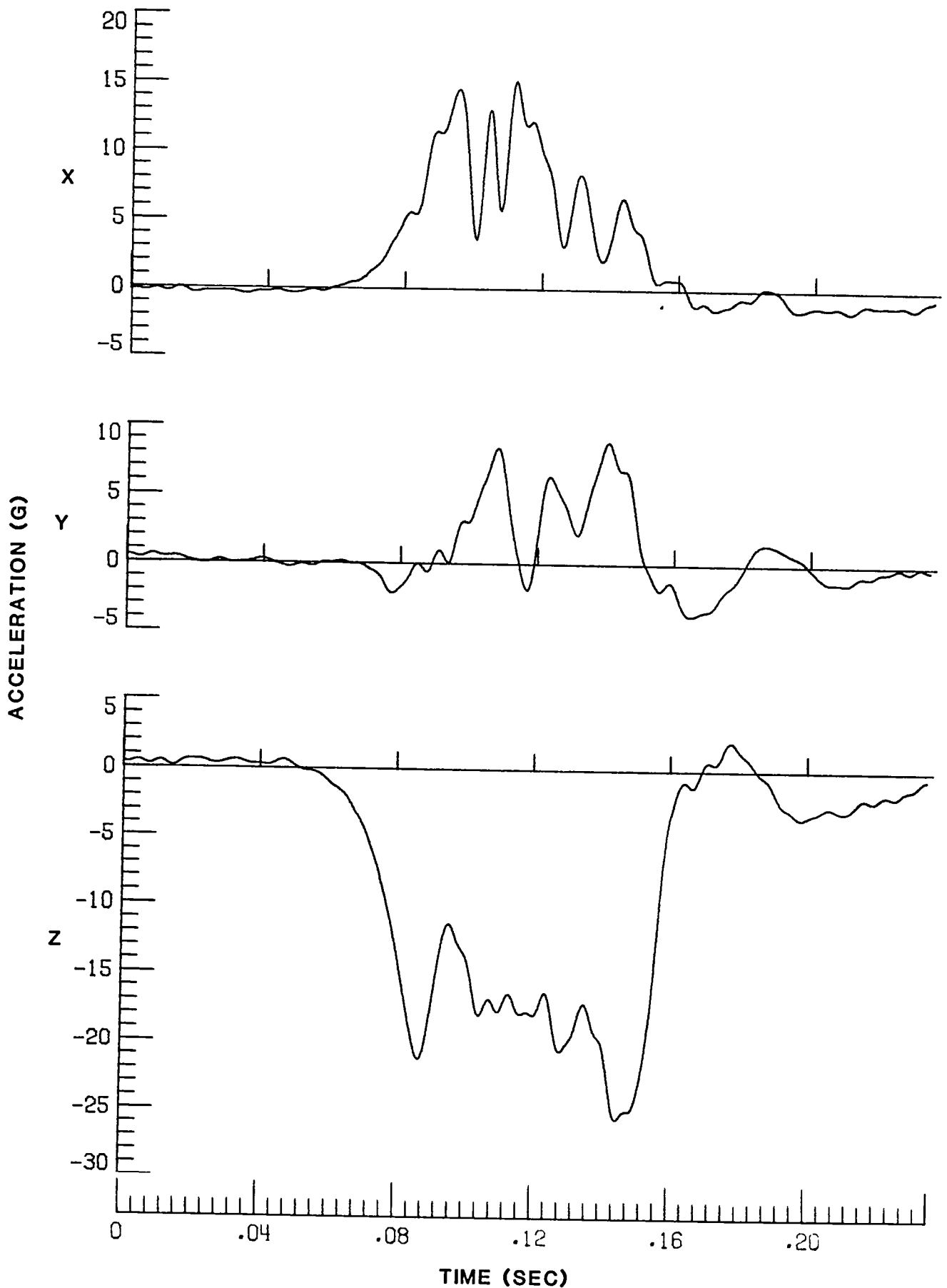


SEAT PAN ALONG BACK TANGENT LINE  
NASA TEST NO. 2

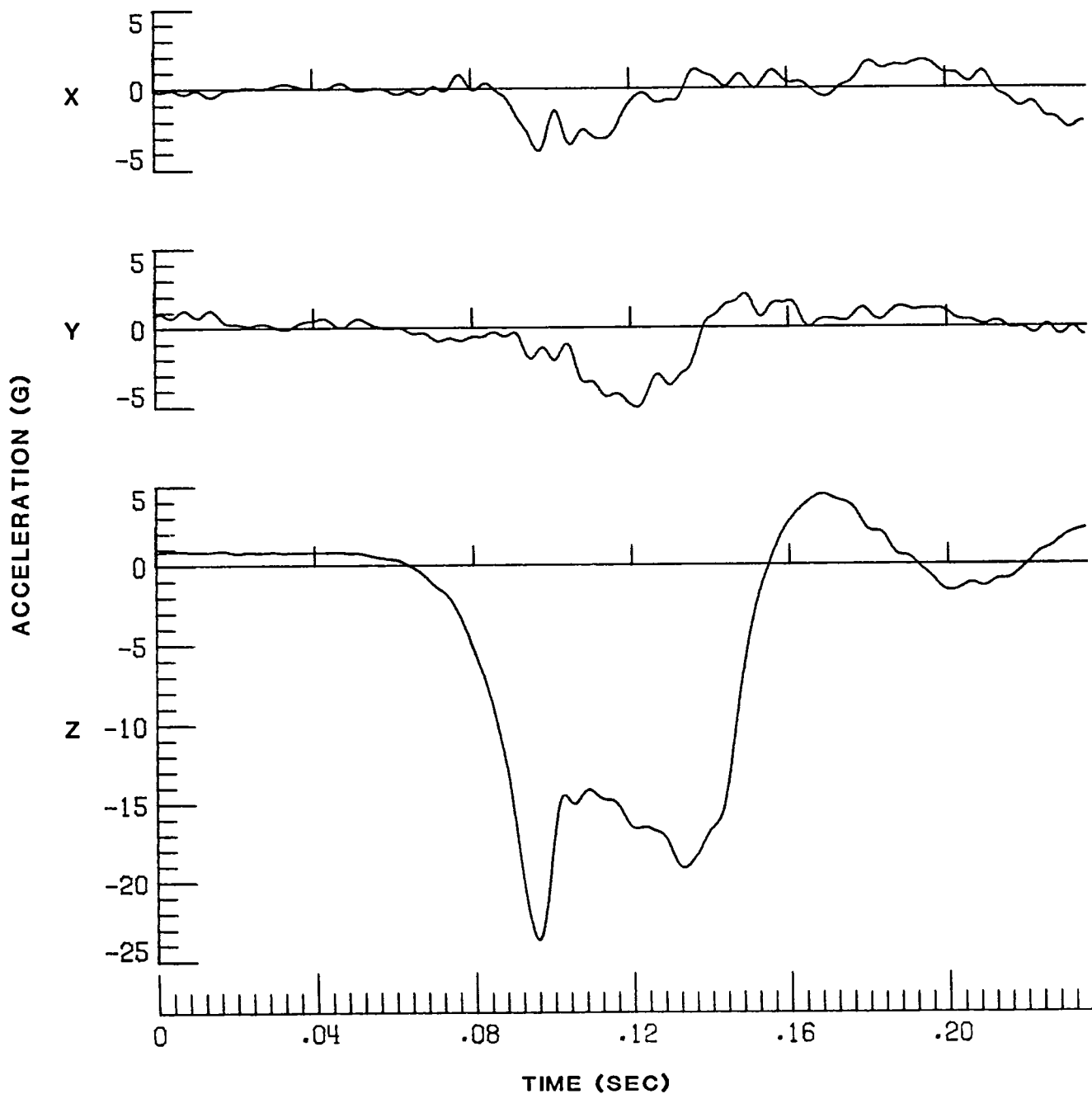




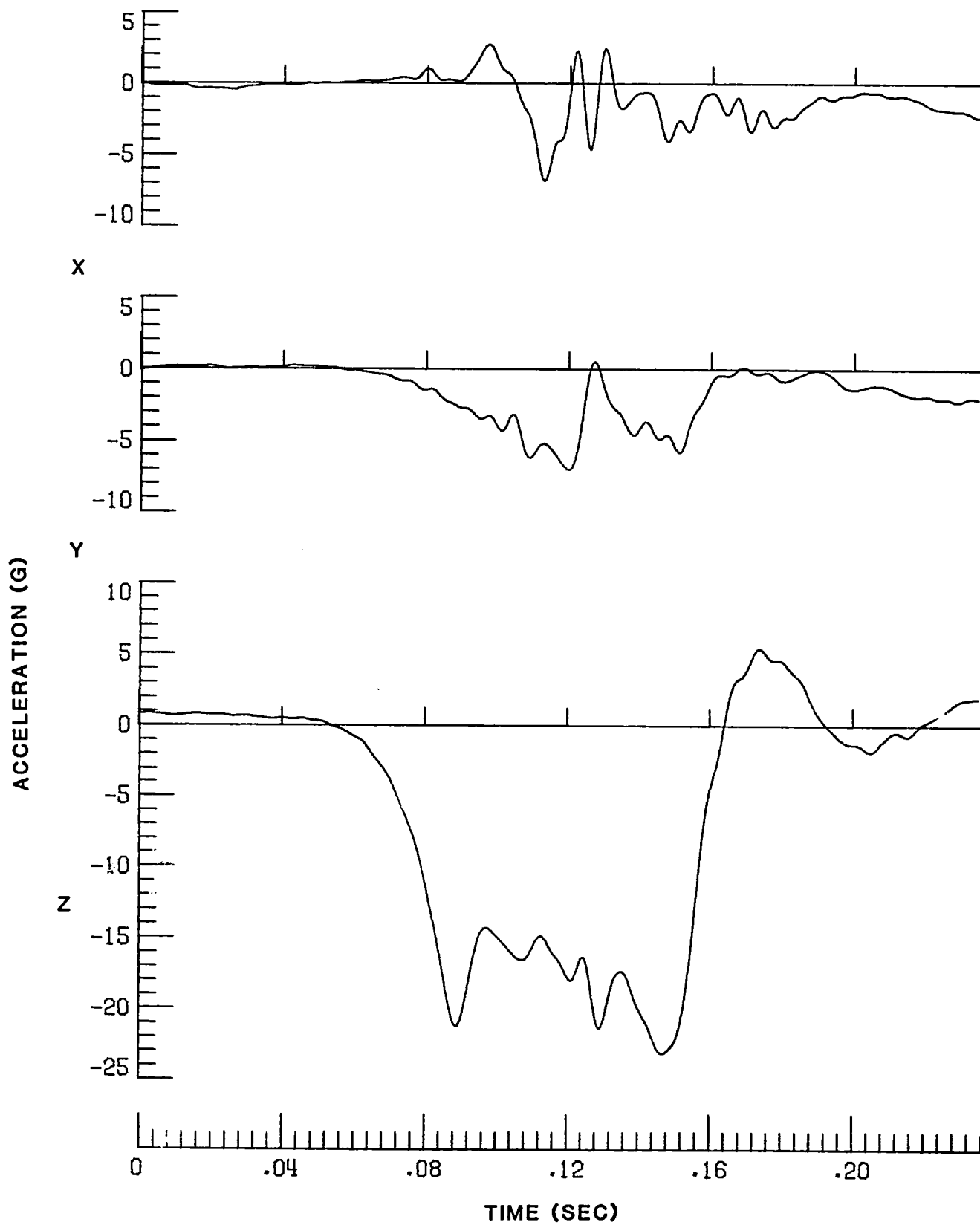
LEFT DUMMY PELVIS  
NASA TEST NO. 2



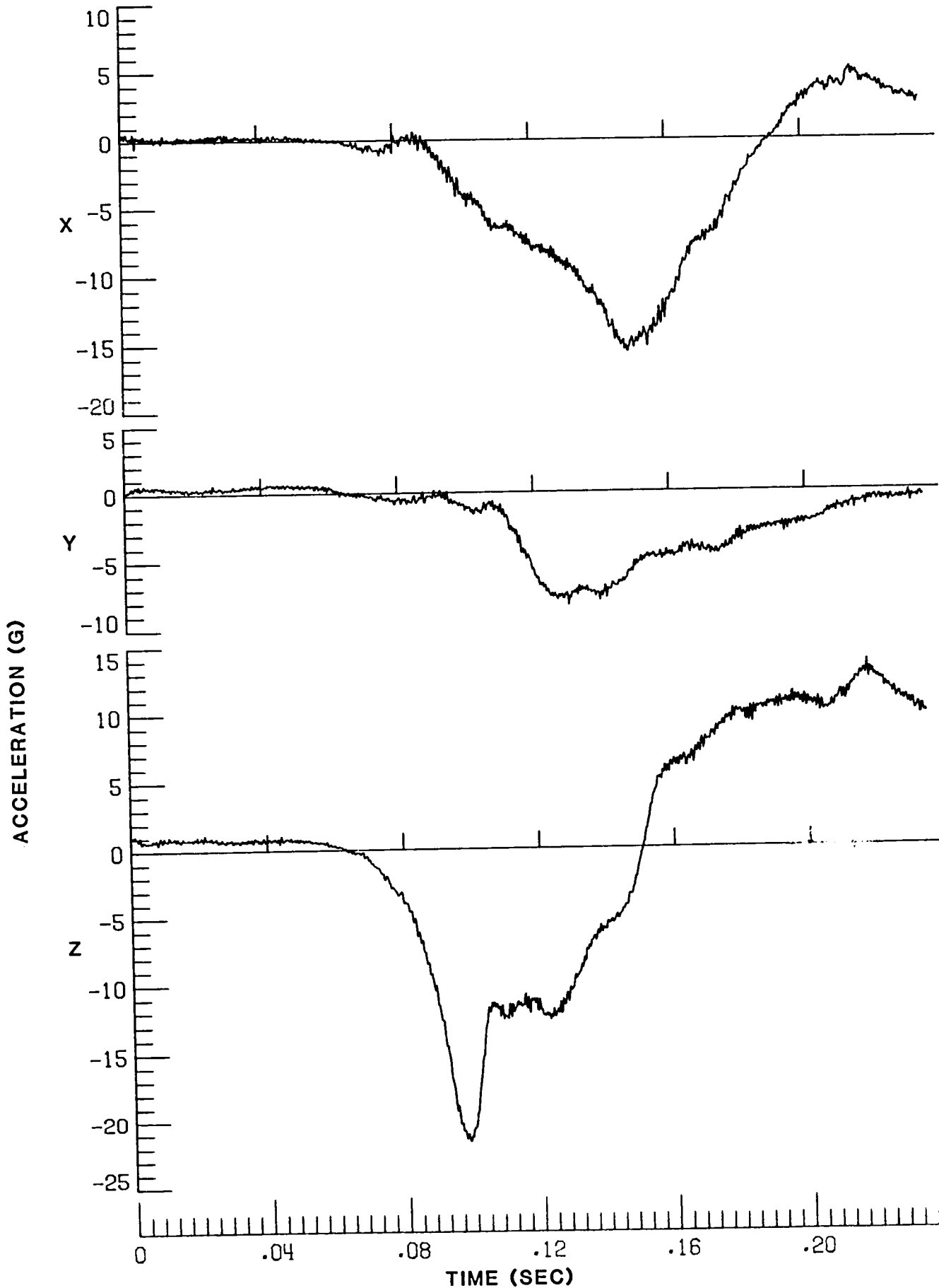
RIGHT DUMMY PELVIS  
NASA TEST NO. 2



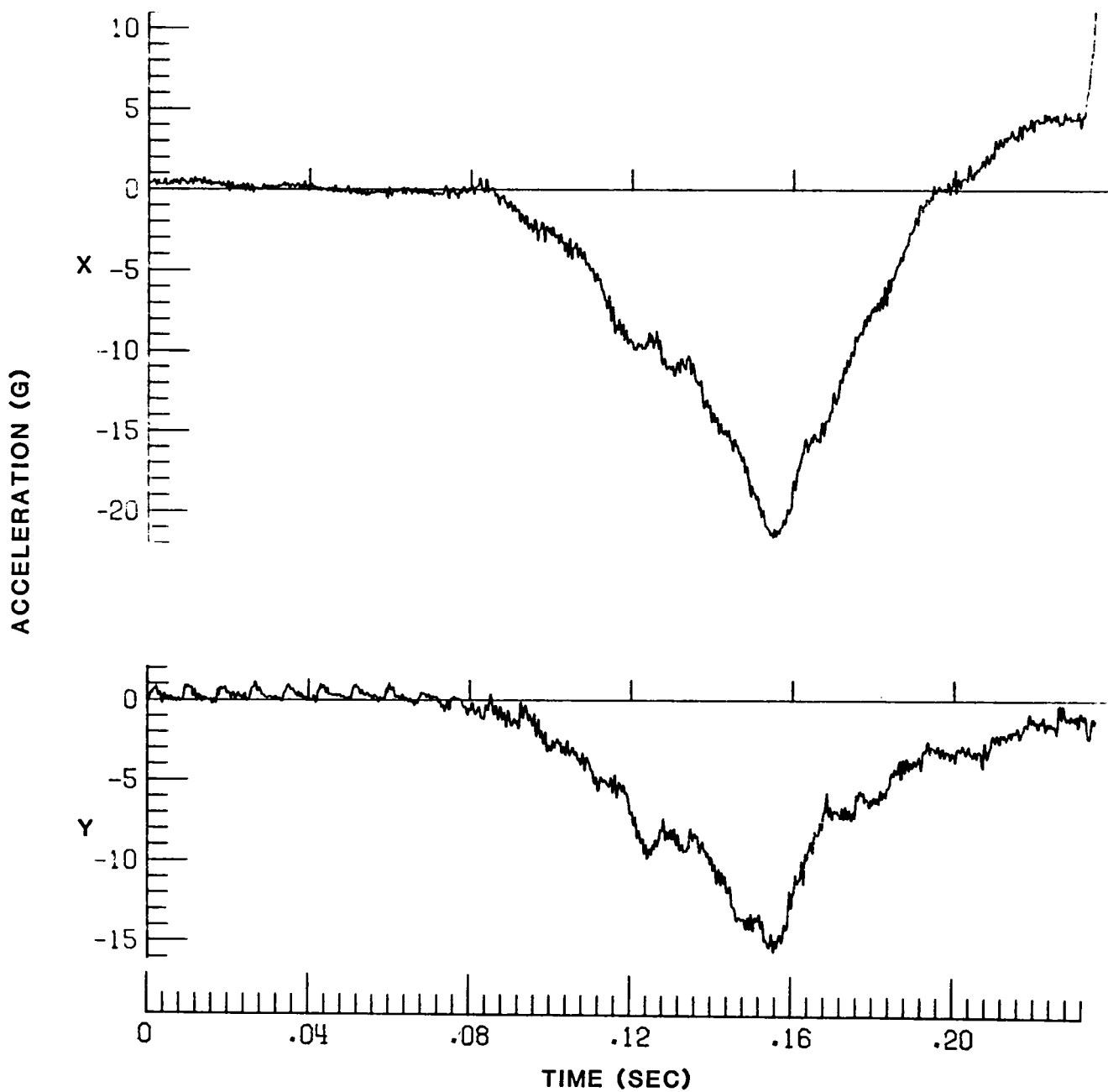
LEFT DUMMY CHEST  
NASA TEST NO. 2



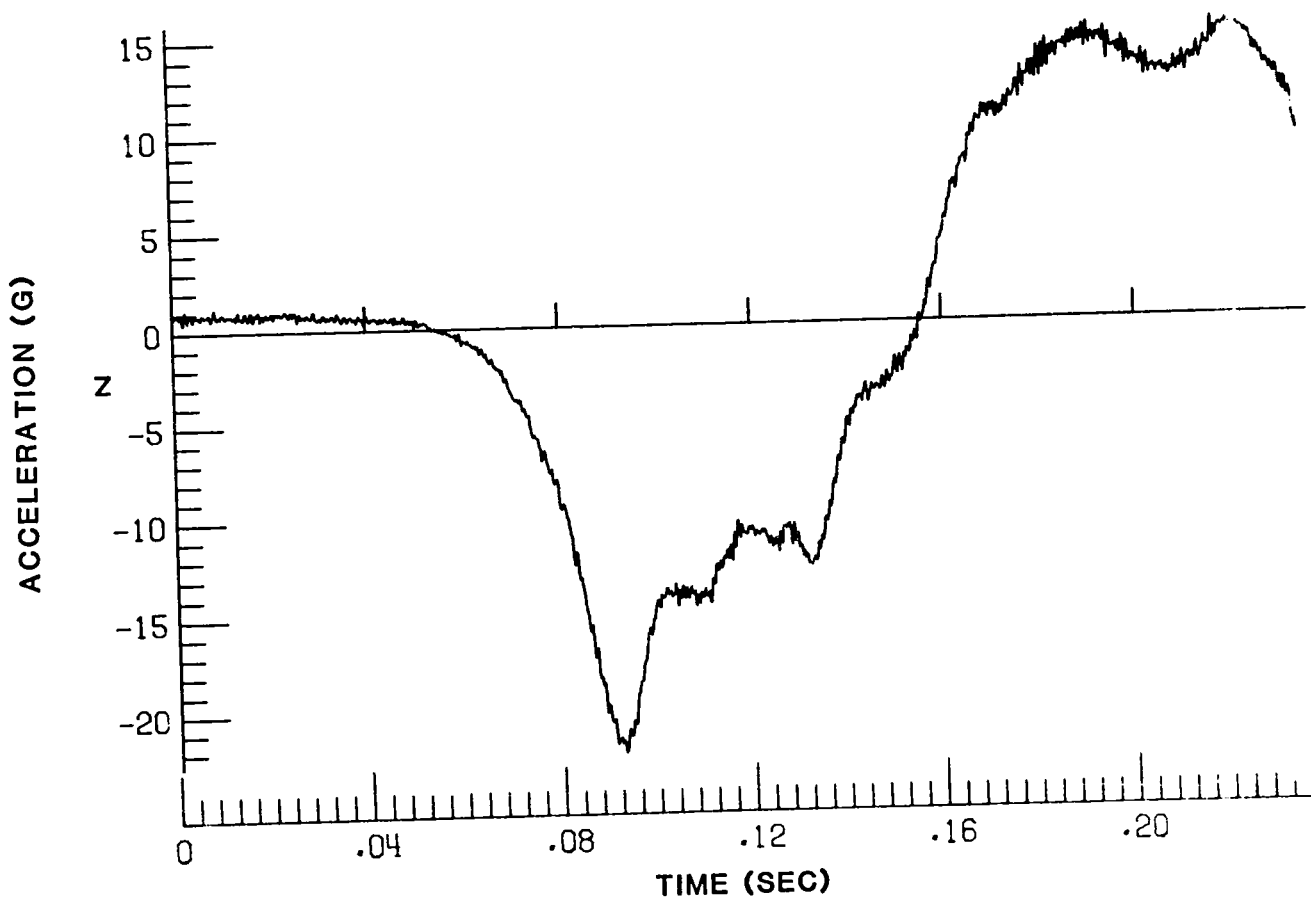
RIGHT DUMMY CHEST  
NASA TEST NO. 2



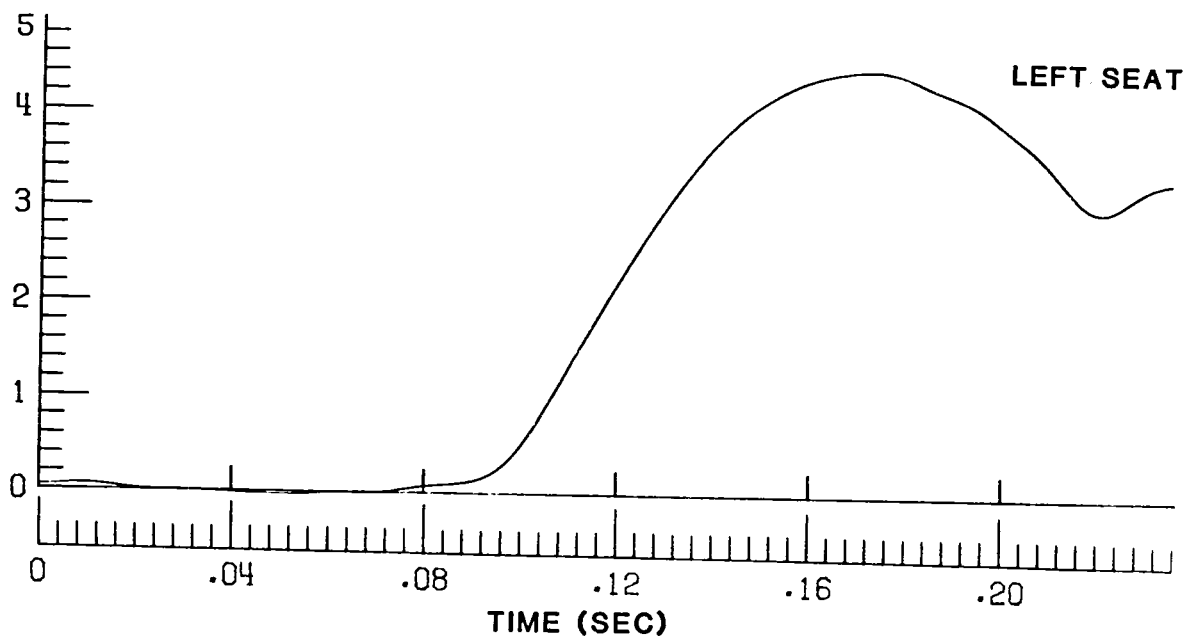
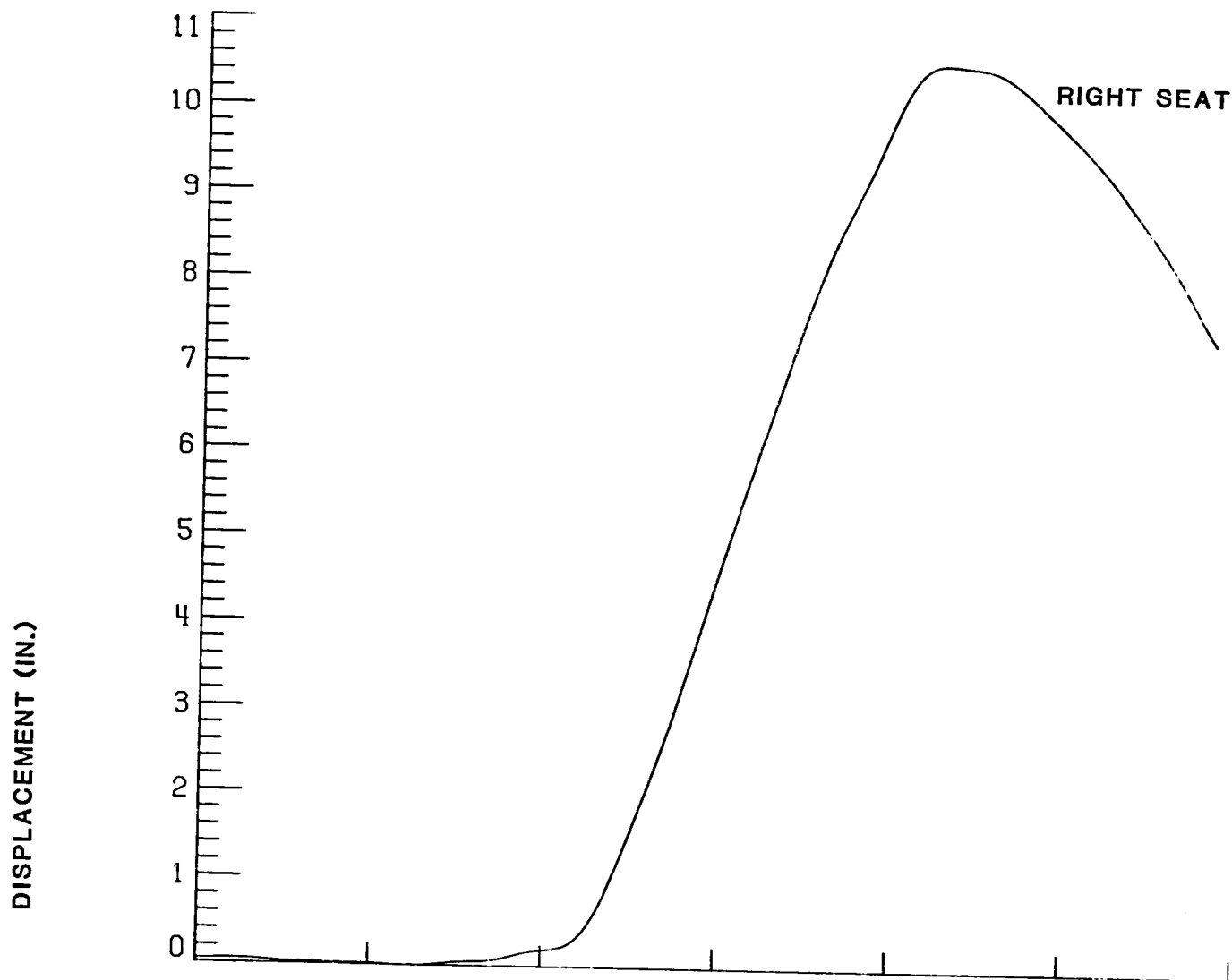
LEFT DUMMY HEAD  
NASA TEST NO. 2



RIGHT DUMMY HEAD  
NASA TEST NO. 2

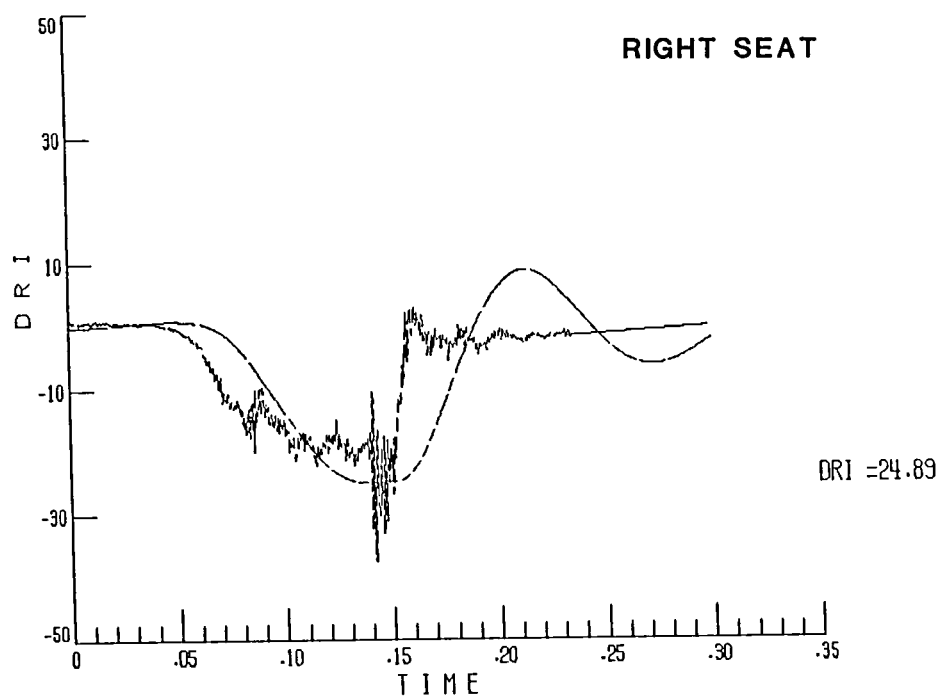
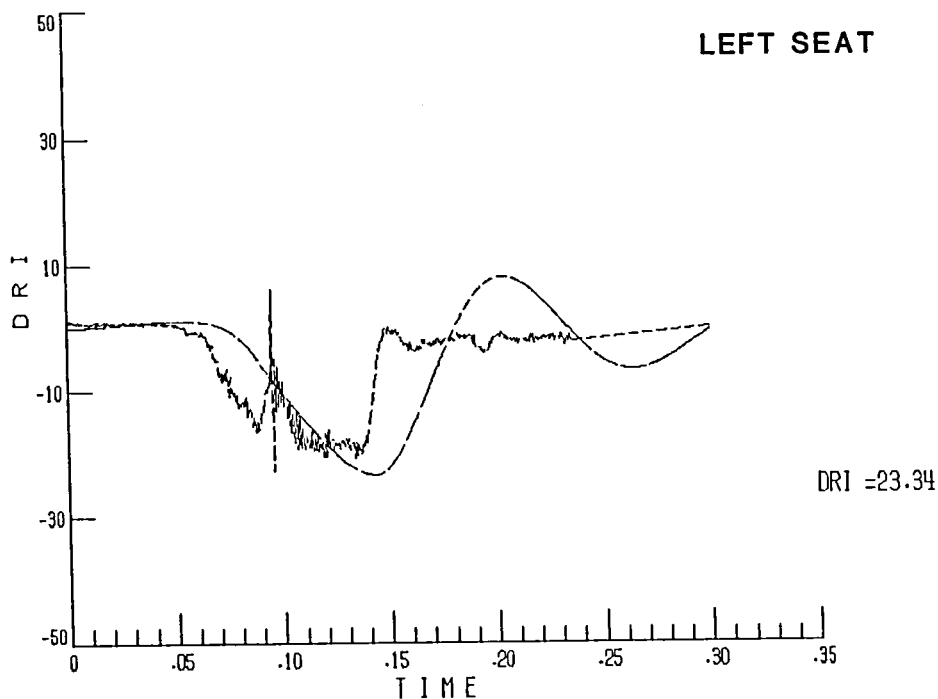


RIGHT DUMMY HEAD  
NASA TEST NO. 2

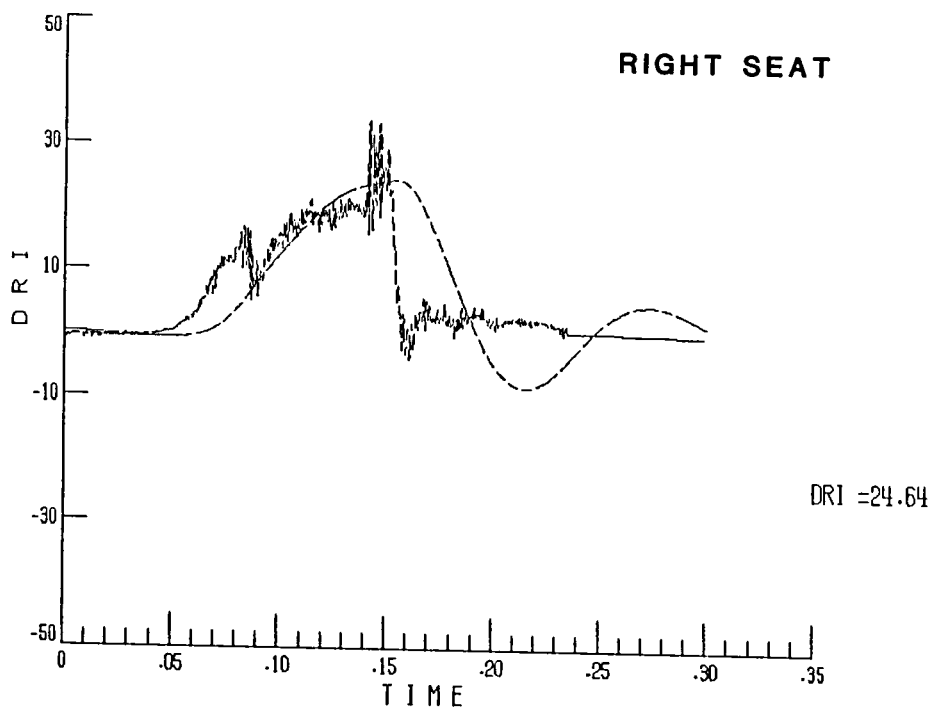
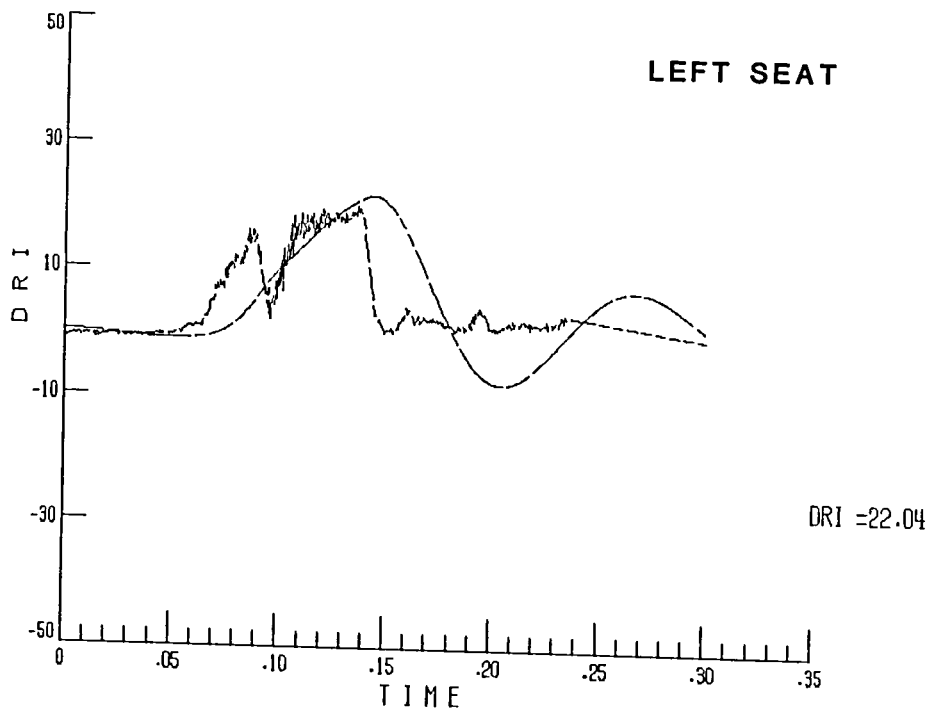


SEAT STROKE  
NASA TEST NO. 2

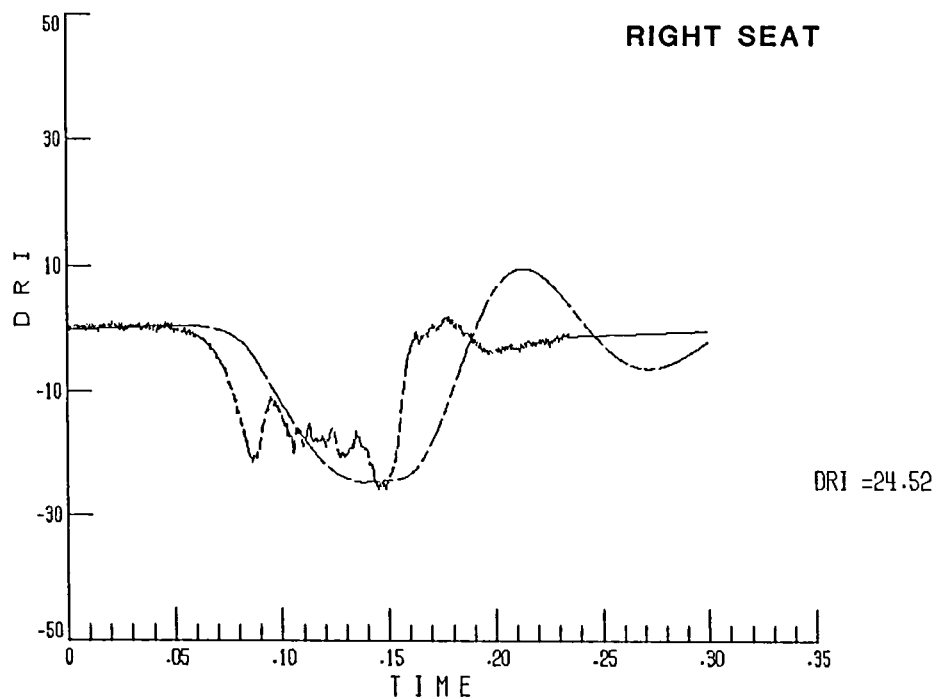
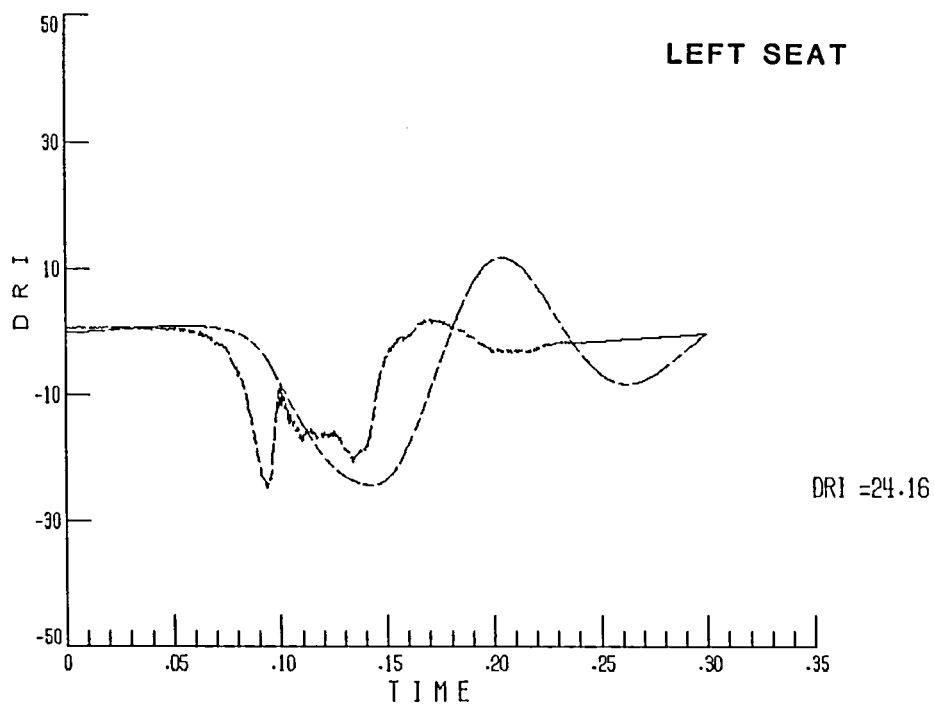




**DYNAMIC RESPONSE INDEX  
SEAT PAN Z-AXIS  
NASA TEST NO. 2**



**DYNAMIC RESPONSE INDEX  
BACK TANGENT LINE  
NASA TEST NO. 2**



DYNAMIC RESPONSE INDEX  
DUMMY PELVIS Z-AXIS  
NASA TEST NO. 2

#### 4.0 NASA TEST NO. 3

HORIZONTAL VELOCITY: 34 ft/sec

VERTICAL VELOCITY: 32 ft/sec

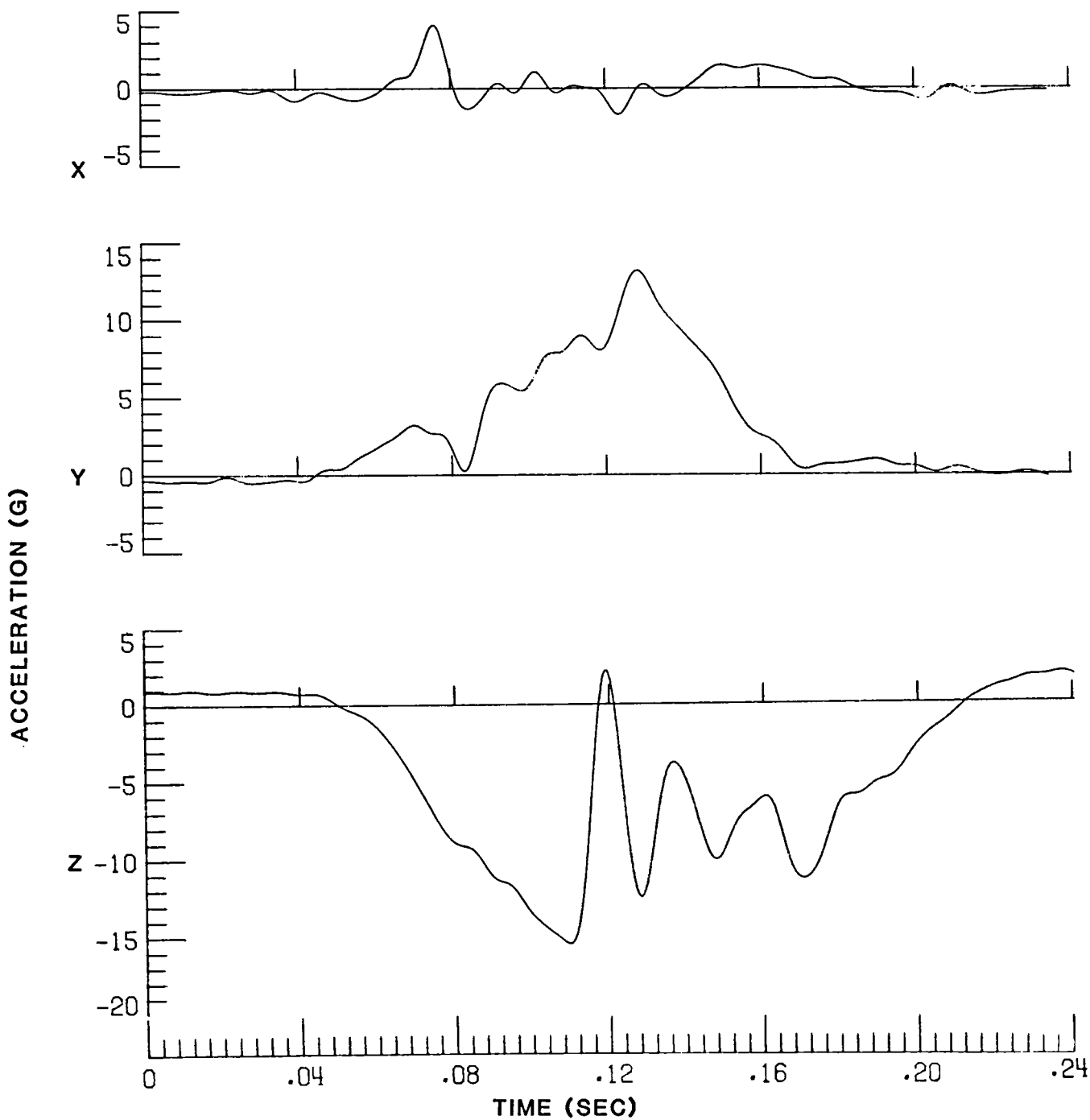
CREW MODULE PITCH:  $17^{\circ}$

CREW MODULE ROLL:  $0^{\circ}$

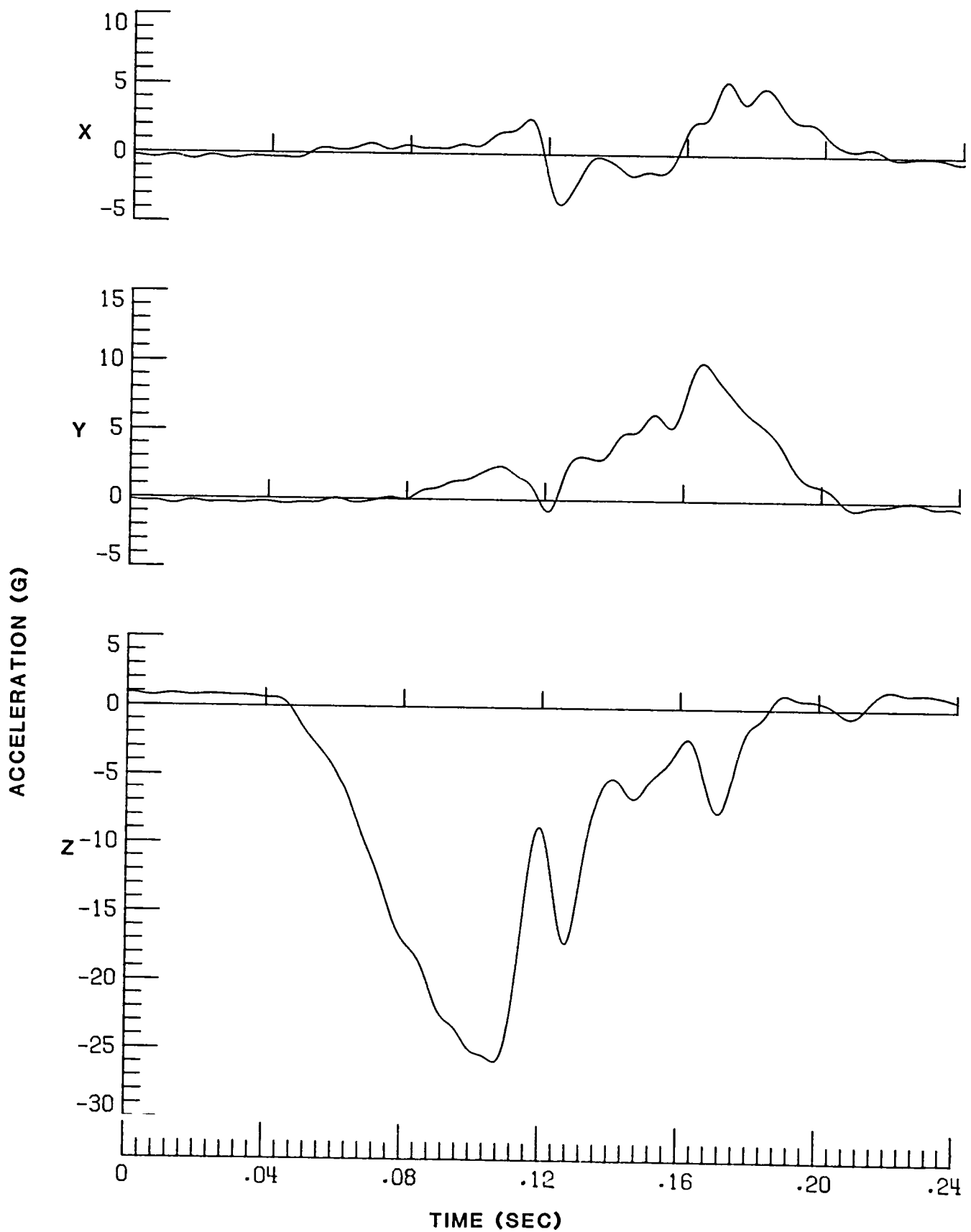
CREW MODULE YAW:  $90^{\circ}$

LEFT SEAT: Energy Absorbing

RIGHT SEAT: Energy Absorbing

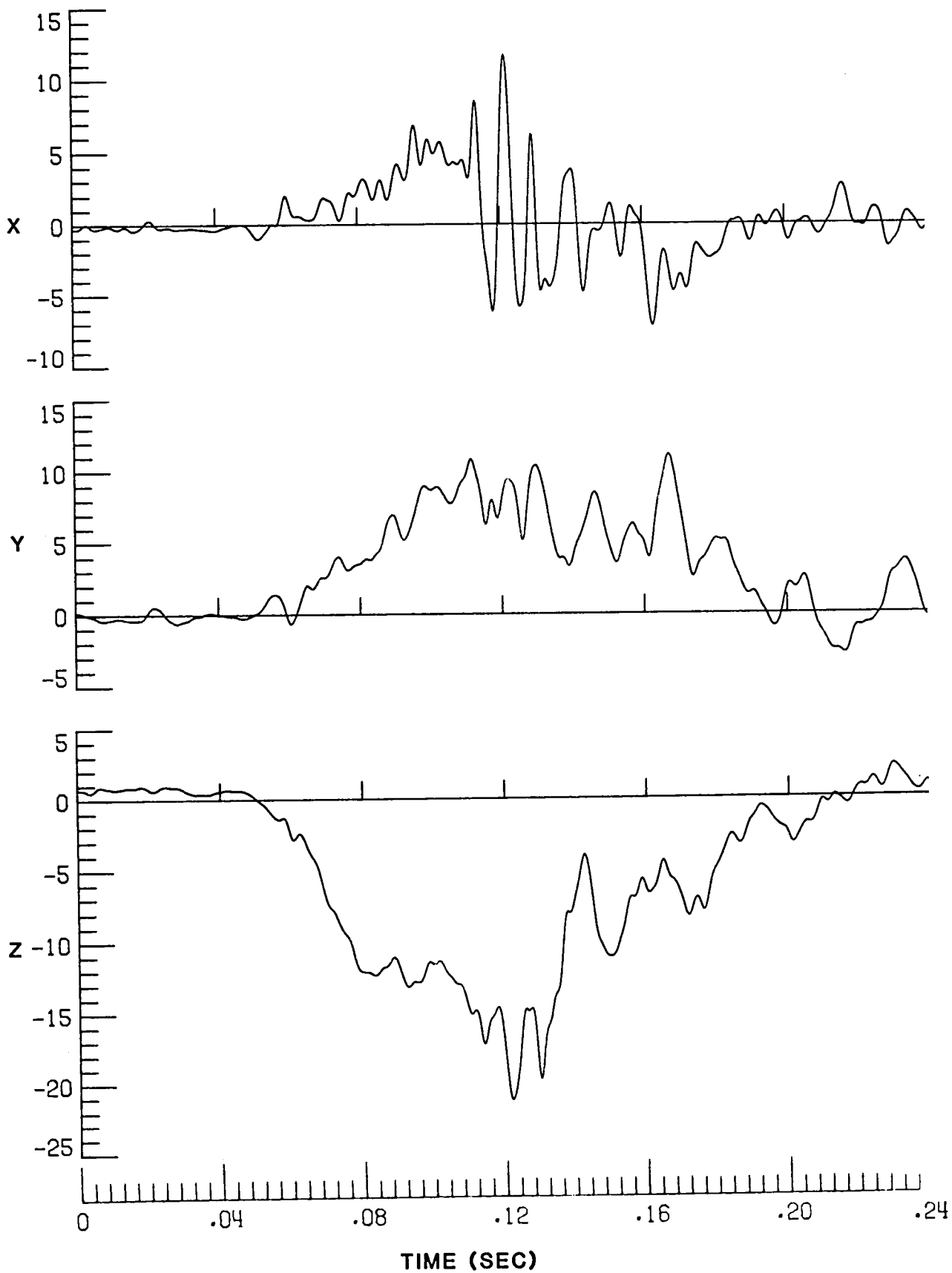


LEFT BULKHEAD  
NASA TEST NO. 3

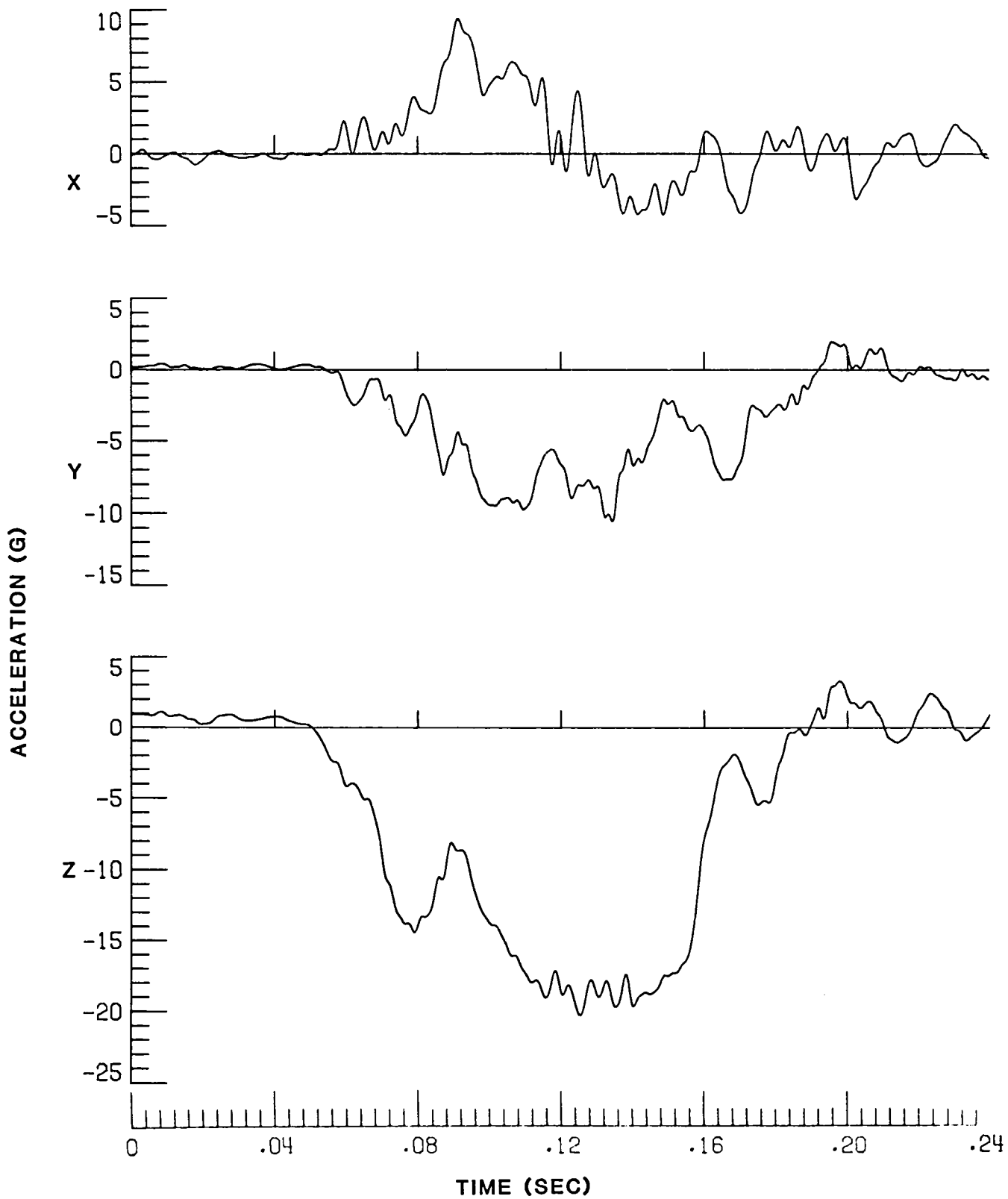


RIGHT BULKHEAD  
NASA TEST NO. 3

ACCELERATION (G)

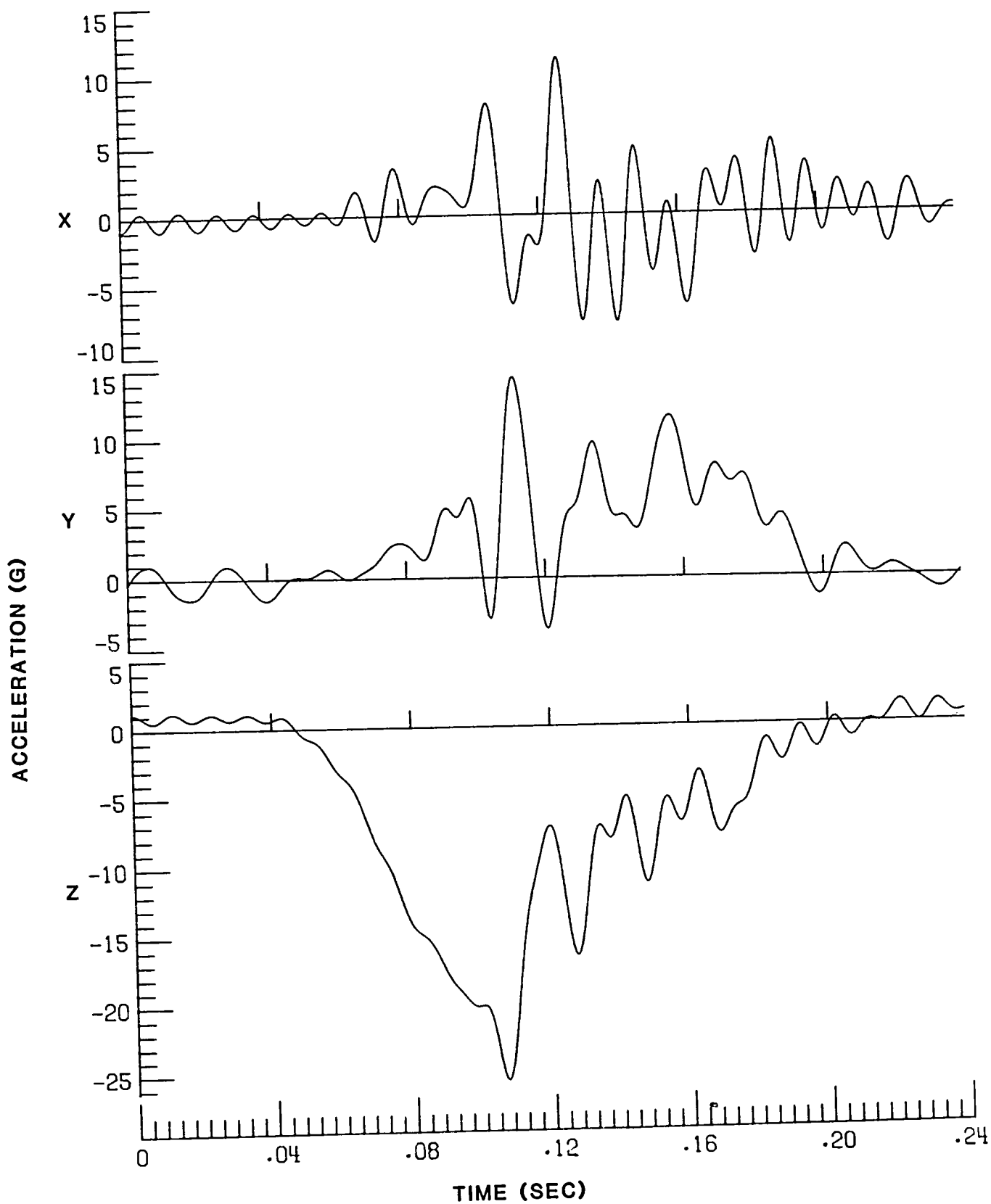


LEFT SEAT PAN  
NASA TEST NO. 3

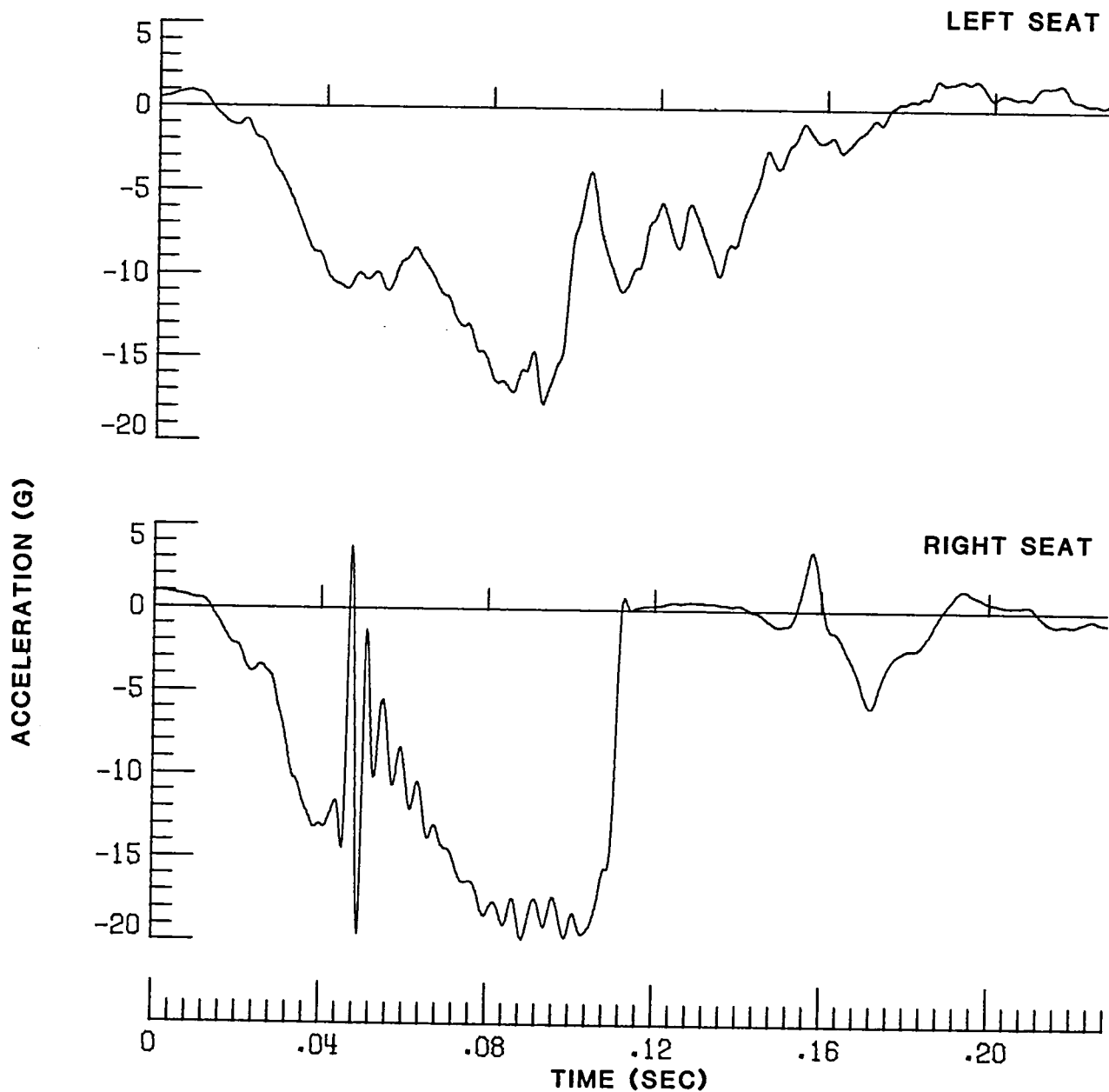


RIGHT SEAT PAN  
NASA TEST NO. 3

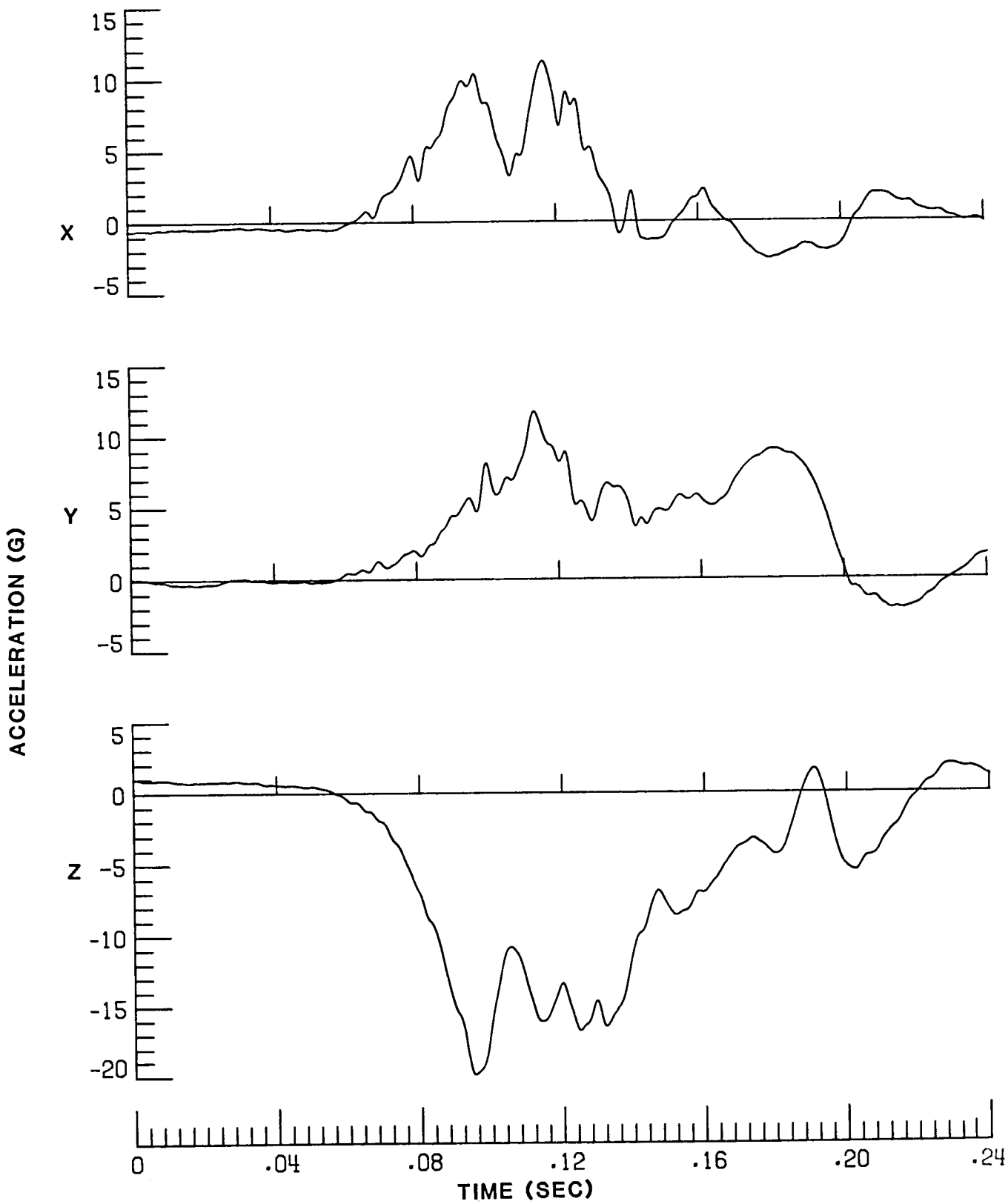




CAPSULE C.G.  
NASA TEST NO. 3

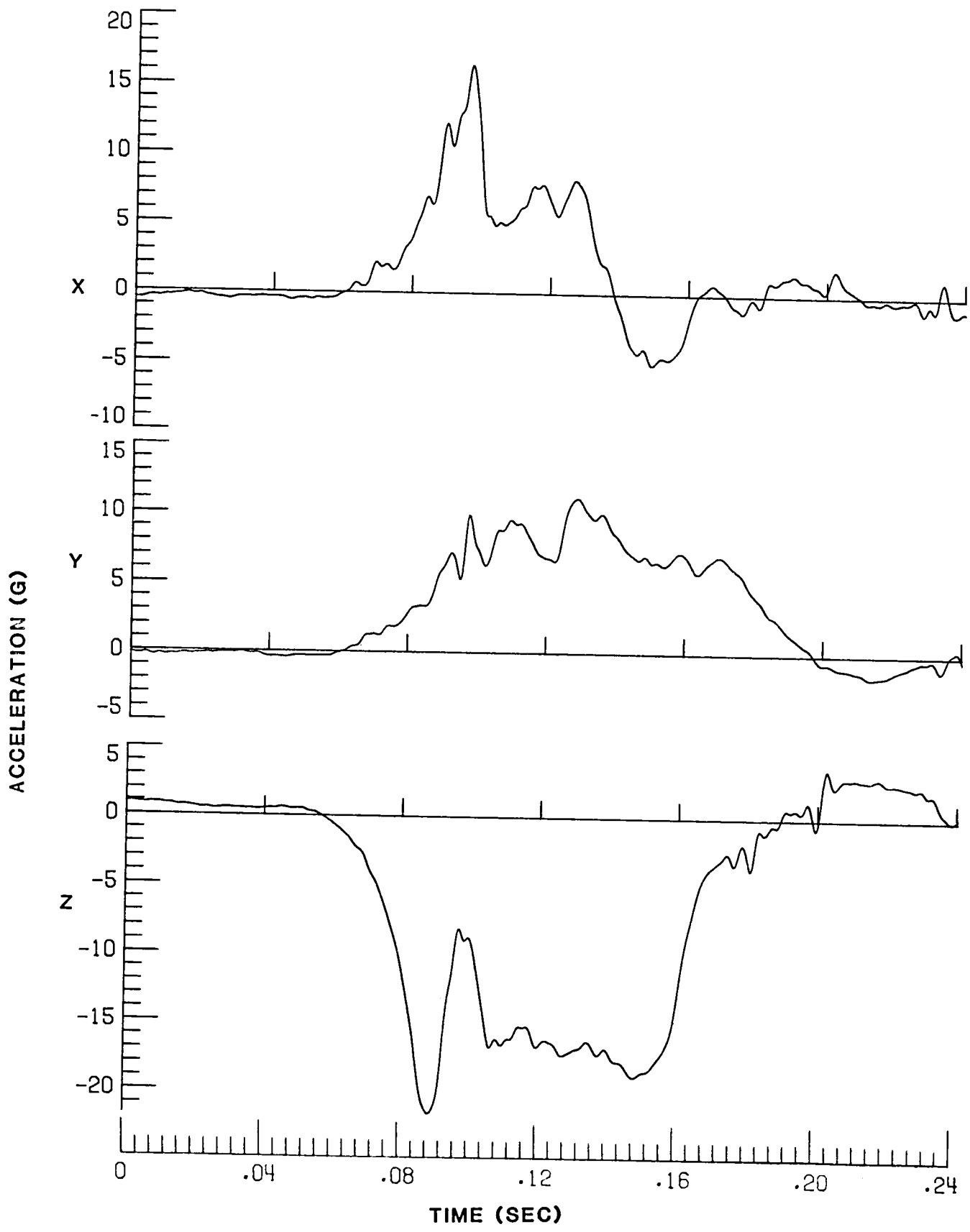


SEAT PAN ALONG BACK TANGENT LINE  
NASA TEST NO. 3



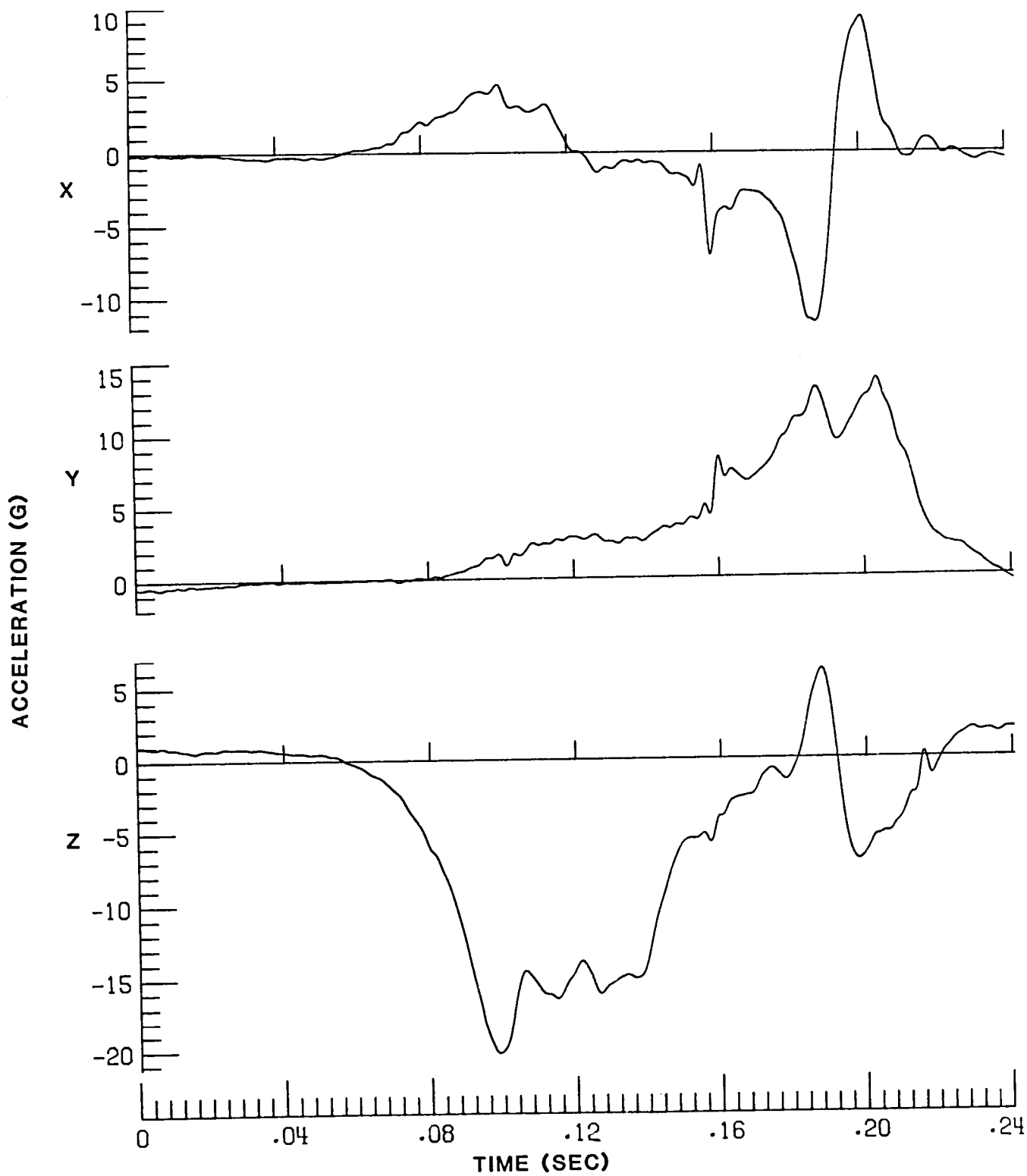
LEFT DUMMY PELVIS

NASA TEST NO. 3

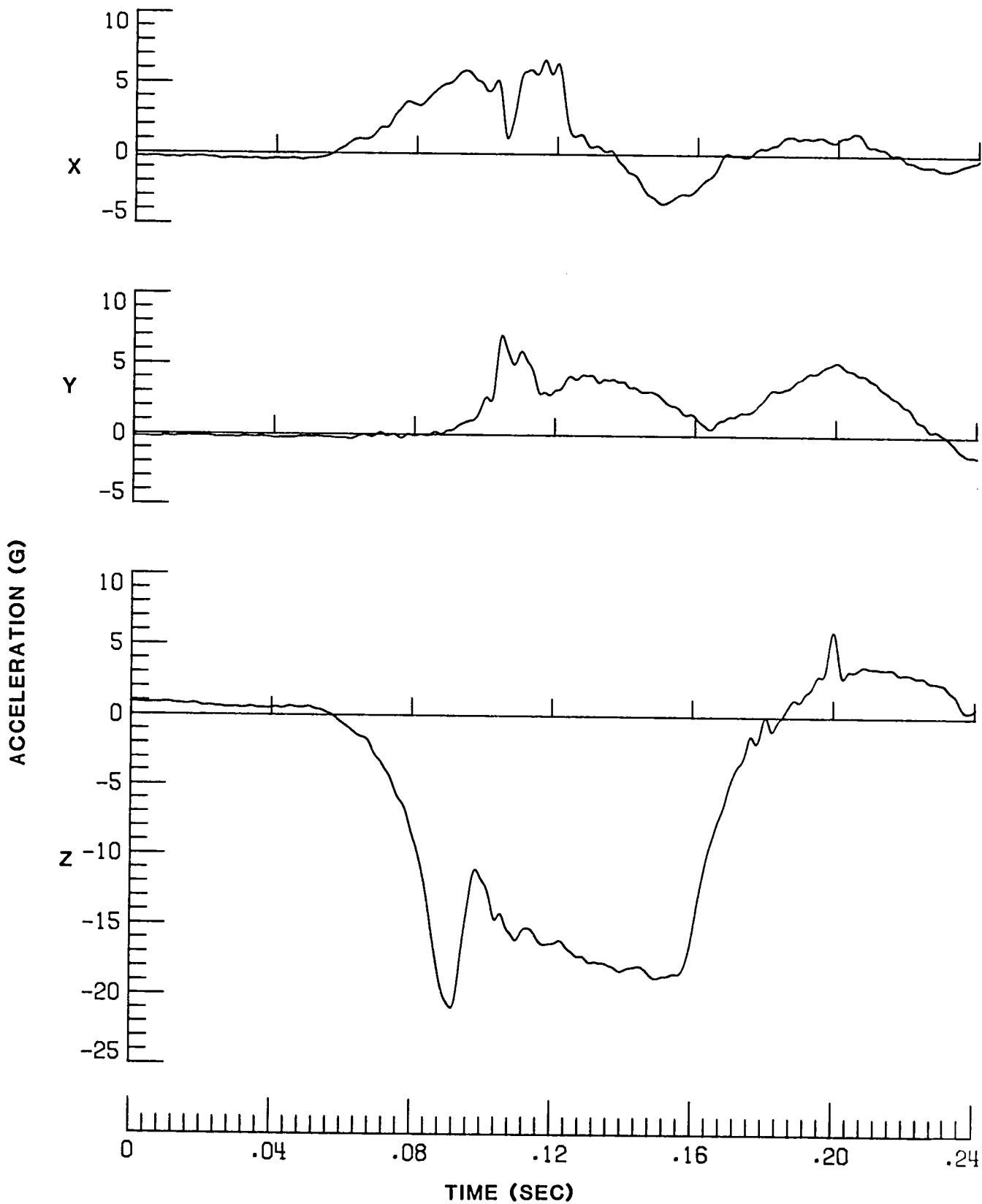


RIGHT DUMMY PELVIS

NASA TEST NO. 3



LEFT DUMMY CHEST  
NASA TEST NO. 3

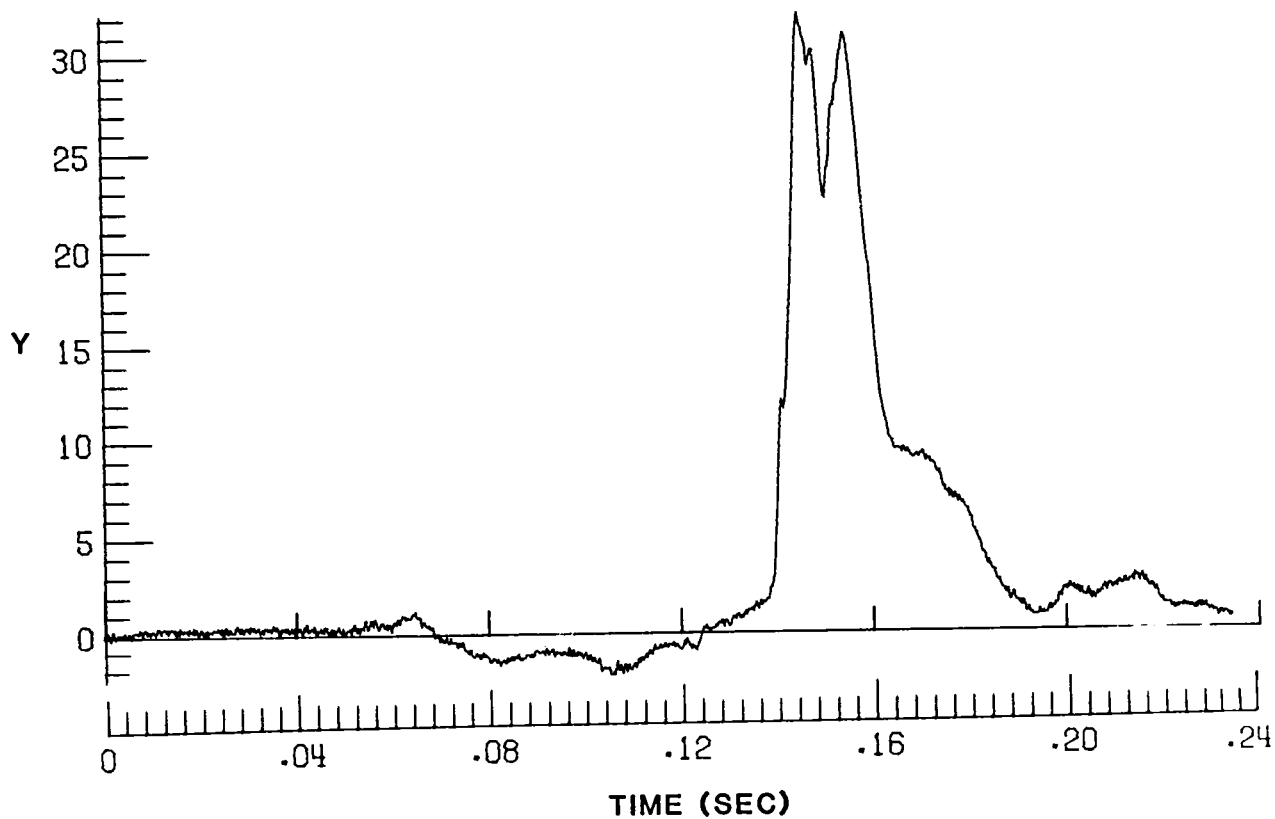


RIGHT DUMMY CHEST

NASA TEST NO. 3

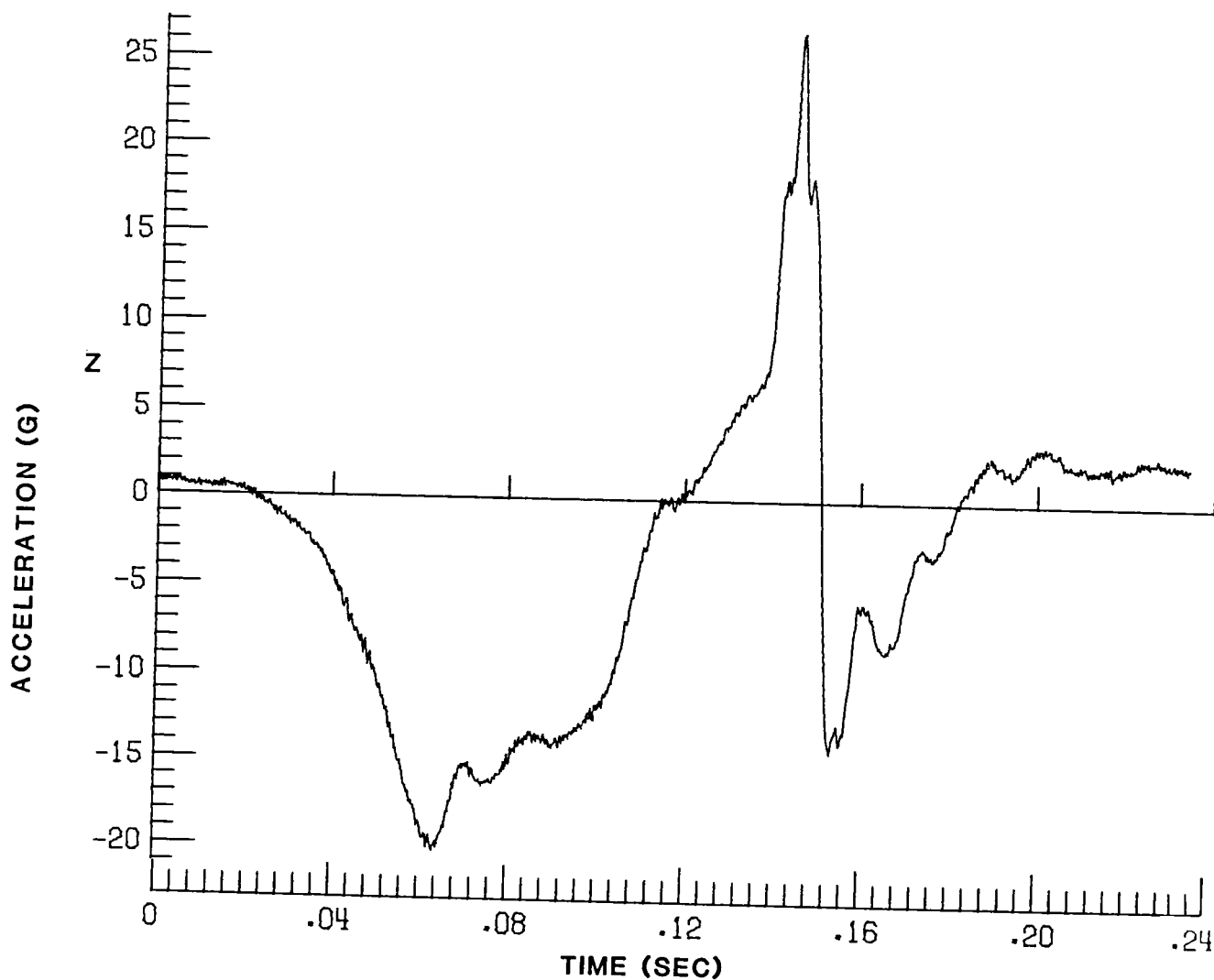


ACCELERATION (G)



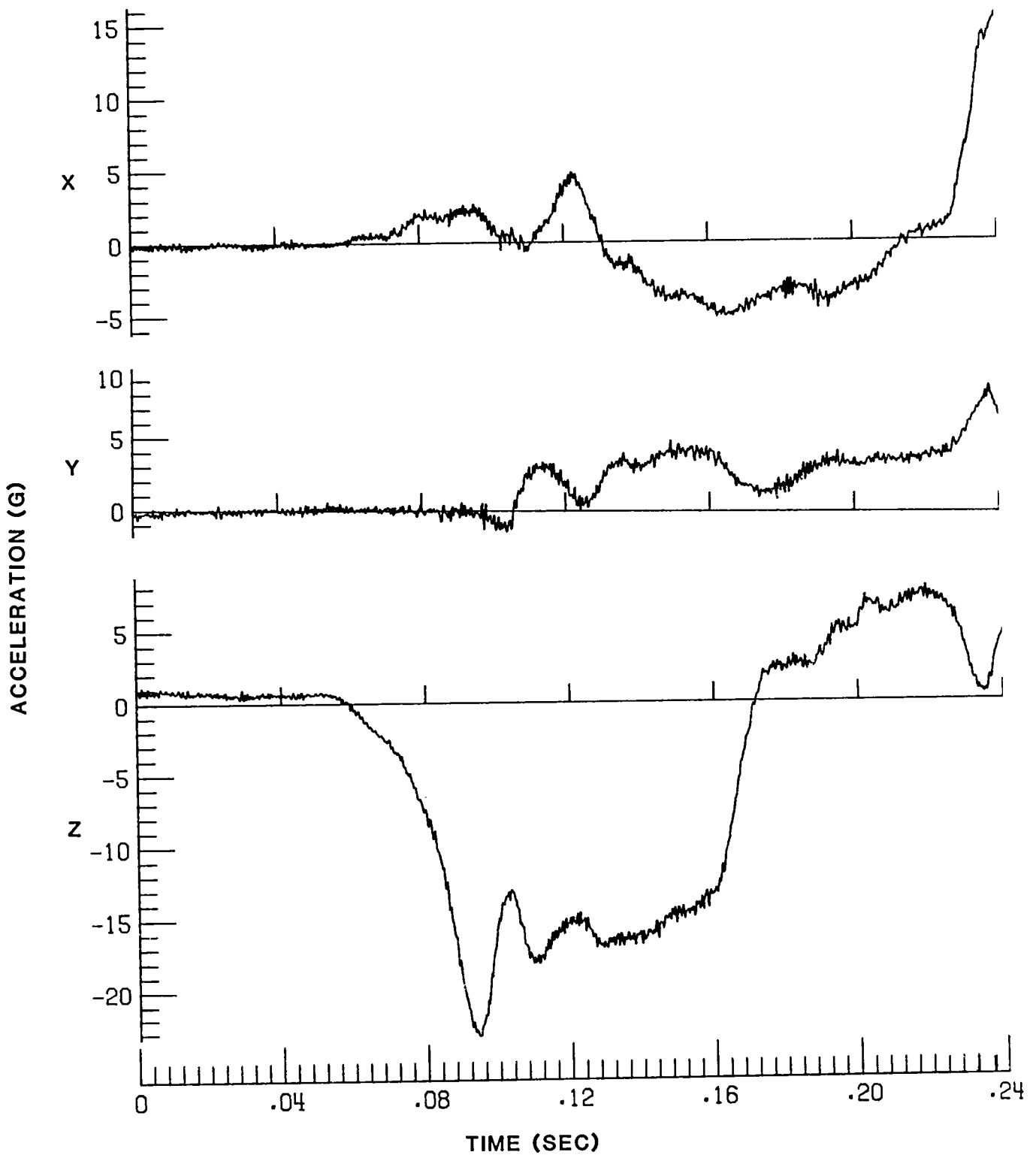
LEFT DUMMY HEAD

NASA TEST NO. 3

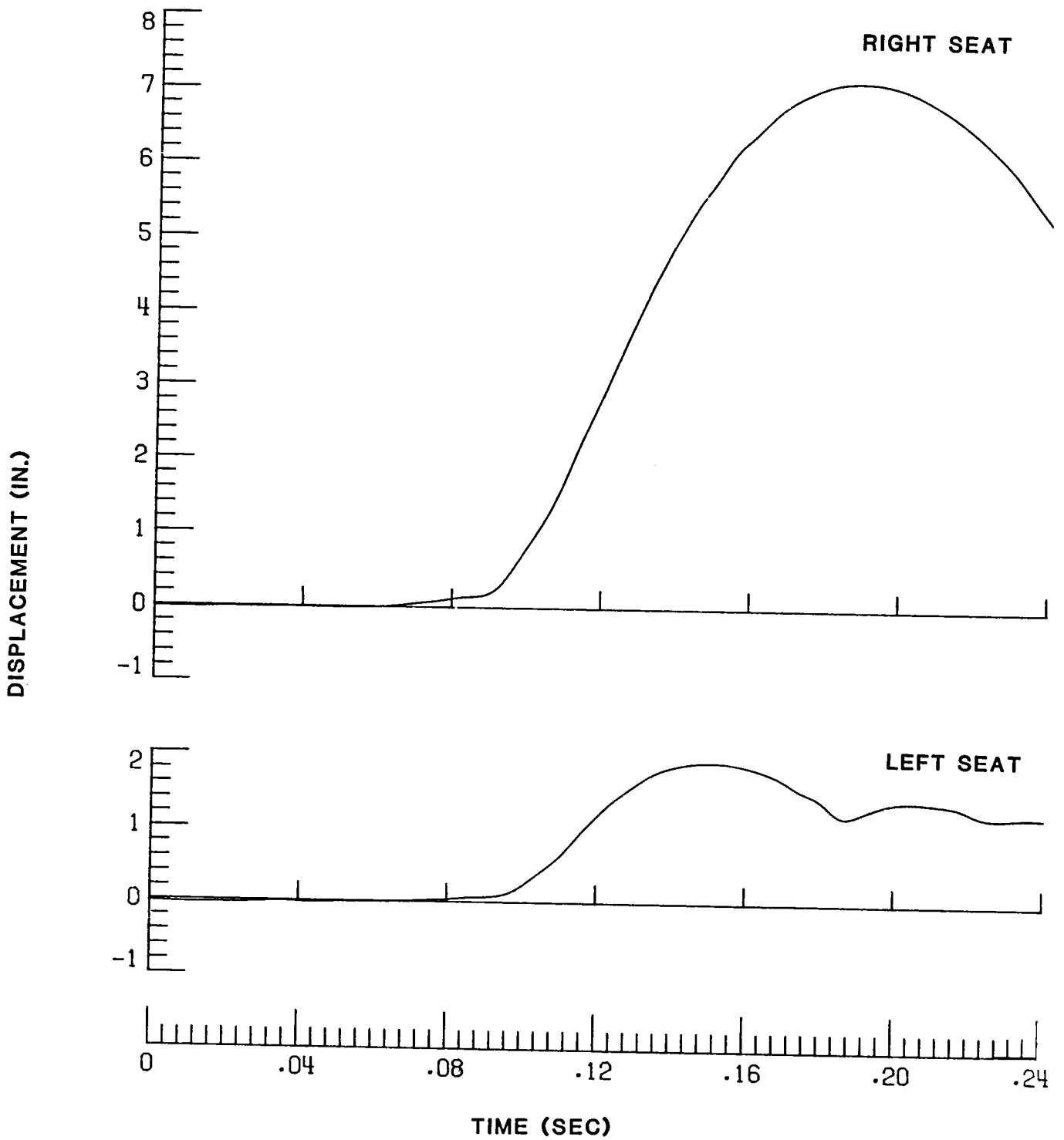


LEFT DUMMY HEAD  
NASA TEST NO. 3

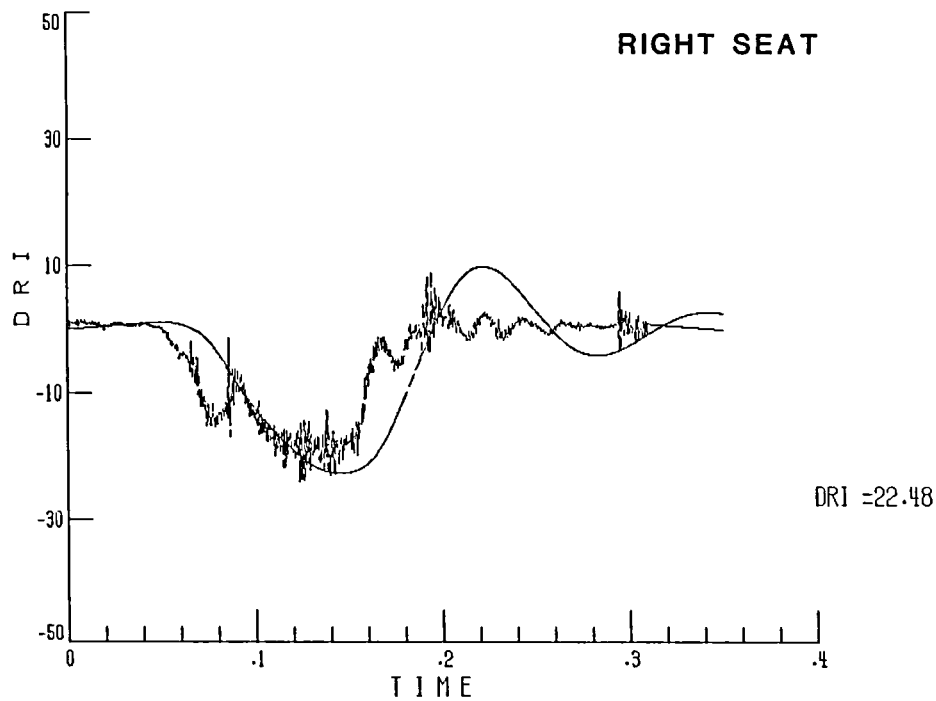
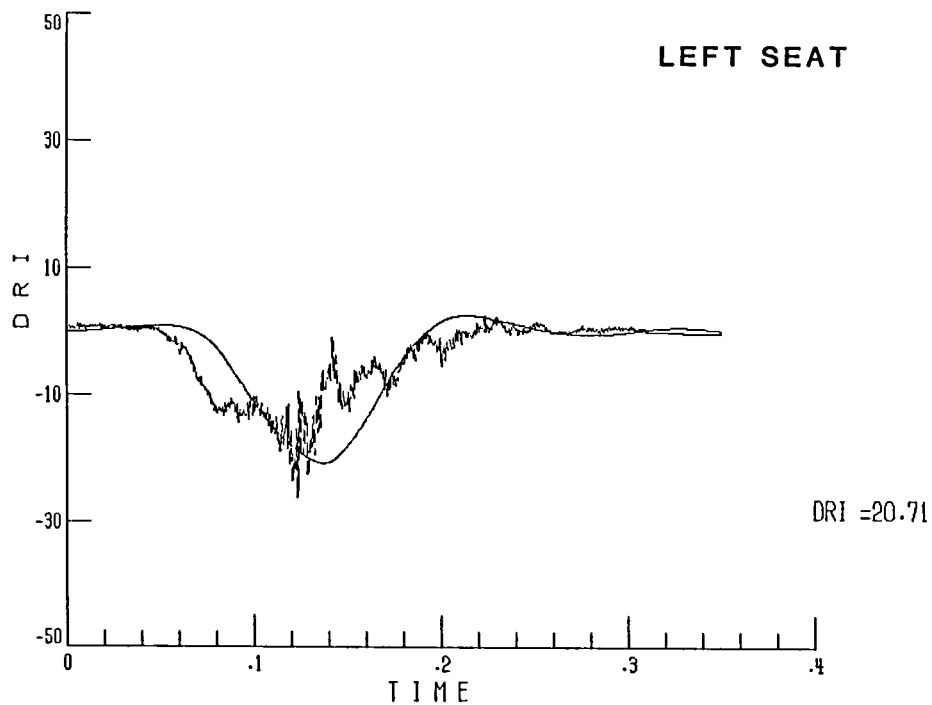




RIGHT DUMMY HEAD  
NASA TEST NO. 3



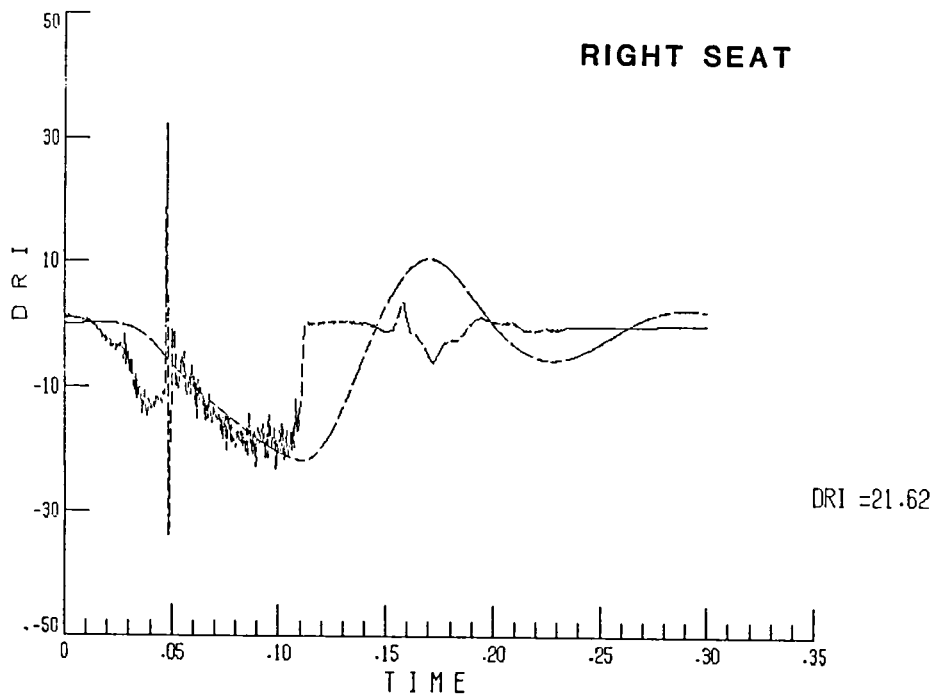
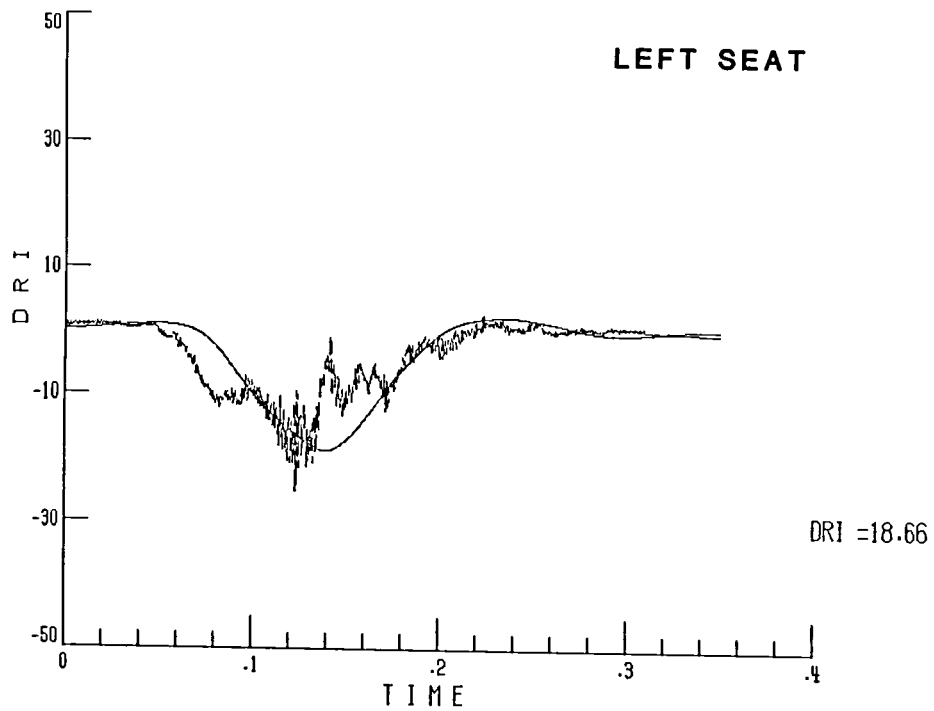
SEAT STROKE  
NASA TEST NO. 3



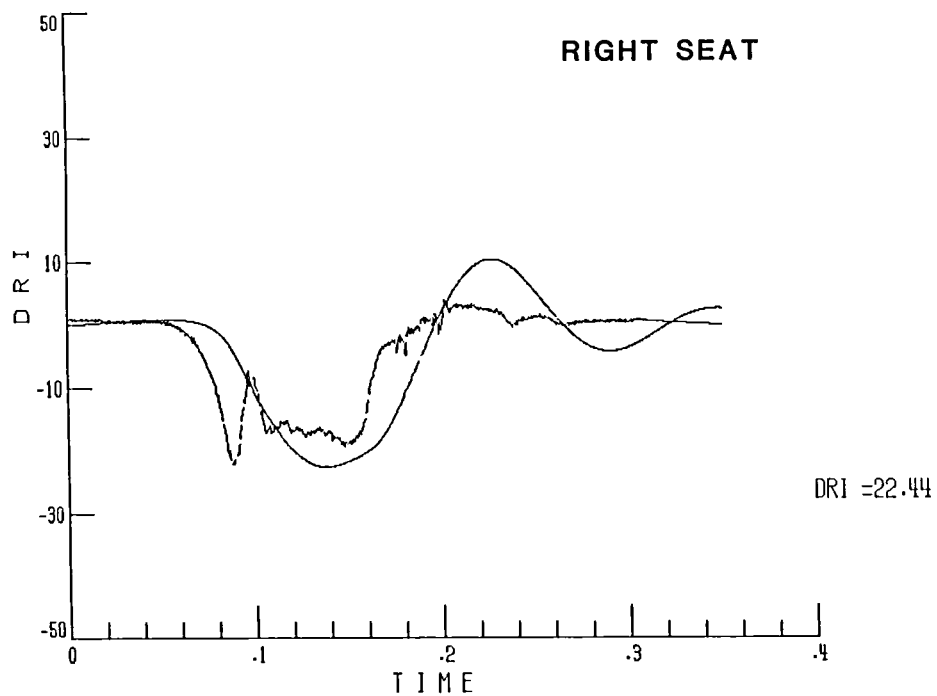
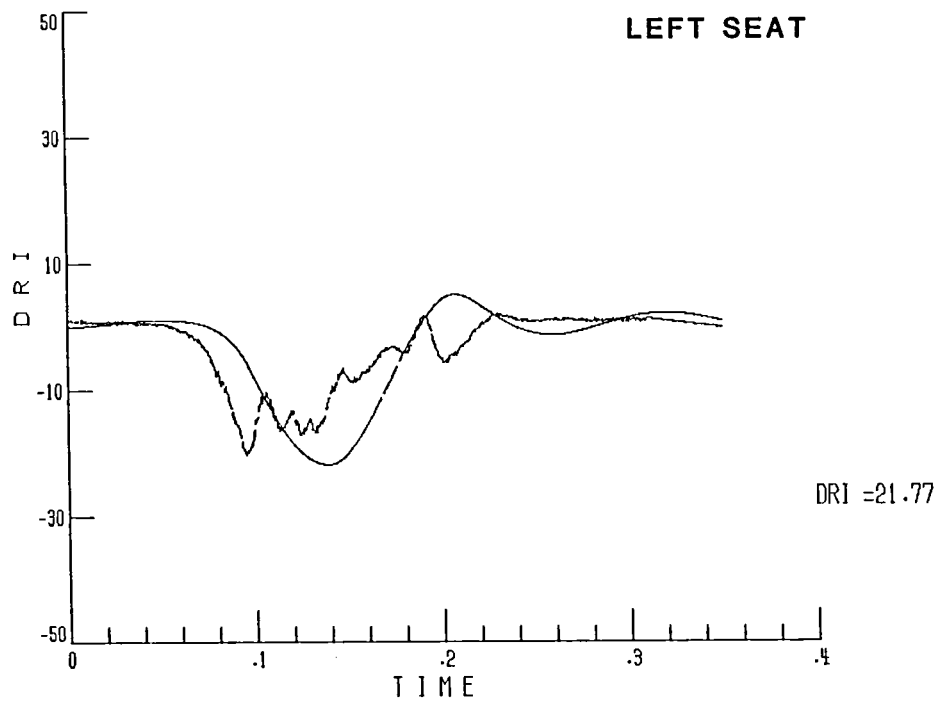
DYNAMIC RESPONSE INDEX

SEAT PAN Z-AXIS

NASA TEST NO. 3



**DYNAMIC RESPONSE INDEX  
BACK TANGENT LINE  
NASA TEST NO. 3**



**DYNAMIC RESPONSE INDEX  
DUMMY PELVIS Z-AXIS  
NASA TEST NO. 3**

5.0 NASA TEST NO. 4

HORIZONTAL VELOCITY: 34 ft/sec

VERTICAL VELOCITY: 32 ft/sec

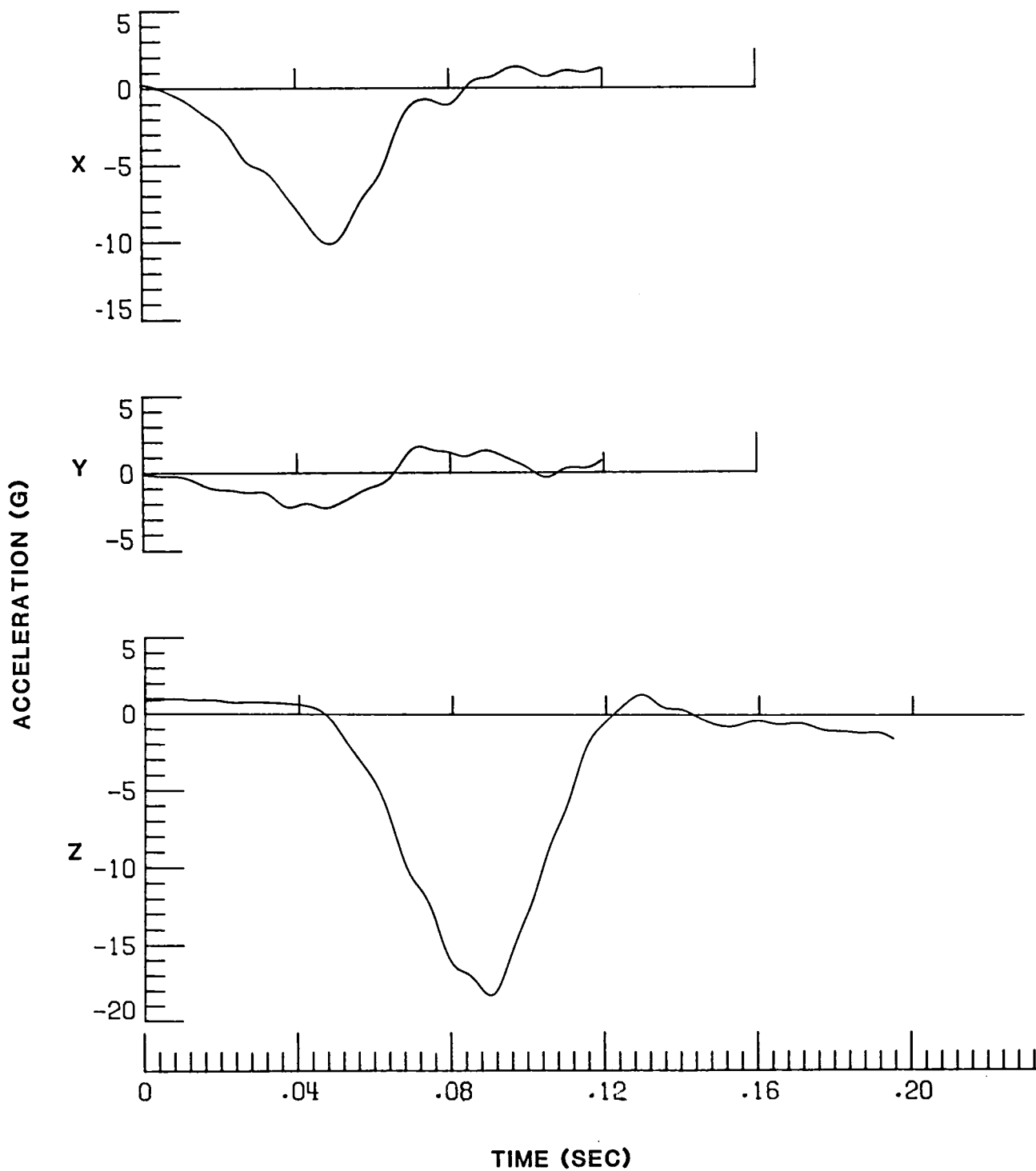
CREW MODULE PITCH:  $-13^{\circ}$

CREW MODULE ROLL:  $0^{\circ}$

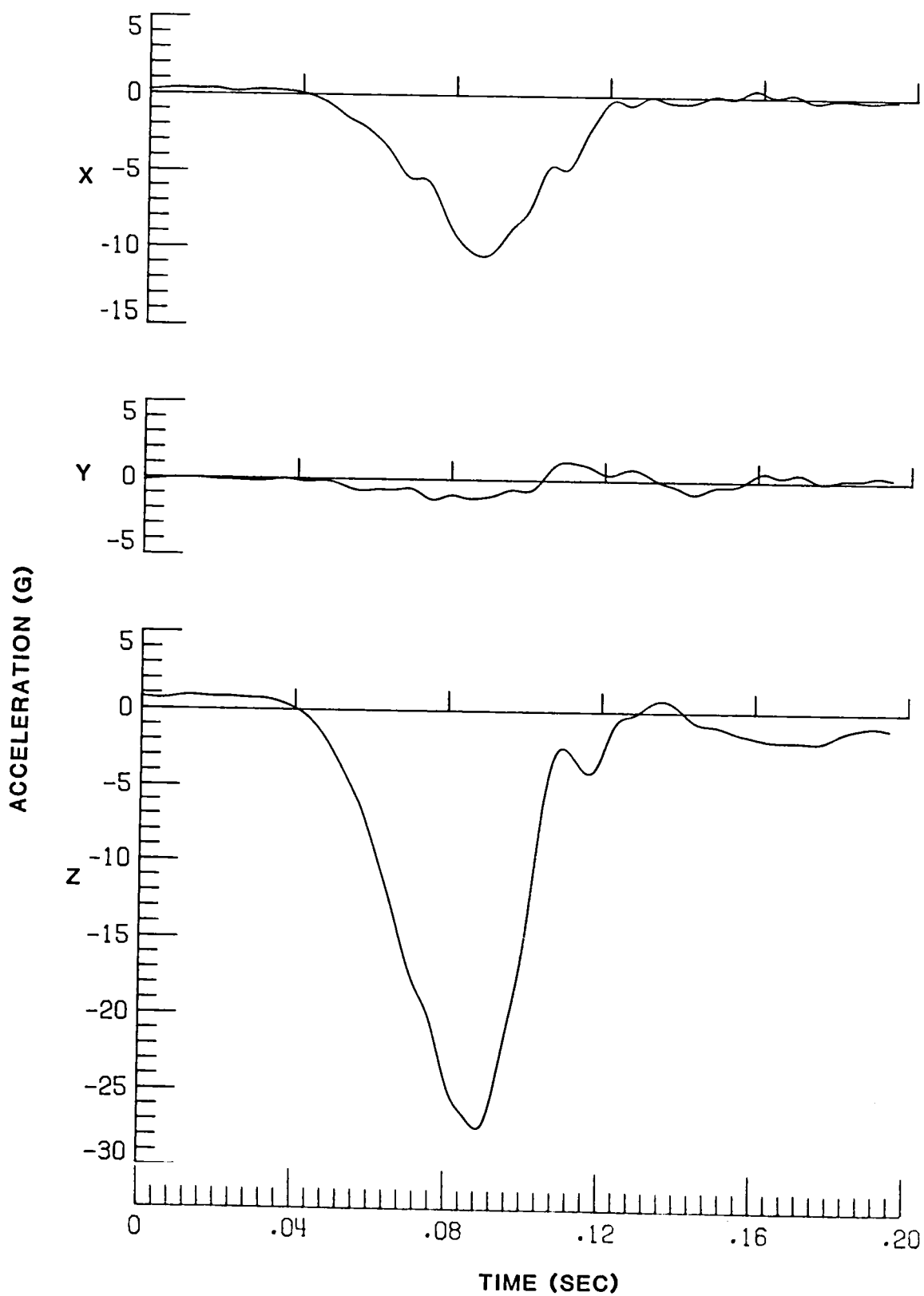
CREW MODULE YAW:  $75^{\circ}$

LEFT SEAT: Energy Absorbing

RIGHT SEAT: Energy Absorbing

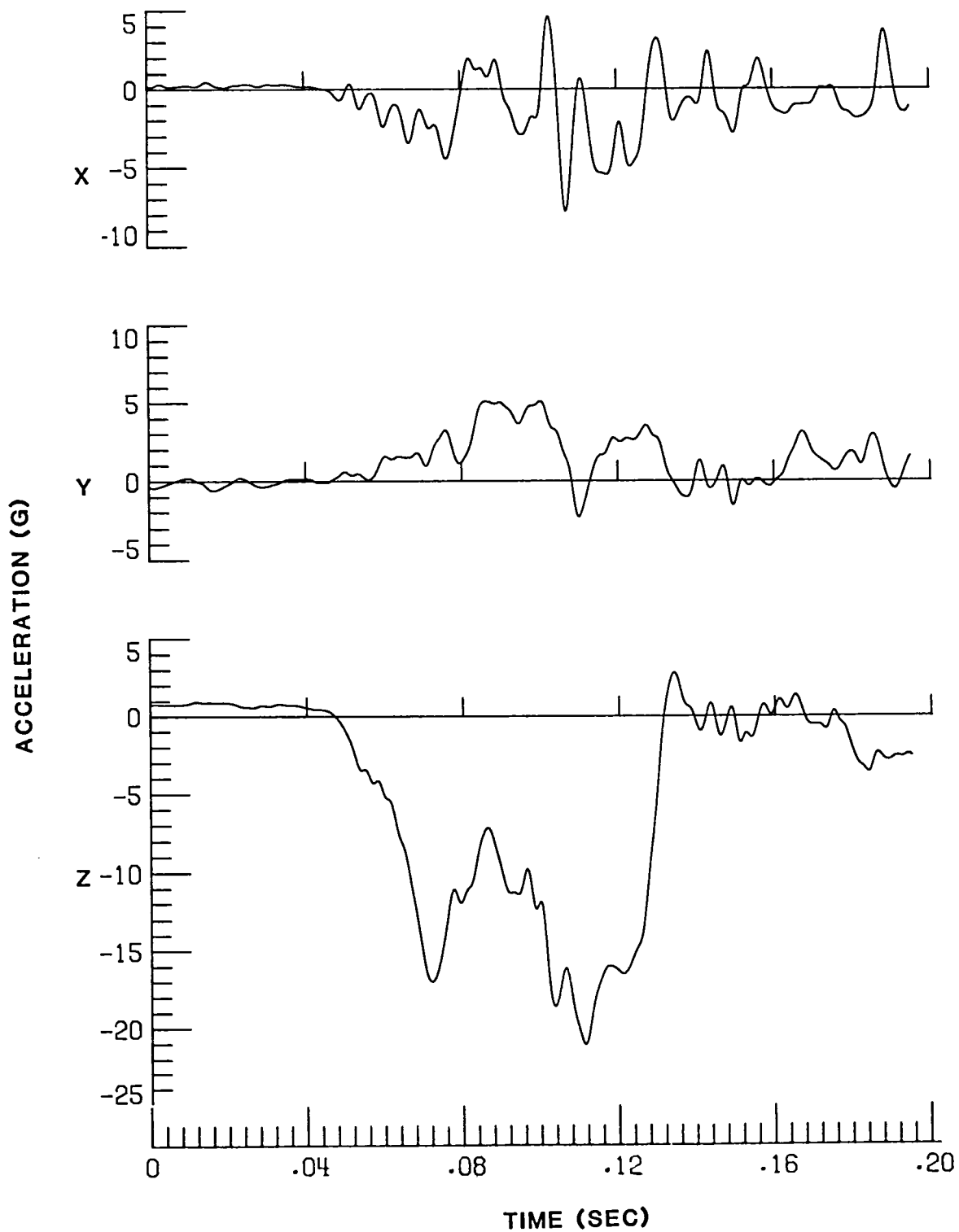


LEFT BULKHEAD  
NASA TEST NO. 4

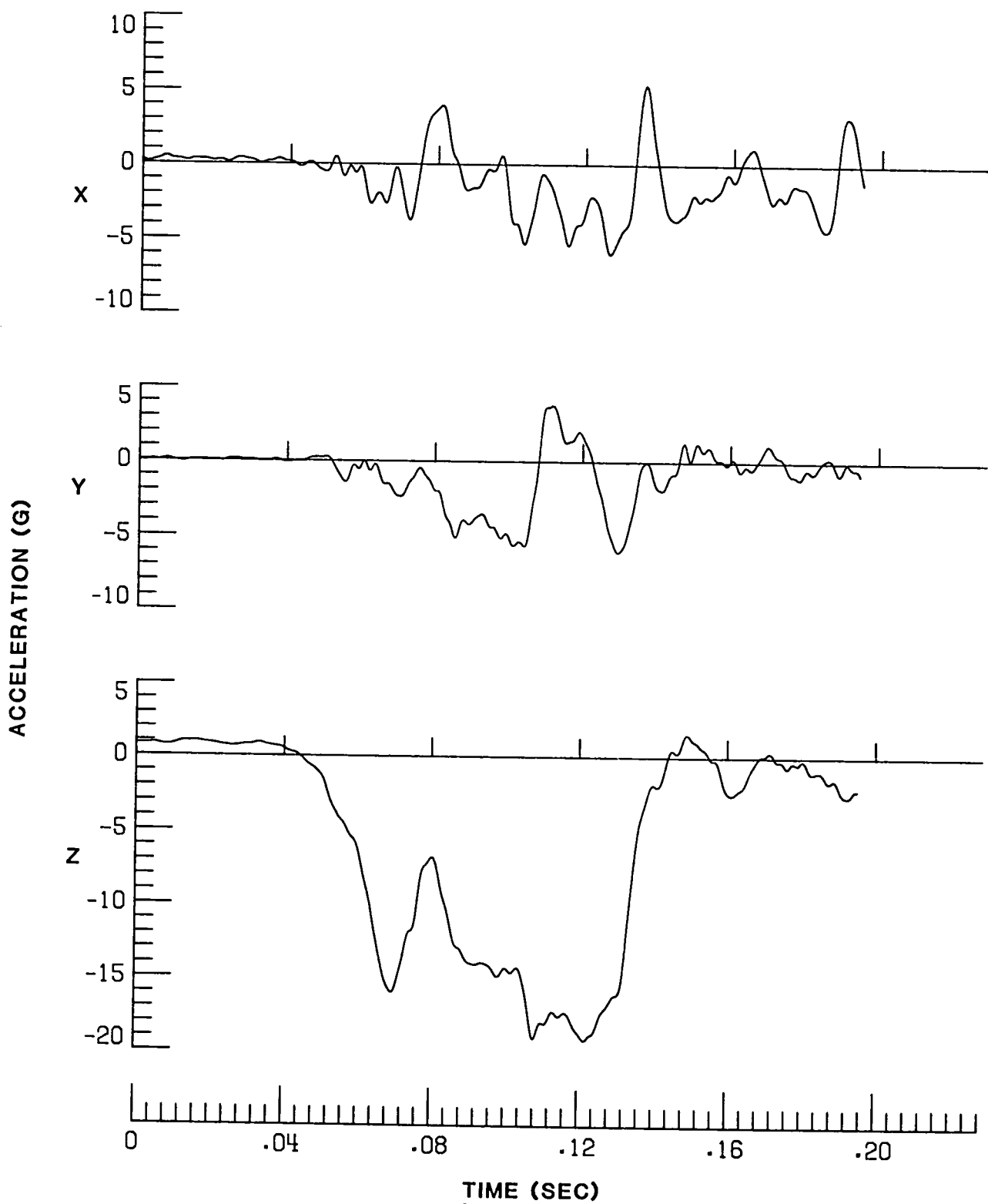


RIGHT BULKHEAD  
NASA TEST NO. 4

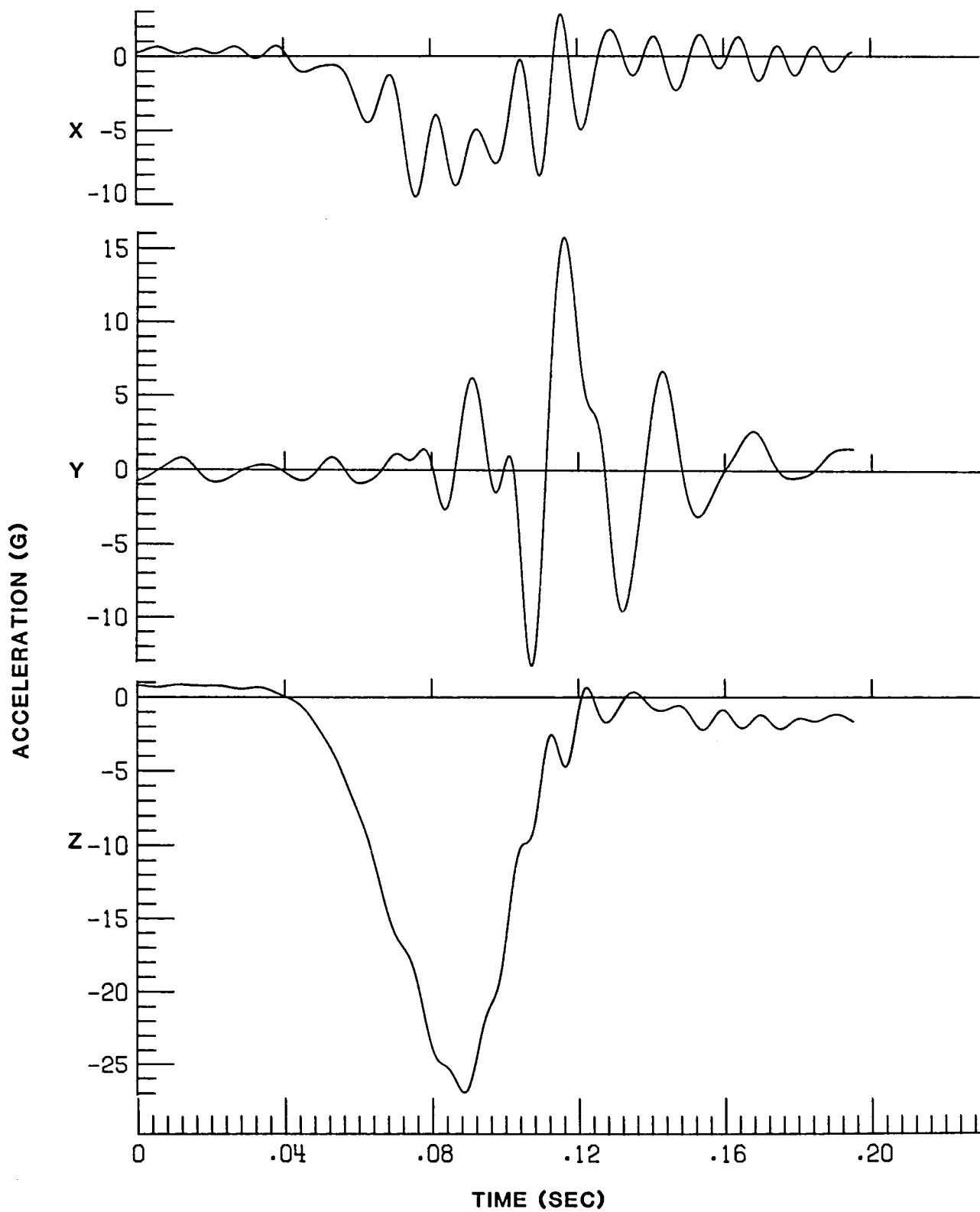




LEFT SEAT PAN  
NASA TEST NO. 4

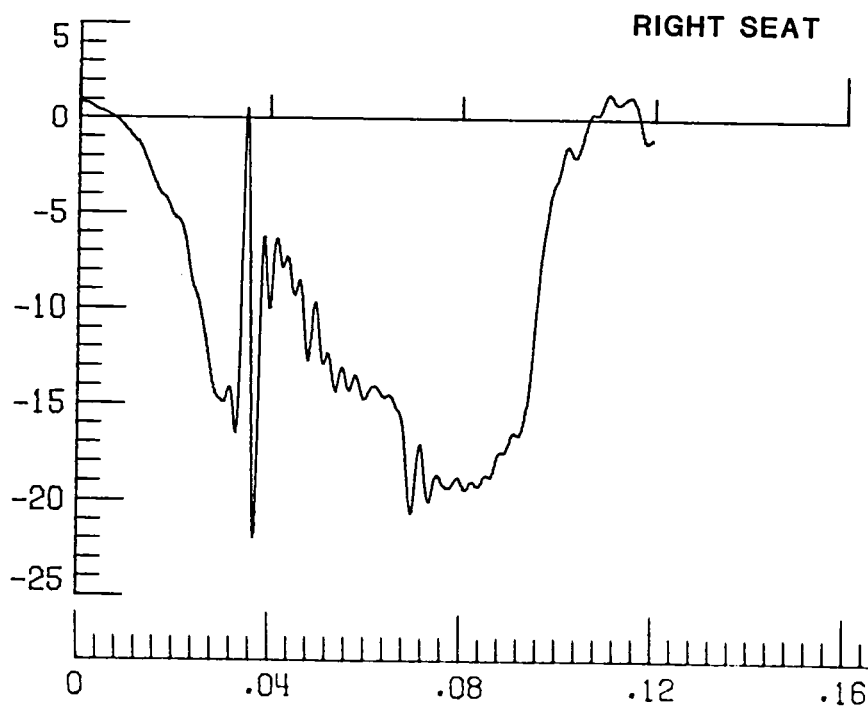
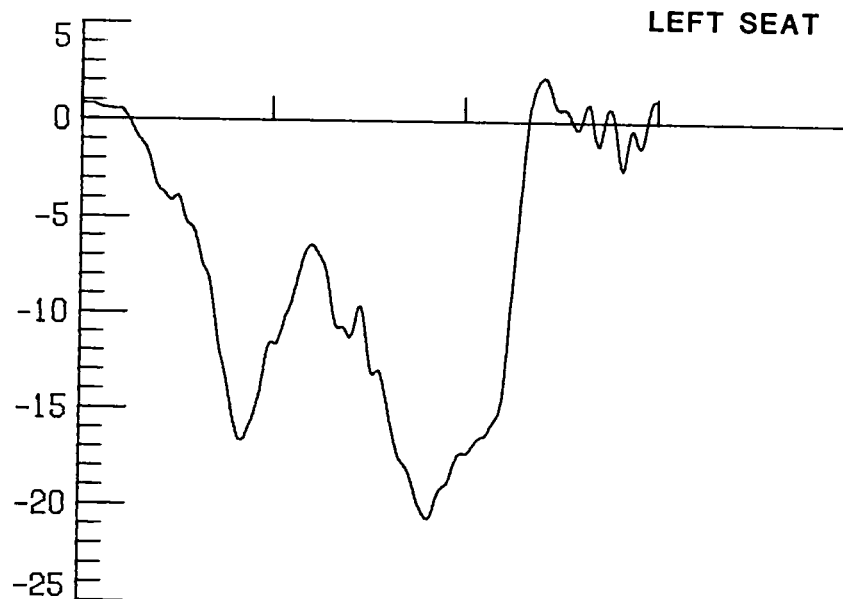


RIGHT SEAT PAN  
NASA TEST NO. 4



CAPSULE C.G.  
NASA TEST NO. 4

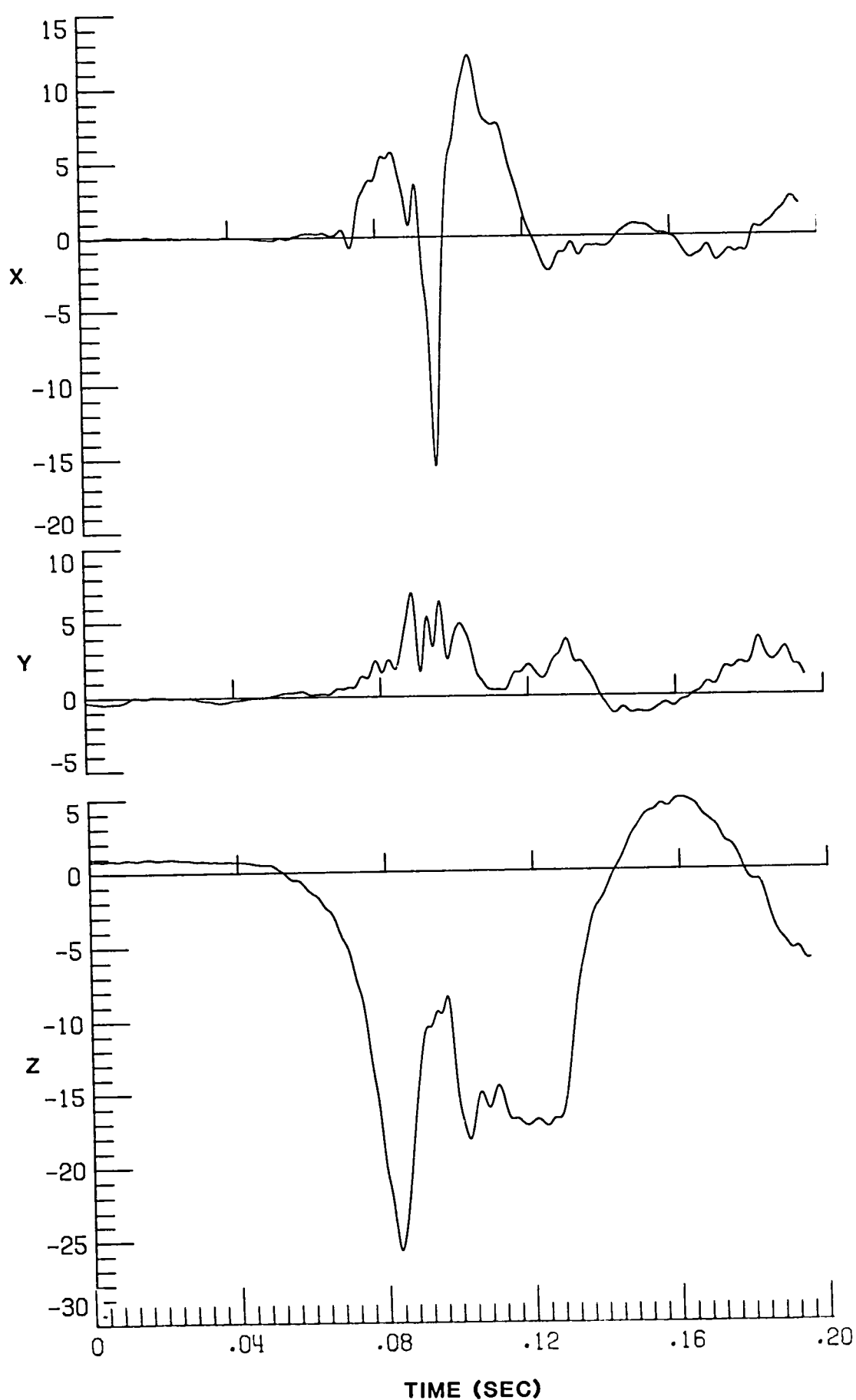
ACCELERATION (G)



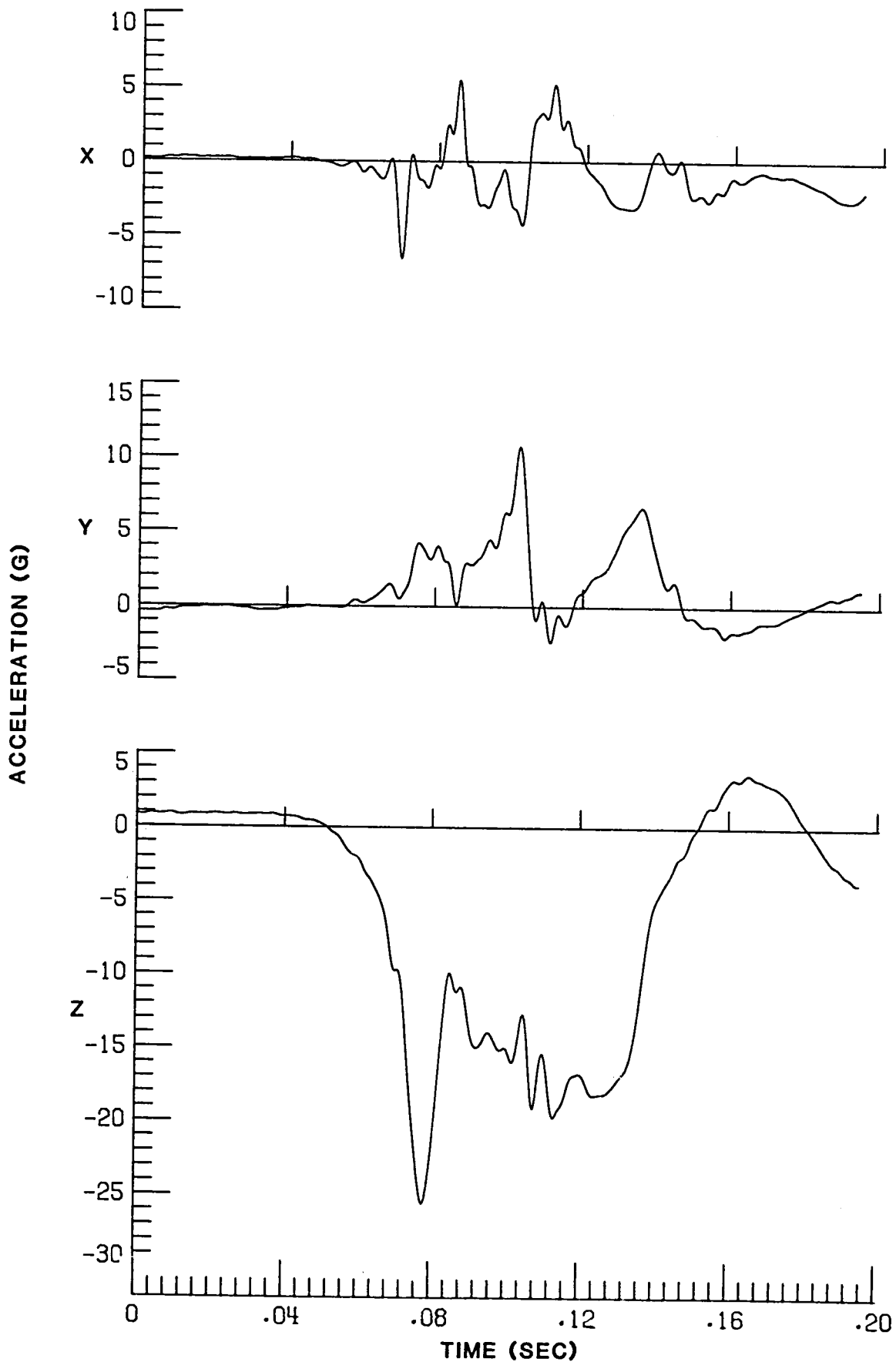
TIME (SEC)

SEAT PAN ALONG BACK TANGENT LINE  
NASA TEST NO. 4

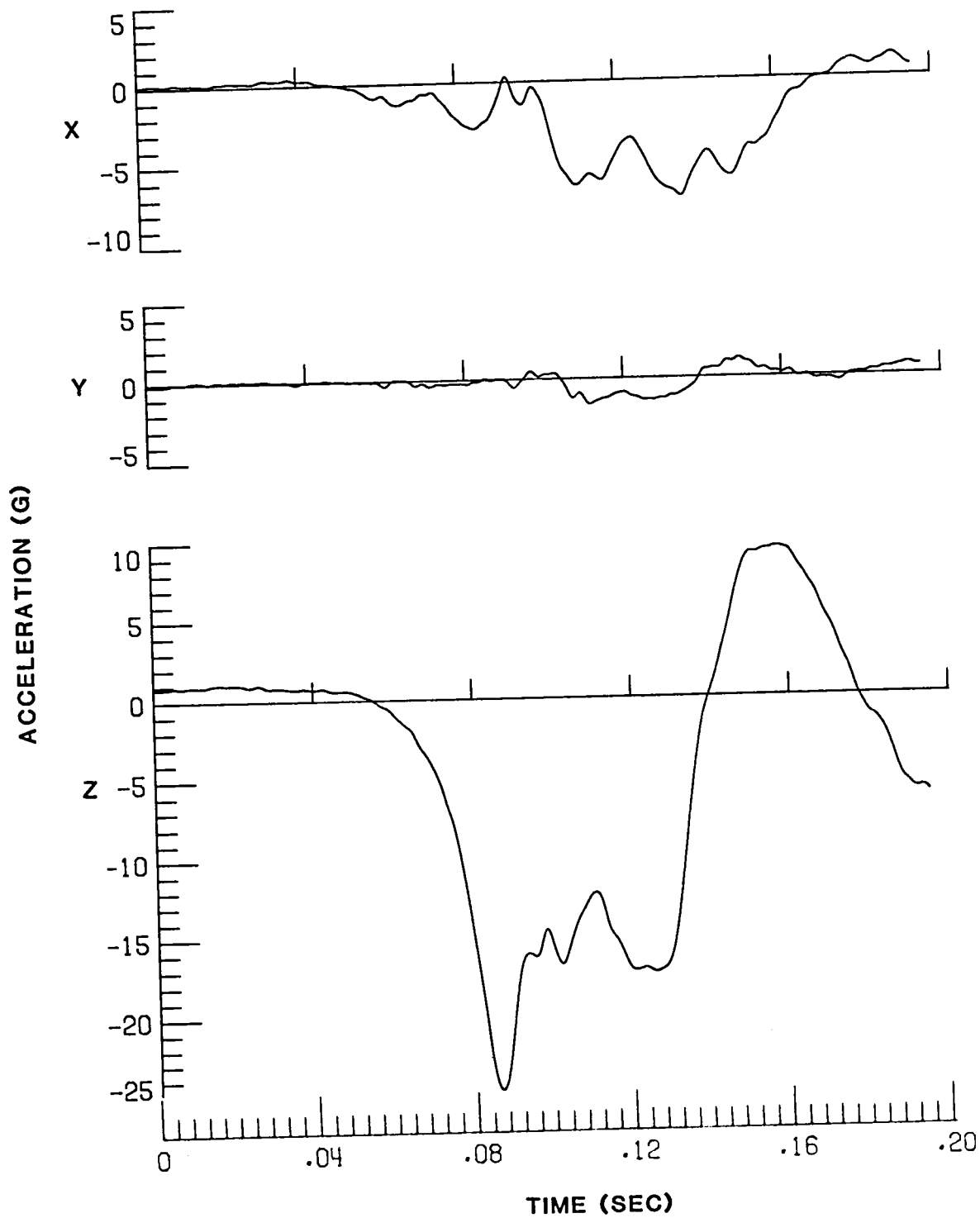
ACCELERATION (G)



LEFT DUMMY PELVIS  
NASA TEST NO. 4

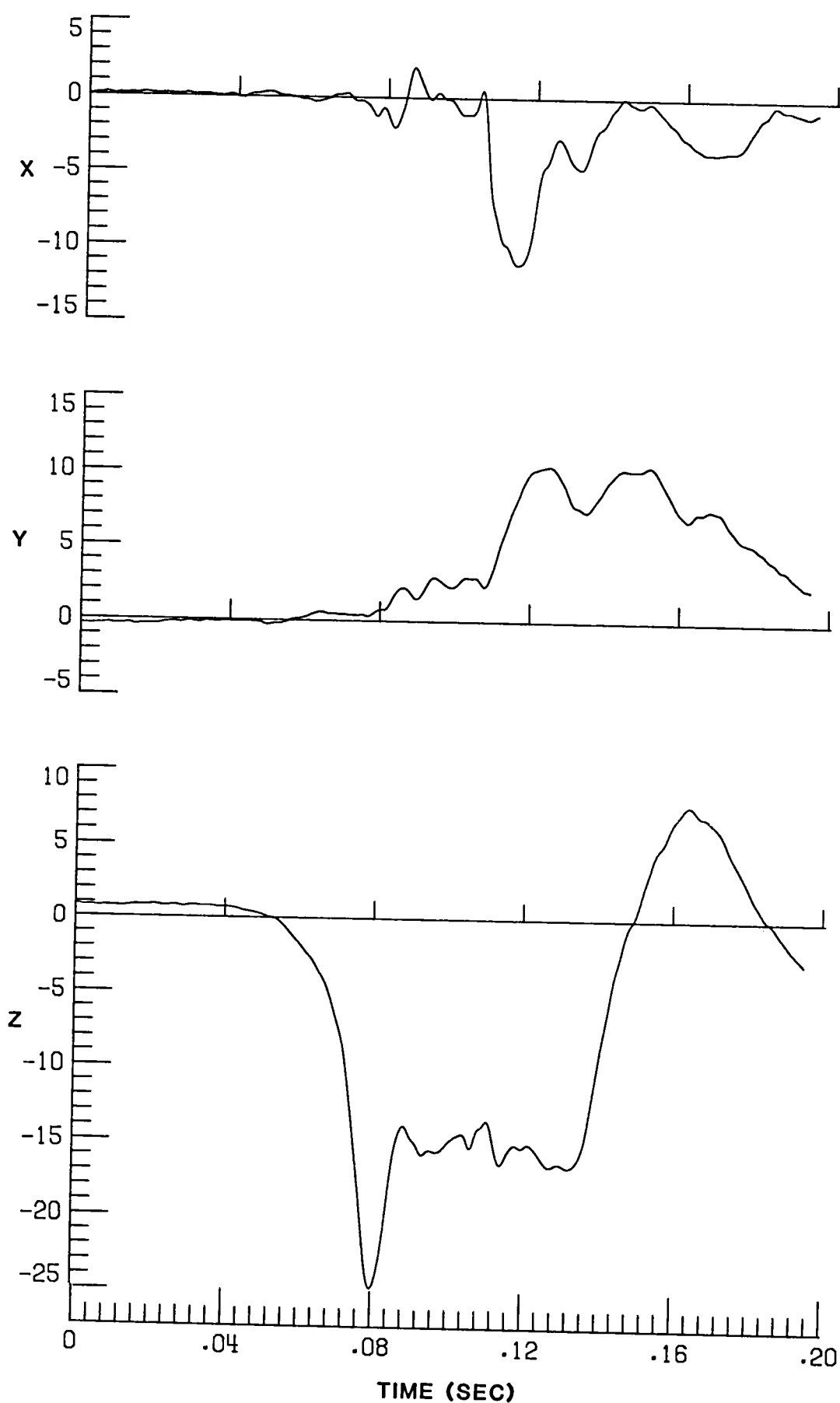


RIGHT DUMMY PELVIS  
NASA TEST NO. 4



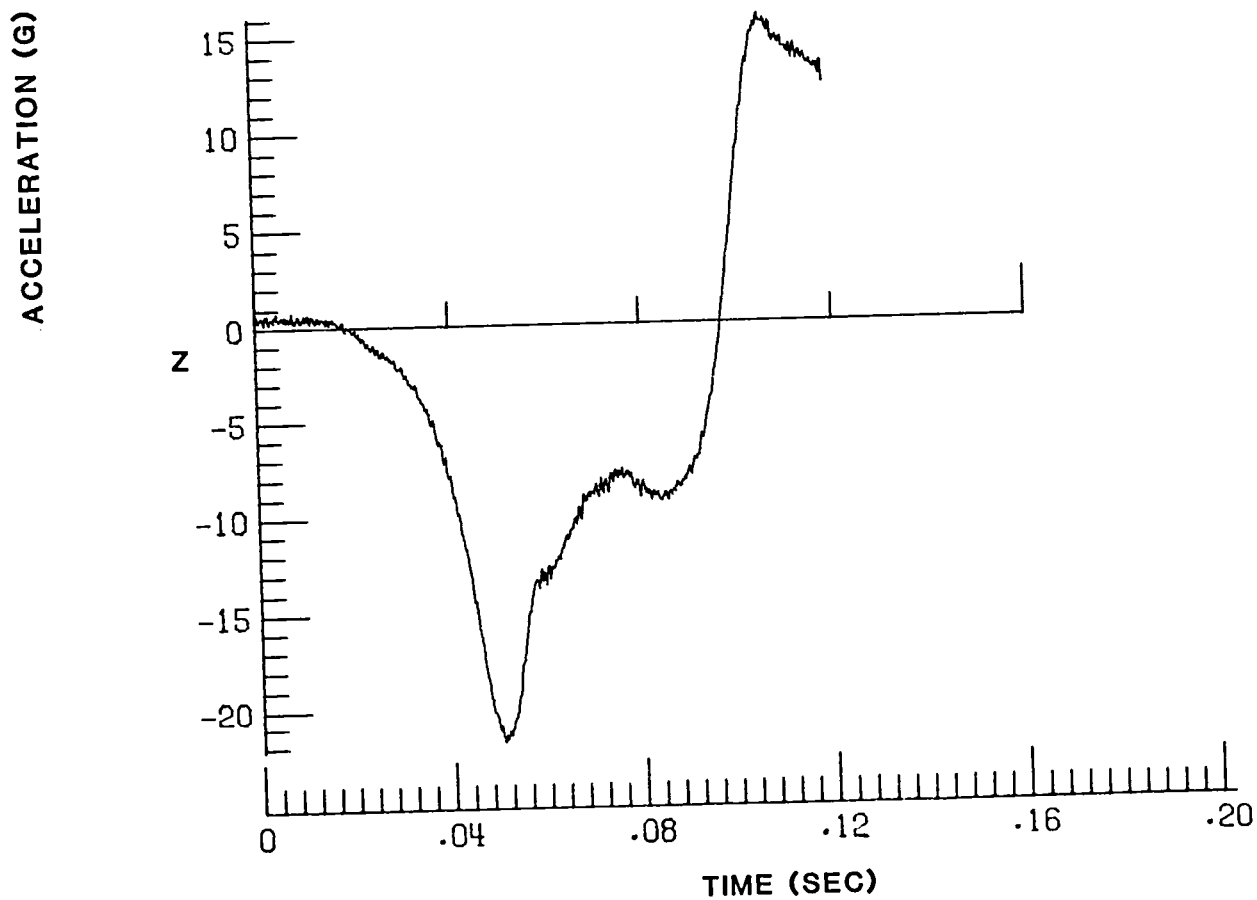
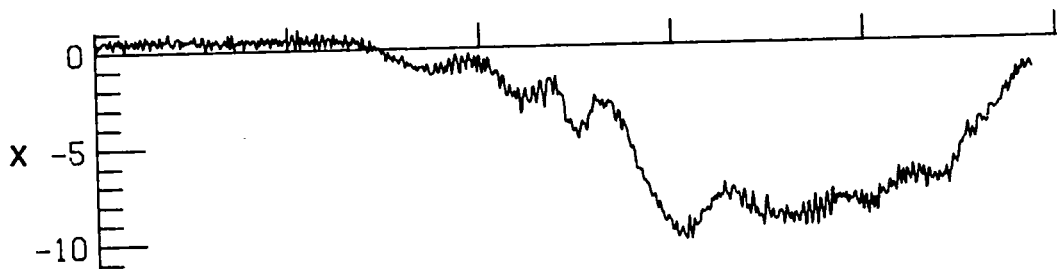
LEFT DUMMY CHEST  
NASA TEST NO. 4

ACCELERATION (G)

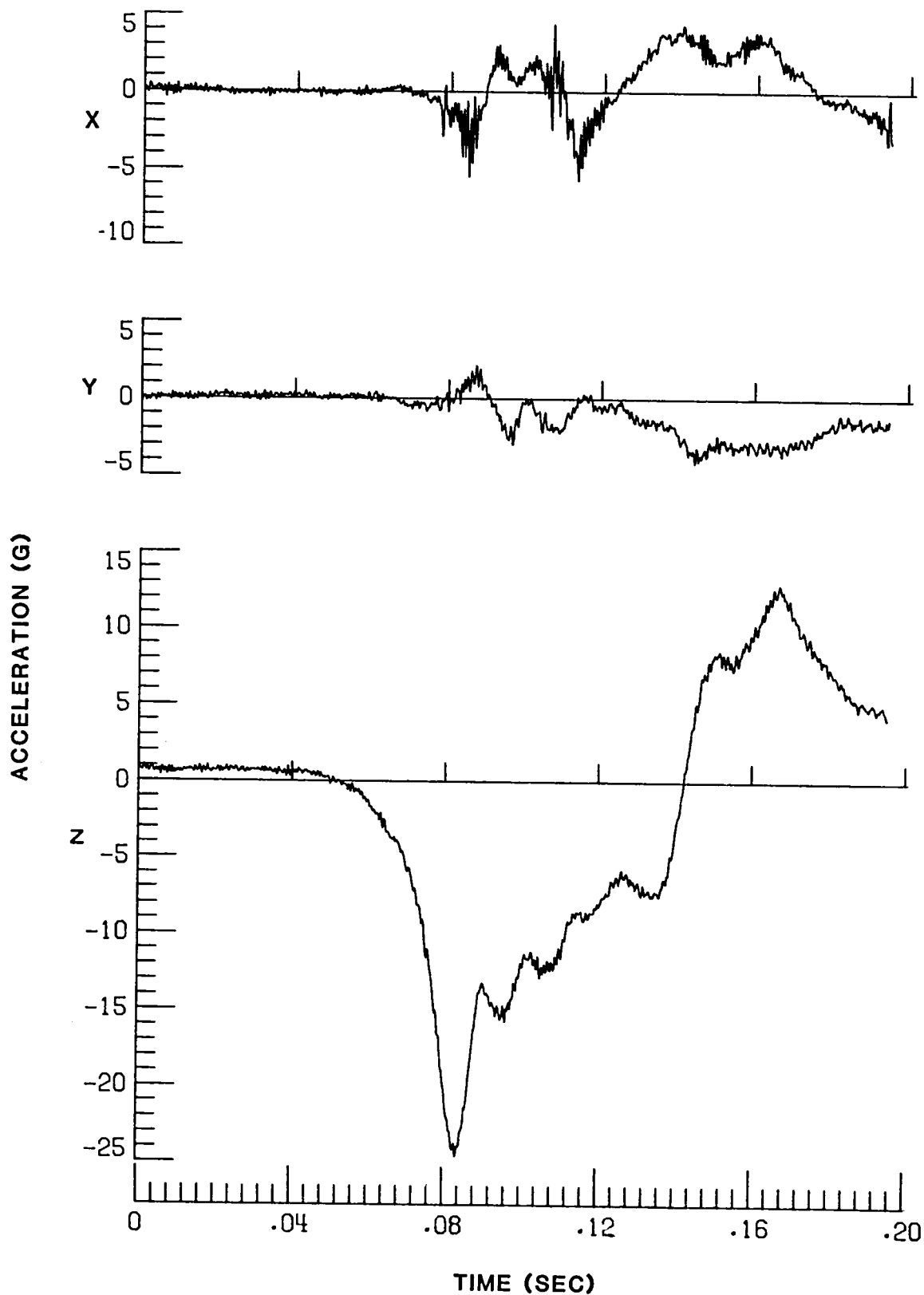


RIGHT DUMMY CHEST  
NASA TEST NO. 4

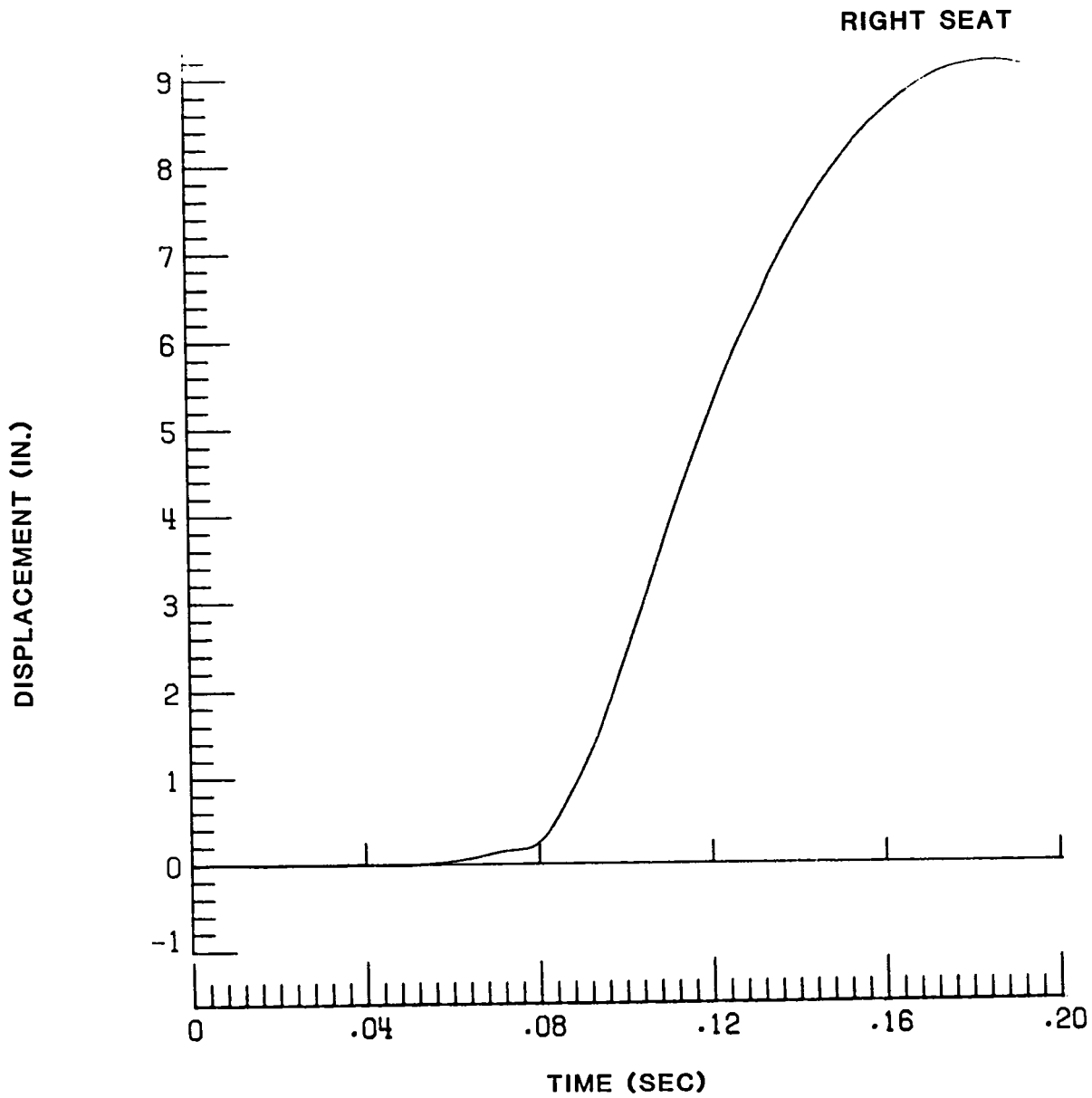




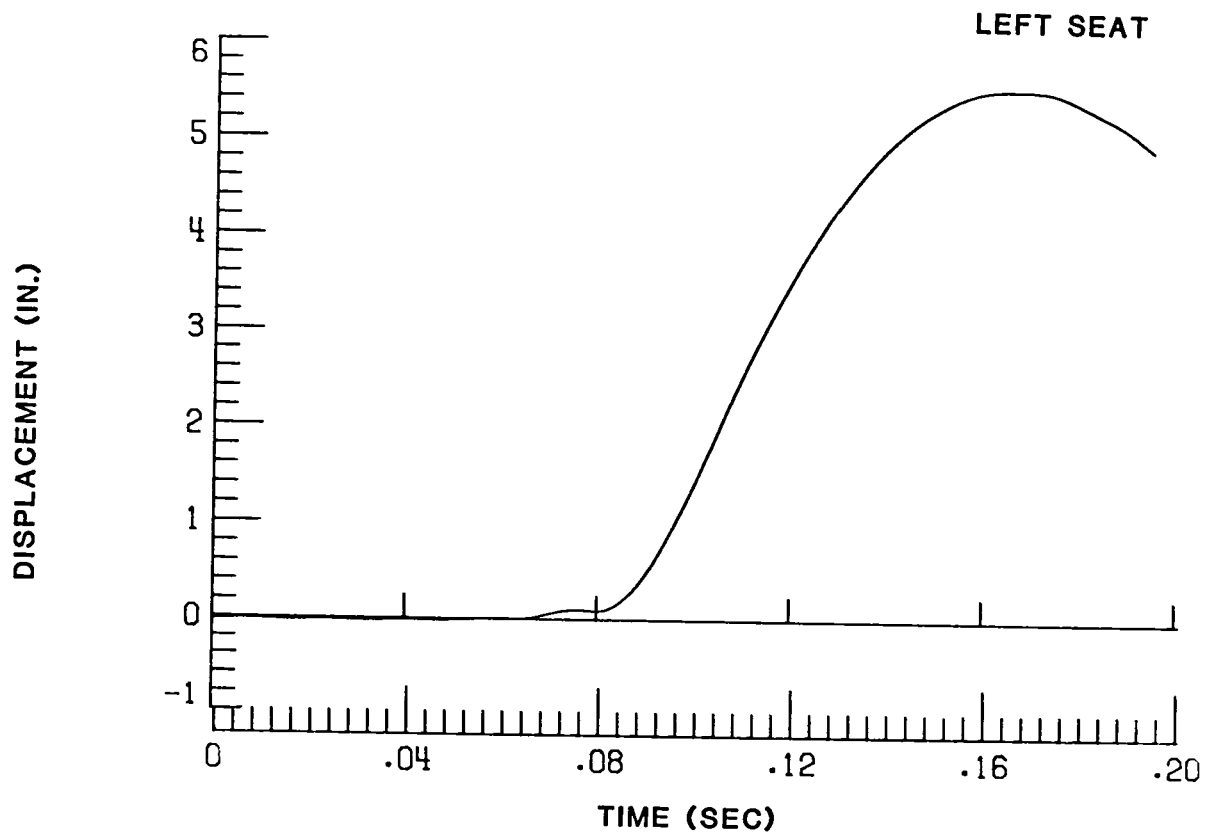
LEFT DUMMY HEAD  
NASA TEST NO. 4



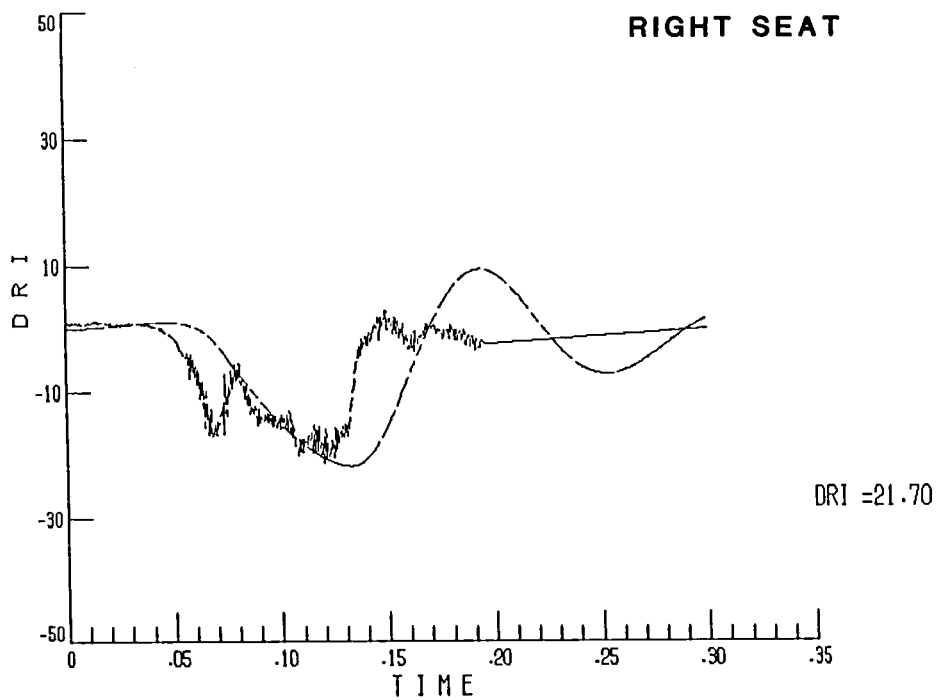
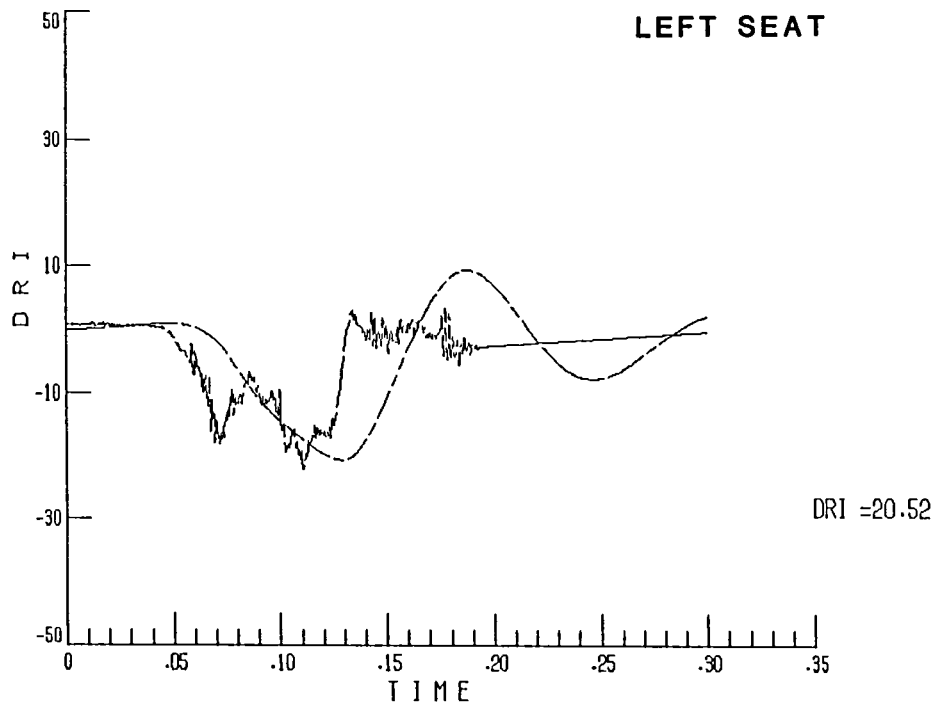
RIGHT DUMMY HEAD  
NASA TEST NO. 4



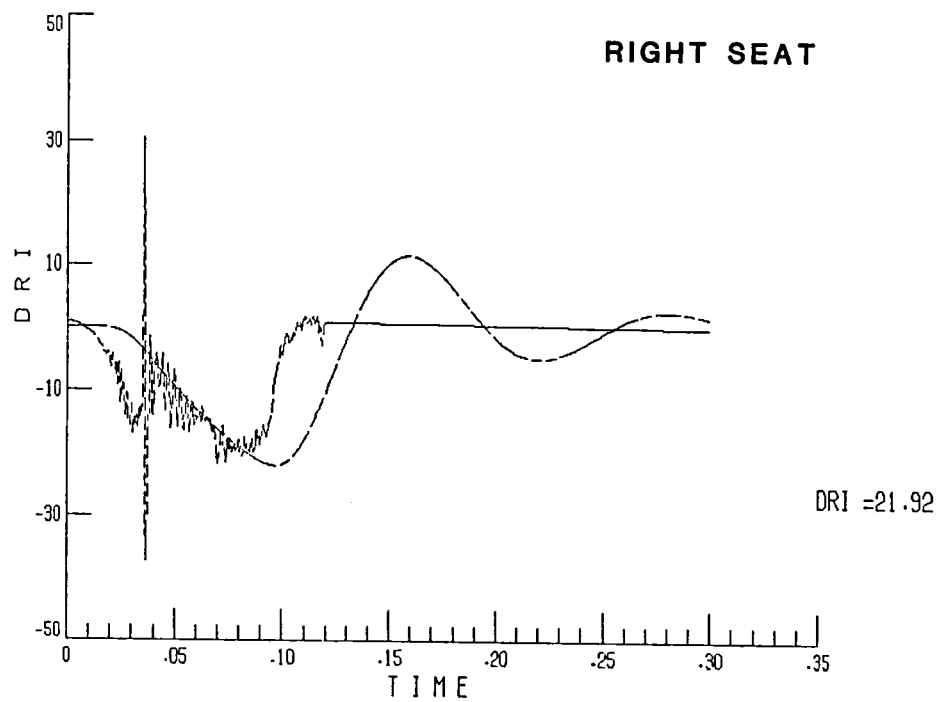
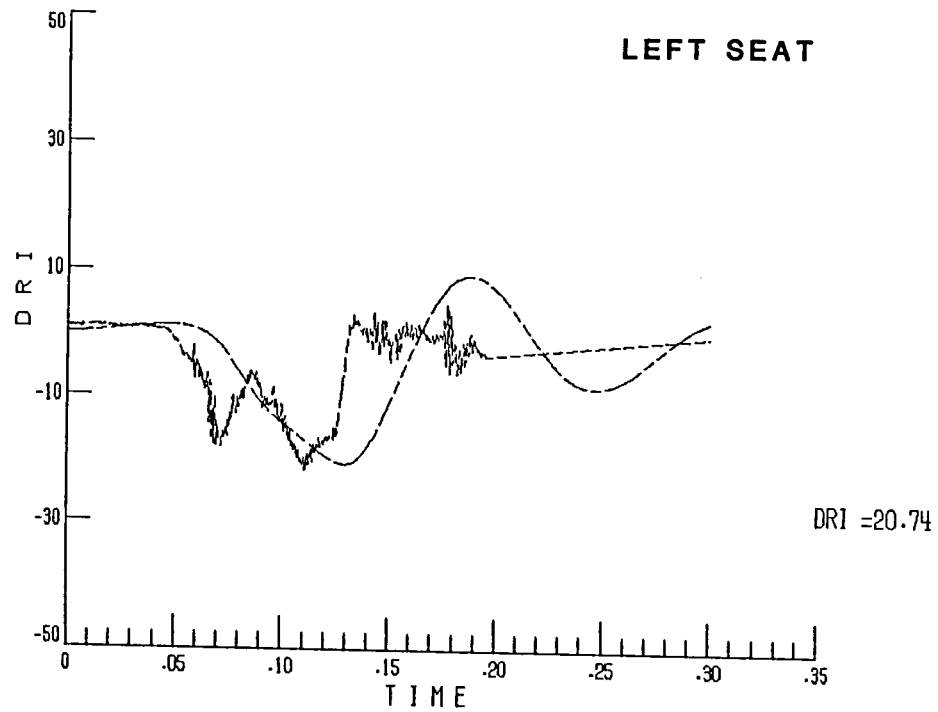
**SEAT STROKE  
NASA TEST NO. 4**



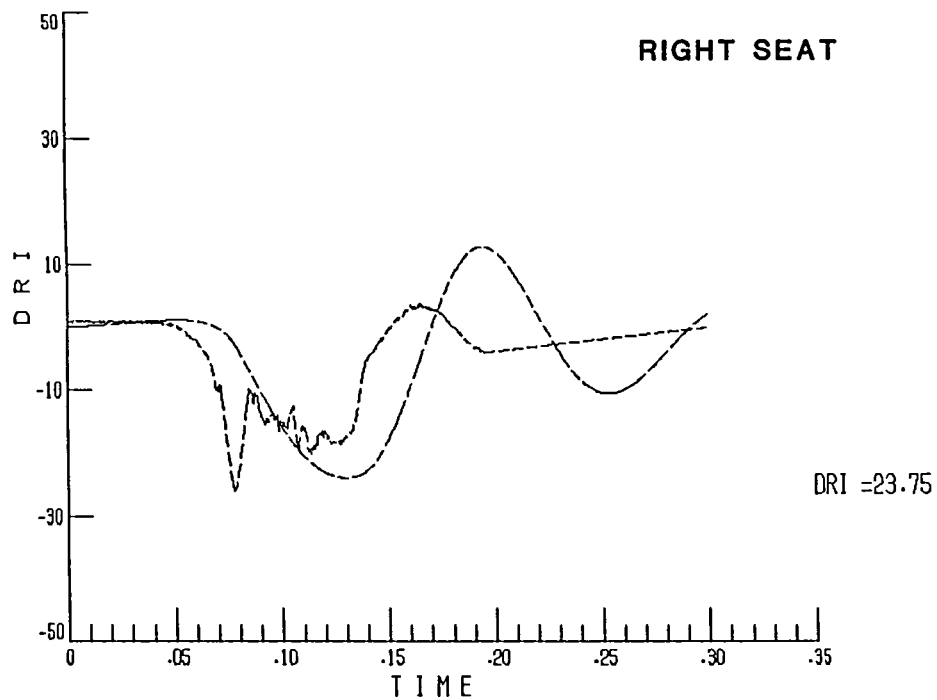
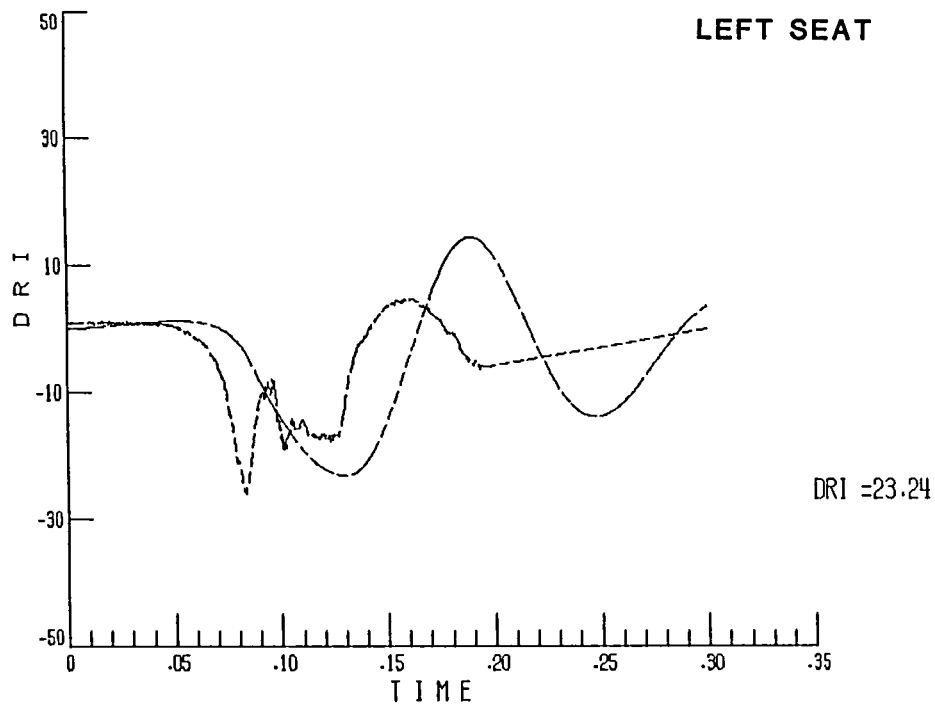
**SEAT STROKE  
NASA TEST NO. 4**



**DYNAMIC RESPONSE INDEX  
SEAT PAN Z-AXIS  
NASA TEST NO. 4**



**DYNAMIC RESPONSE INDEX  
BACK TANGENT LINE  
NASA TEST NO. 4**



**DYNAMIC RESPONSE INDEX  
DUMMY PELVIS Z-AXIS  
NASA TEST NO. 4**

#### 6.0 NASA TEST NO. 5

HORIZONTAL VELOCITY: 34 ft/sec

VERTICAL VELOCITY: 32 ft/sec

CREW MODULE PITCH:  $17^0$

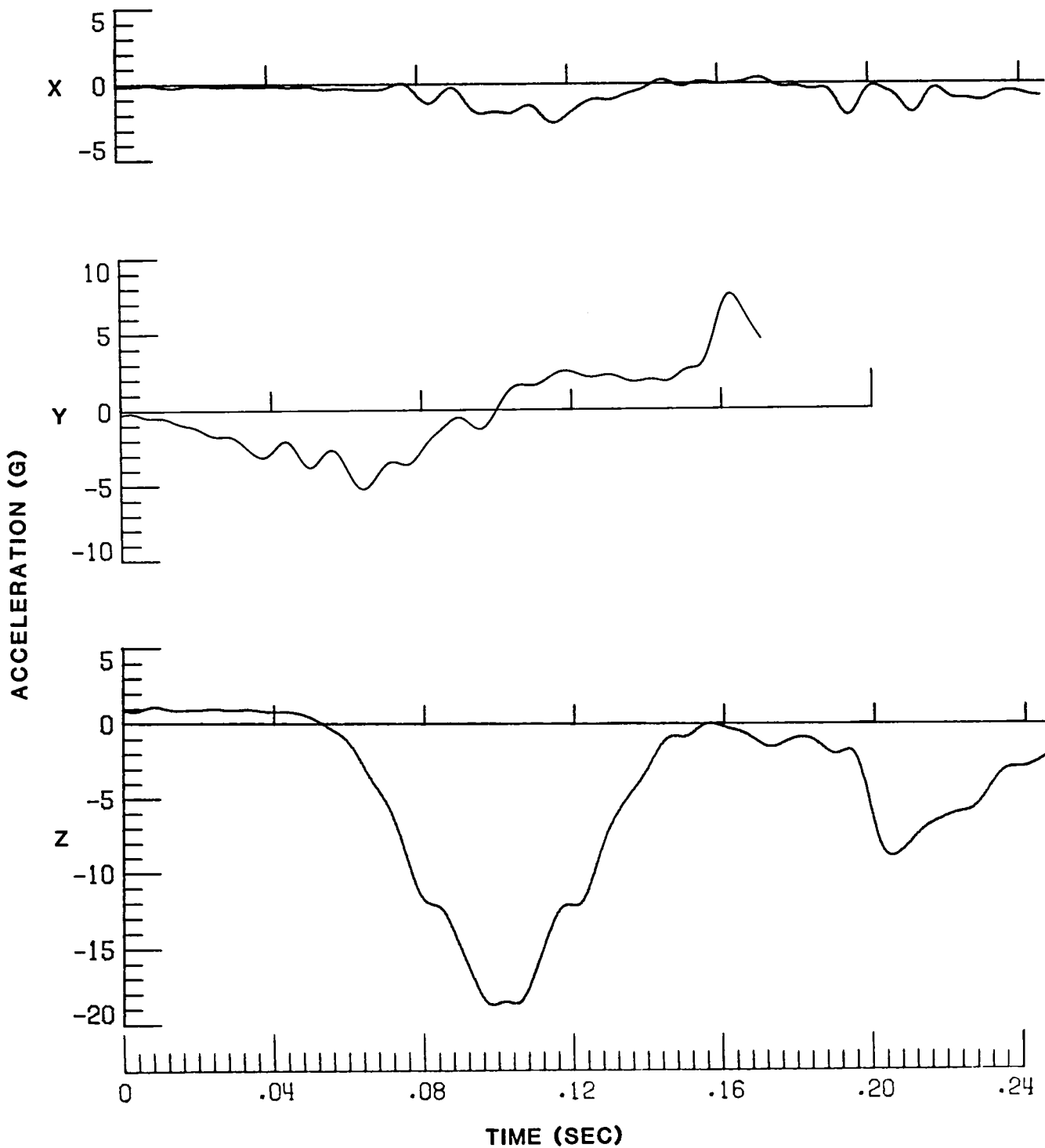
CREW MODULE ROLL:  $0^0$

CREW MODULE YAW:  $45^0$

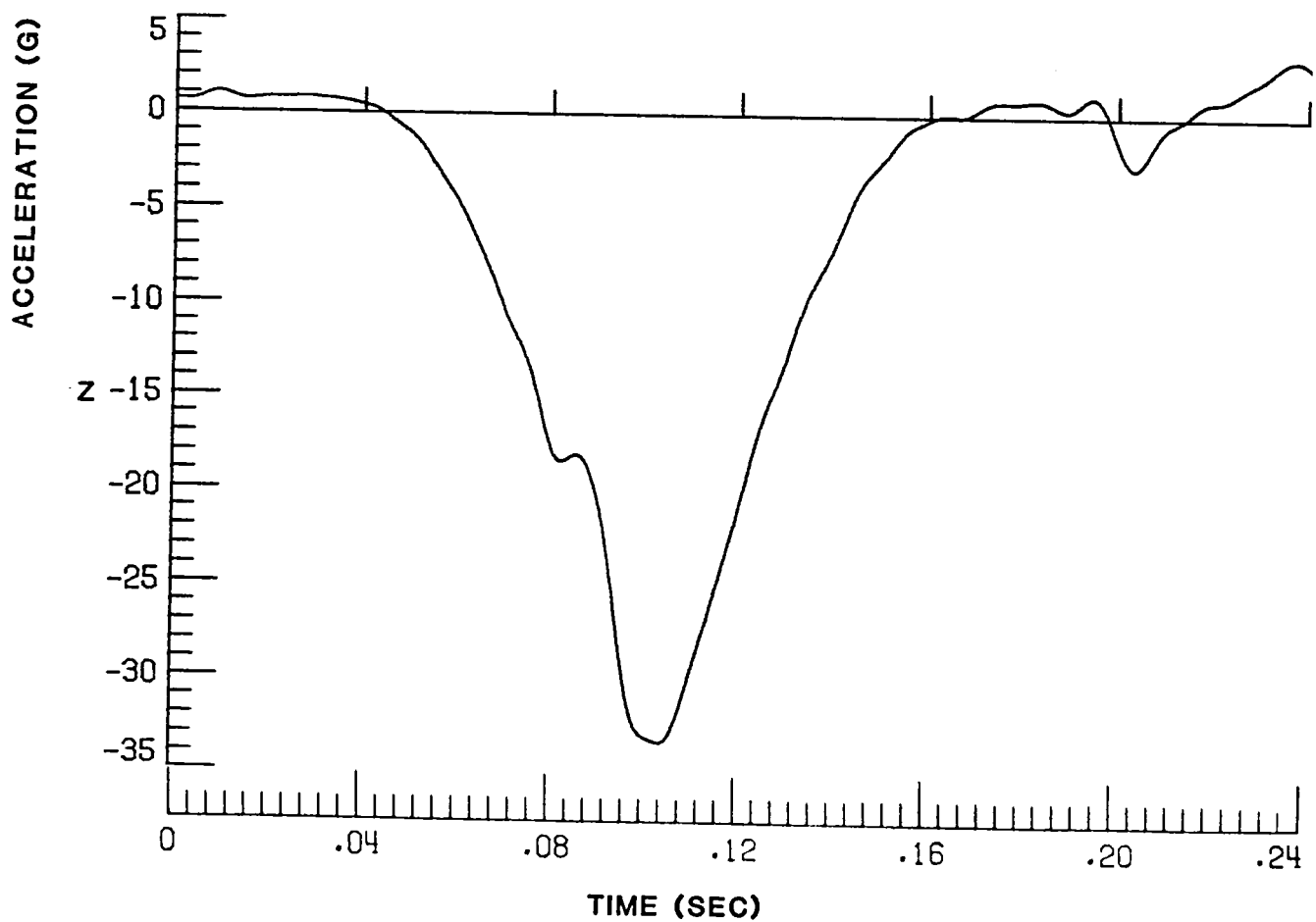
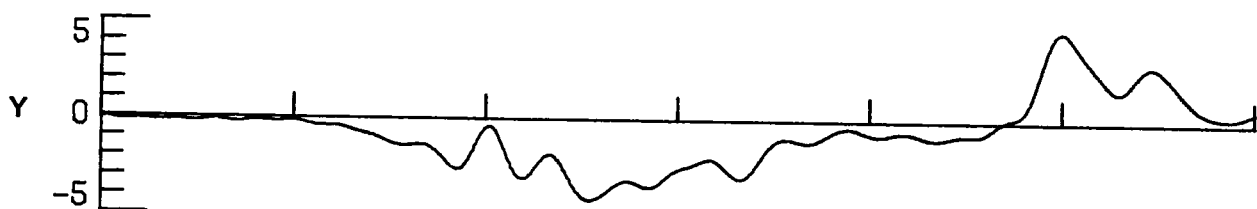
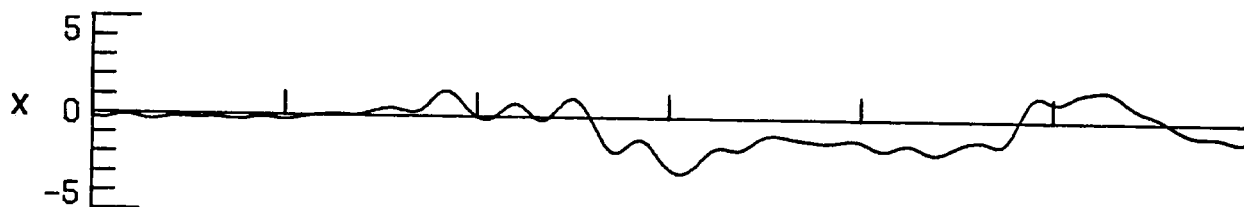
LEFT SEAT: Energy Absorbing

RIGHT SEAT: Energy Absorbing



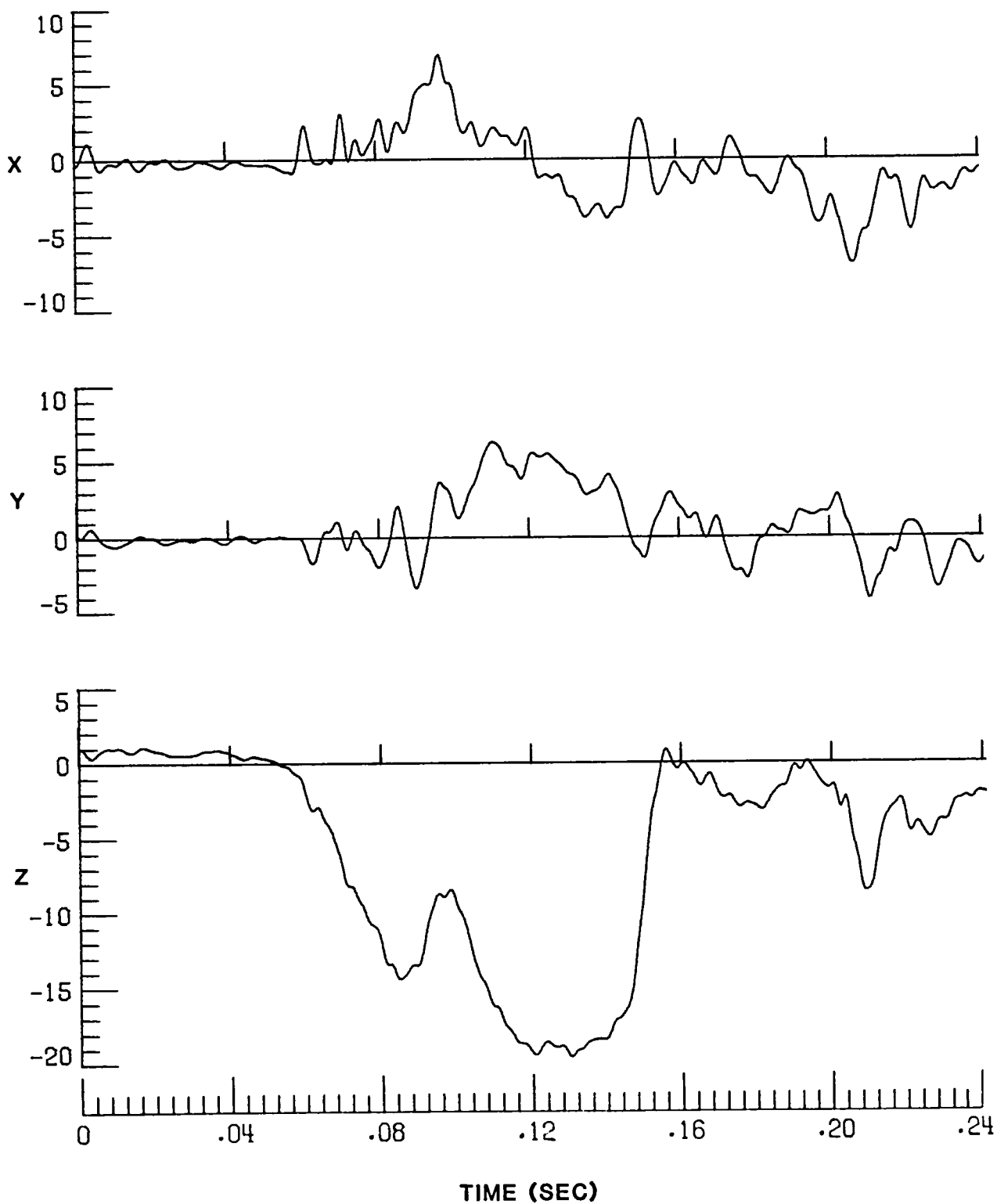


LEFT BULKHEAD  
NASA TEST NO. 5

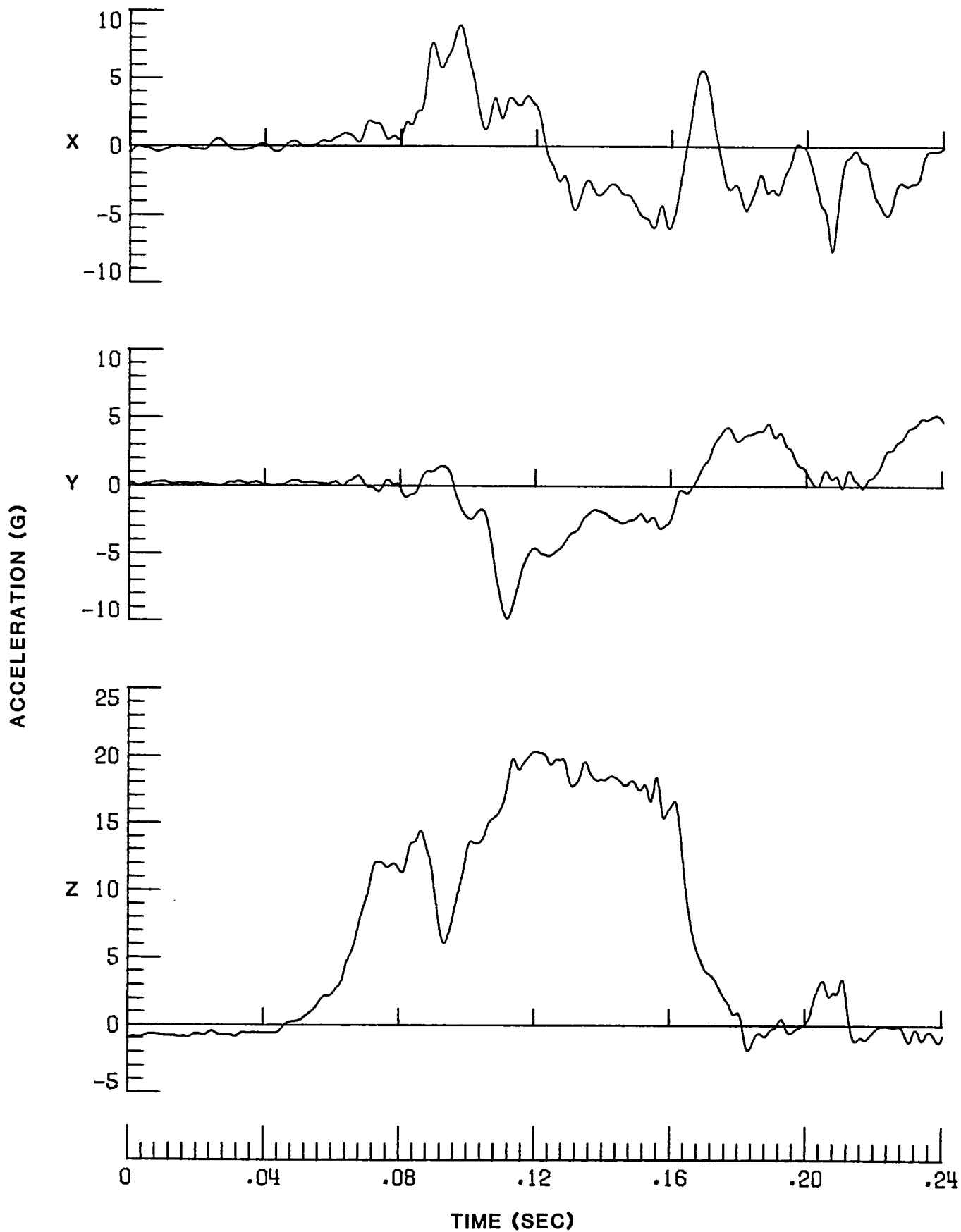


RIGHT BULKHEAD  
NASA TEST NO. 5

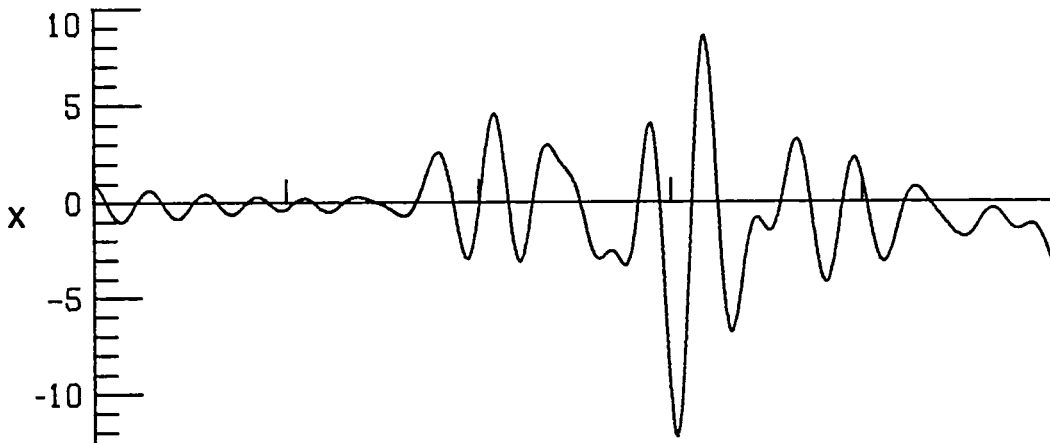
ACCELERATION (G)



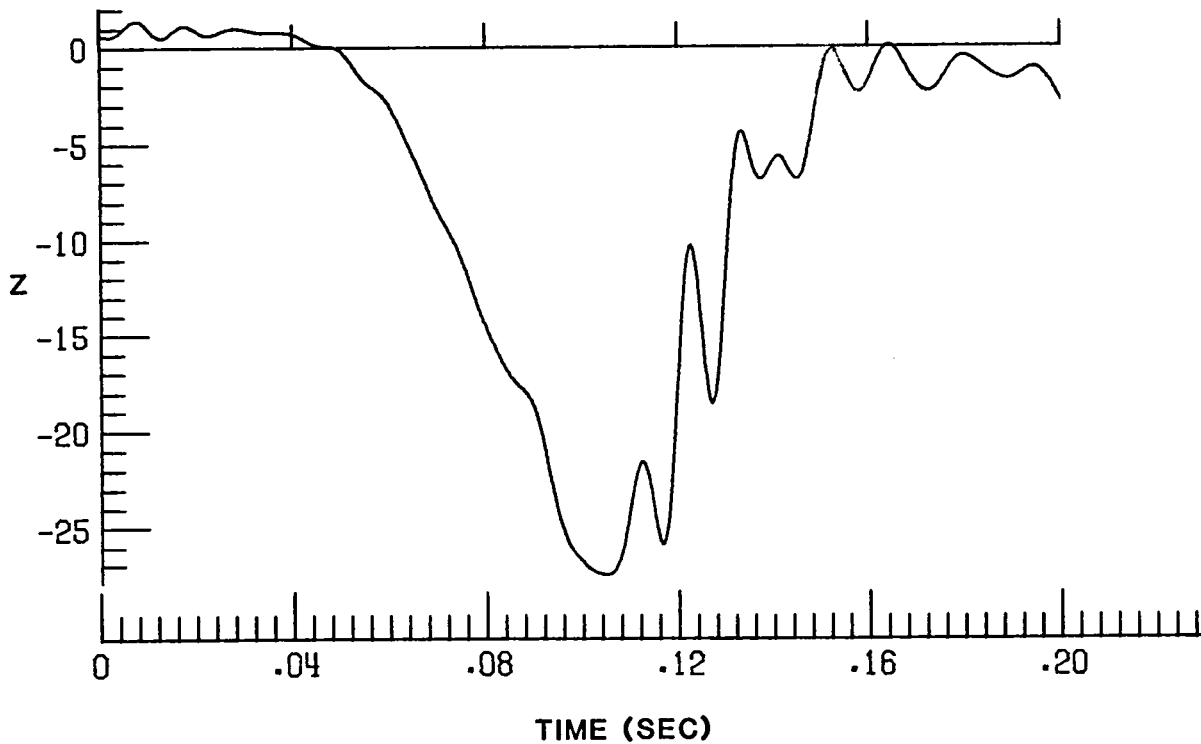
LEFT SEAT PAN  
NASA TEST NO. 5



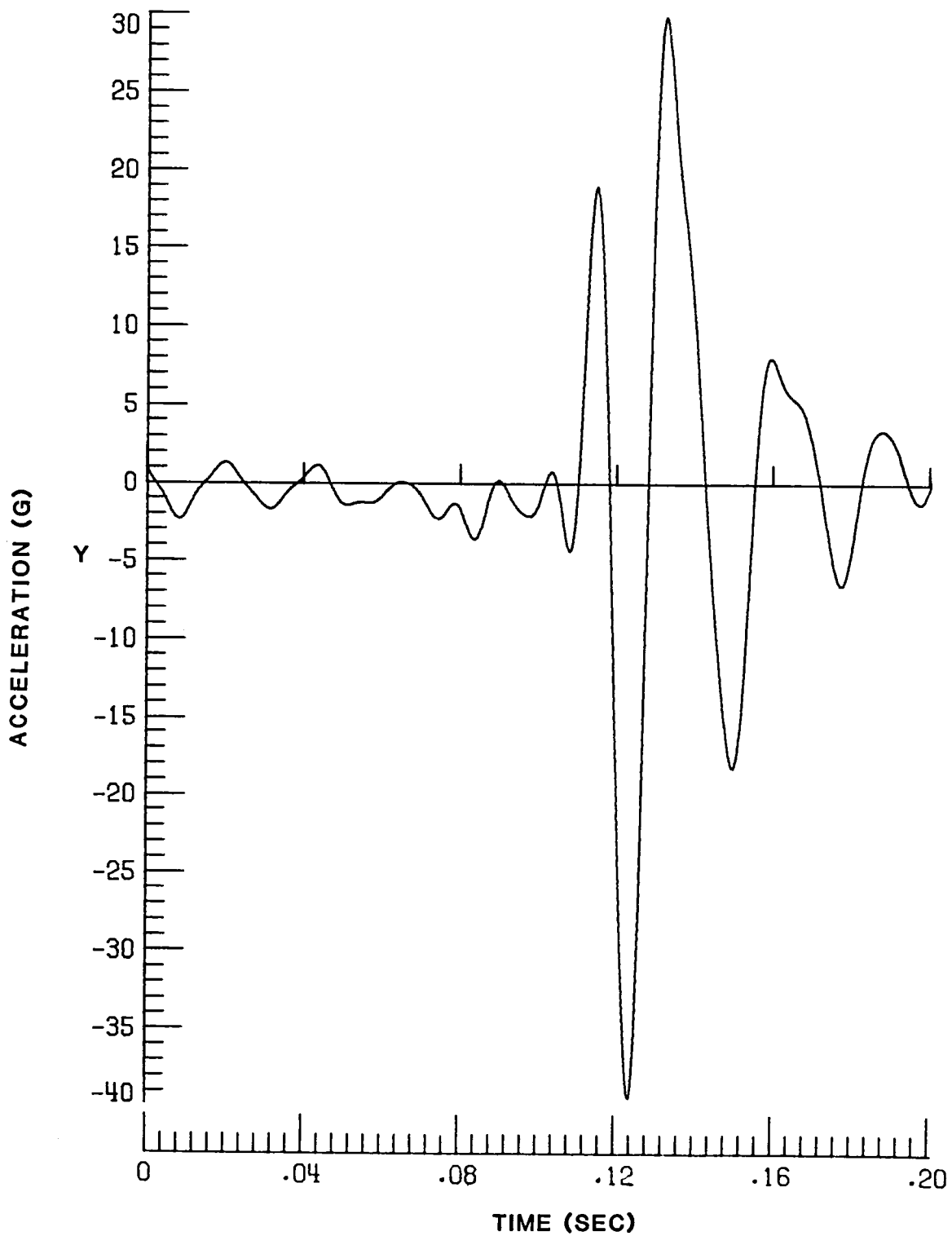
RIGHT SEAT PAN  
NASA TEST NO. 5



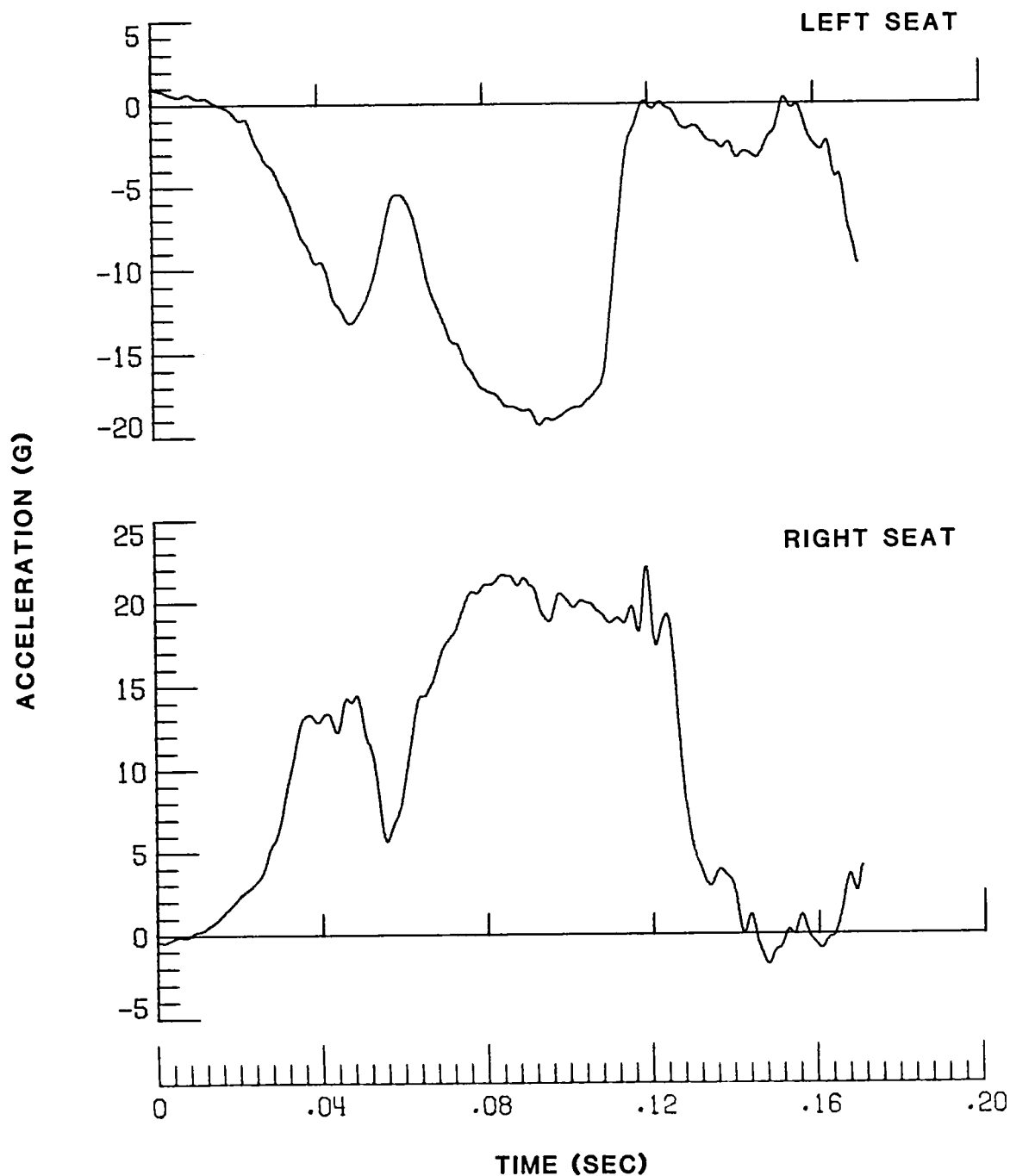
ACCELERATION (G)



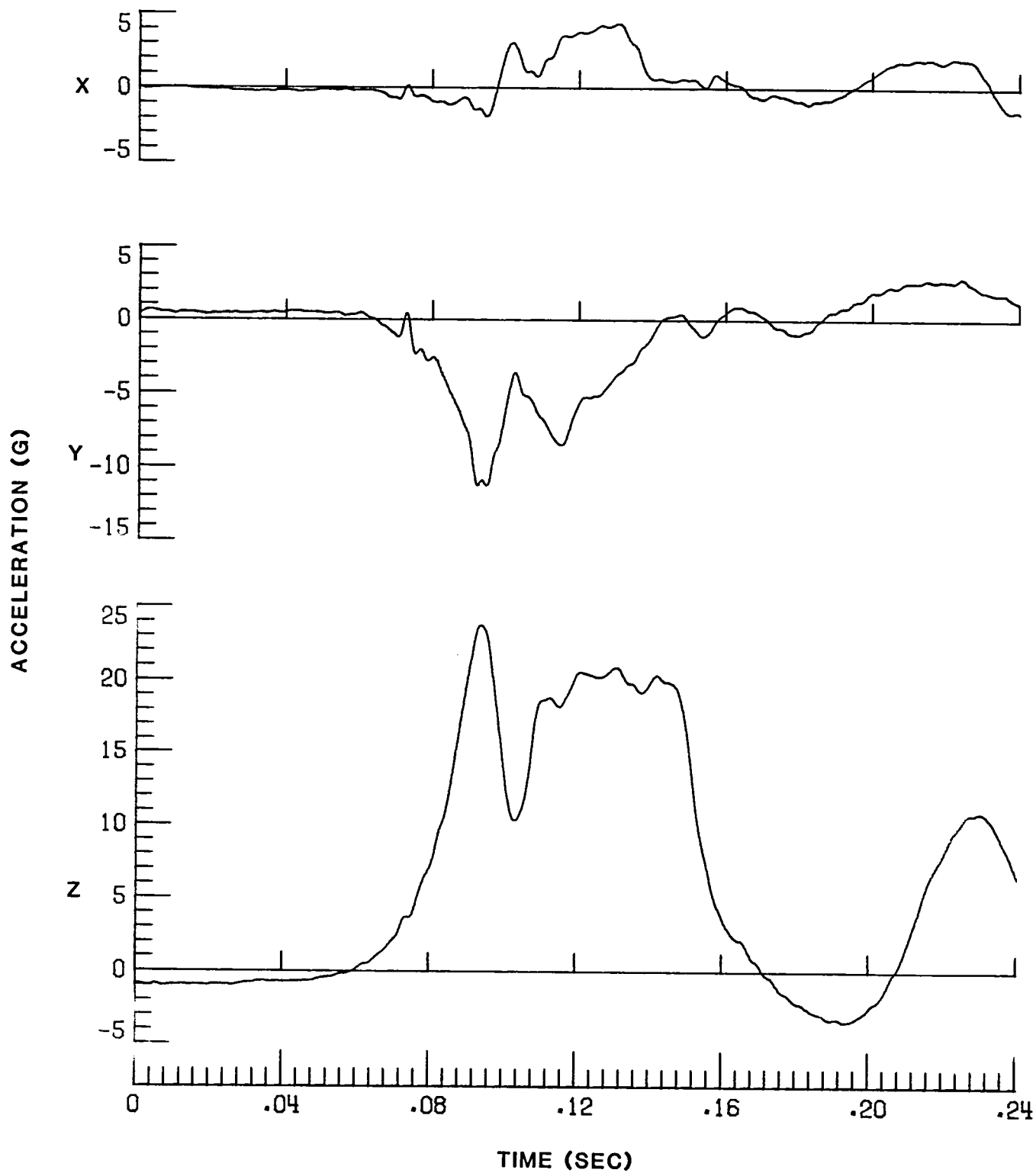
CAPSULE C.G.  
NASA TEST NO. 5



CAPSULE C.G.  
NASA TEST NO. 5

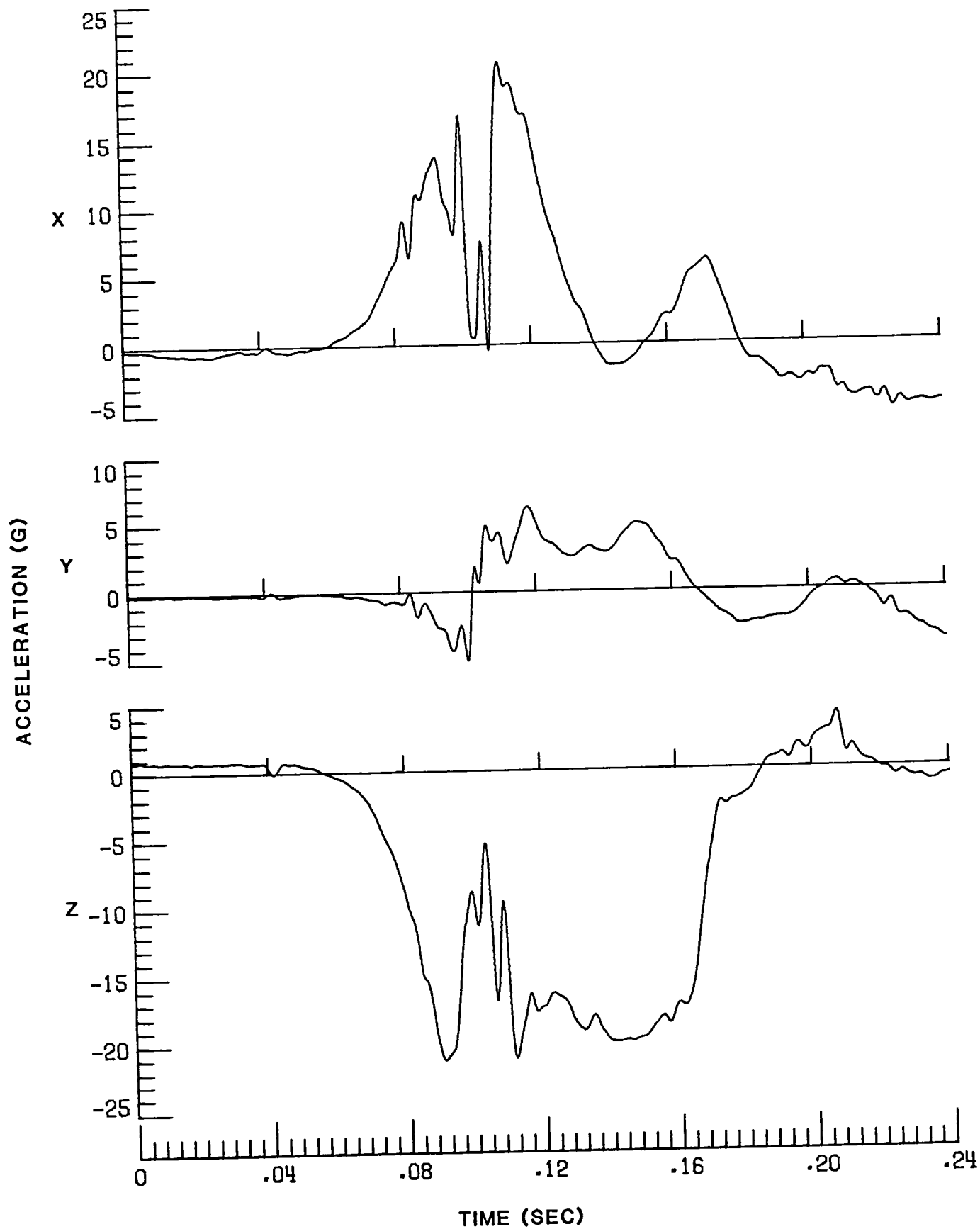


SEAT PAN ALONG BACK TANGENT LINE  
NASA TEST NO. 5

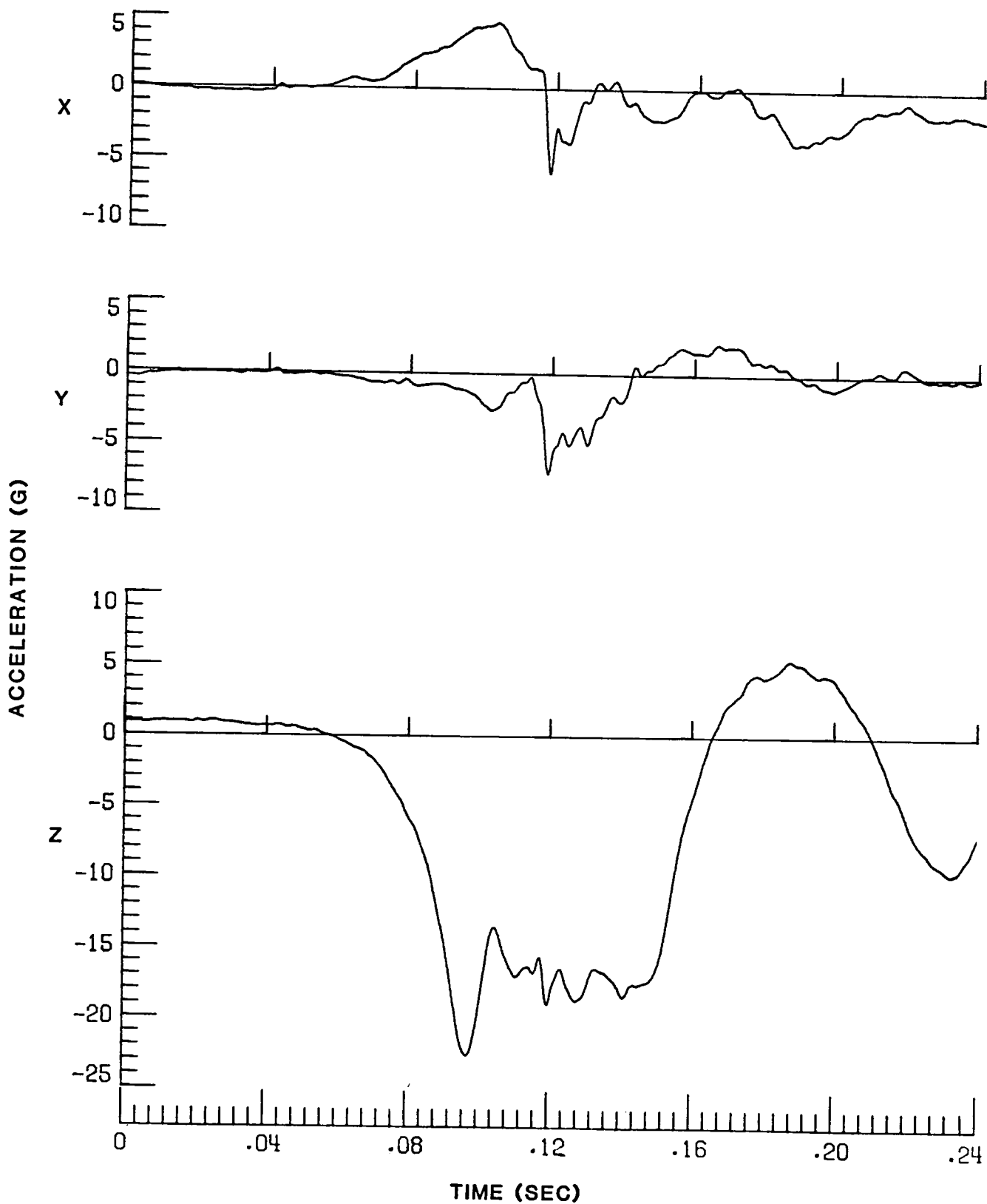


LEFT DUMMY PELVIS  
NASA TEST NO. 5

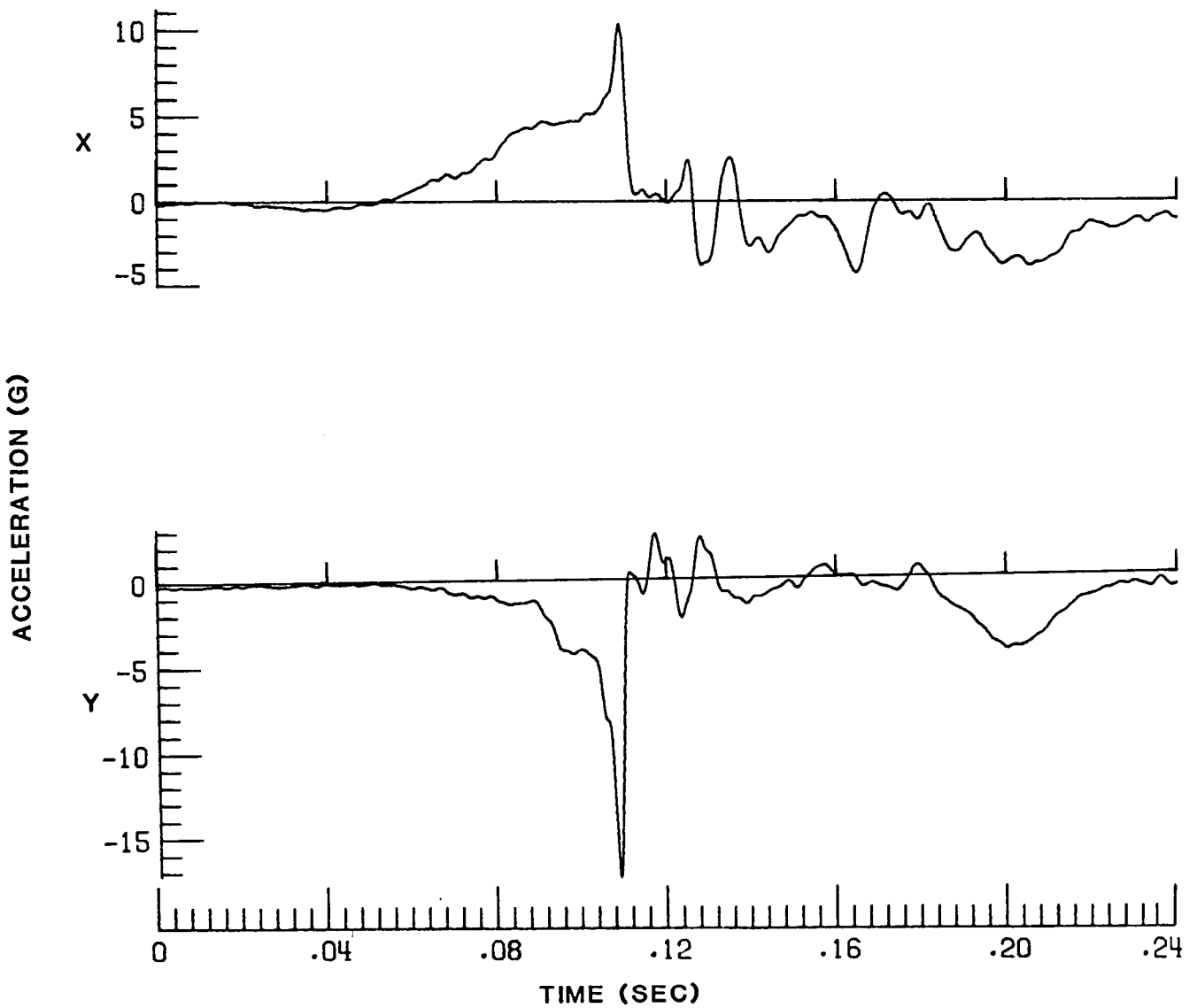




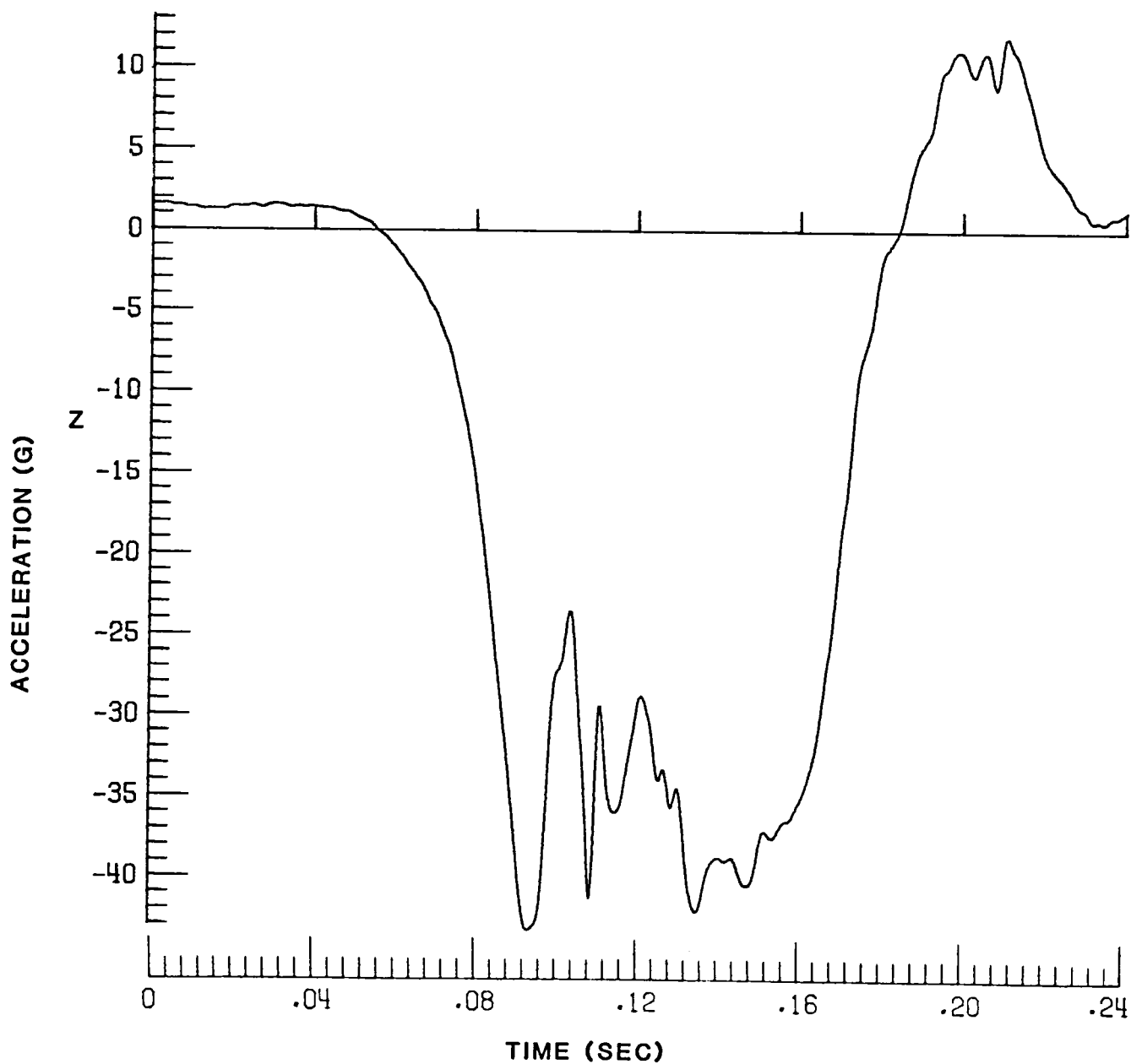
RIGHT DUMMY PELVIS  
NASA TEST NO. 5



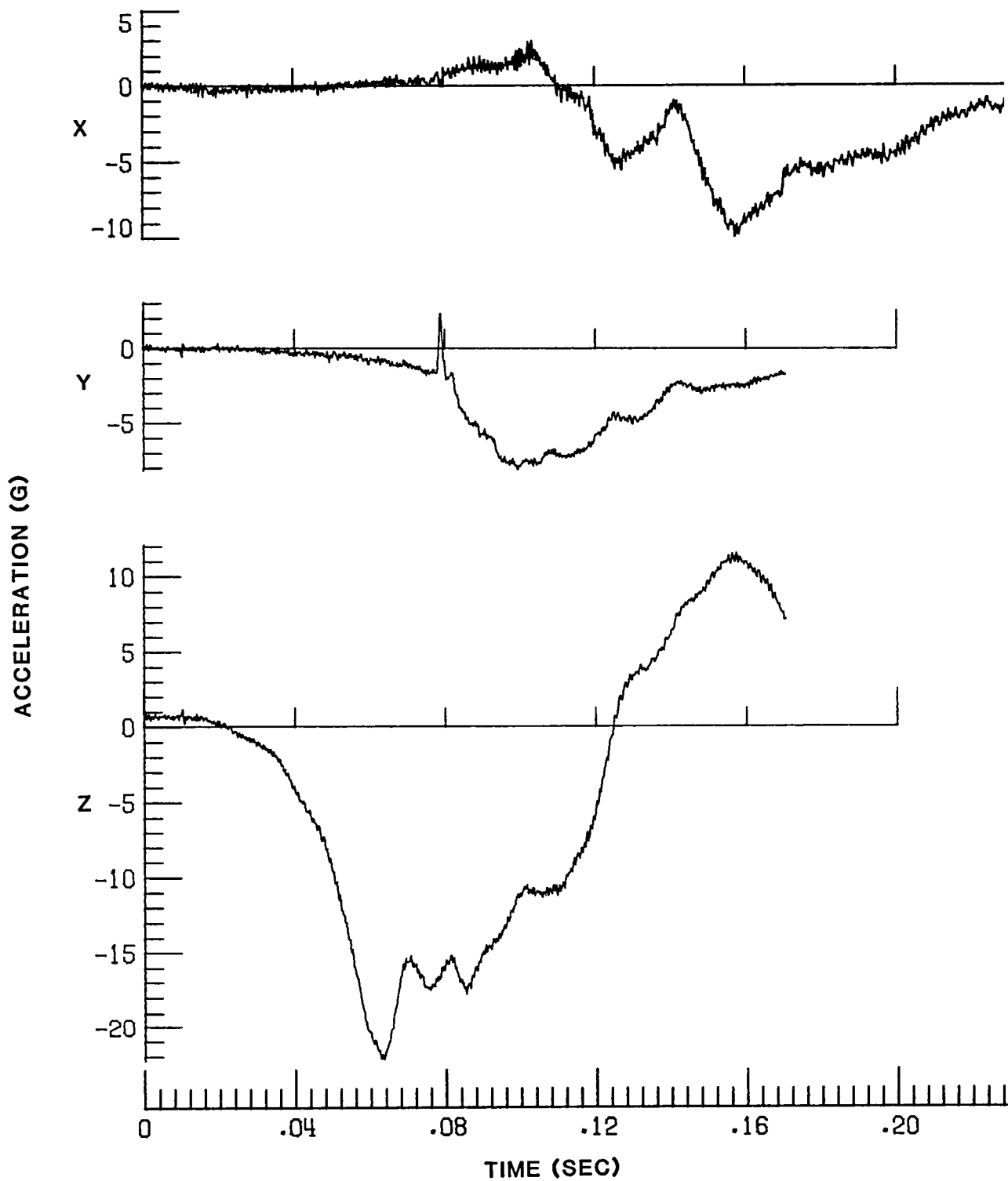
LEFT DUMMY CHEST  
NASA TEST NO. 5



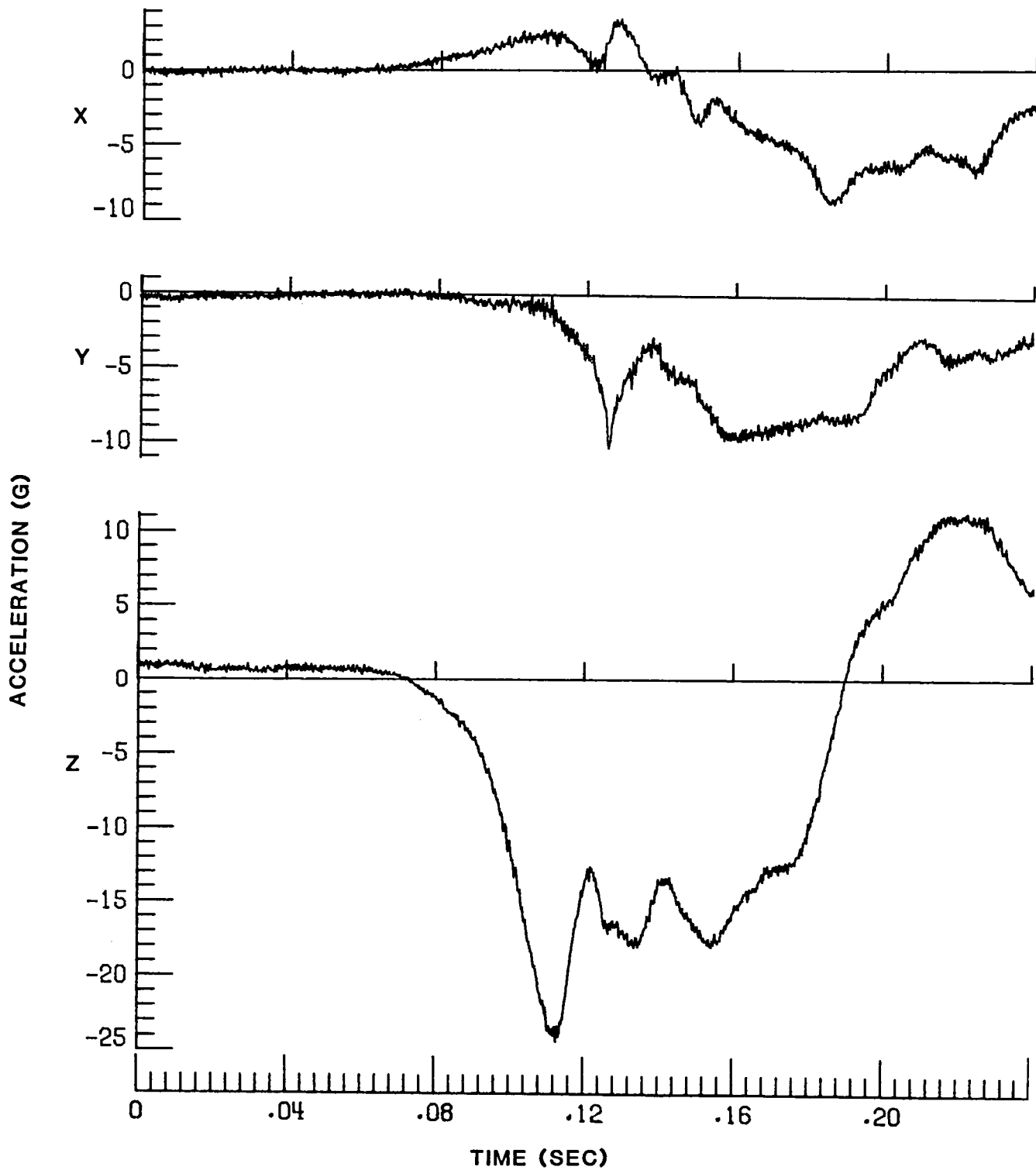
RIGHT DUMMY CHEST  
NASA TEST NO. 5



RIGHT DUMMY CHEST  
NASA TEST NO. 5

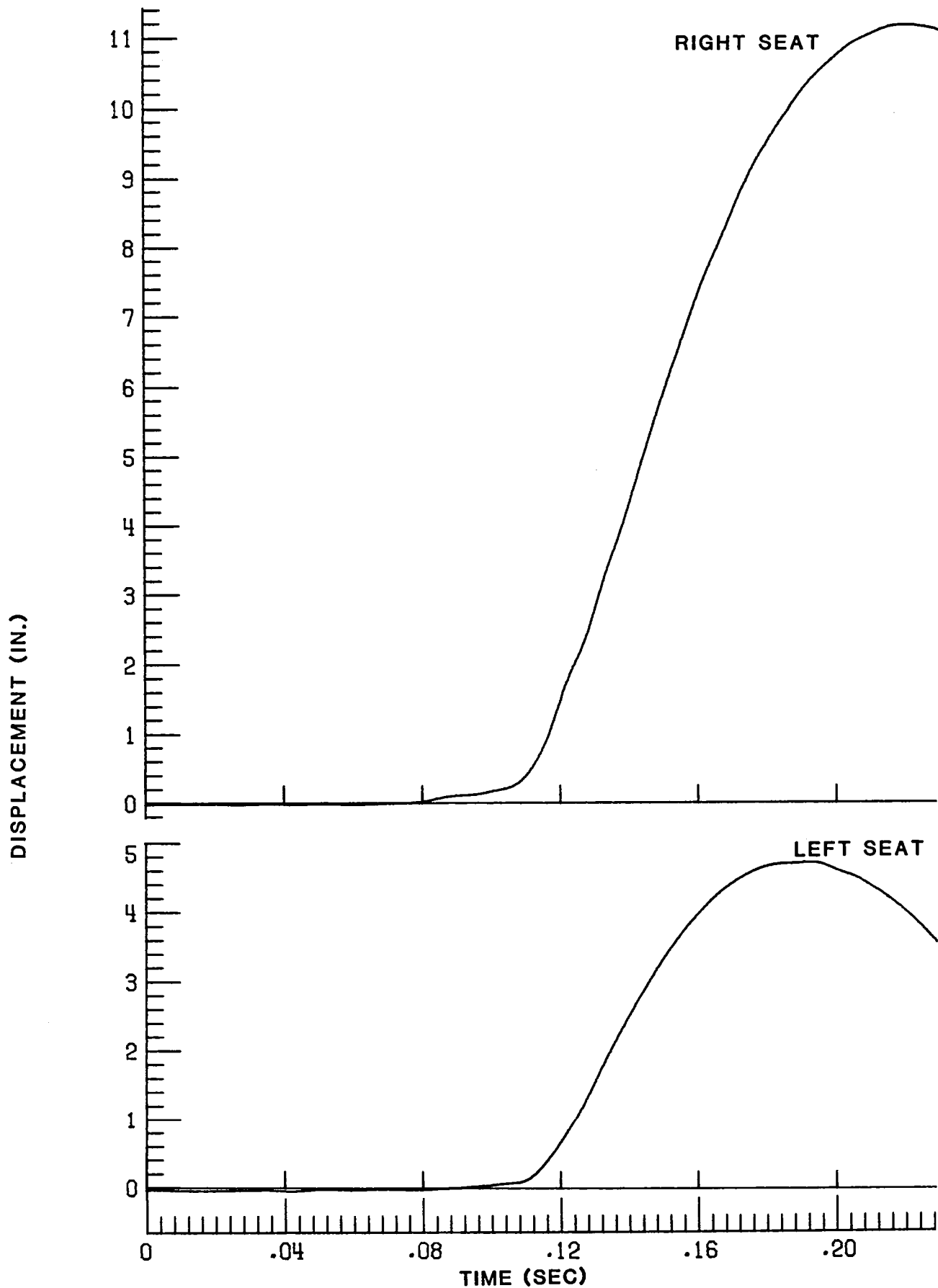


LEFT DUMMY HEAD  
NASA TEST NO. 5

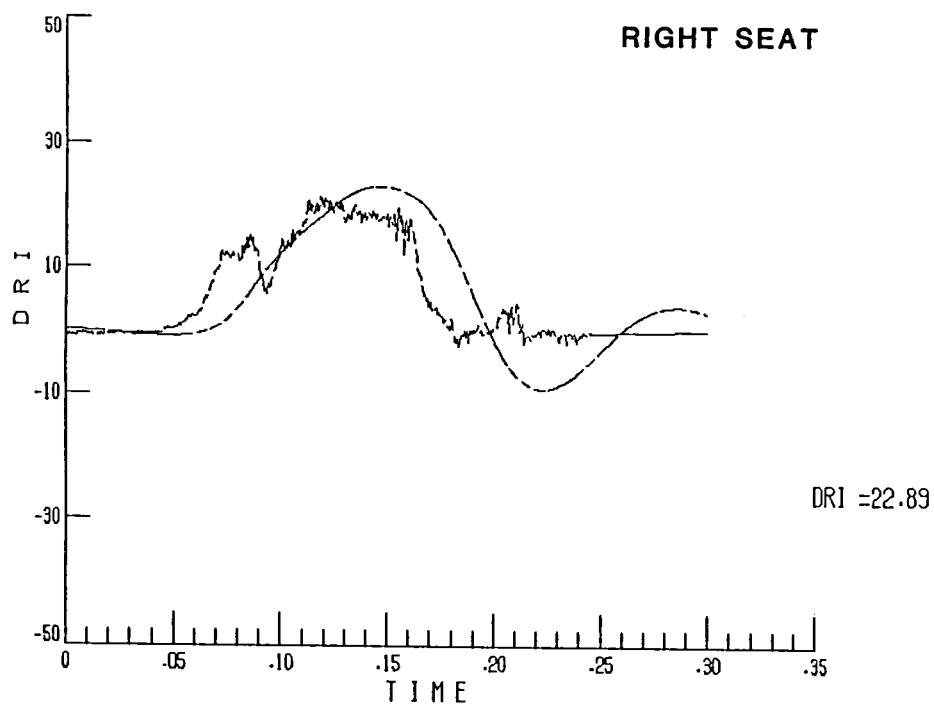
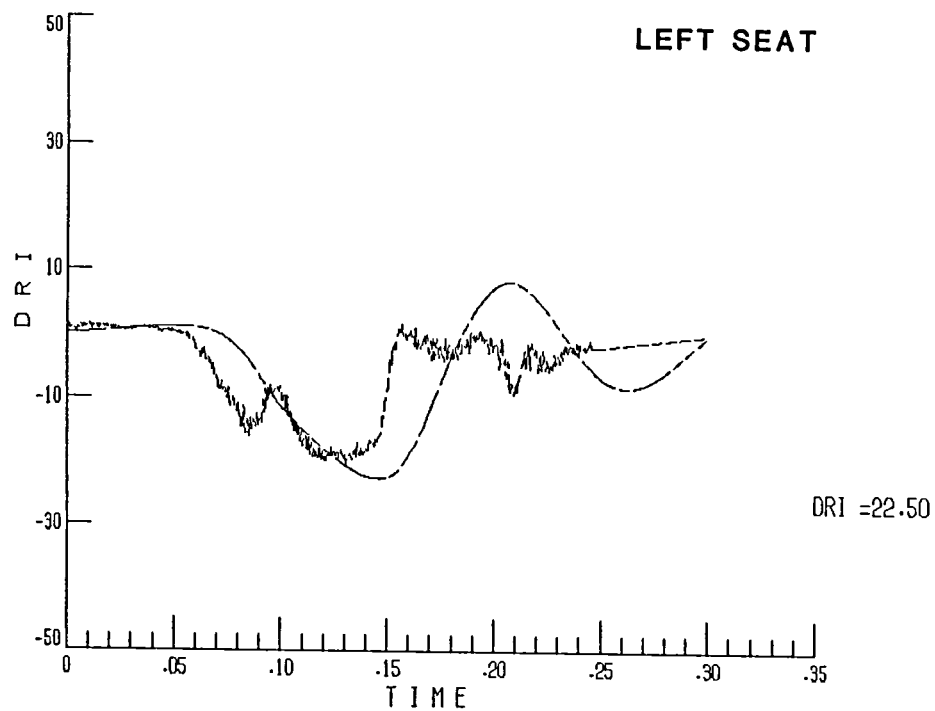


RIGHT DUMMY HEAD

NASA TEST NO. 5

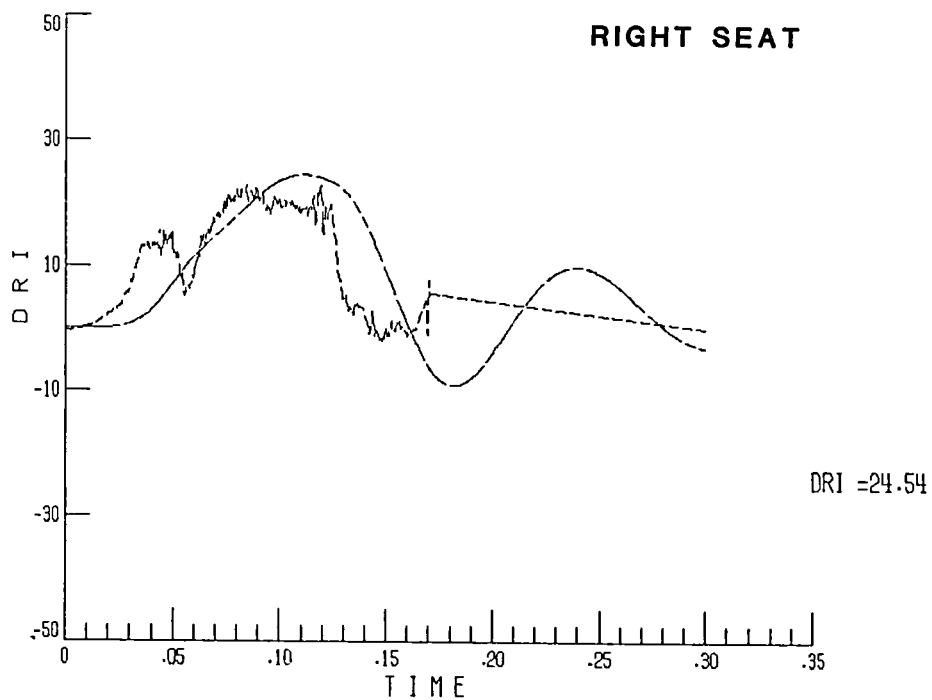
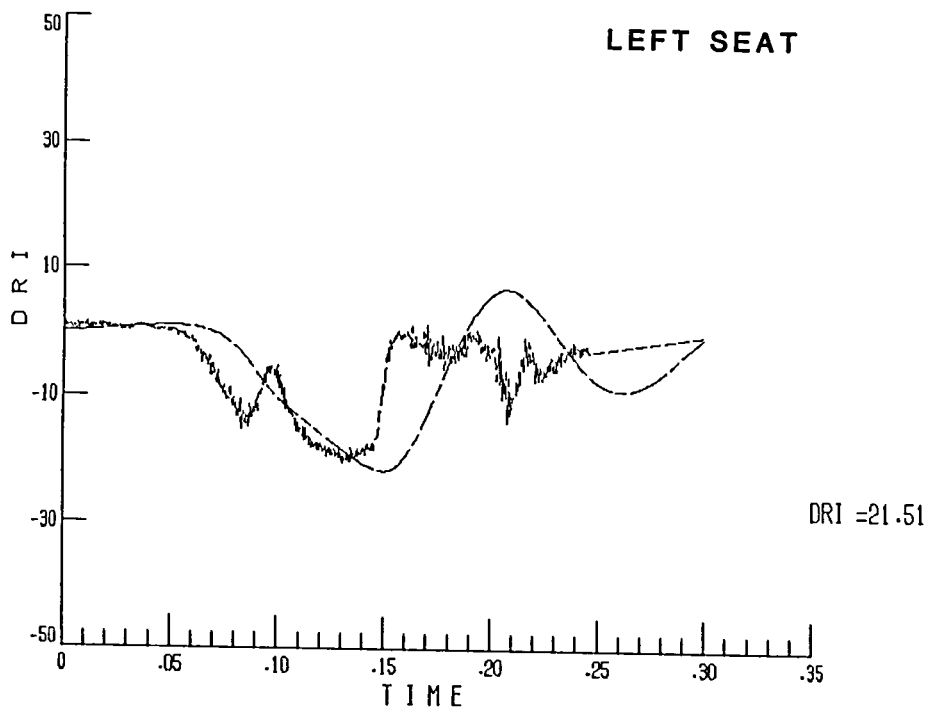


SEAT STROKE  
NASA TEST NO. 5

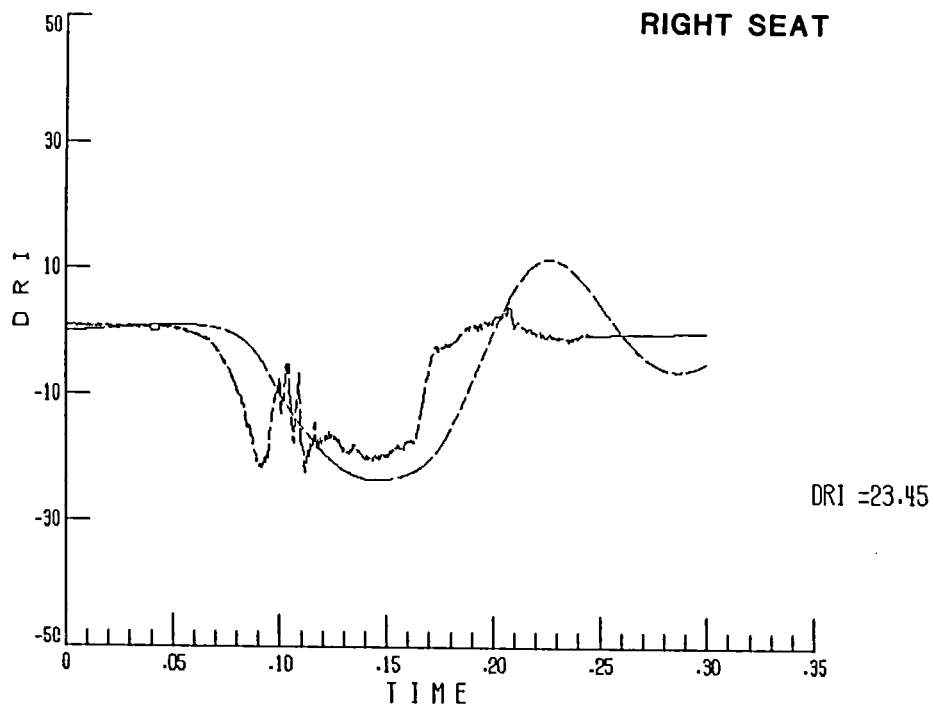
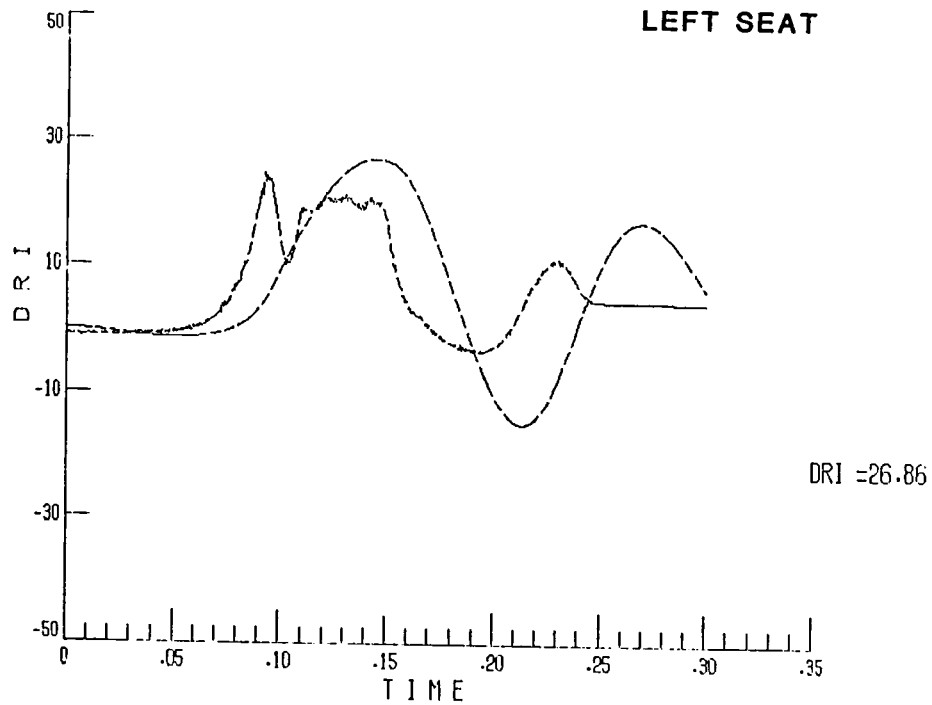


**DYNAMIC RESPONSE INDEX  
SEAT PAN Z-AXIS  
NASA TEST NO. 5**





**DYNAMIC RESPONSE INDEX  
BACK TANGENT LINE  
NASA TEST NO. 5**



**DYNAMIC RESPONSE INDEX  
DUMMY PELVIS Z-AXIS  
NASA TEST NO. 5**

## 7.0 NASA TEST NO. 6

HORIZONTAL VELOCITY: 43 ft/sec

VERTICAL VELOCITY: 32 ft/sec

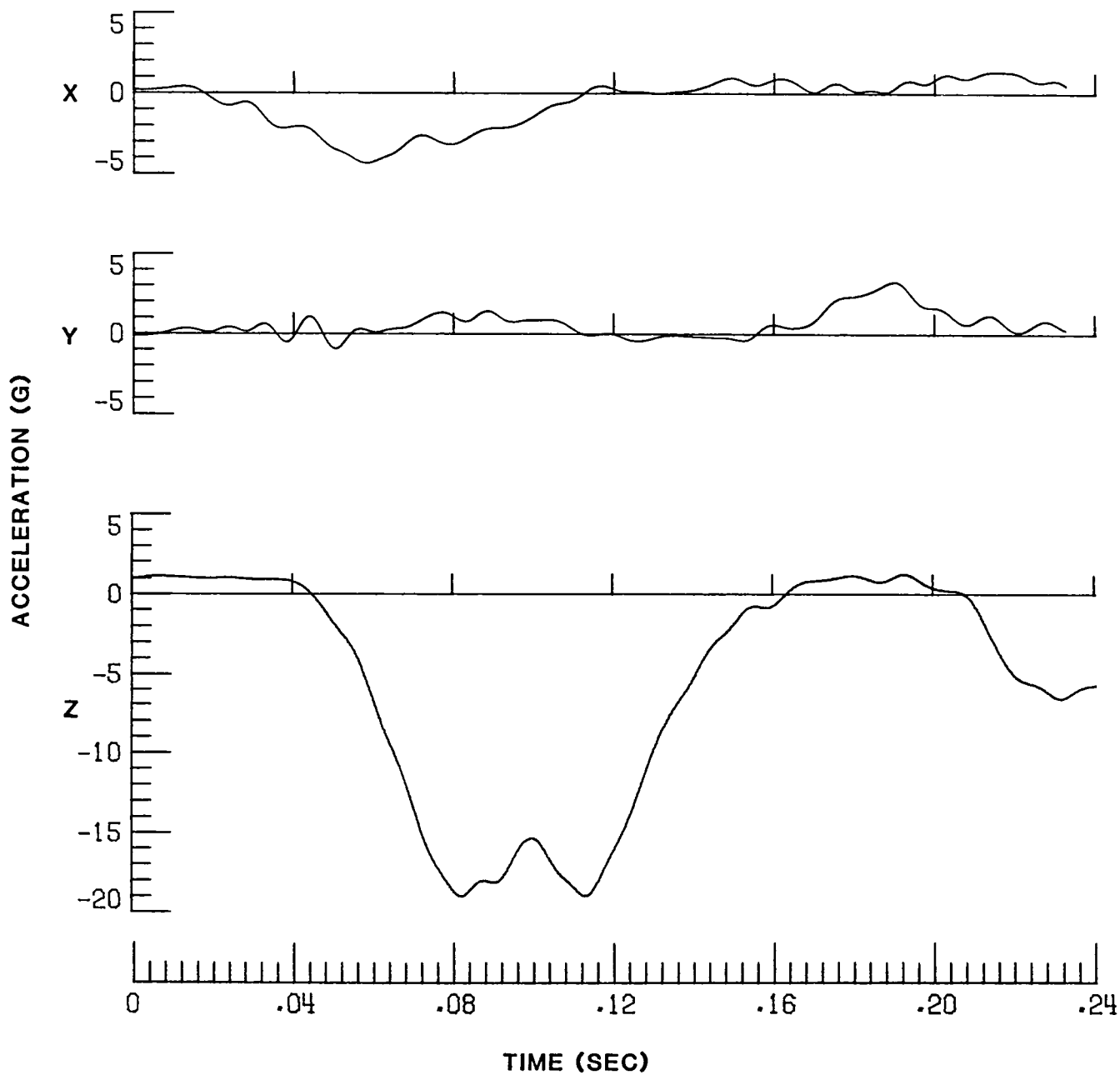
CREW MODULE PITCH:  $2^0$

CREW MODULE ROLL:  $0^0$

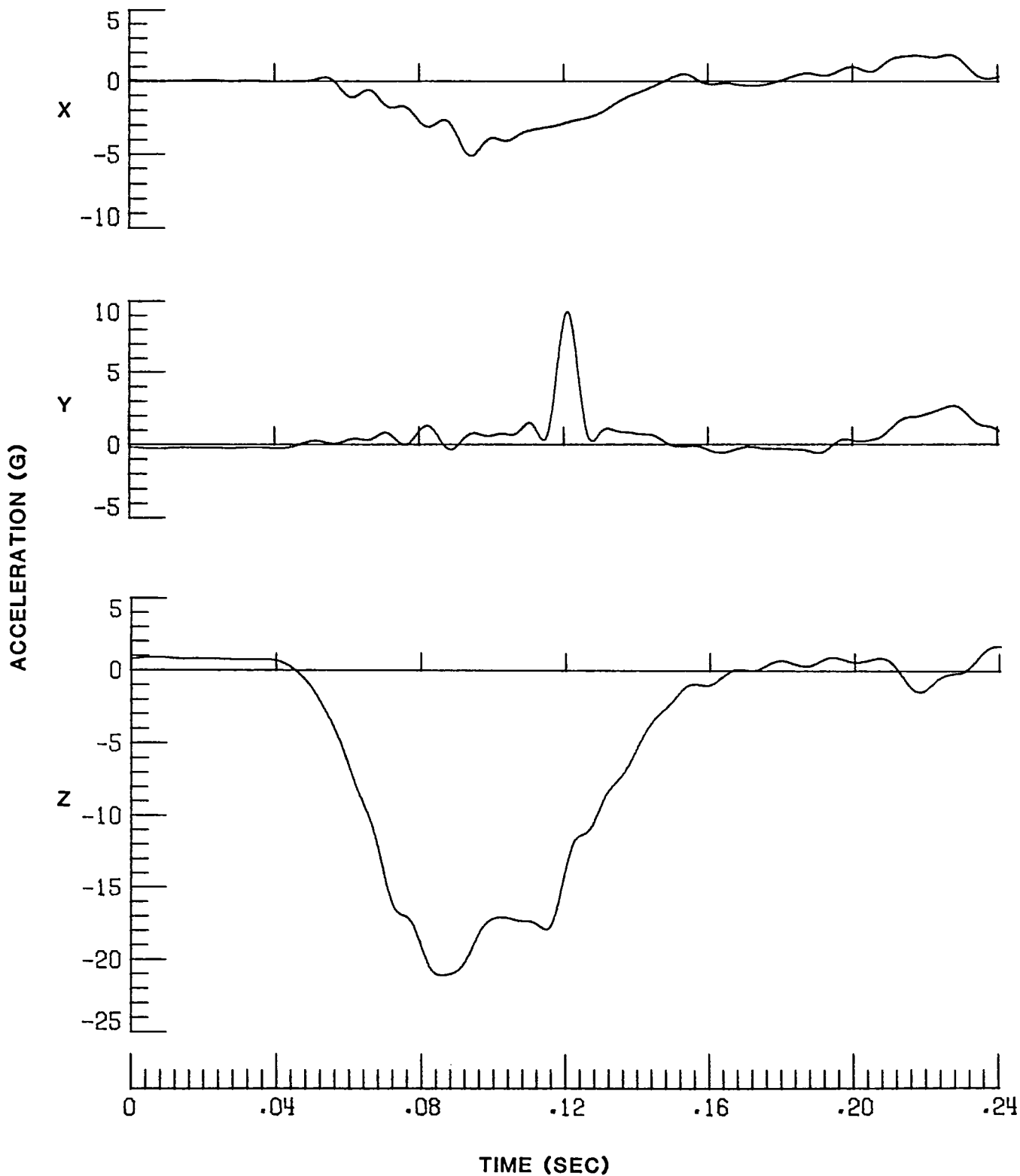
CREW MODULE YAW:  $0^0$

LEFT SEAT: Energy Absorbing

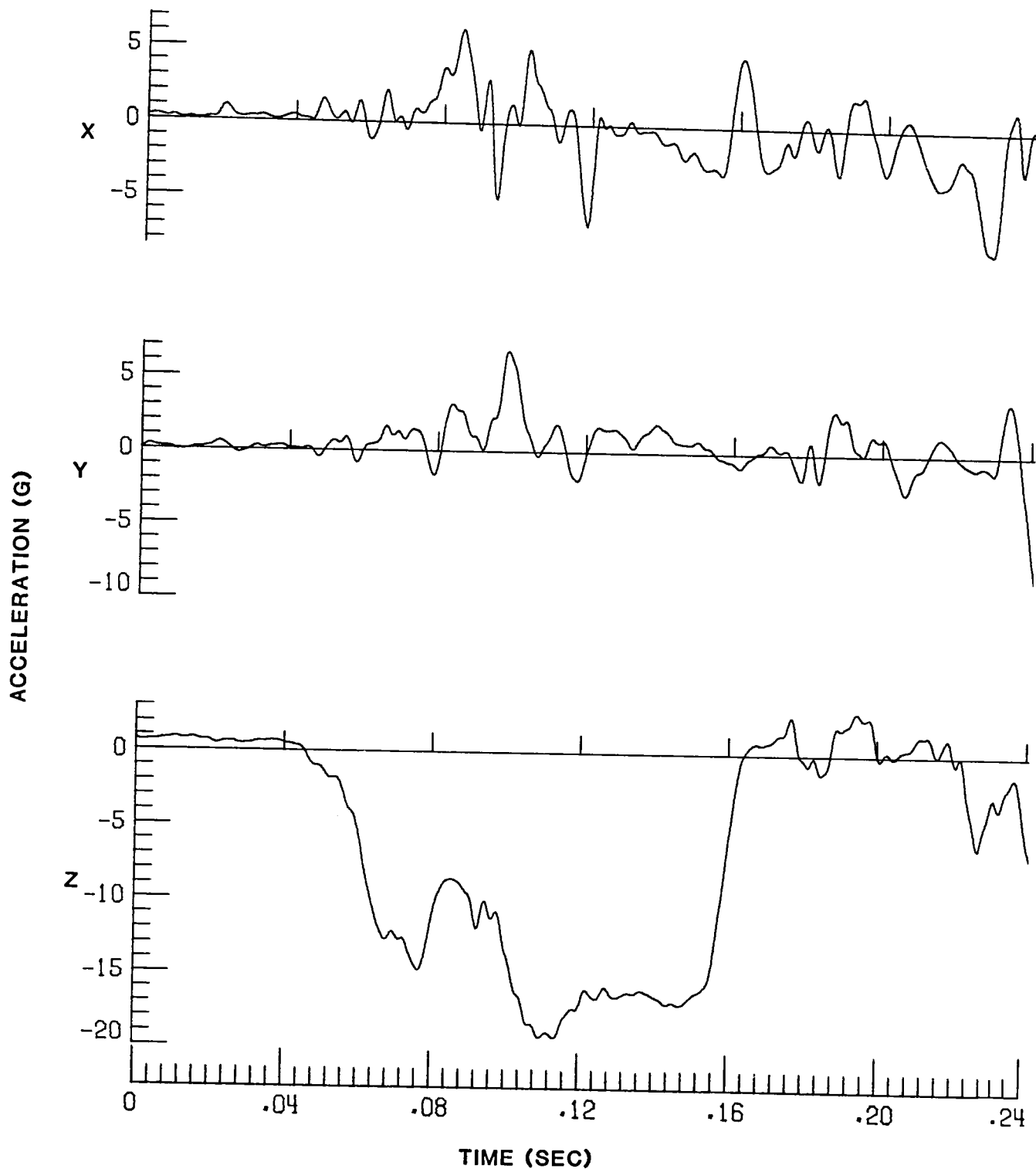
RIGHT SEAT: Energy Absorbing



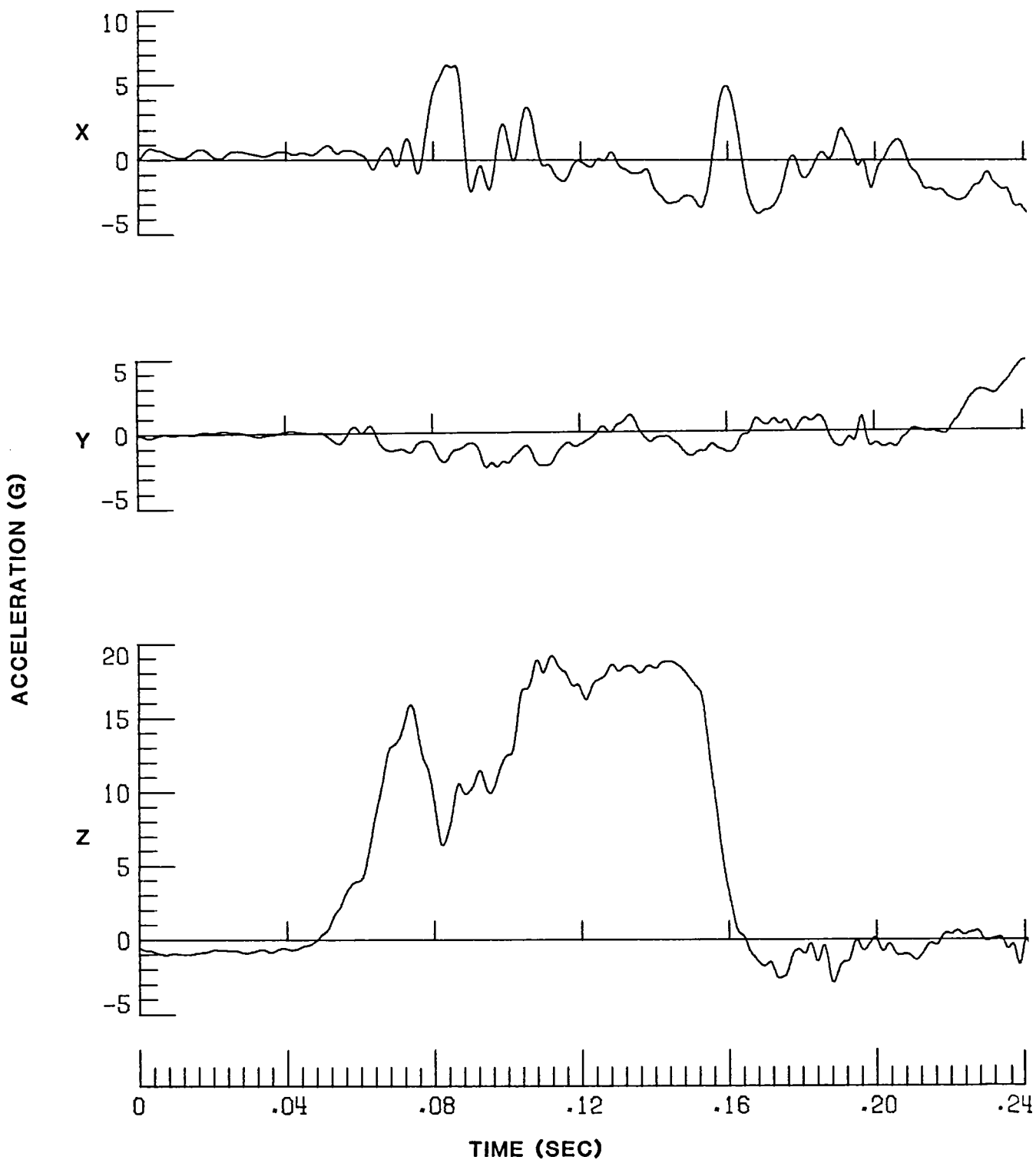
LEFT BULKHEAD  
NASA TEST NO. 6



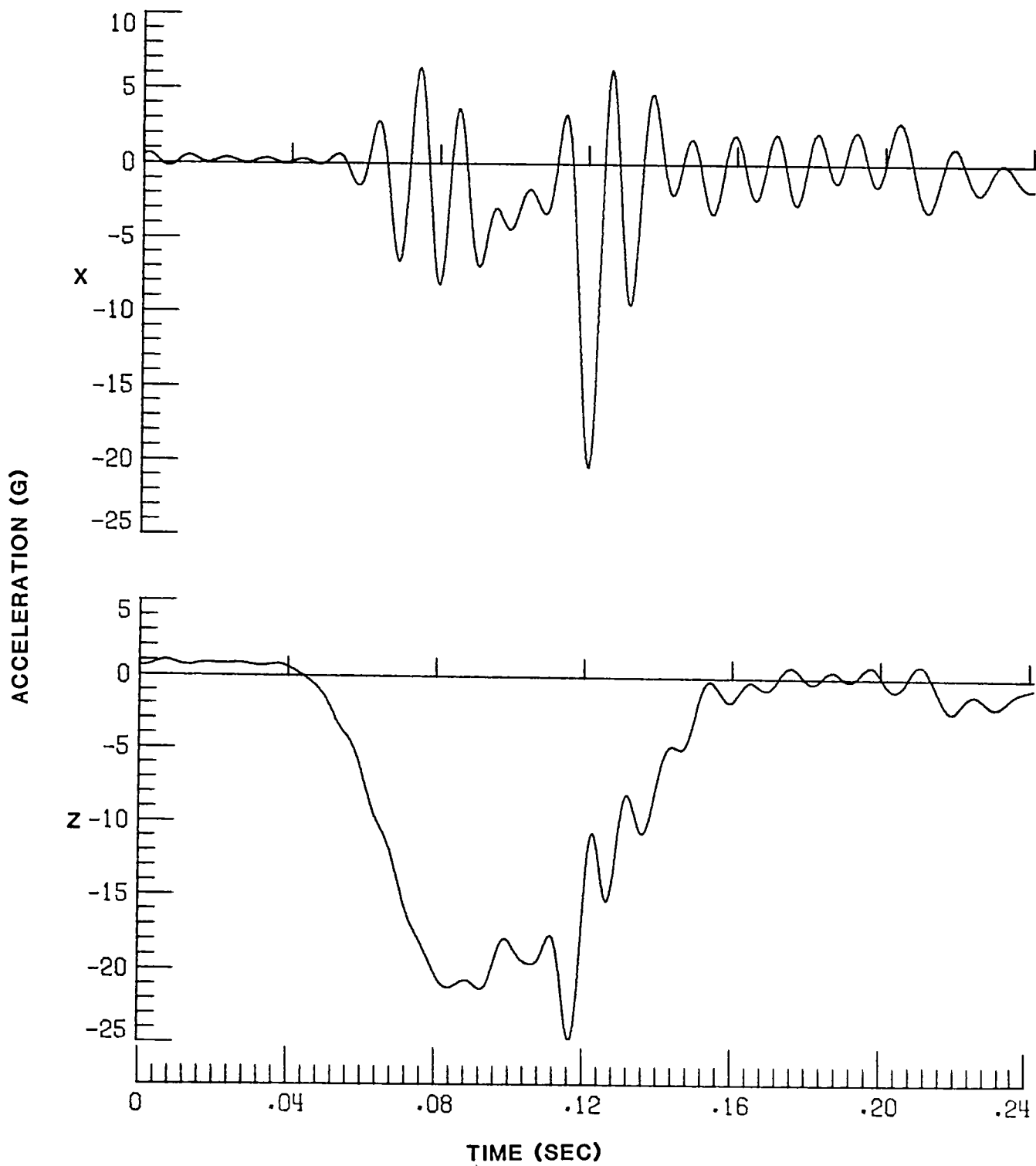
RIGHT BULKHEAD  
NASA TEST NO. 6



LEFT SEAT PAN  
NASA TEST NO. 6



RIGHT SEAT PAN  
NASA TEST NO. 6

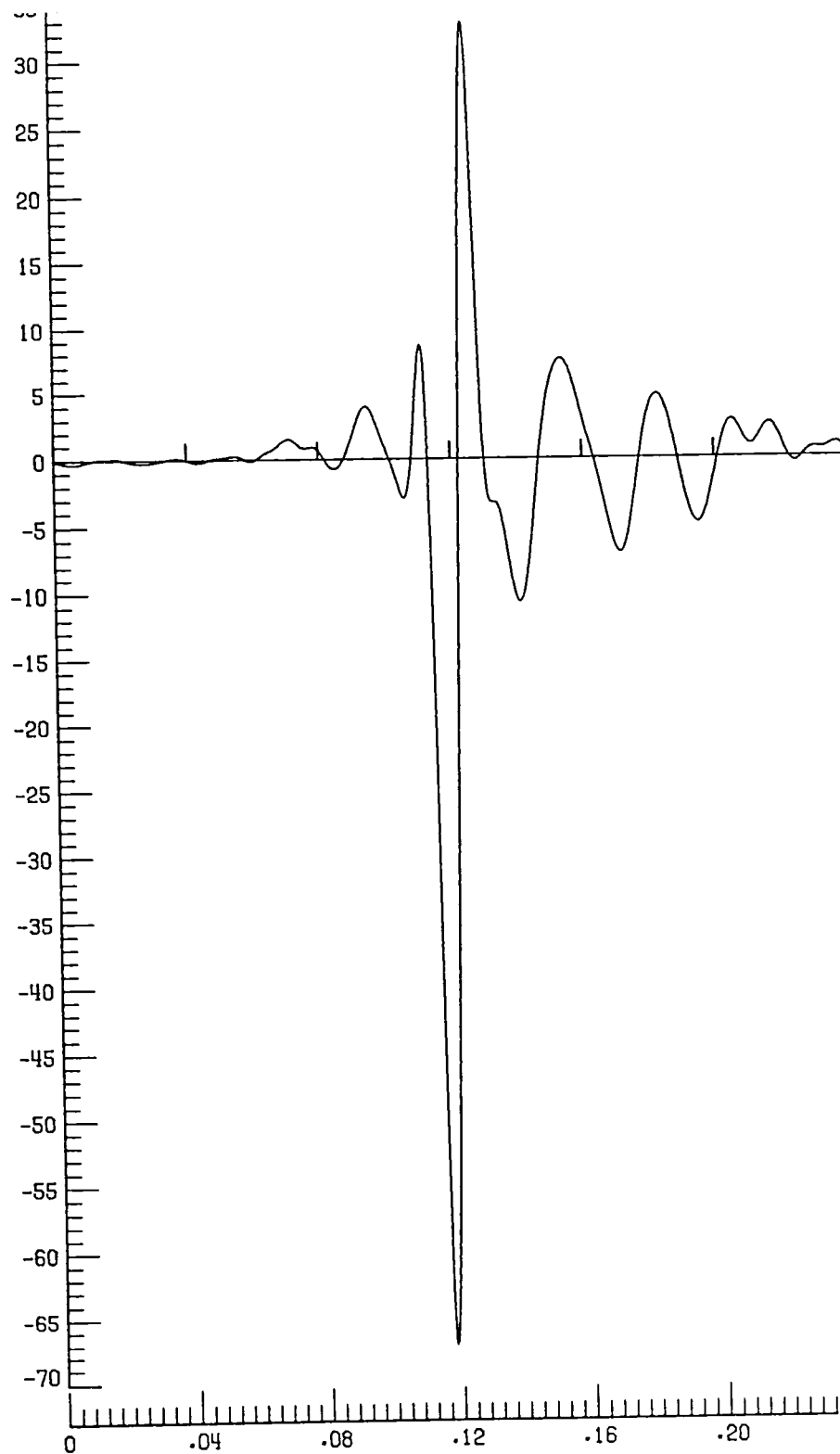


CAPSULE C.G.  
NASA TEST NO. 6



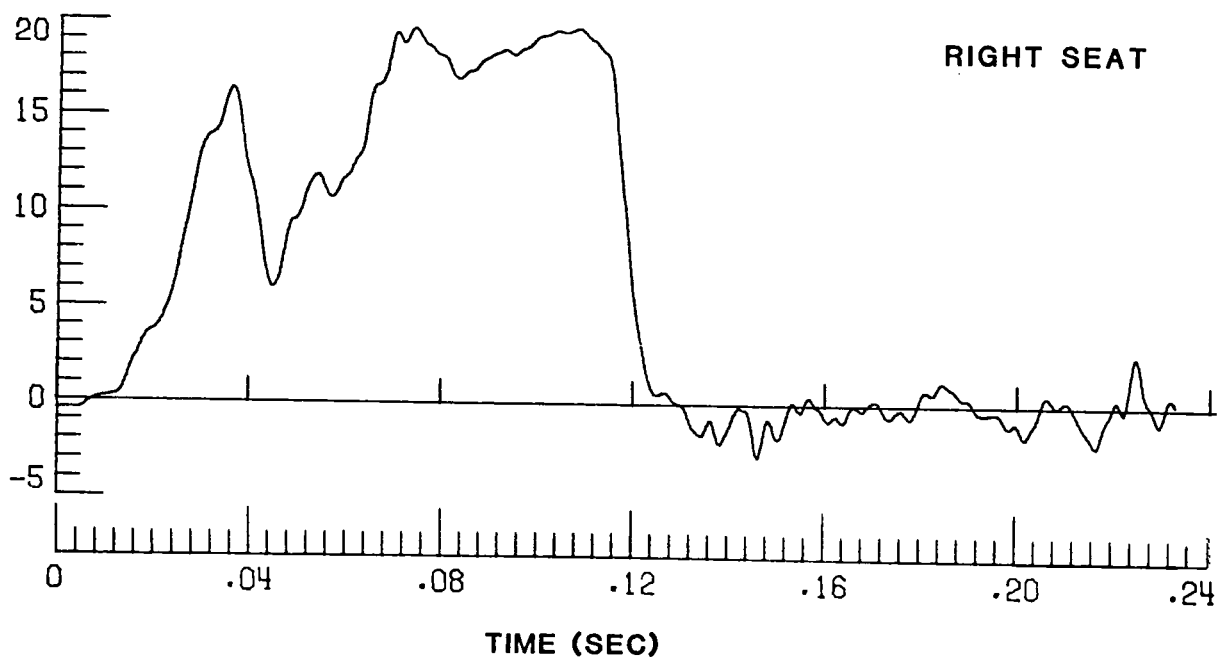
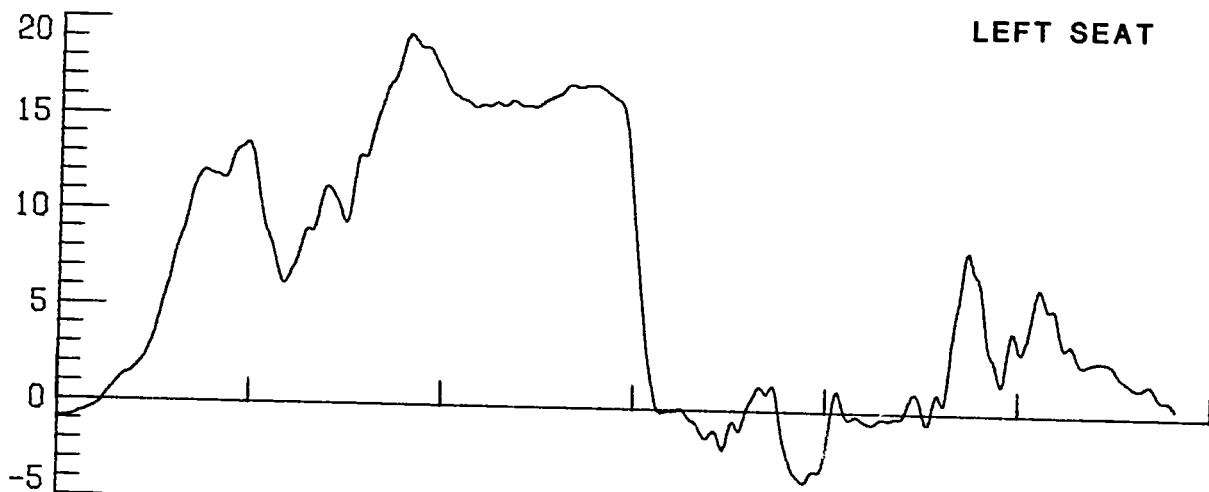
ACCELERATION (G)

Y

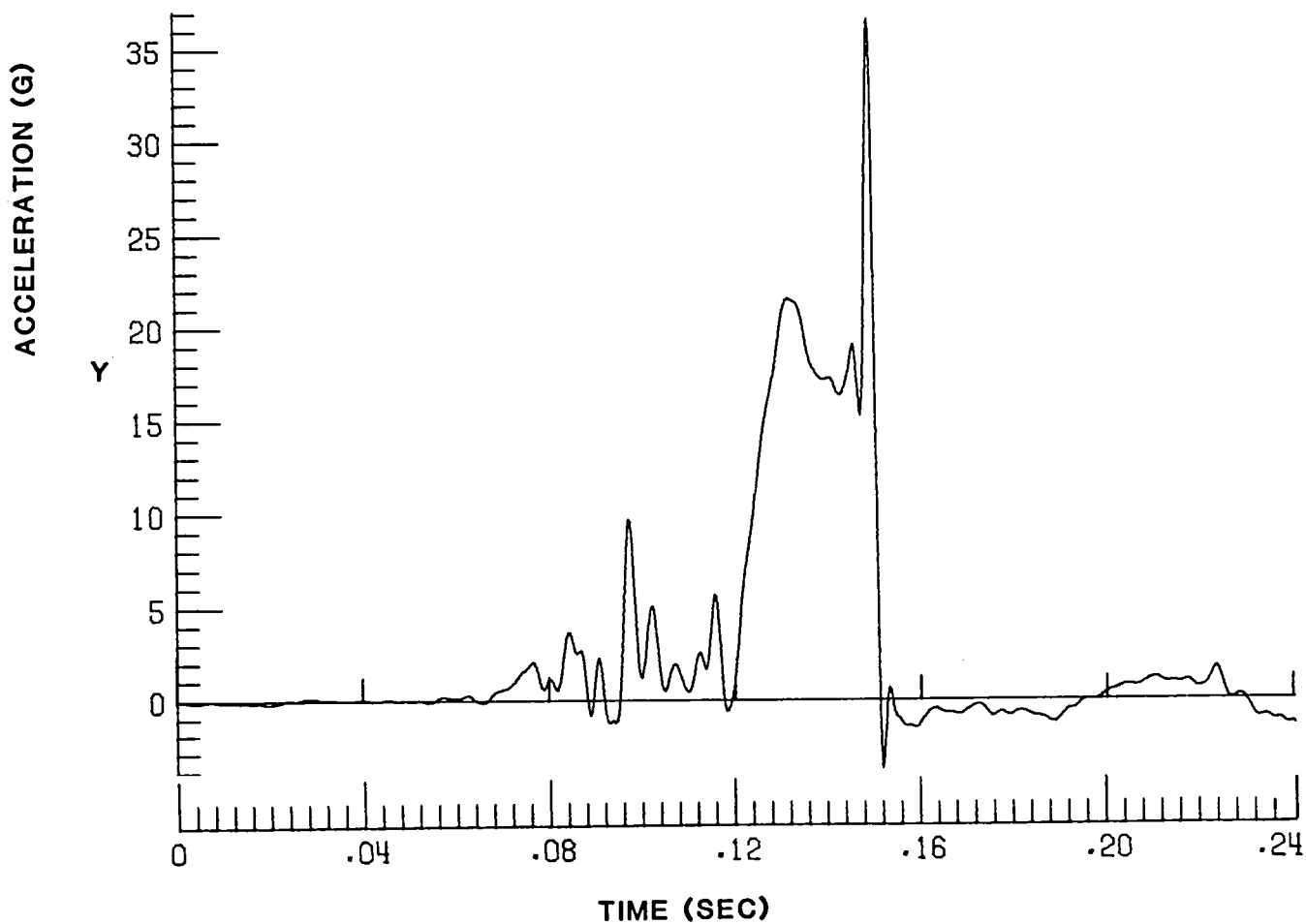
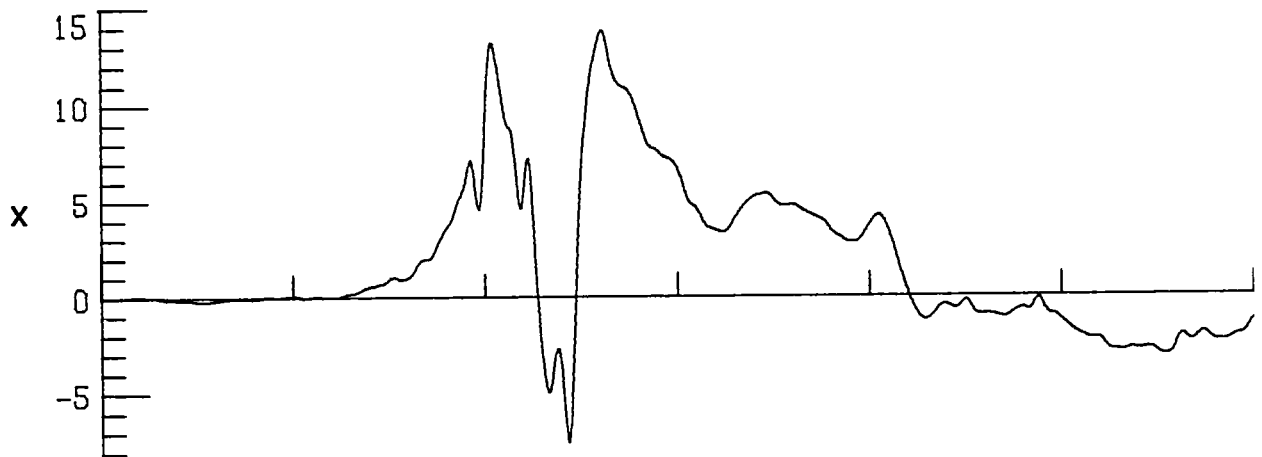


CAPSULE C.G.  
NASA TEST NO. 6

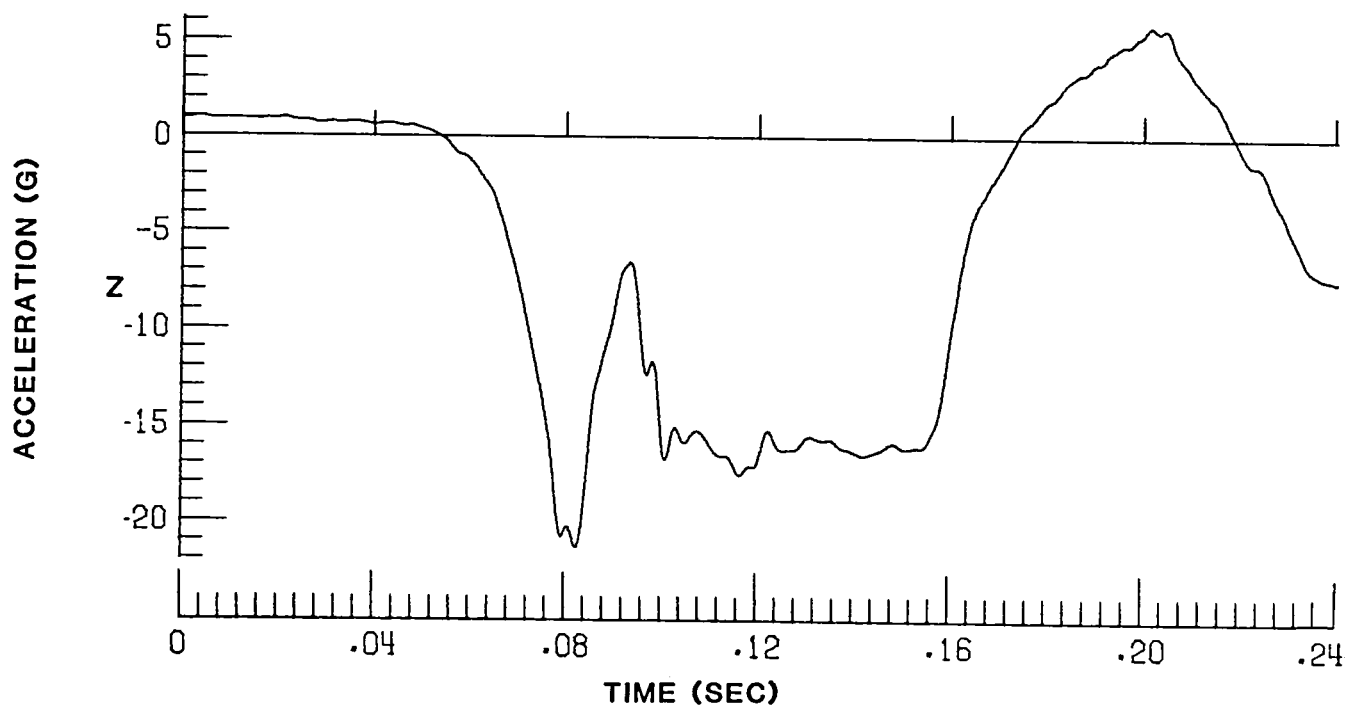
ACCELERATION (G)



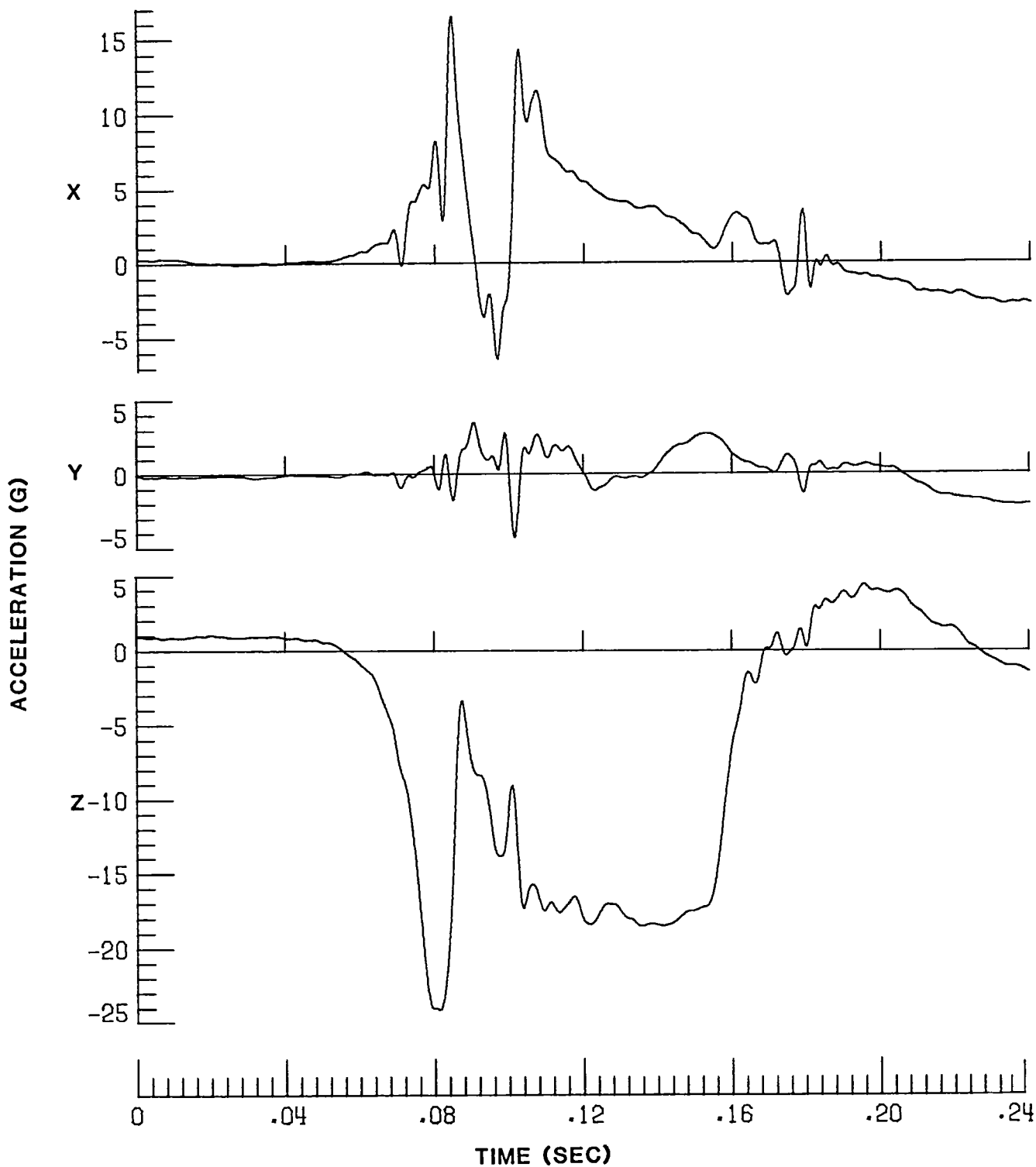
SEAT PAN ALONG BACK TANGENT LINE  
NASA TEST NO. 6  
100



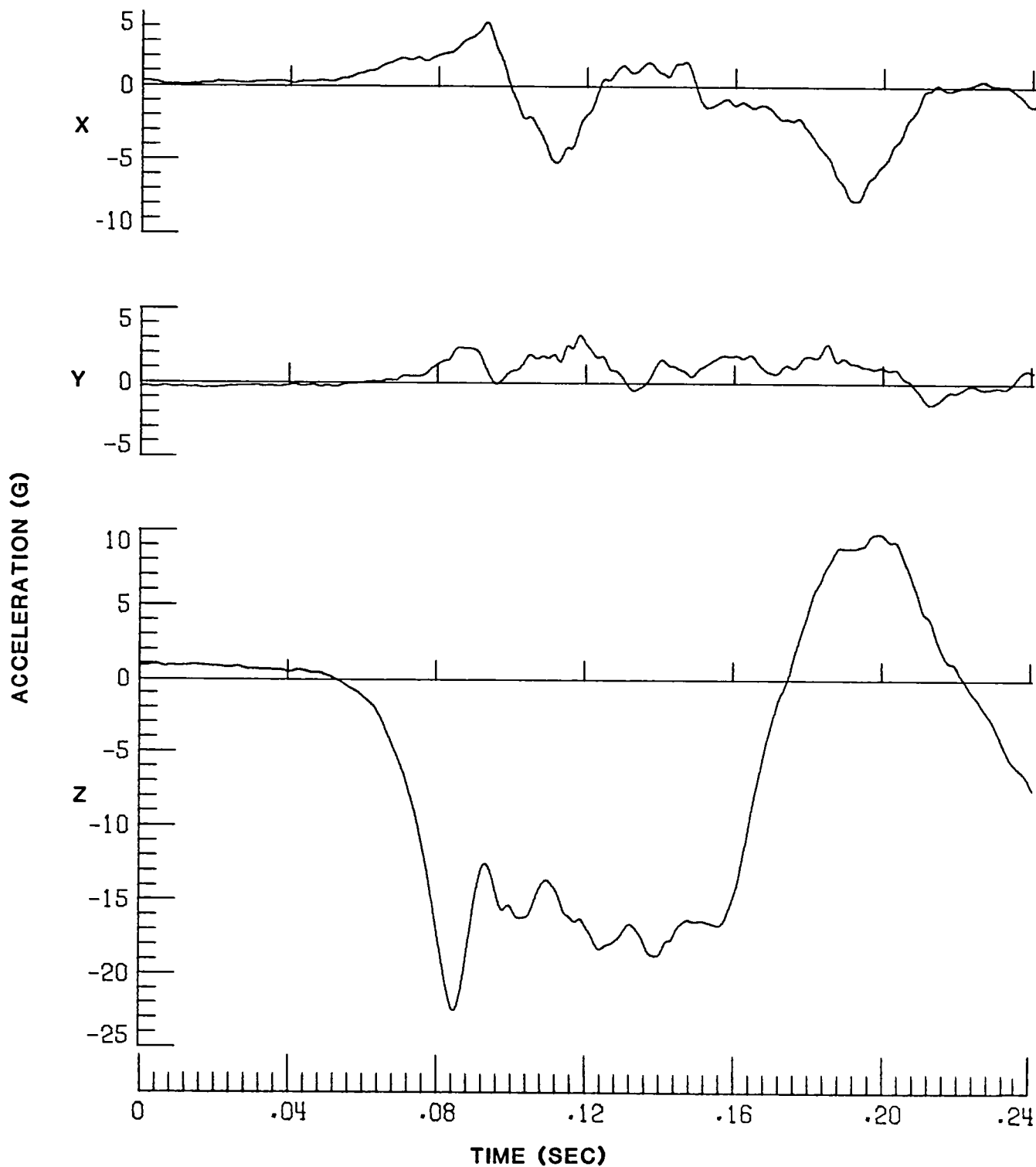
LEFT DUMMY PELVIS  
NASA TEST NO. 6



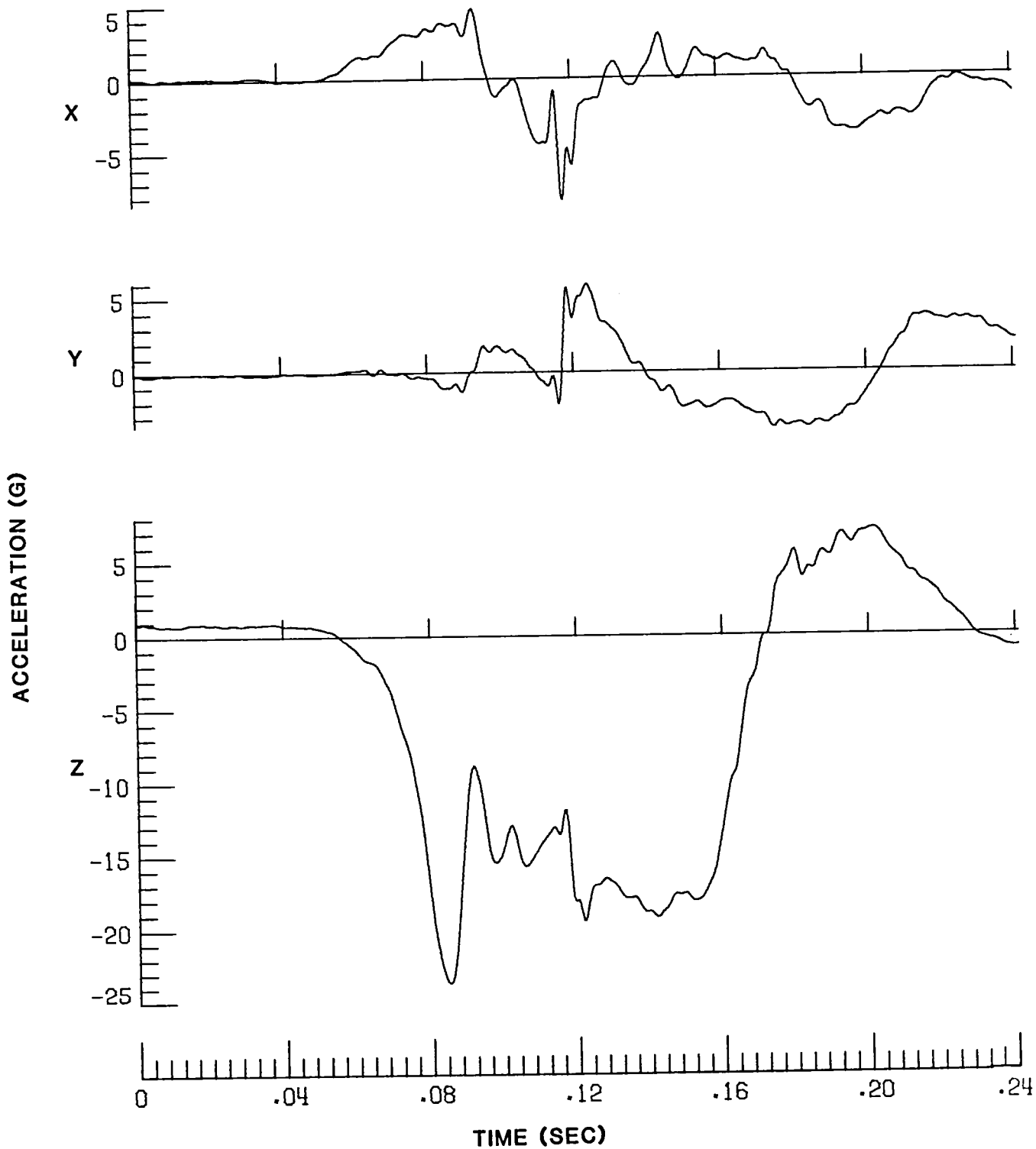
LEFT DUMMY PELVIS  
NASA TEST NO. 6



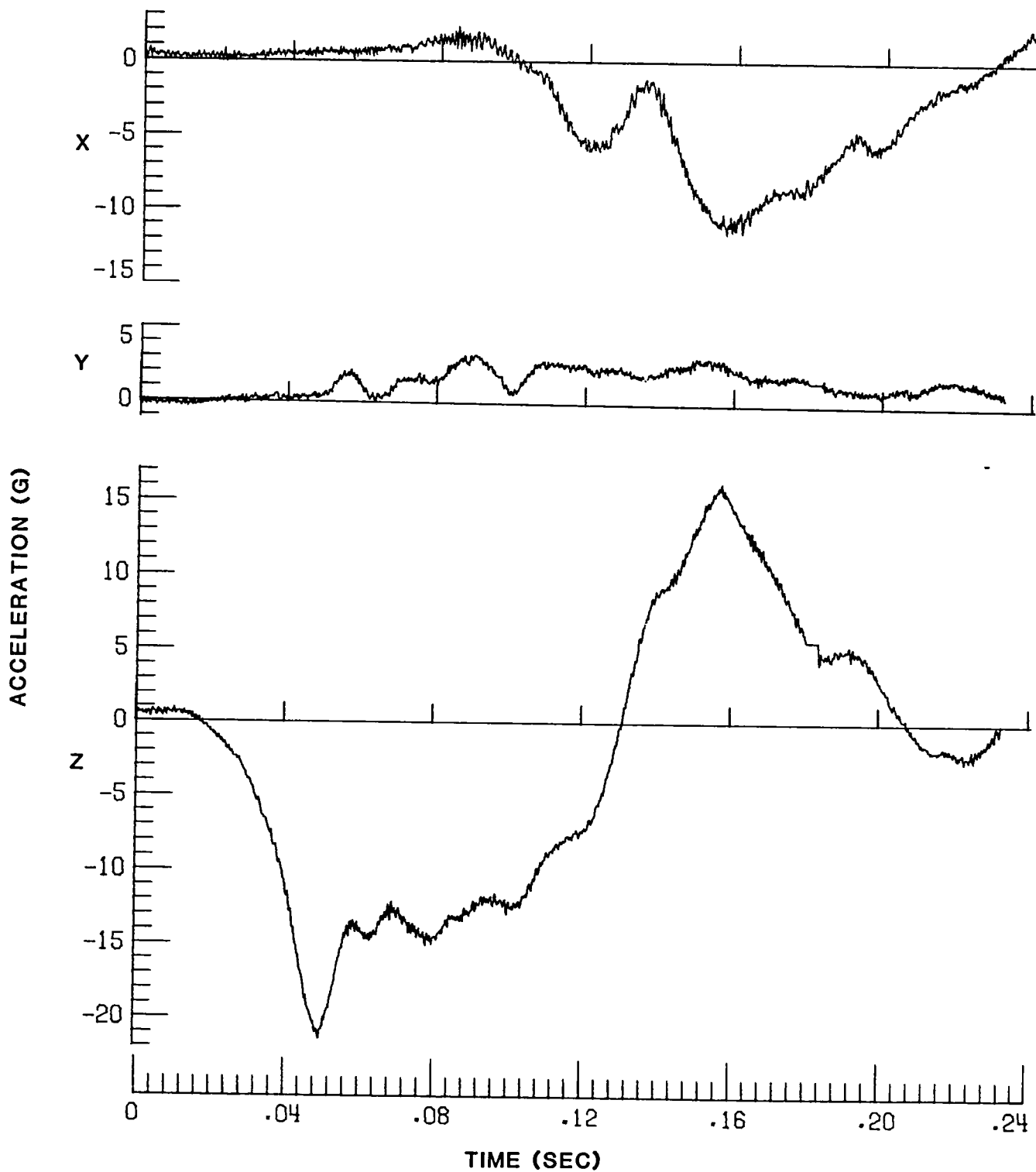
RIGHT DUMMY PELVIS  
NASA TEST NO. 6



LEFT DUMMY CHEST  
NASA TEST NO. 6

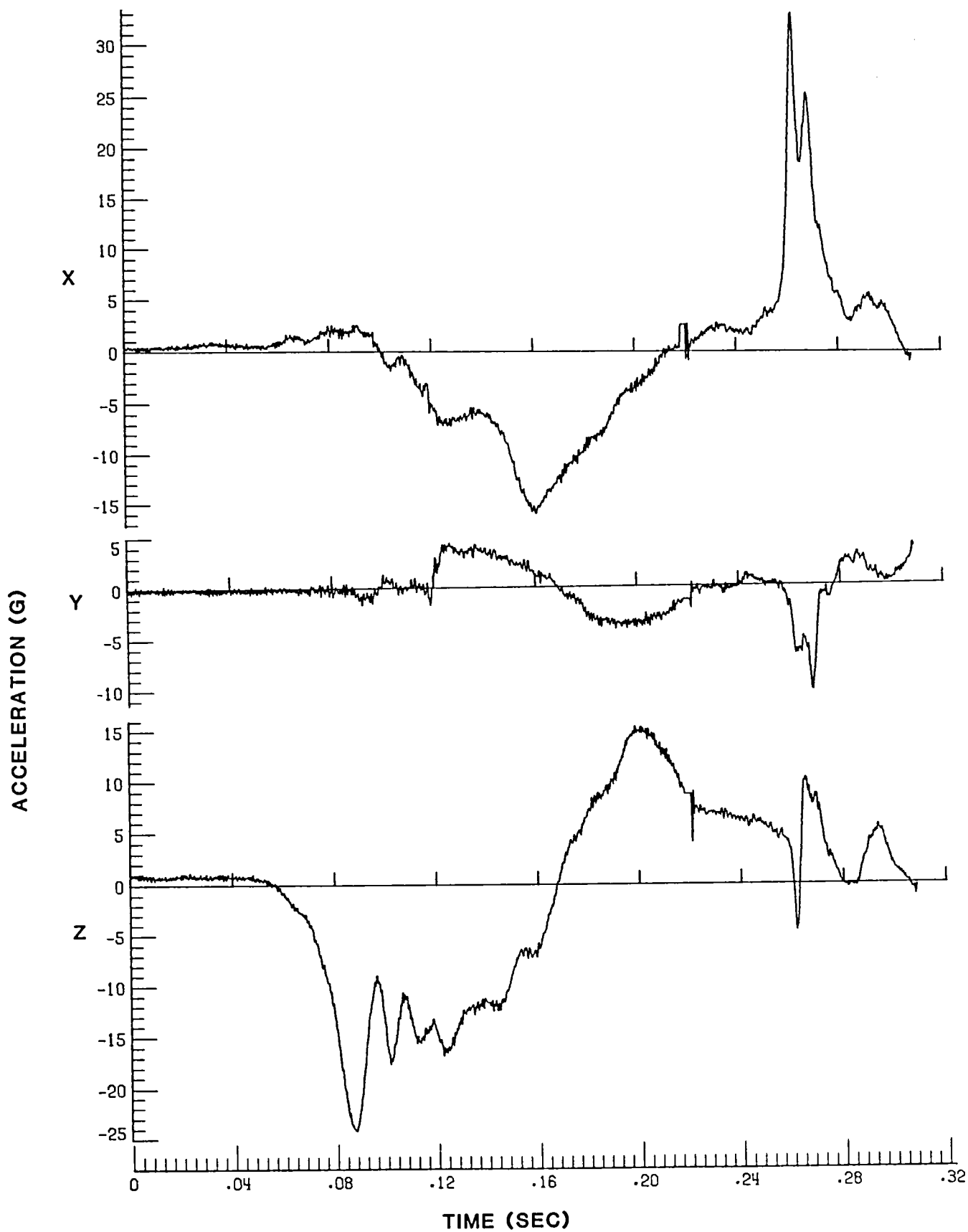


RIGHT DUMMY CHEST  
NASA TEST NO. 6

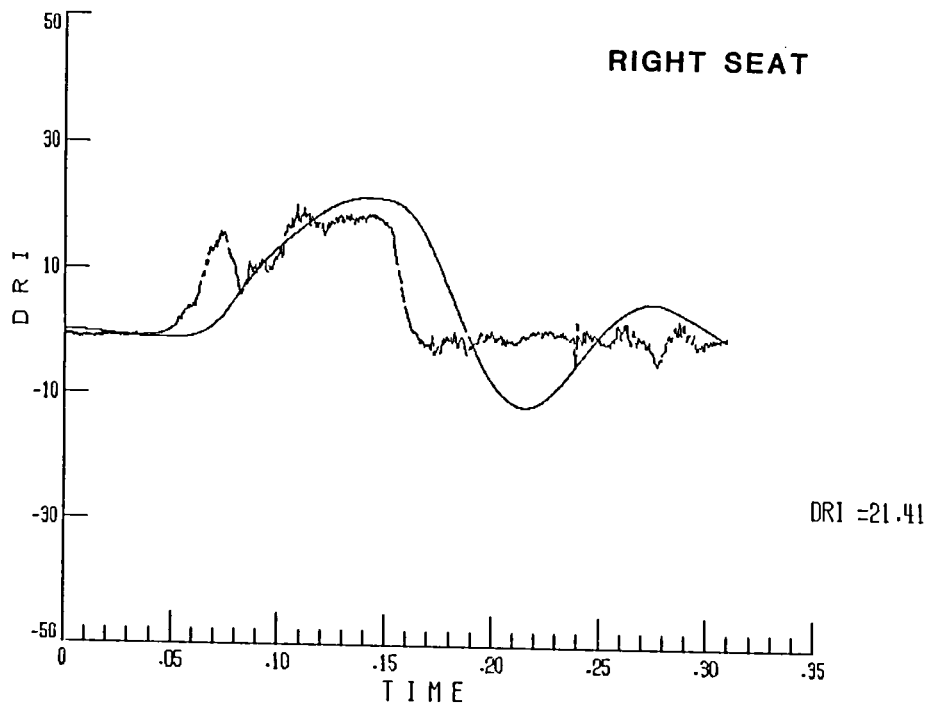
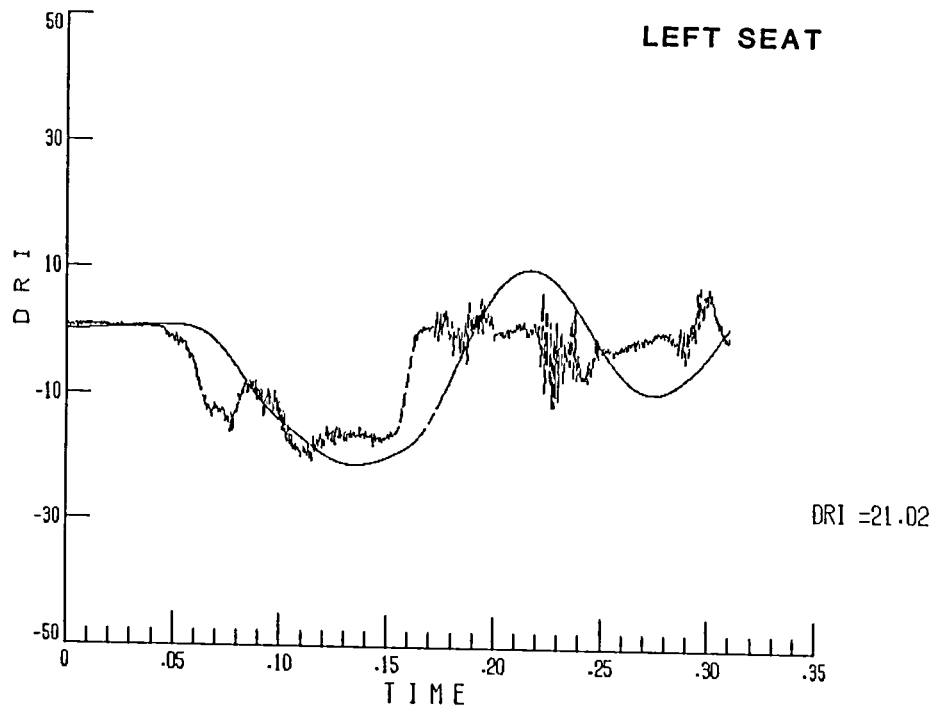


LEFT DUMMY HEAD  
NASA TEST NO. 6

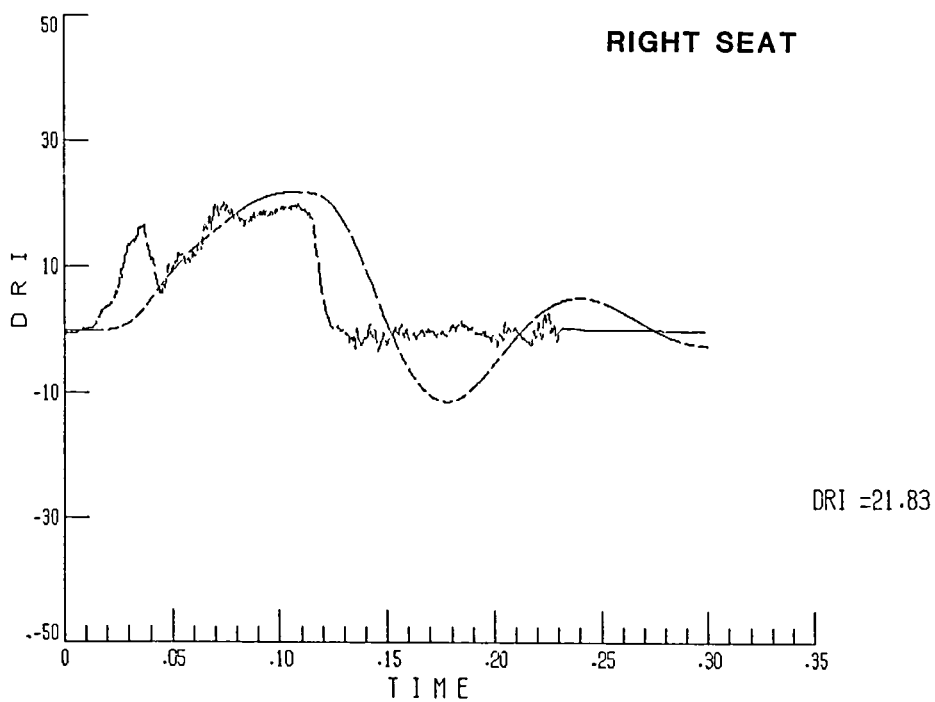
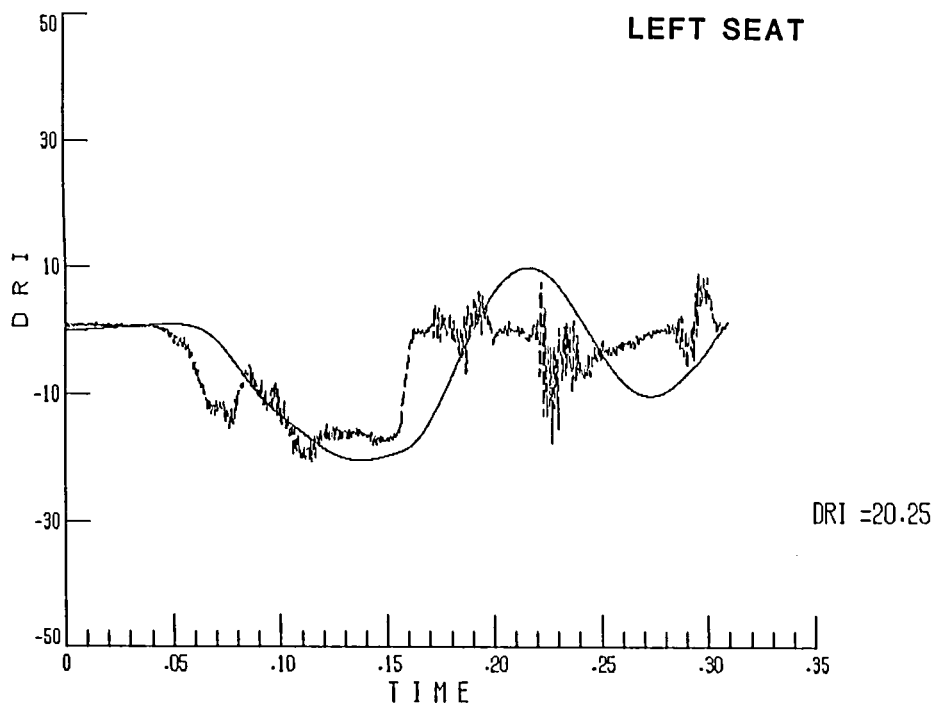




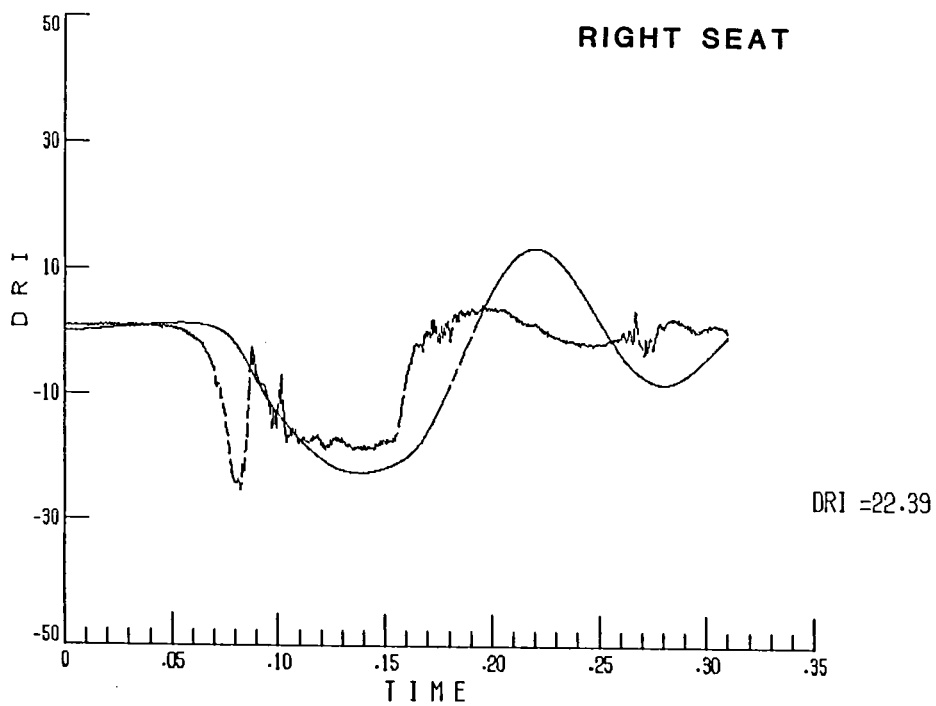
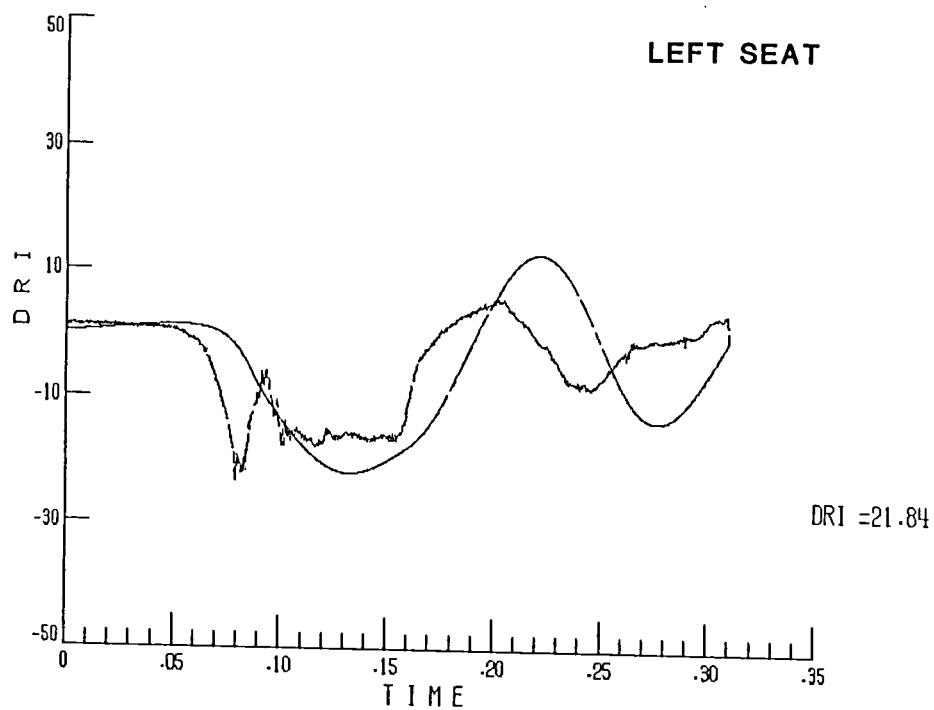
RIGHT DUMMY HEAD  
NASA TEST NO. 6



**DYNAMIC RESPONSE INDEX  
SEAT PAN Z-AXIS  
NASA TEST NO. 6  
108**



**DYNAMIC RESPONSE INDEX  
BACK TANGENT LINE  
NASA TEST NO. 6**



**DYNAMIC RESPONSE INDEX  
DUMMY PELVIS Z-AXIS  
NASA TEST NO. 6**

#### 8.0 NASA TEST NO. 7

HORIZONTAL VELOCITY: 43 ft/sec

VERTICAL VELOCITY: 32 ft/sec

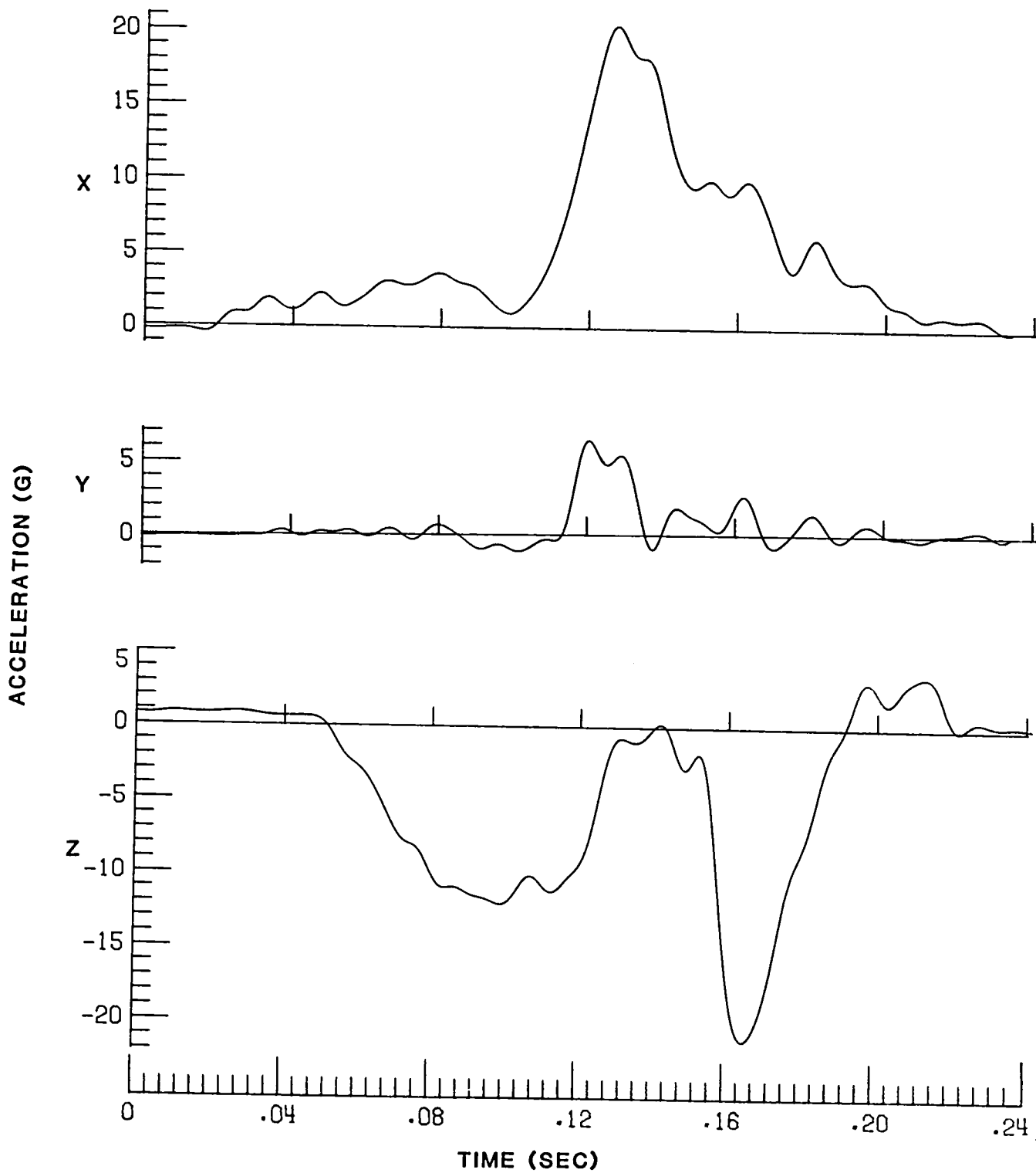
CREW MODULE PITCH:  $7^{\circ}$

CREW MODULE ROLL:  $0^{\circ}$

CREW MODULE YAW:  $180^{\circ}$

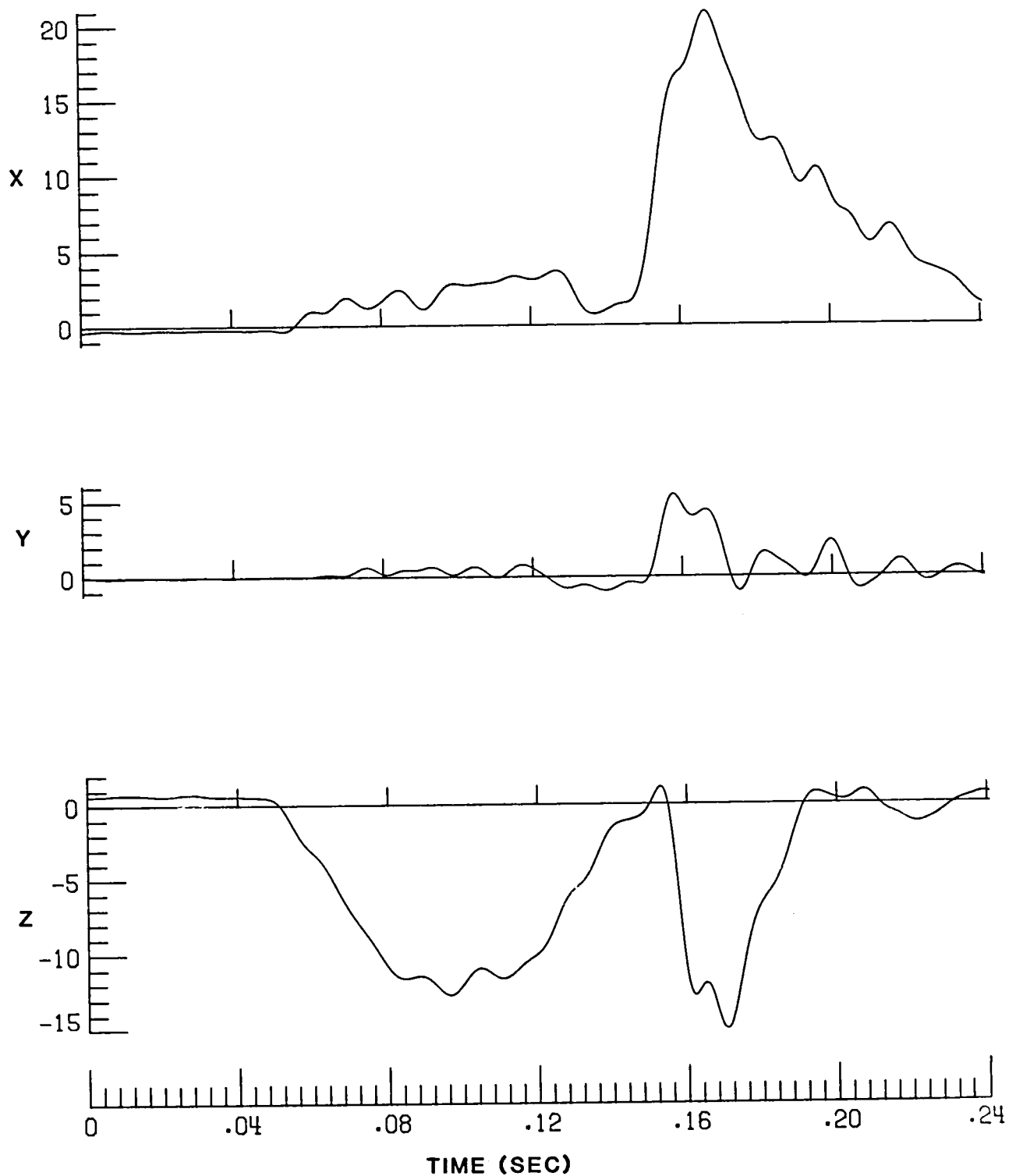
LEFT SEAT: Energy Absorbing

RIGHT SEAT: Energy Absorbing

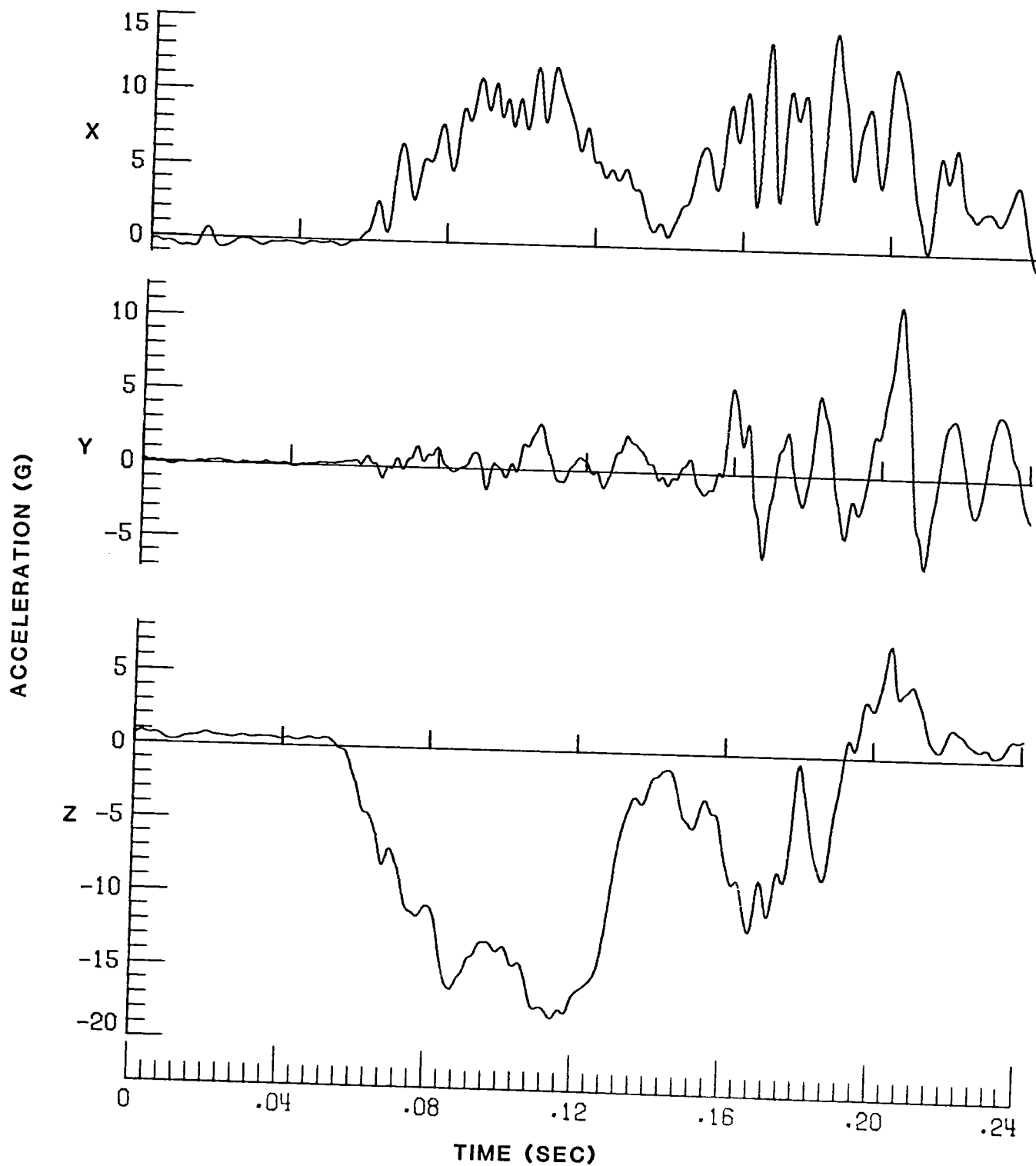


LEFT BULKHEAD  
NASA TEST NO. 7

ACCELERATION (G)

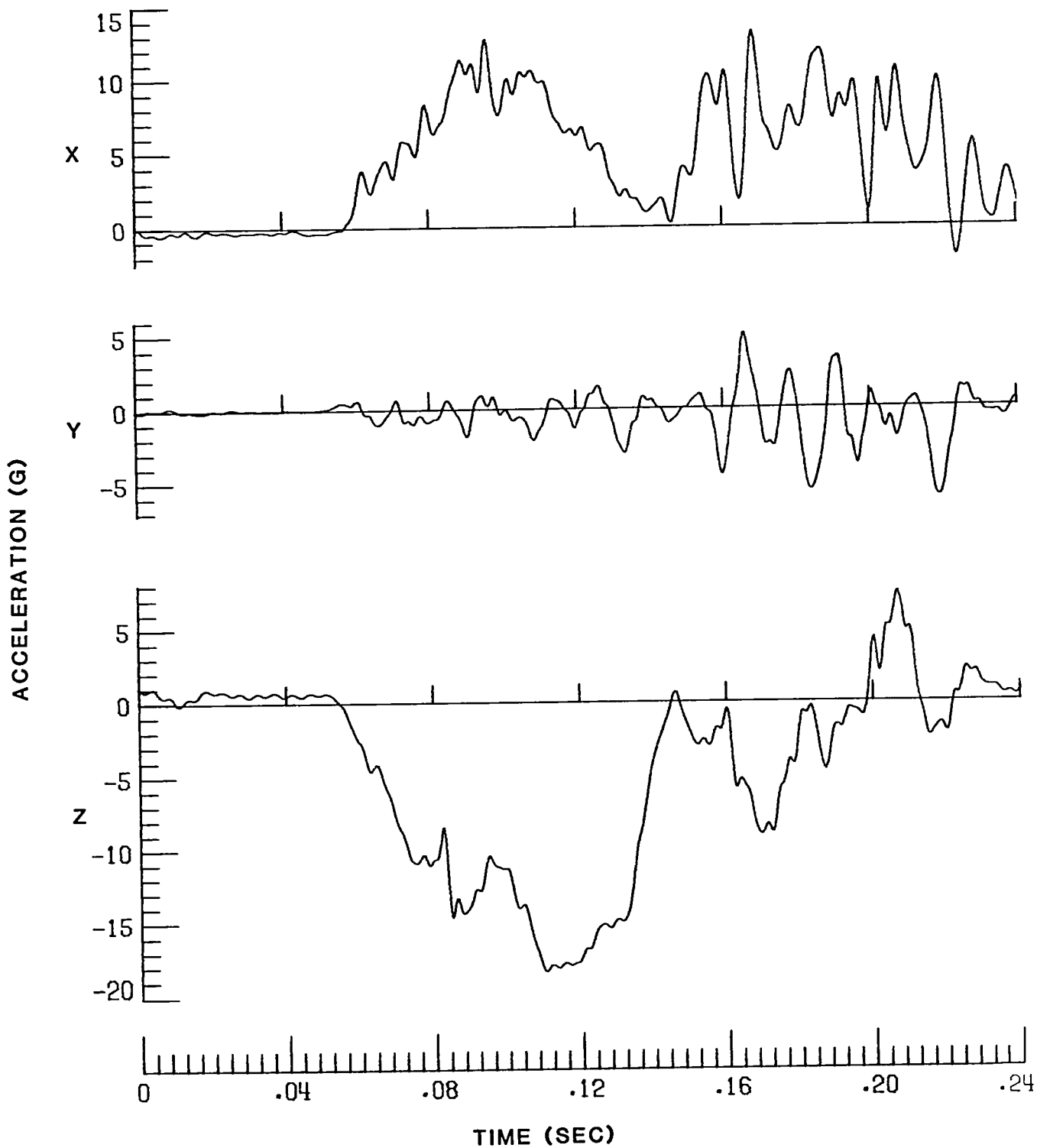


RIGHT BULKHEAD  
NASA TEST NO. 7

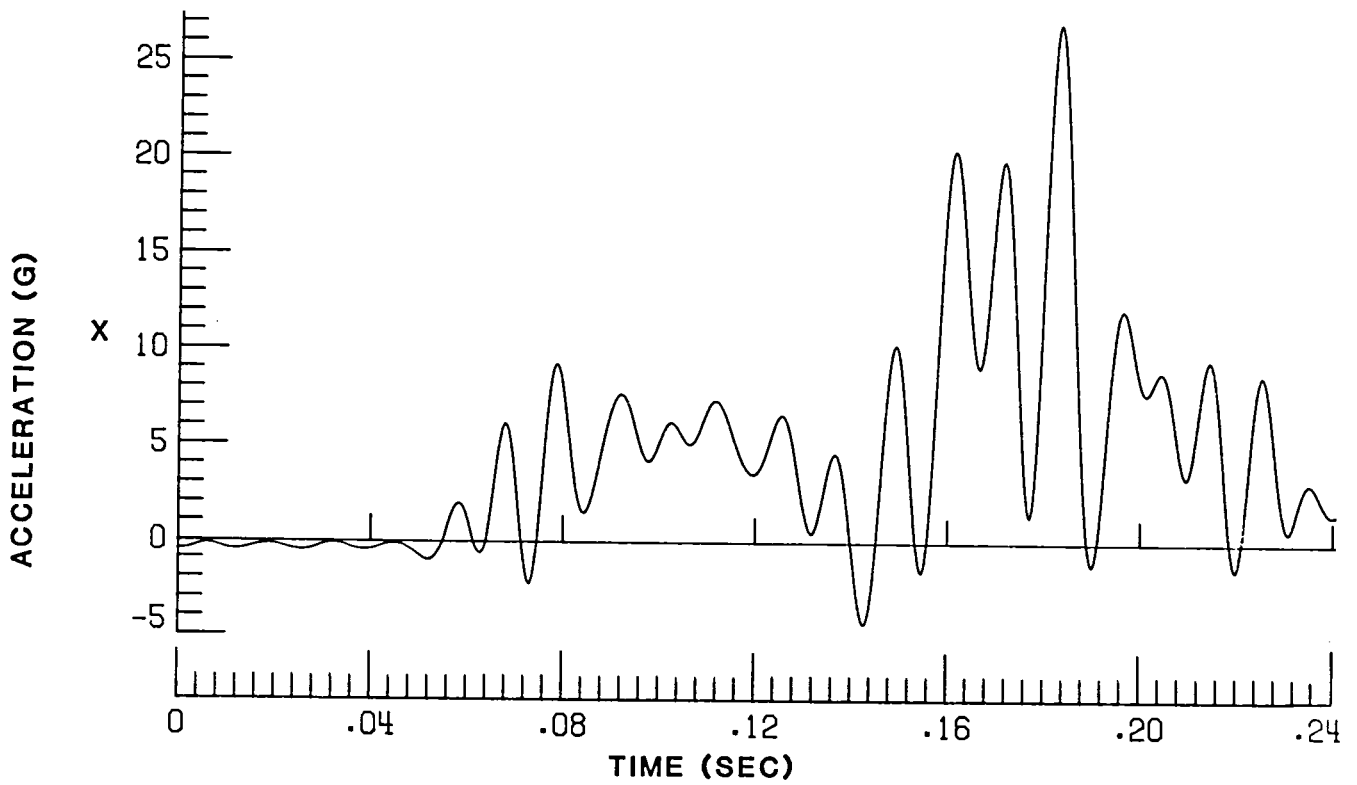


LEFT SEAT PAN  
NASA TEST NO. 7

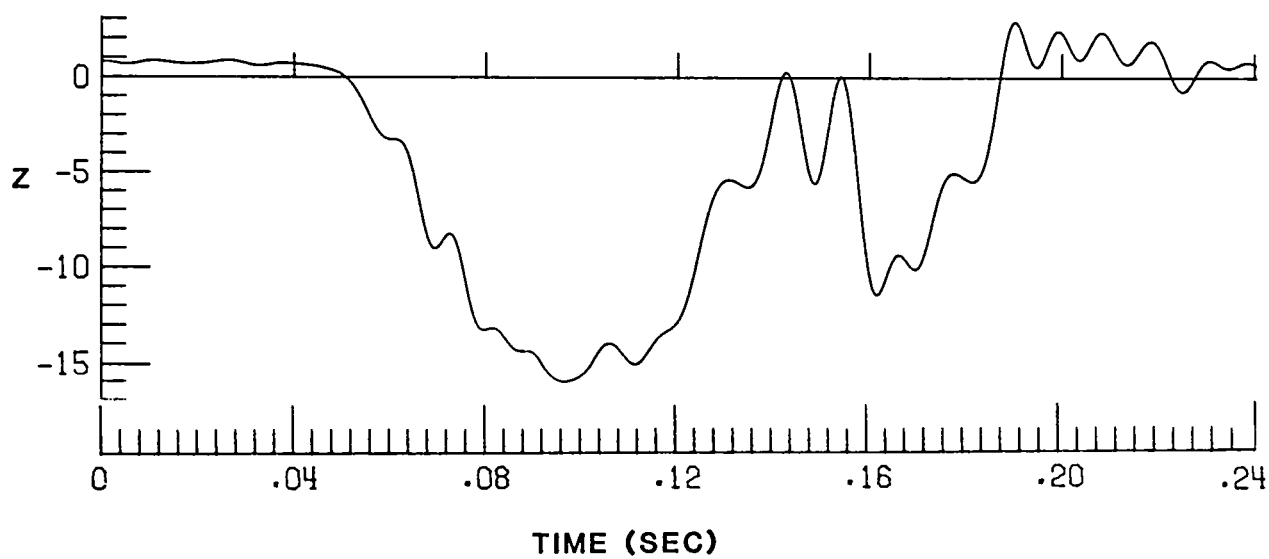
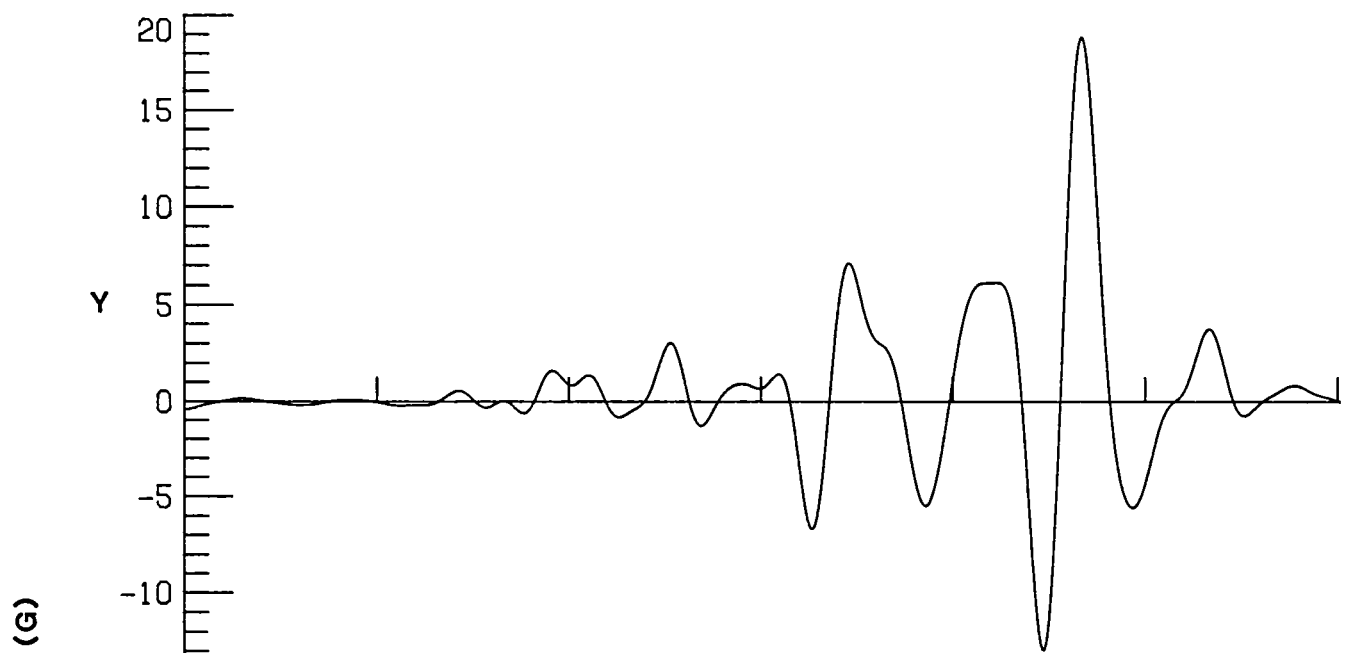




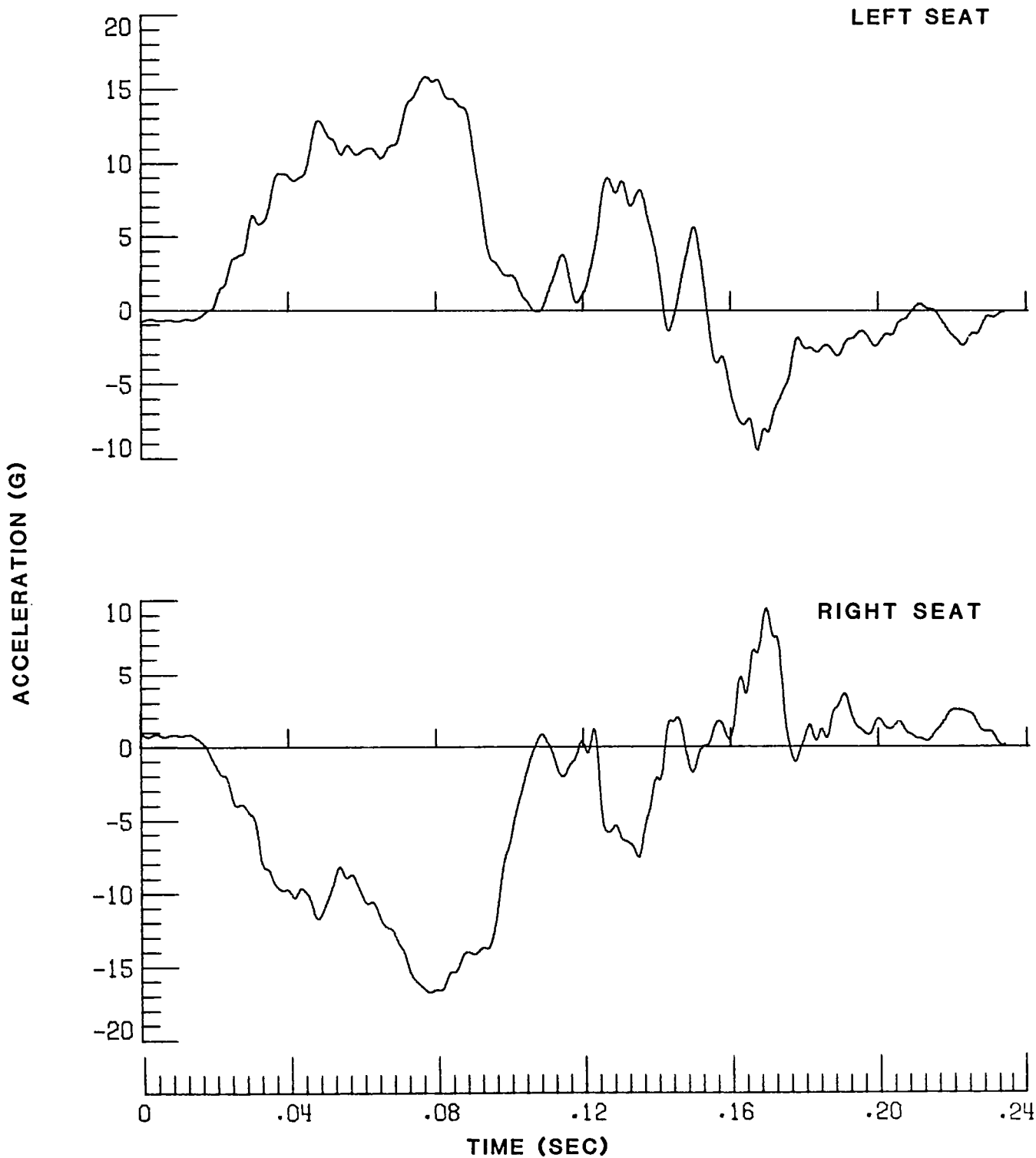
RIGHT SEAT PAN  
NASA TEST NO. 7



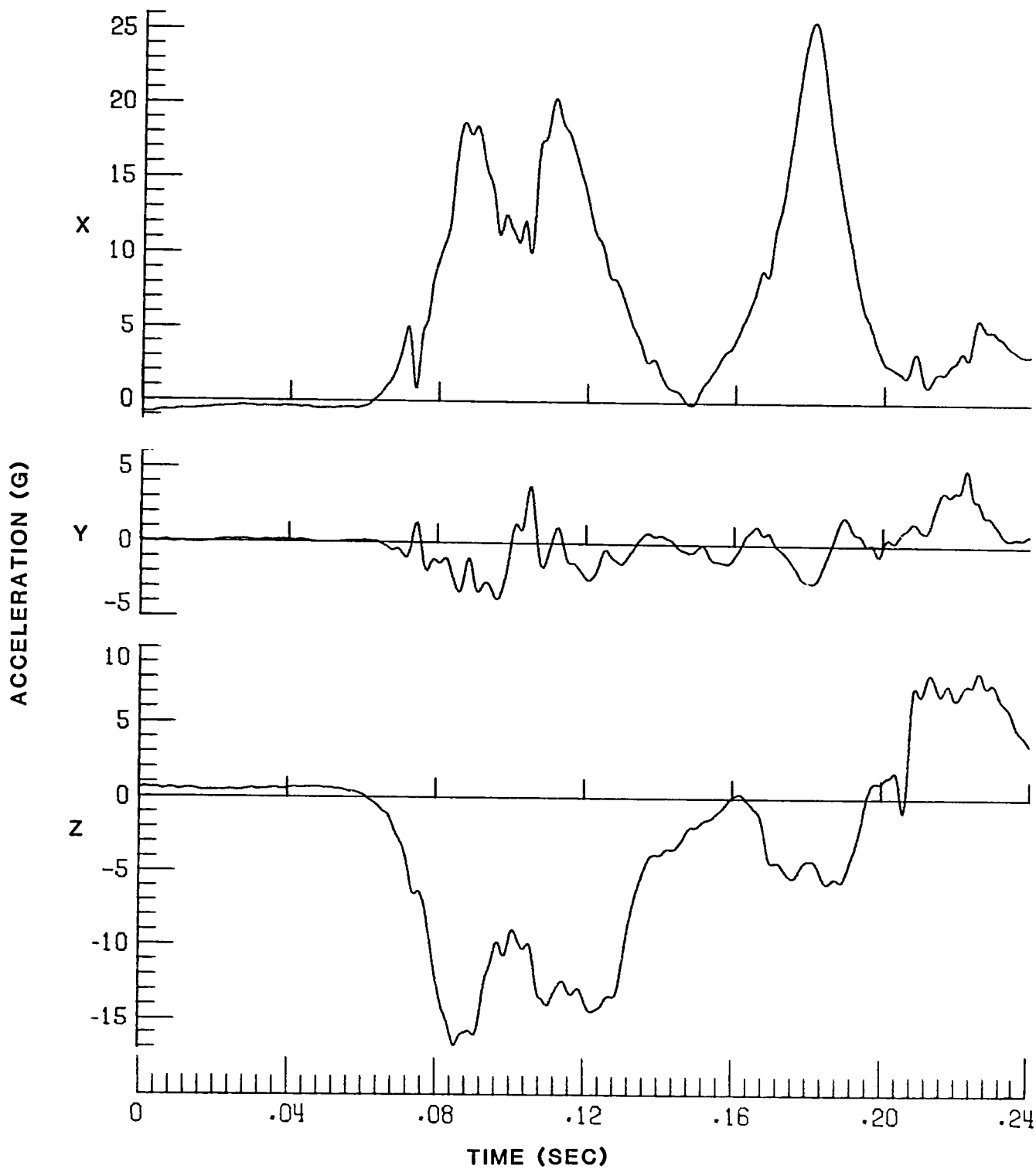
CAPSULE C.G.  
NASA TEST NO. 7



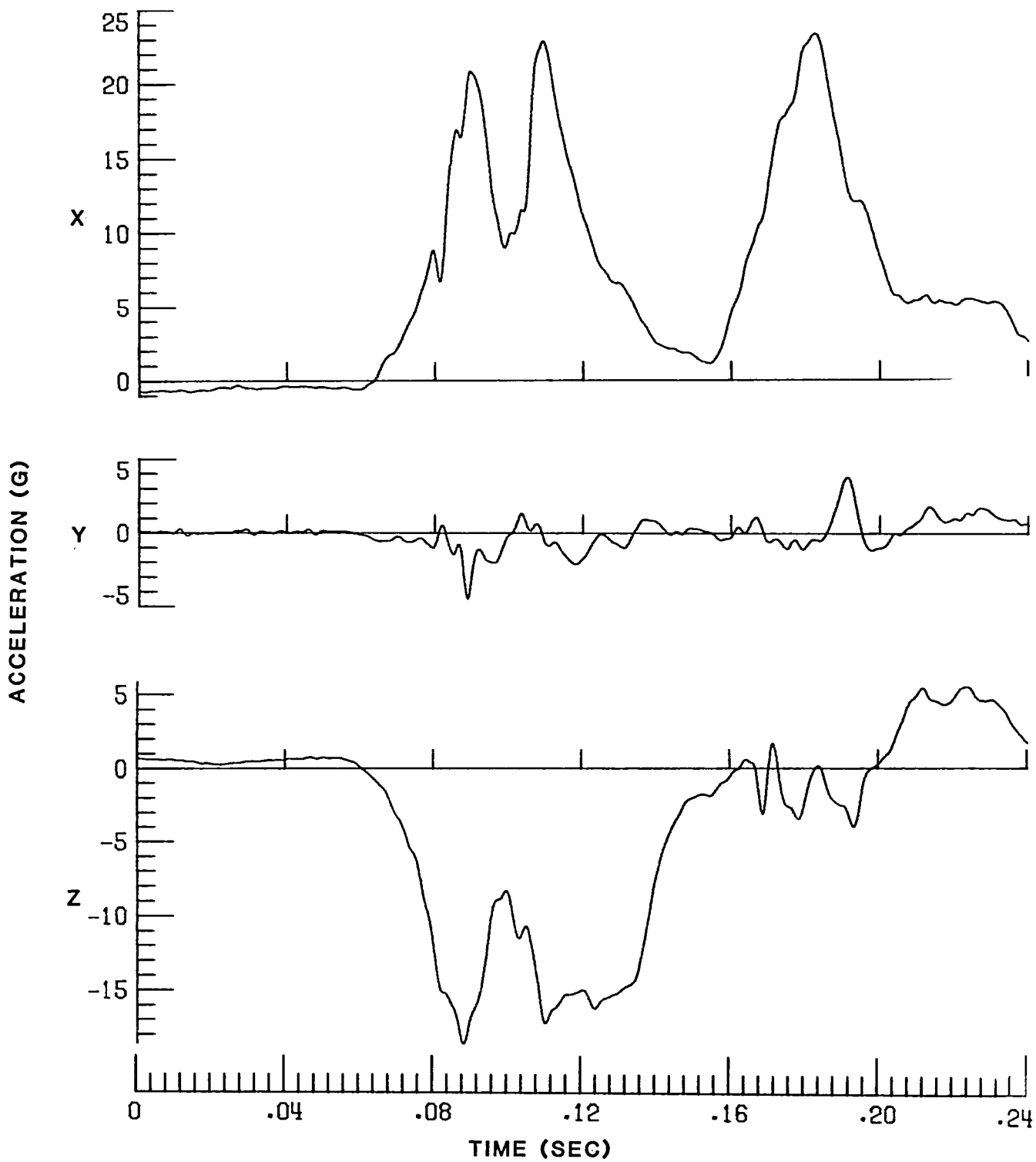
CAPSULE C.G.  
NASA TEST NO. 7



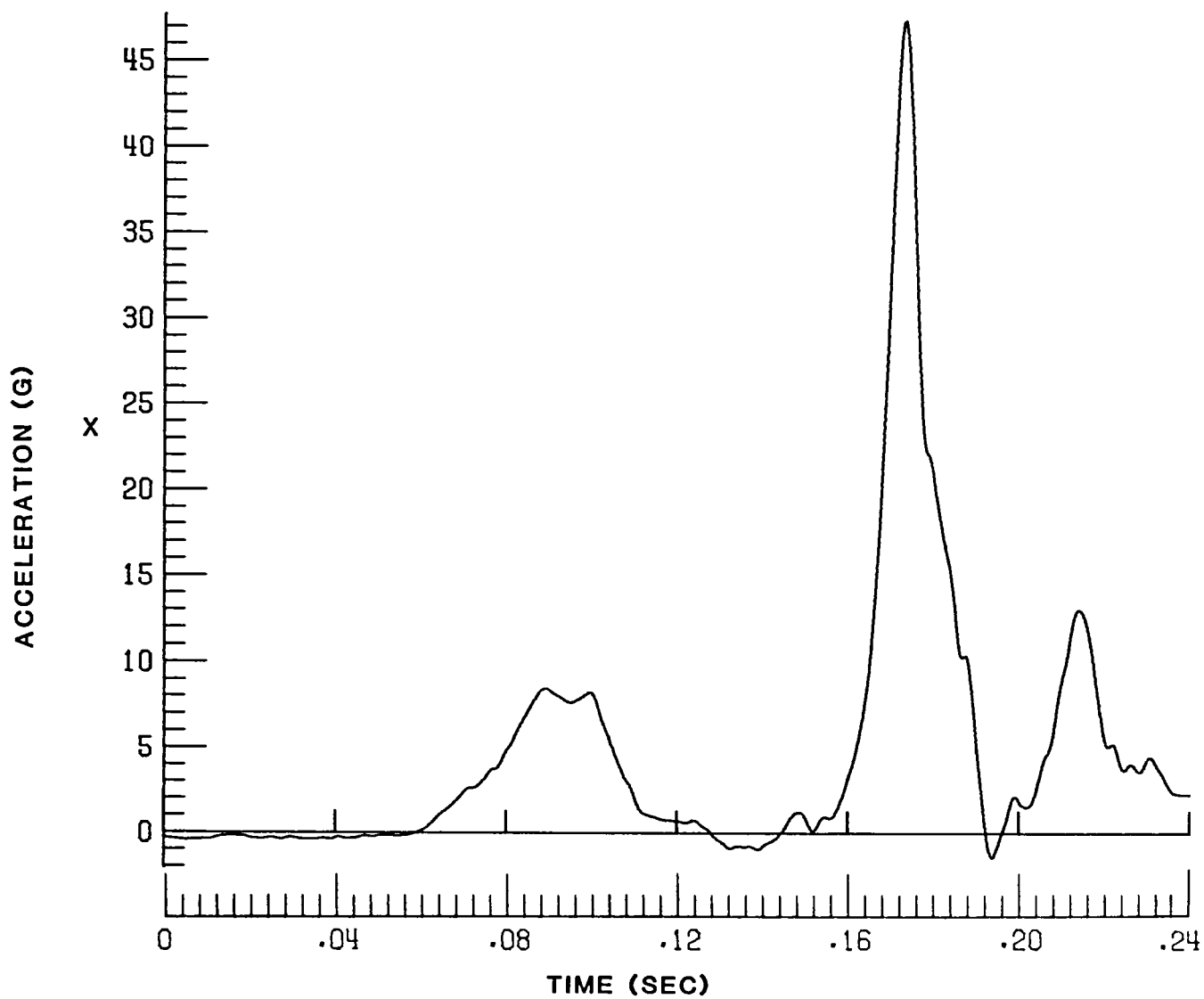
SEAT PAN ALONG BACK TANGENT LINE  
NASA TEST NO. 7



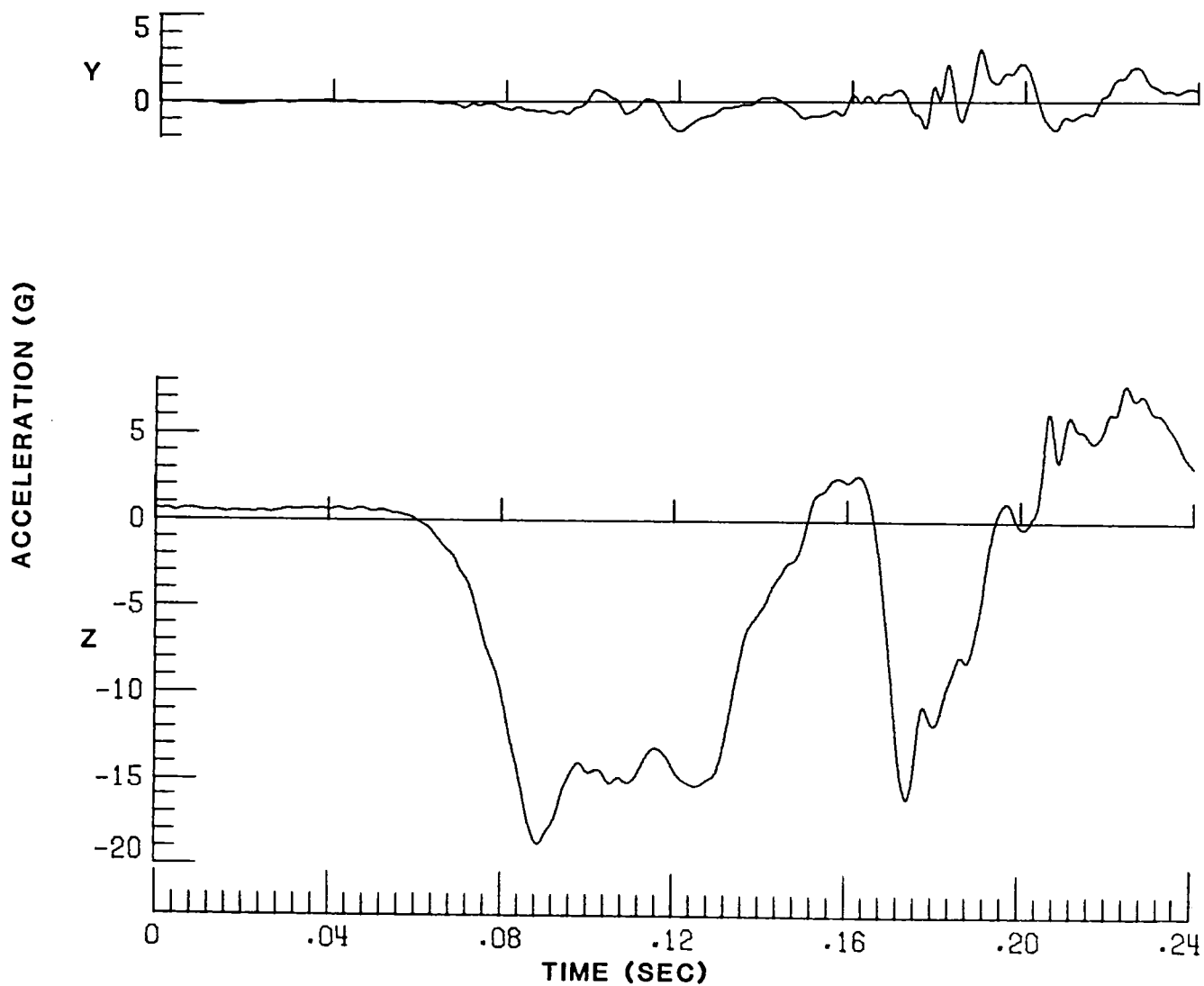
LEFT DUMMY PELVIS  
NASA TEST NO. 7



RIGHT DUMMY PELVIS  
NASA TEST NO. 7



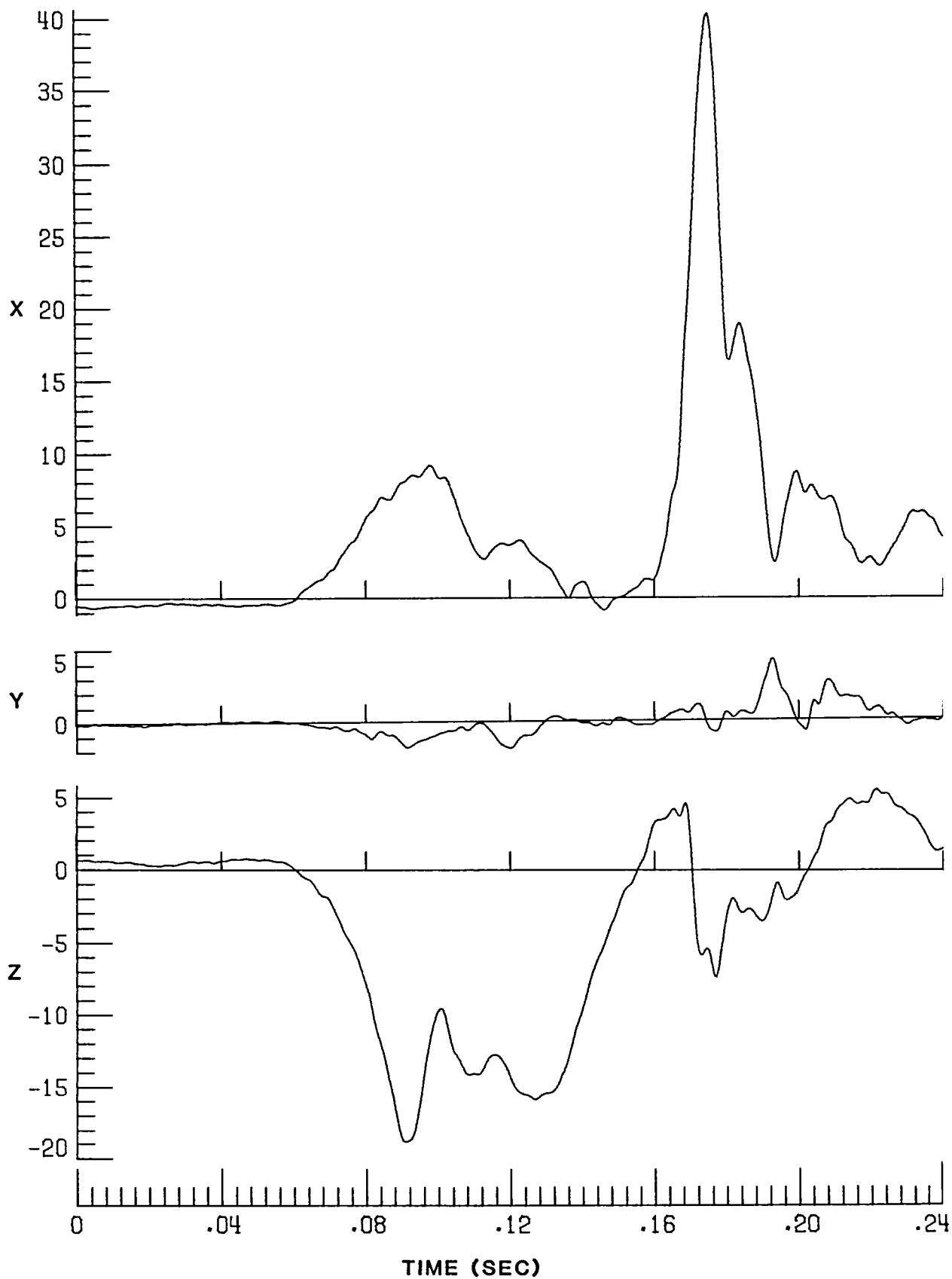
LEFT DUMMY CHEST  
NASA TEST NO. 7



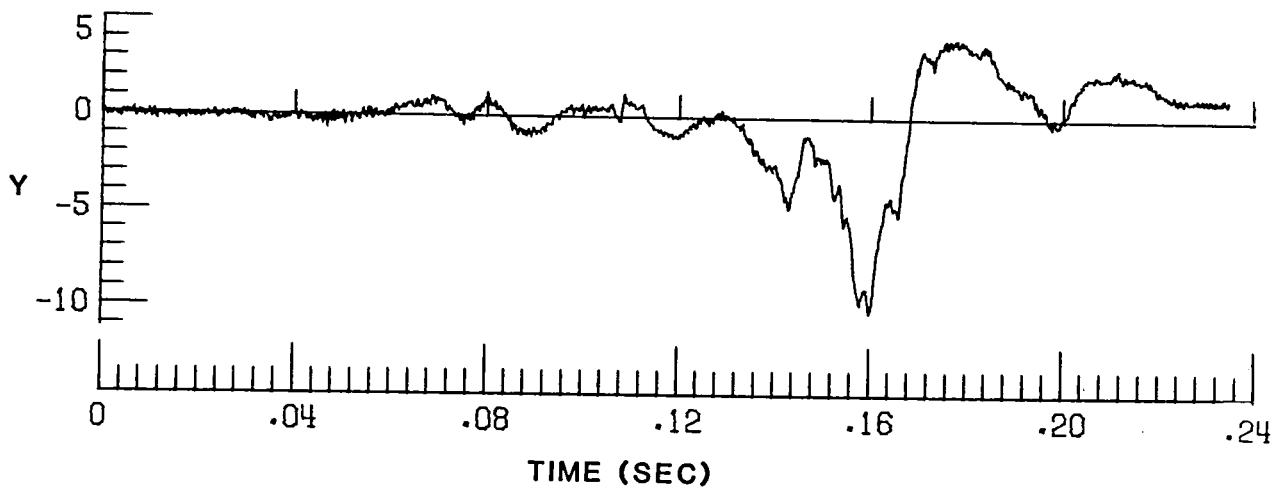
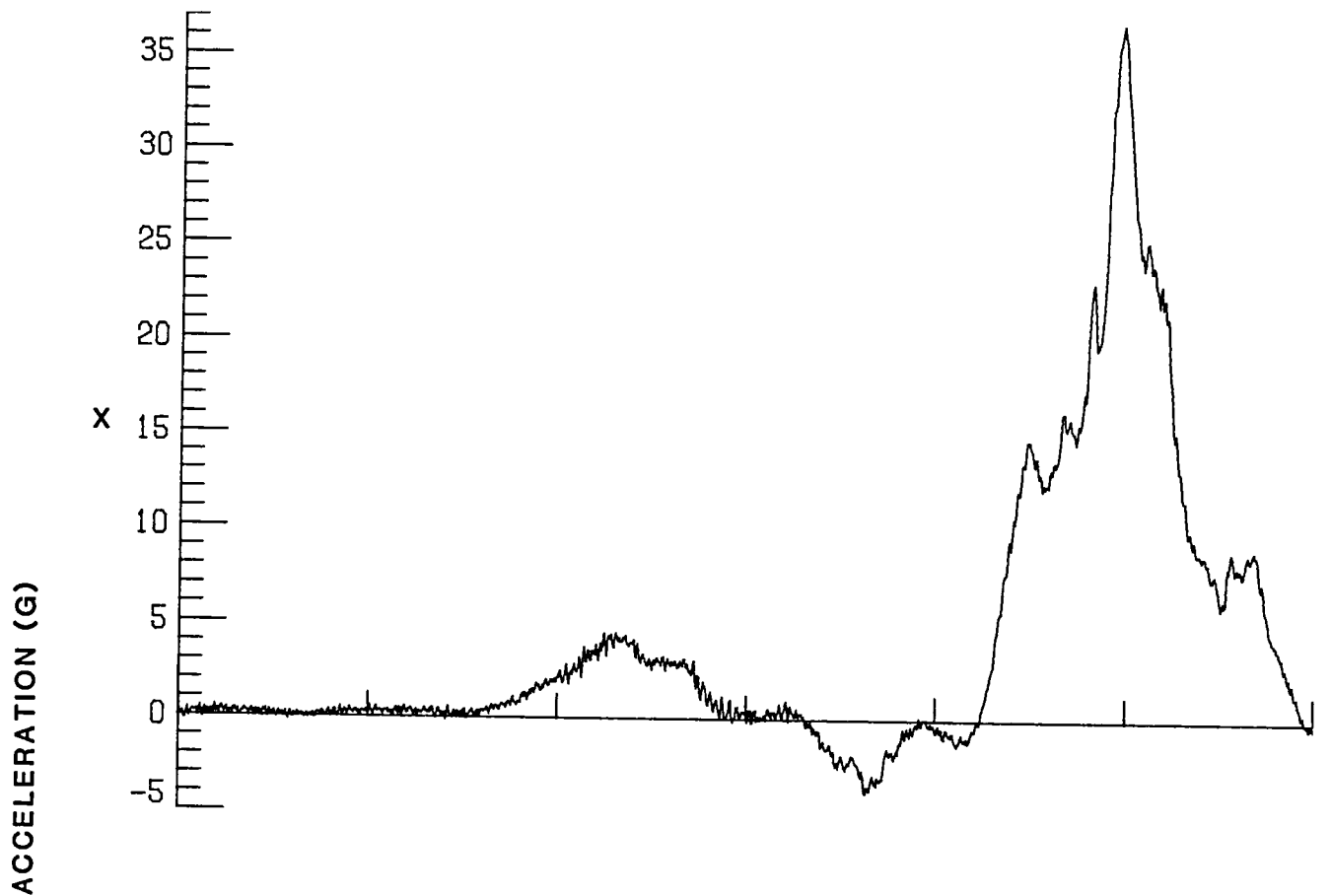
LEFT DUMMY CHEST  
NASA TEST NO. 7



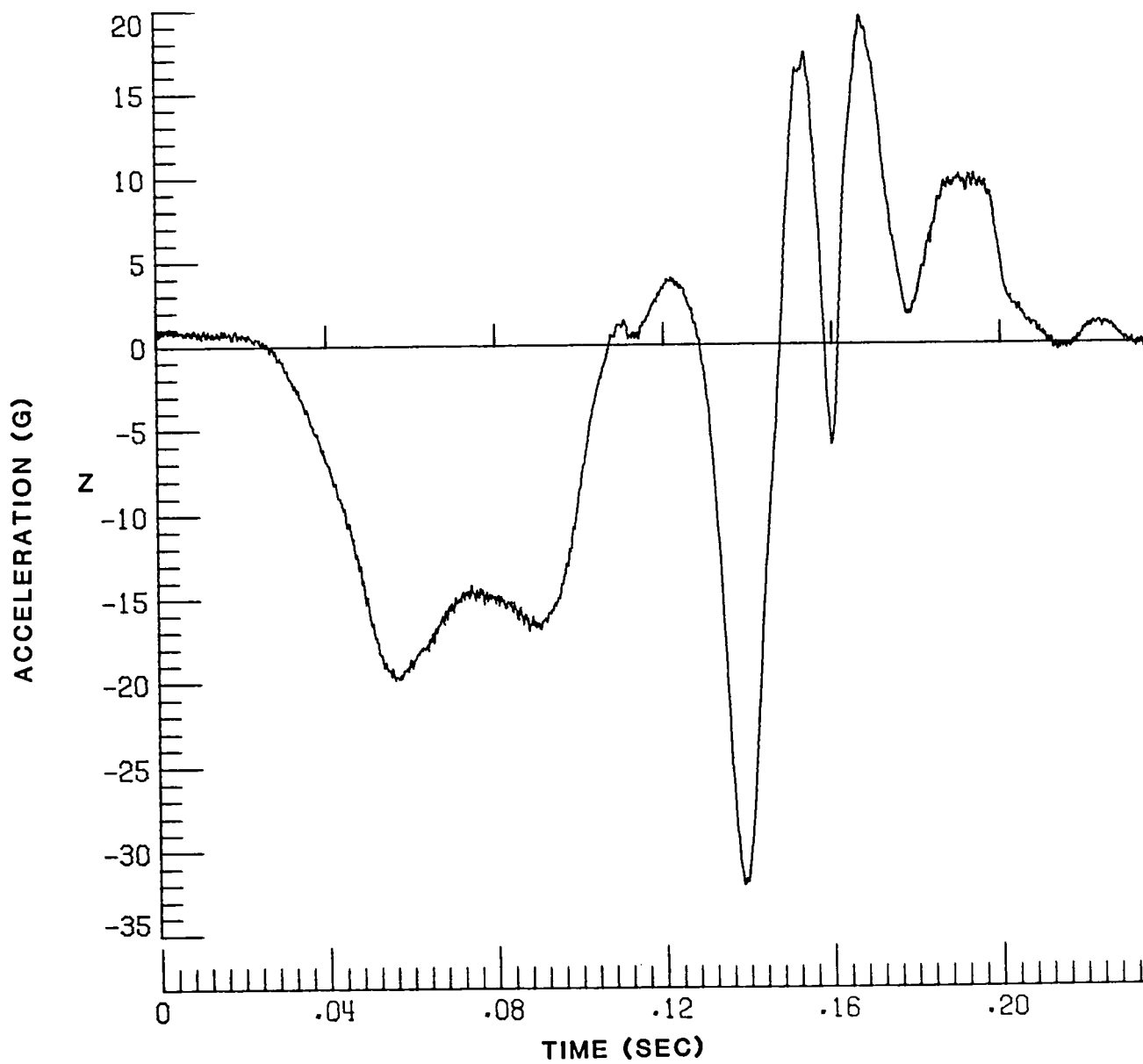
ACCELERATION (G)



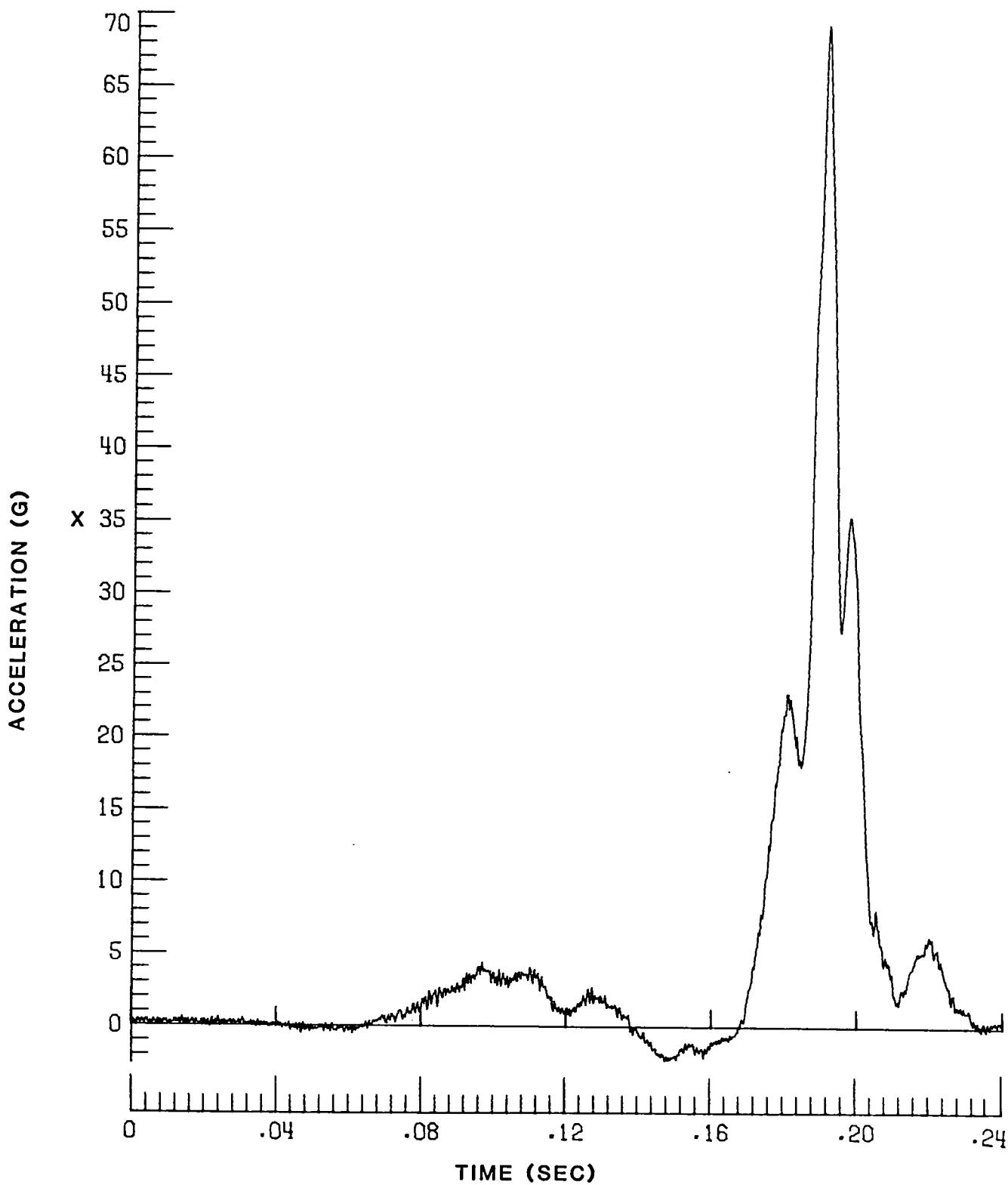
RIGHT DUMMY CHEST  
NASA TEST NO. 7



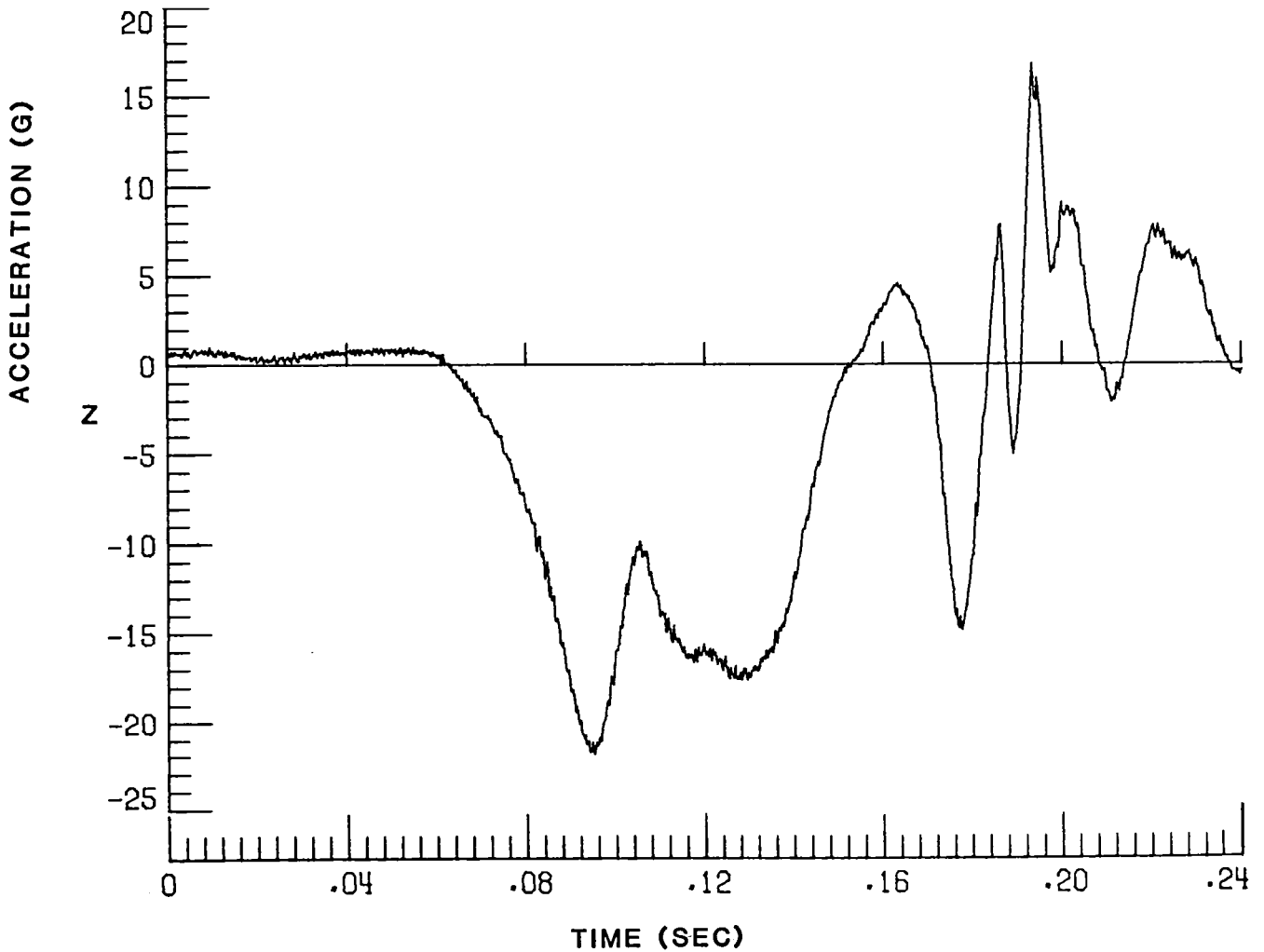
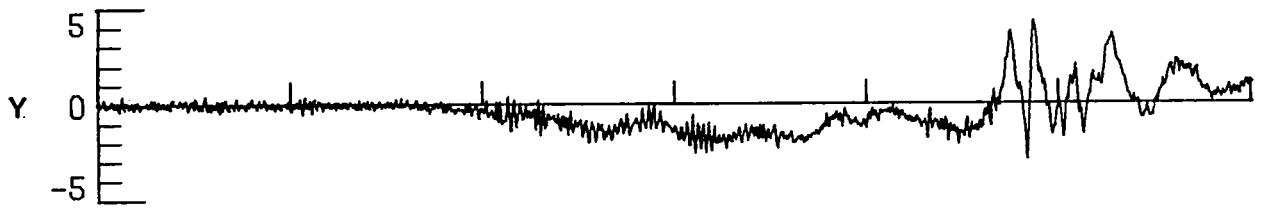
LEFT DUMMY HEAD  
NASA TEST NO. 7



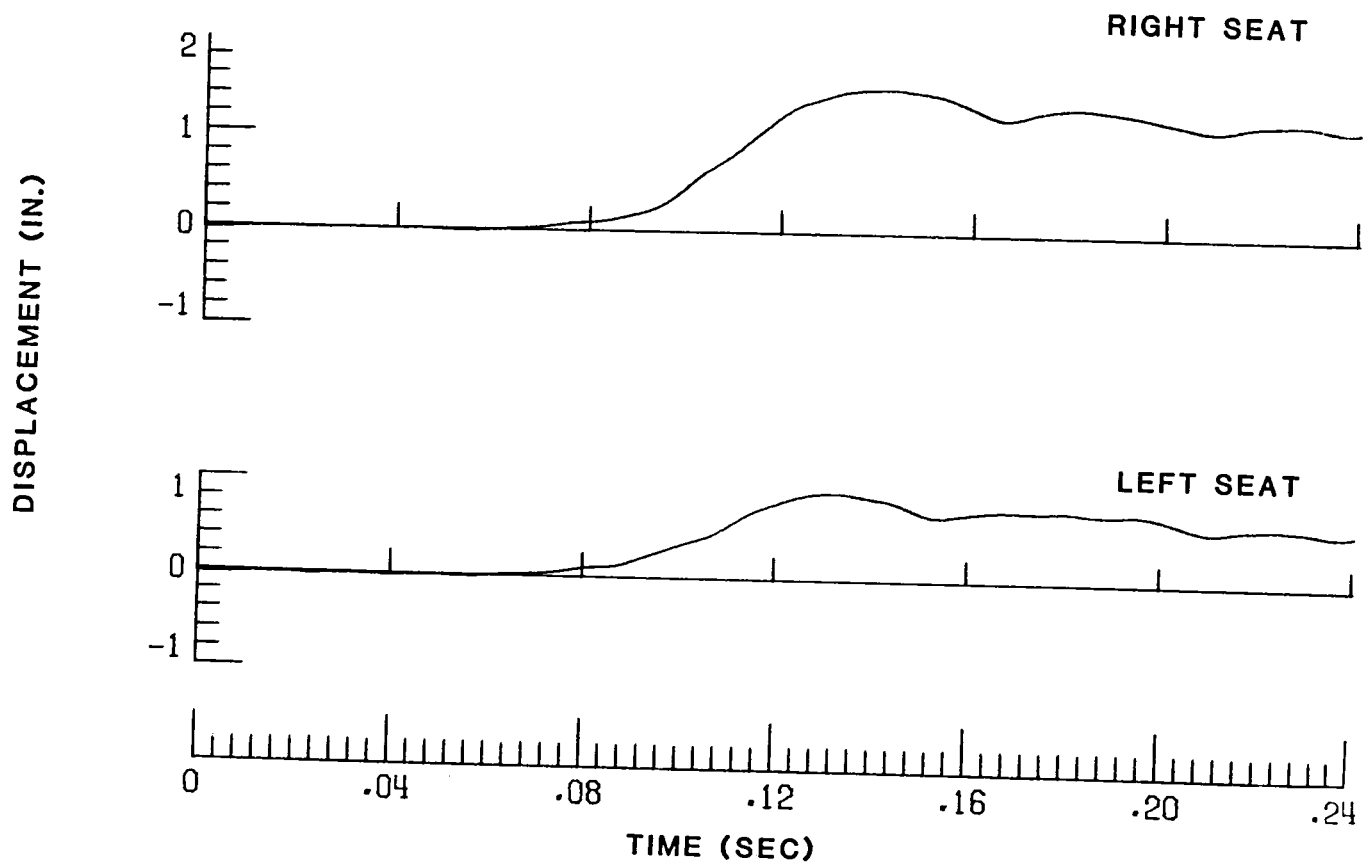
LEFT DUMMY HEAD  
NASA TEST NO. 7



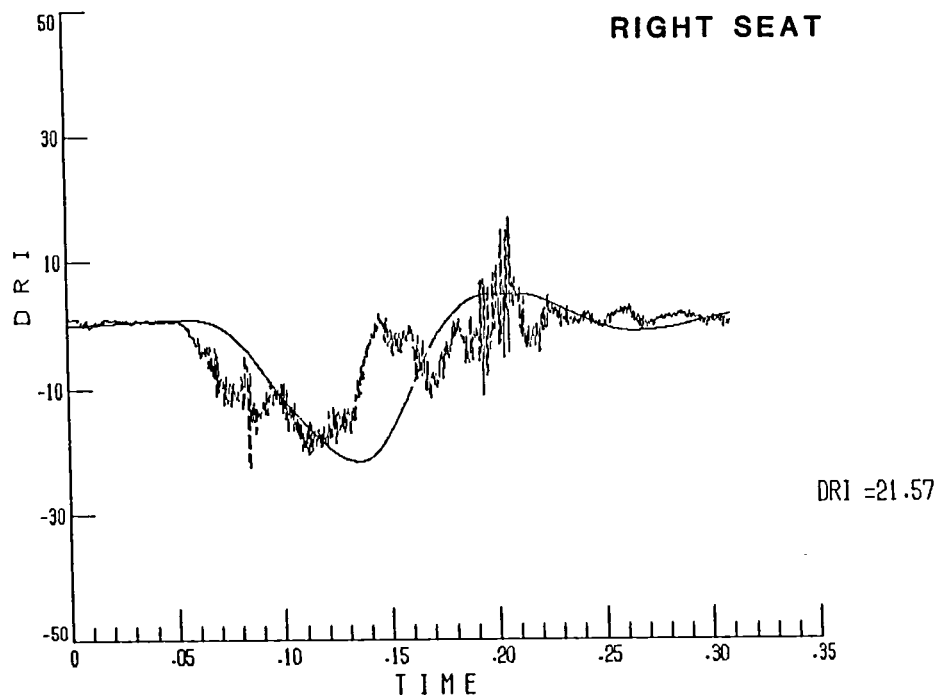
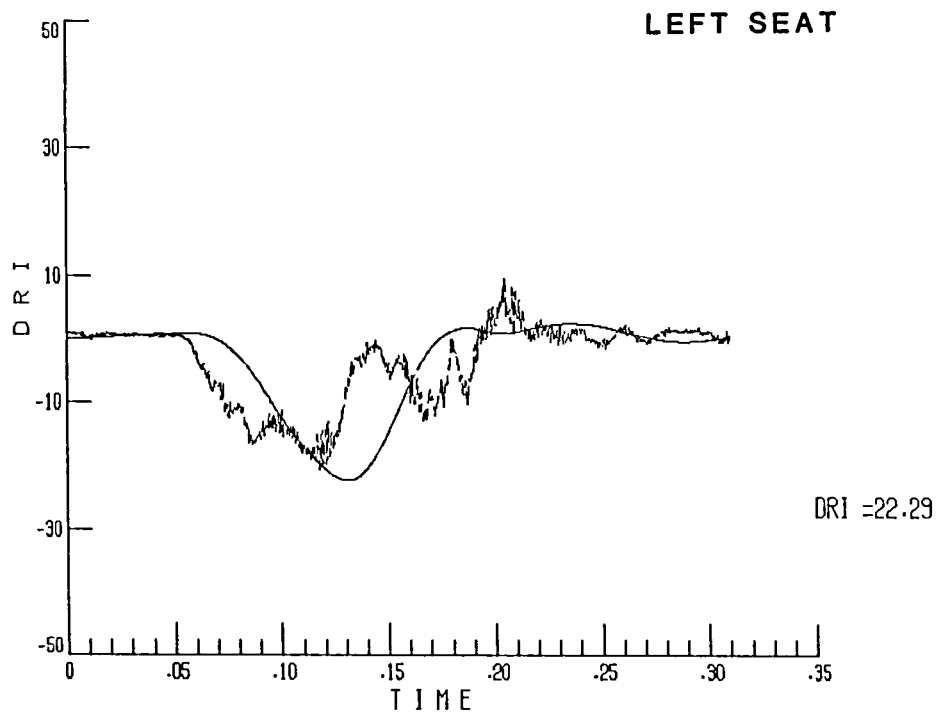
RIGHT DUMMY HEAD  
NASA TEST NO. 7



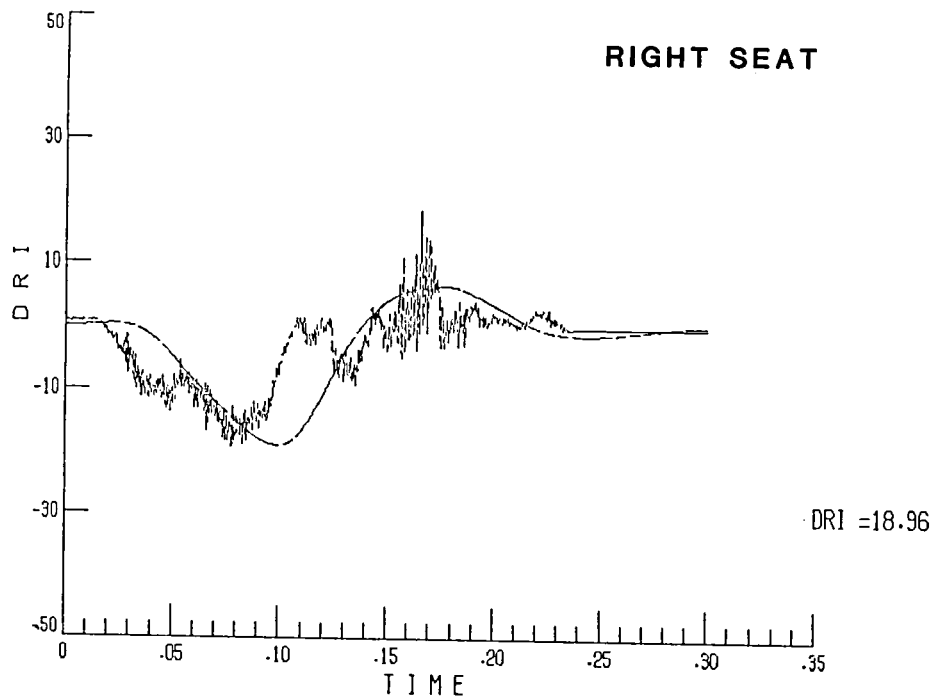
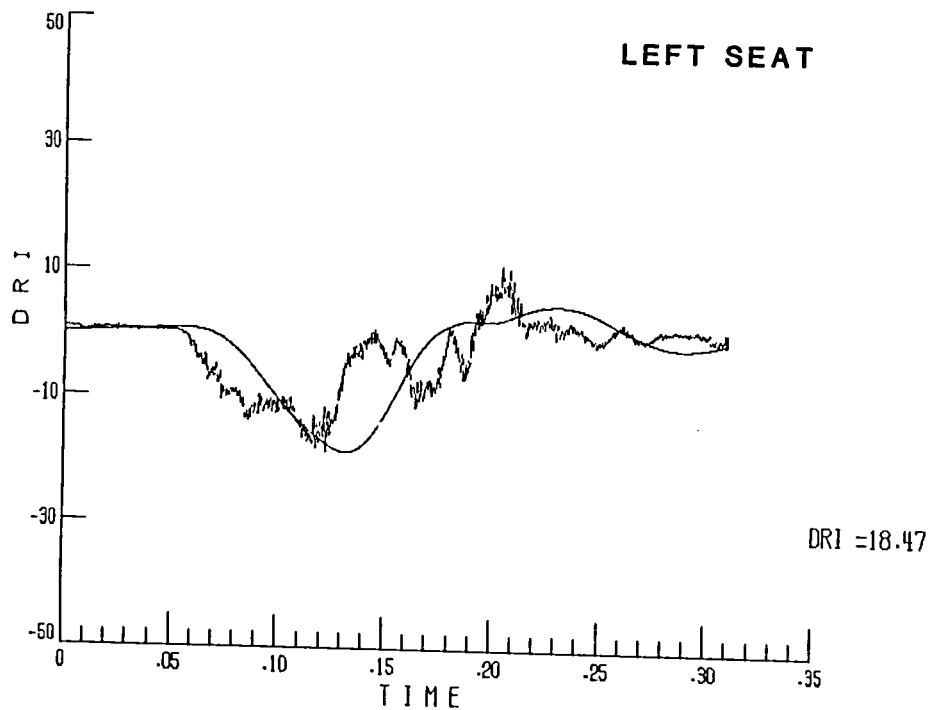
RIGHT DUMMY HEAD  
NASA TEST NO. 7



SEAT STROKE  
NASA TEST NO. 7

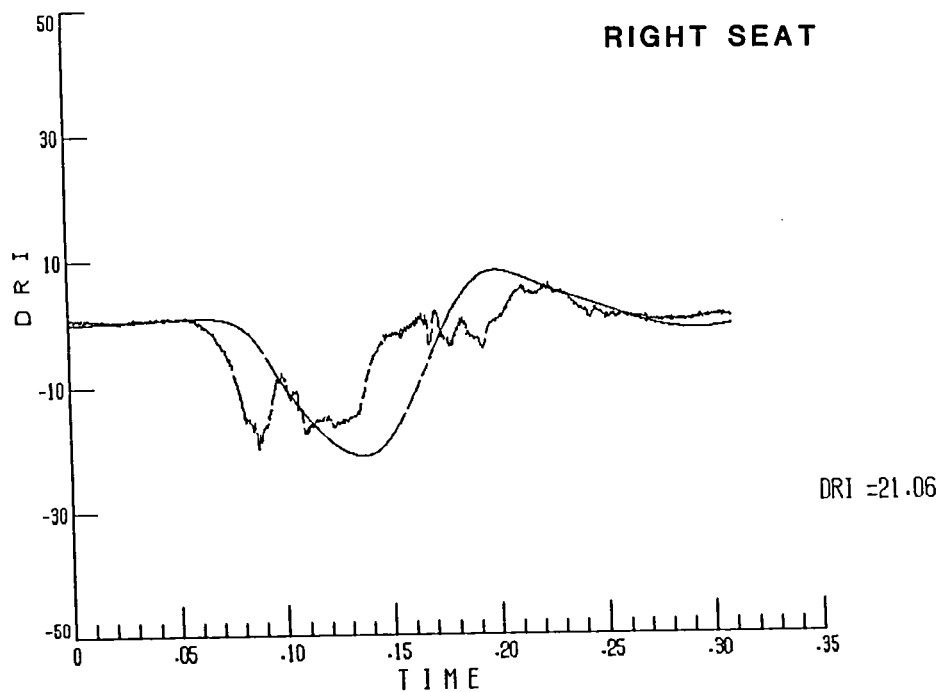
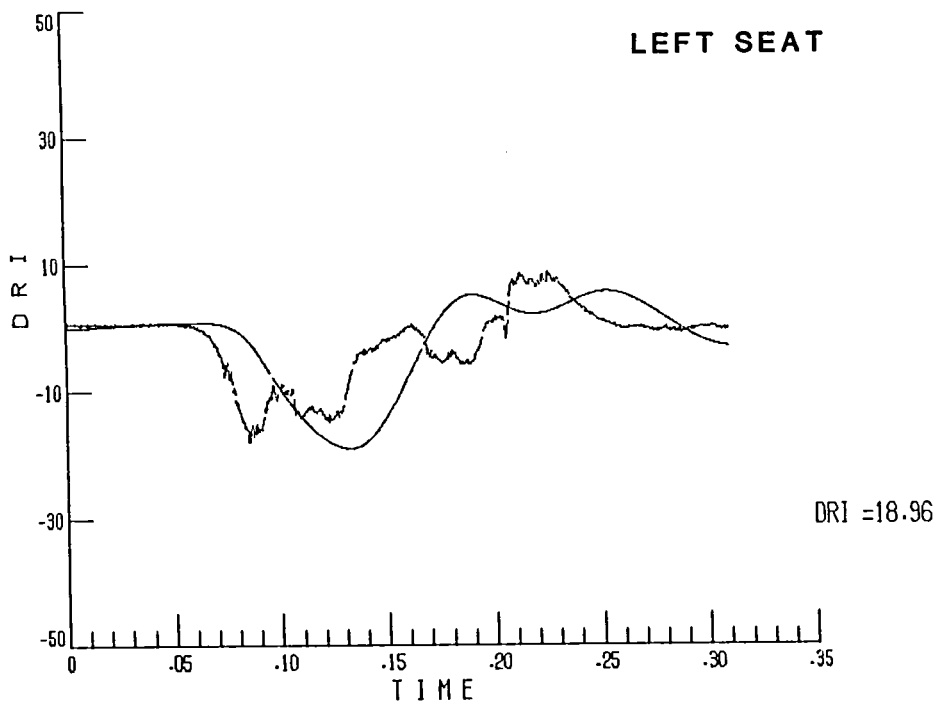


**DYNAMIC RESPONSE INDEX  
SEAT PAN Z-AXIS  
NASA TEST NO. 7**



**DYNAMIC RESPONSE INDEX  
BACK TANGENT LINE  
NASA TEST NO. 7  
130**





**DYNAMIC RESPONSE INDEX  
DUMMY PELVIS Z-AXIS  
NASA TEST NO. 7**

9.0 NASA TEST NO. 8

HORIZONTAL VELOCITY: 0 ft/sec

VERTICAL VELOCITY: 32 ft/sec

CREW MODULE PITCH:  $2^0$

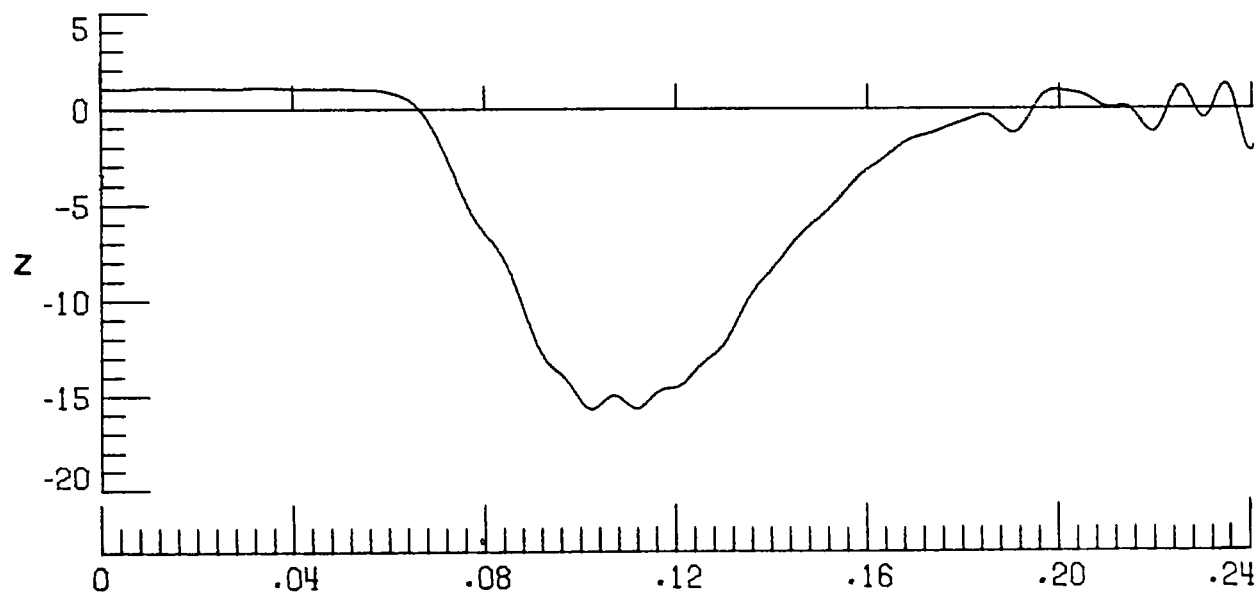
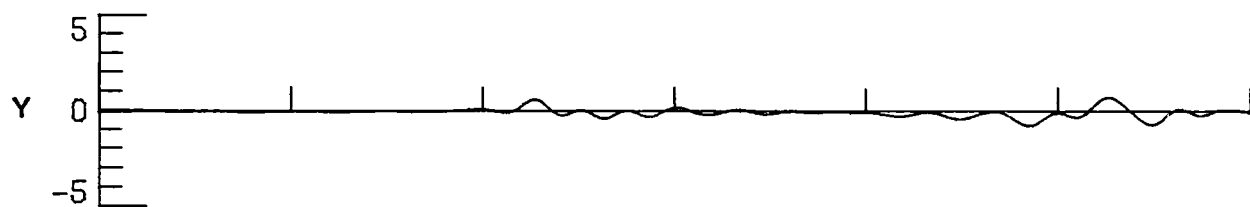
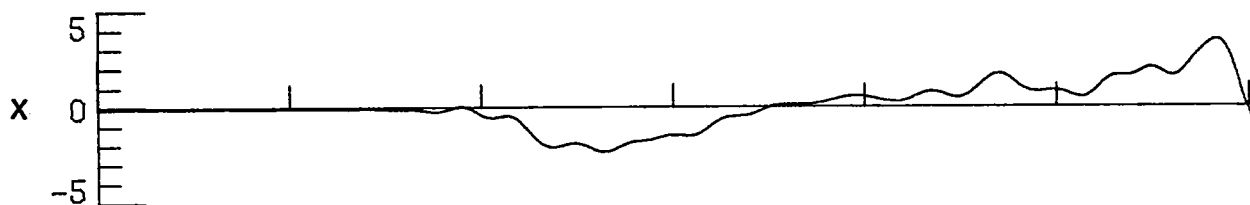
CREW MODULE ROLL:  $0^0$

CREW MODULE YAW:  $0^0$

LEFT SEAT: F-111 Operational

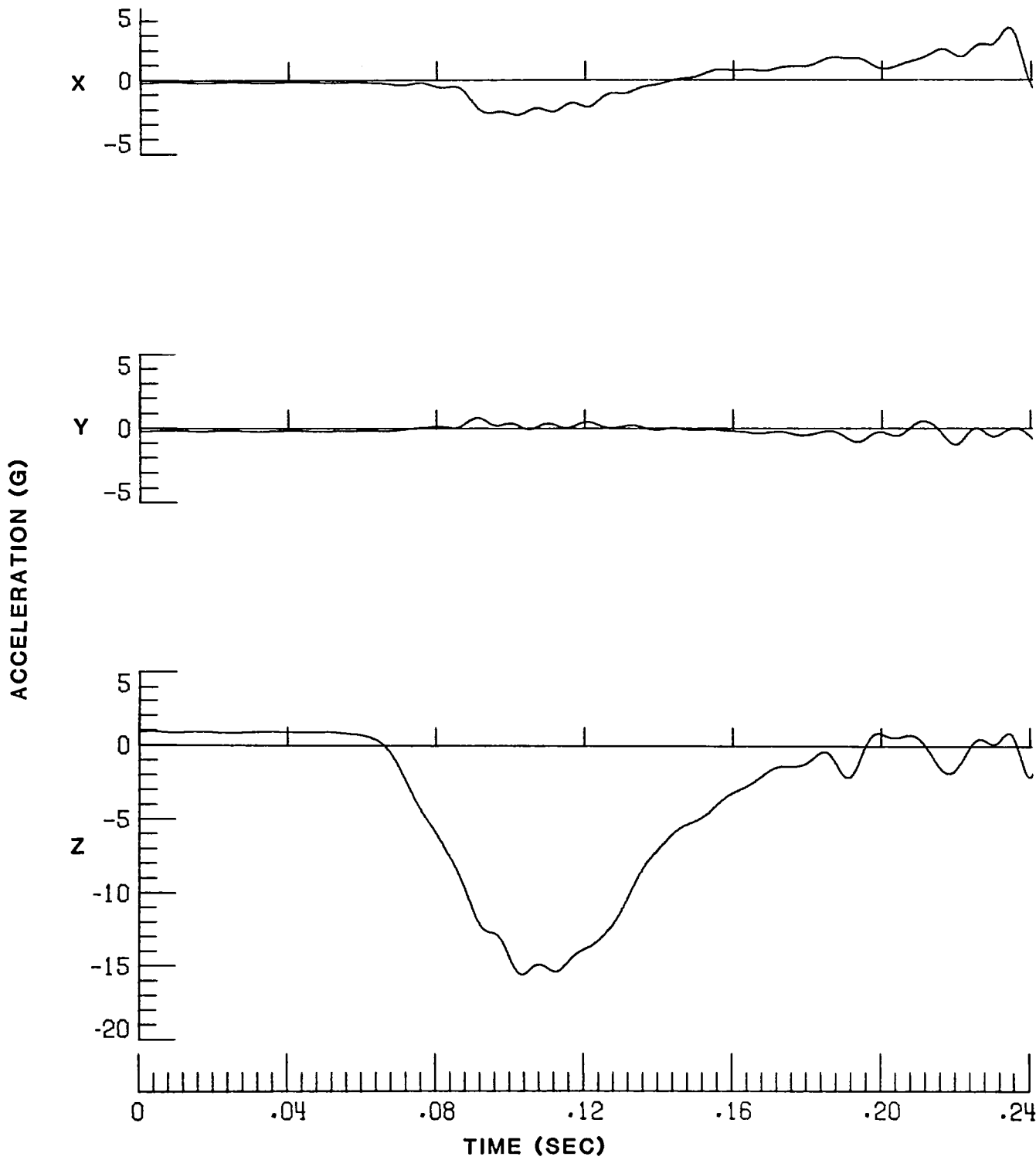
RIGHT SEAT: Energy Absorbing

ACCELERATION (G)

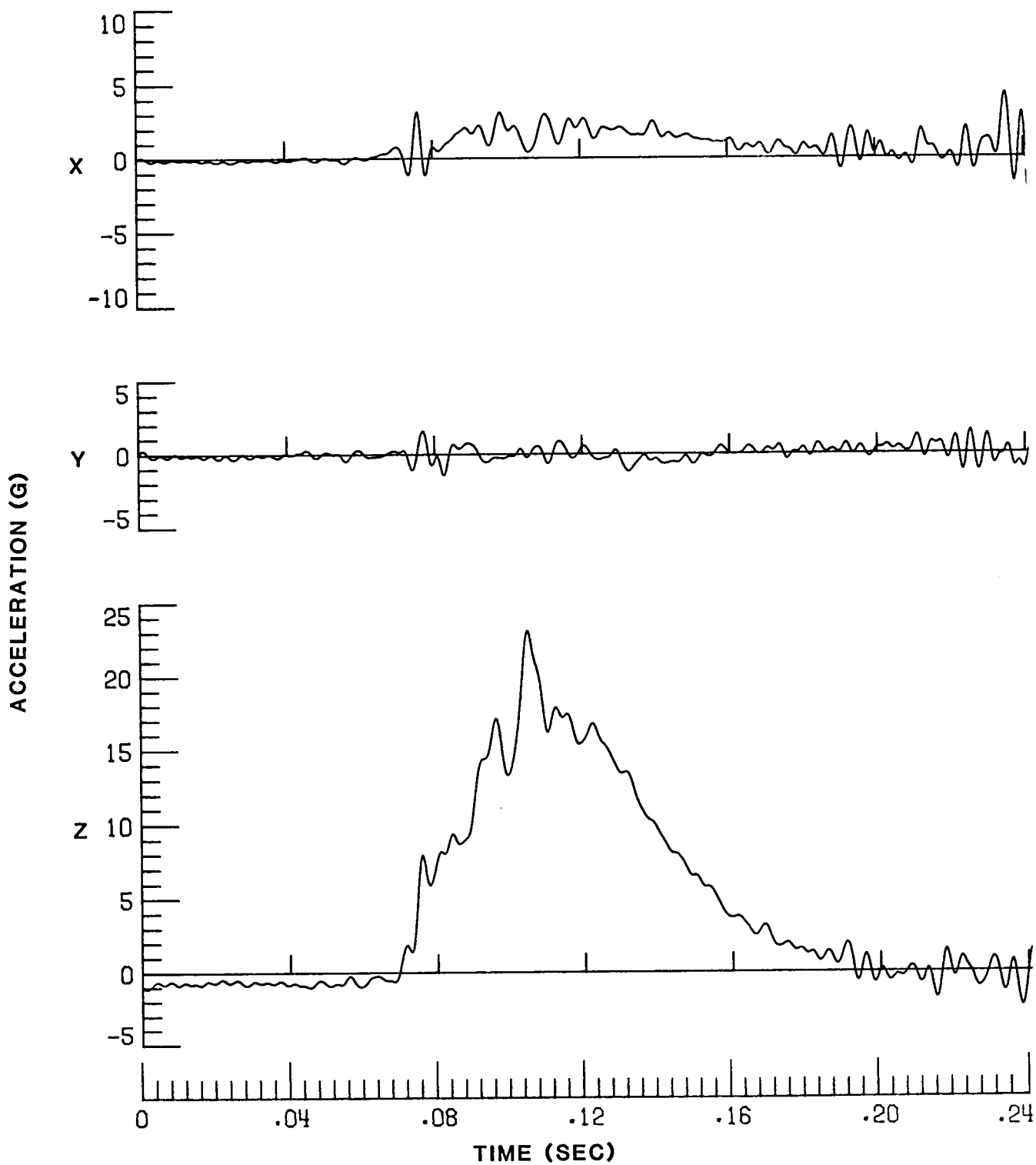


TIME (SEC)

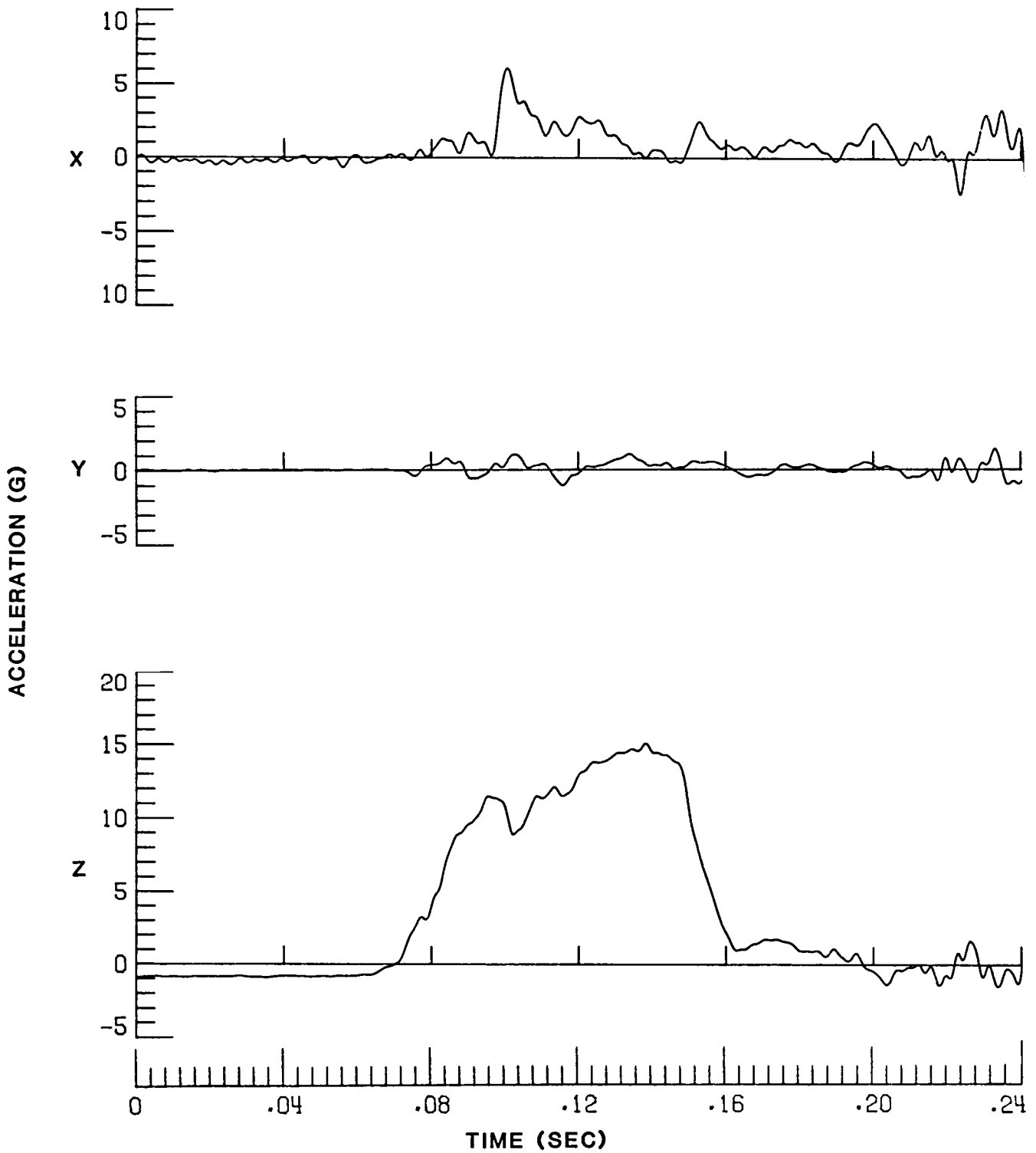
LEFT BULKHEAD  
NASA TEST NO. 8



RIGHT BULKHEAD  
NASA TEST NO. 8

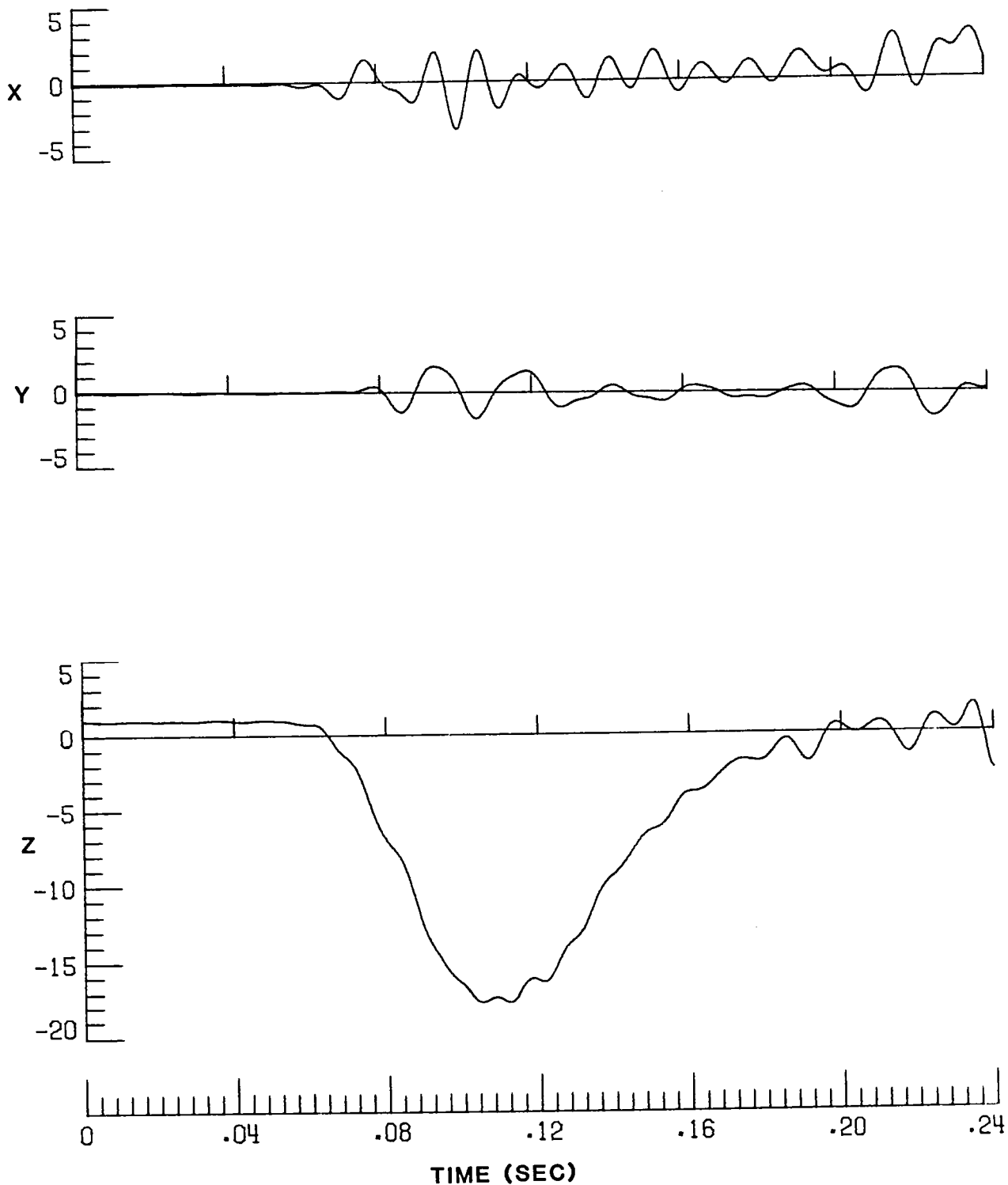


LEFT SEAT PAN  
NASA TEST NO. 8

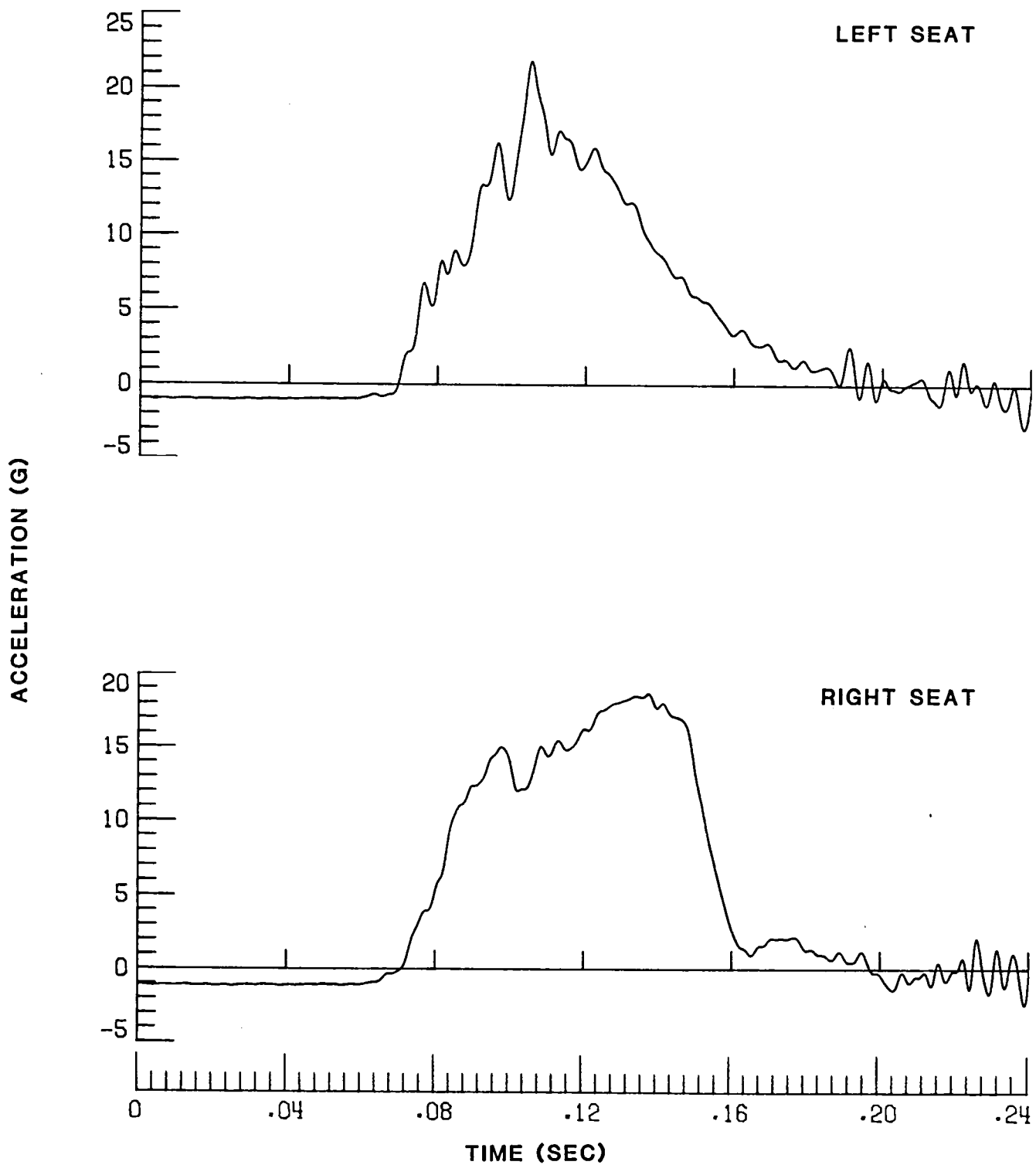


RIGHT SEAT PAN  
NASA TEST NO. 8

ACCELERATION (G)

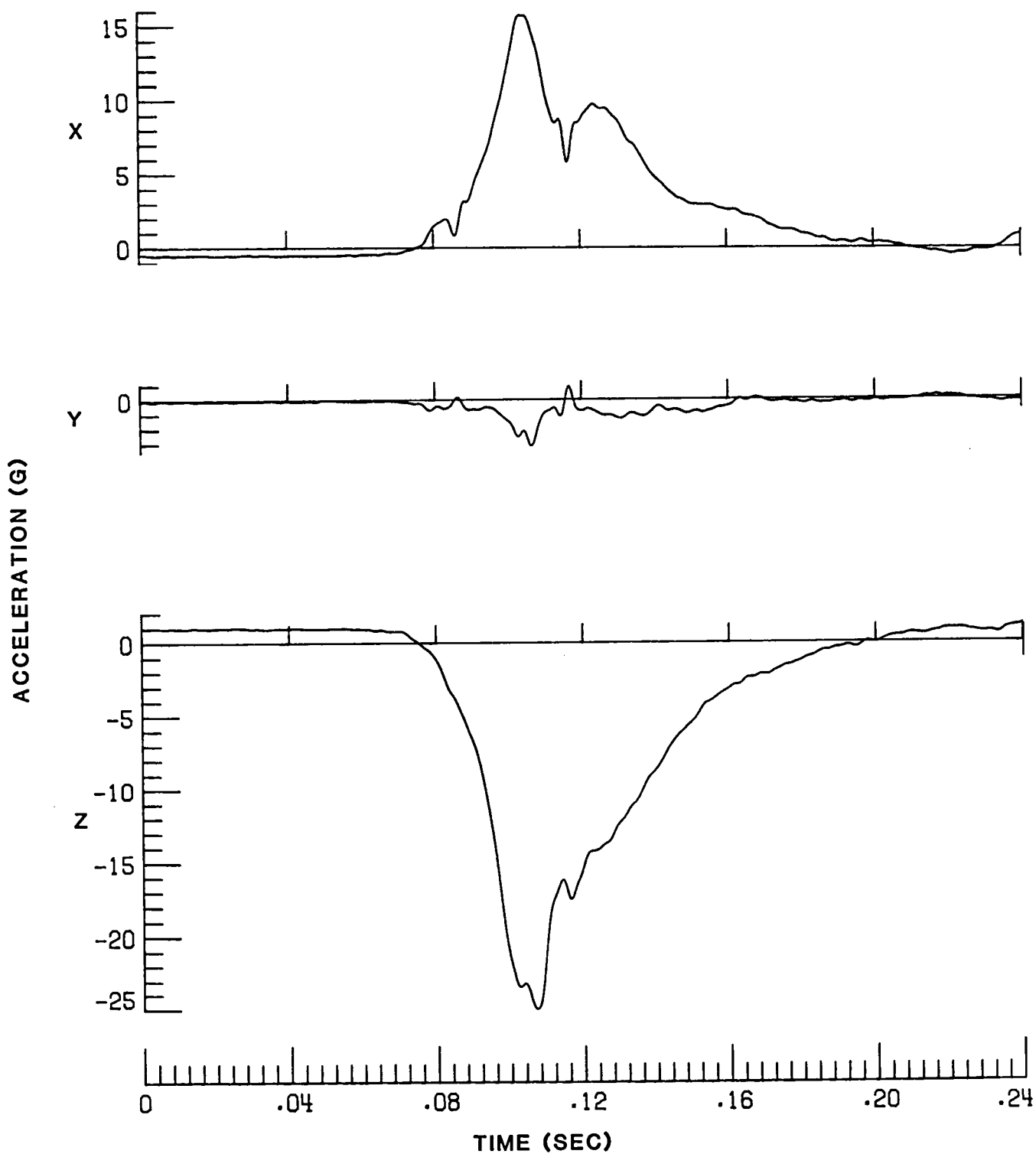


CAPSULE C.G.  
NASA TEST NO. 8

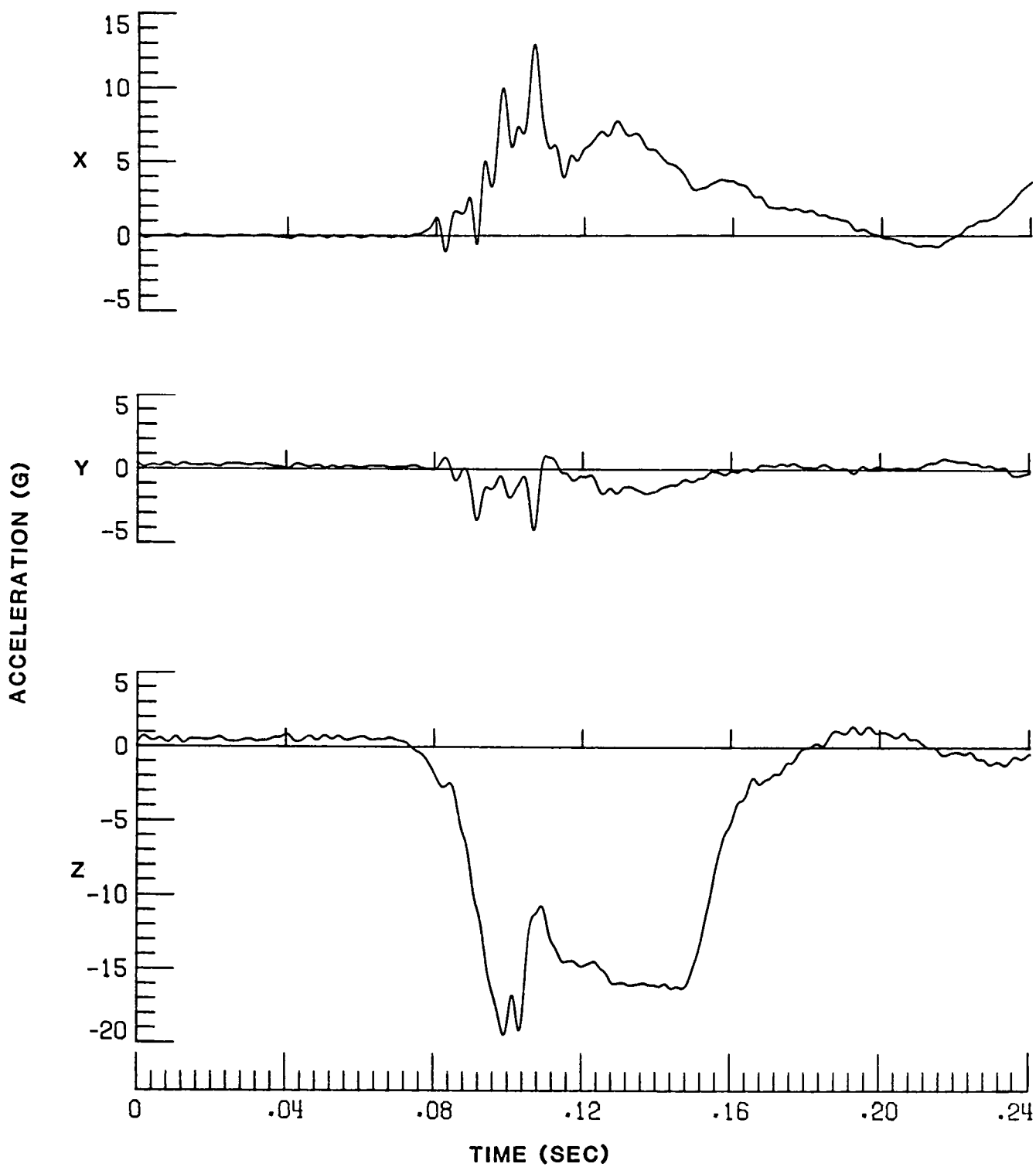


SEAT PAN ALONG BACK TANGENT LINE  
NASA TEST NO. 8

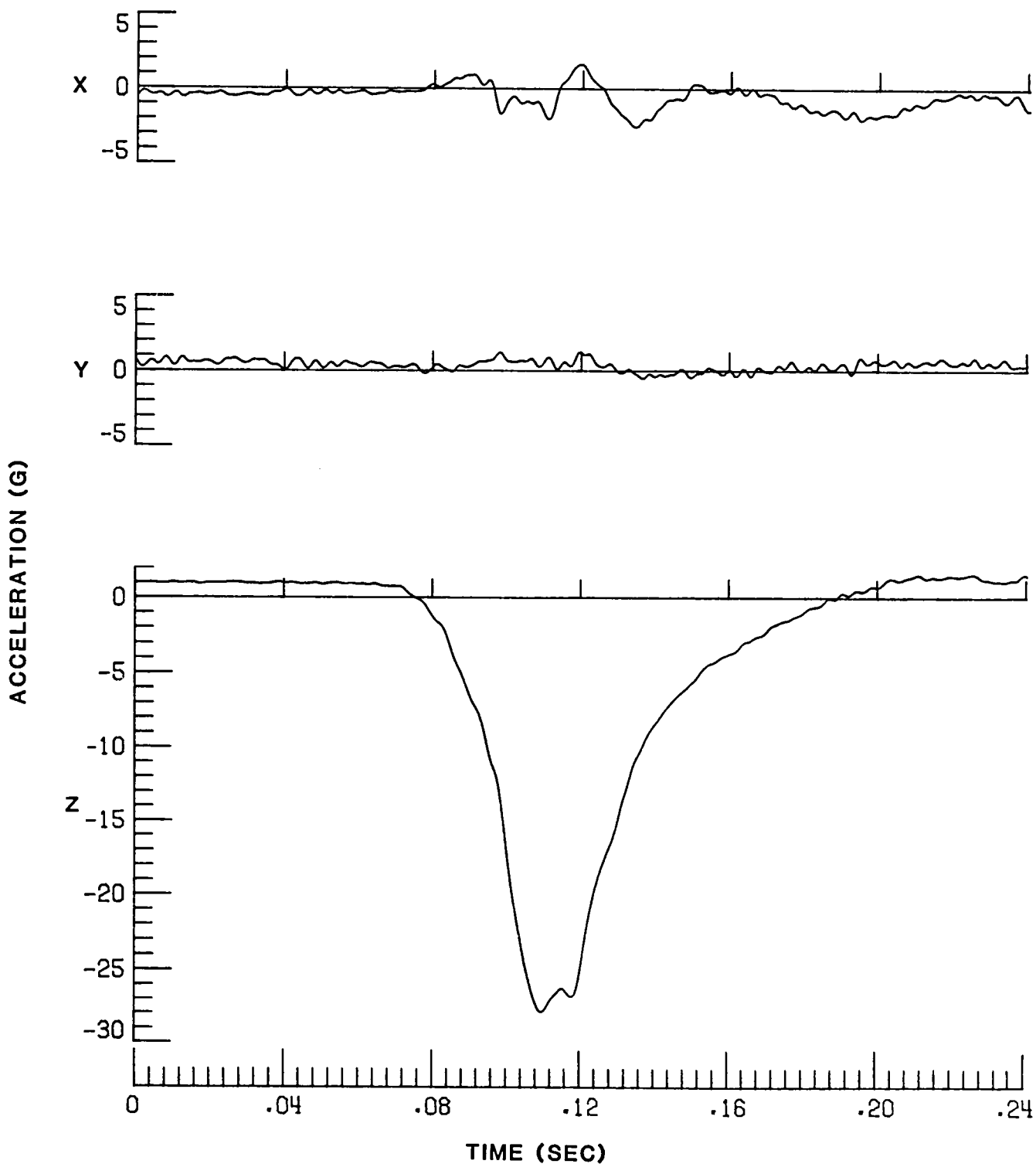




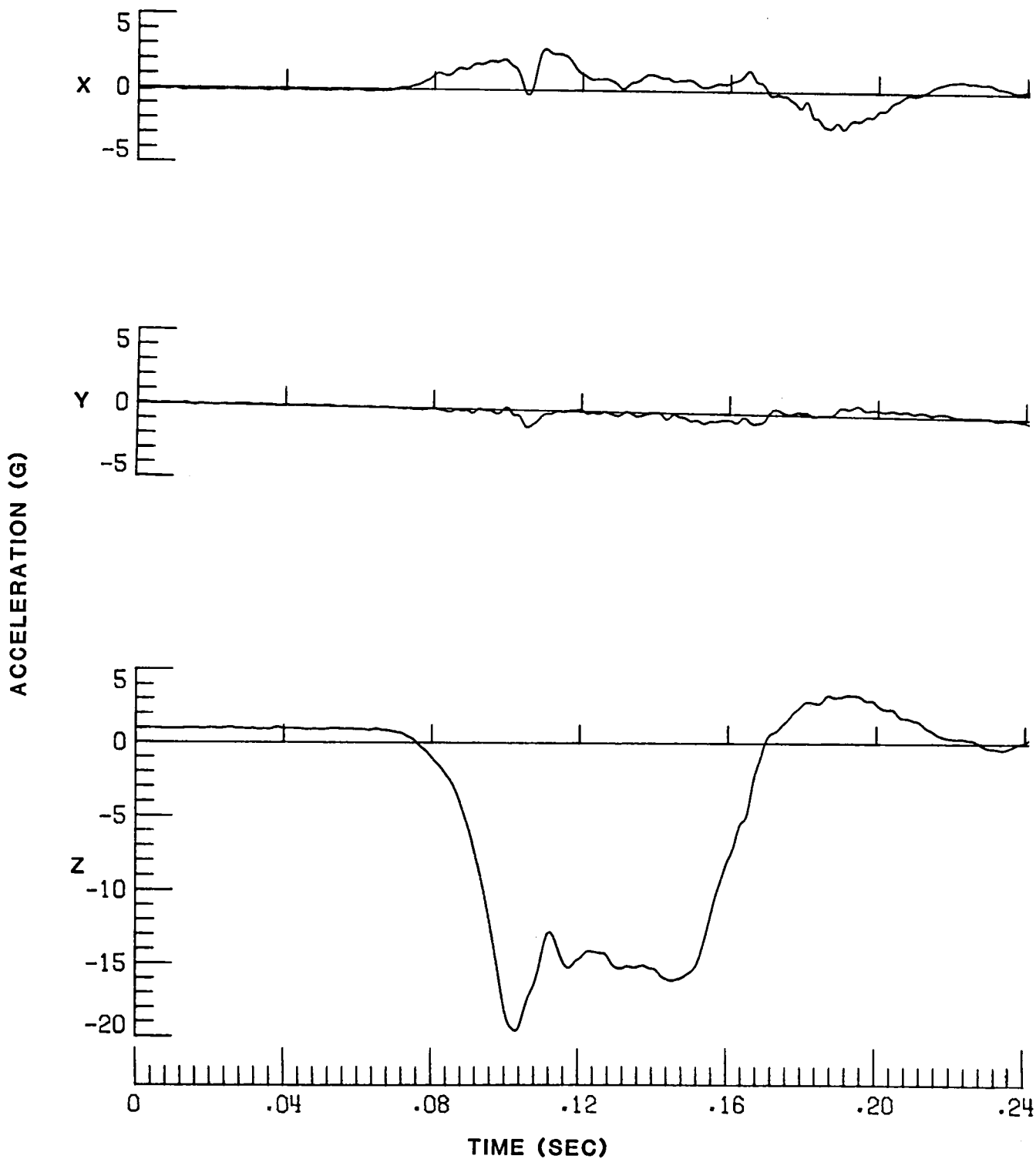
LEFT DUMMY PELVIS  
NASA TEST NO. 8



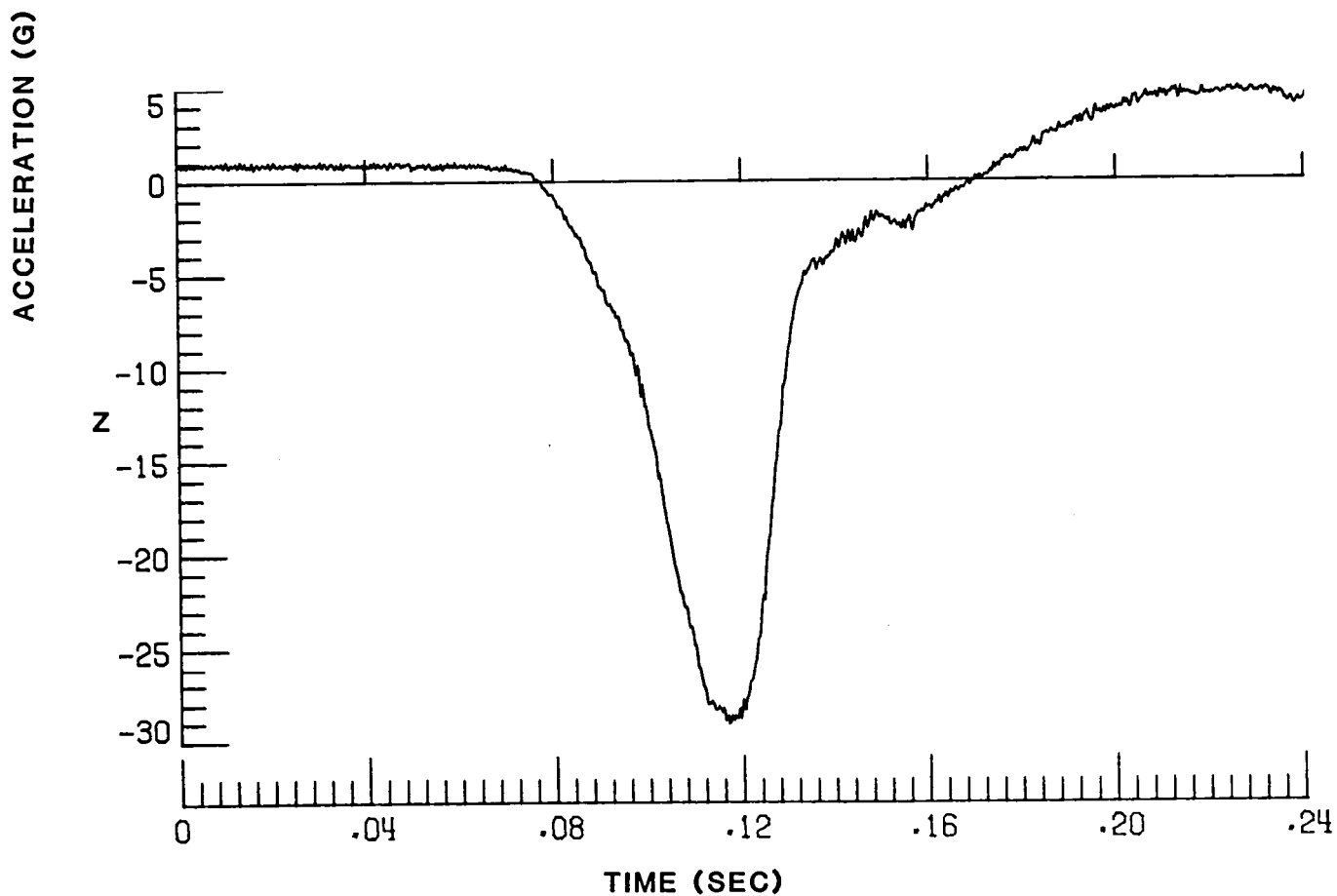
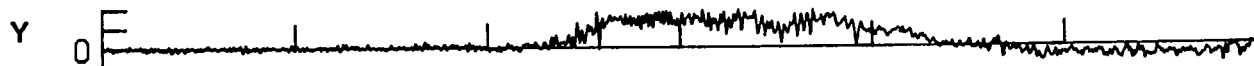
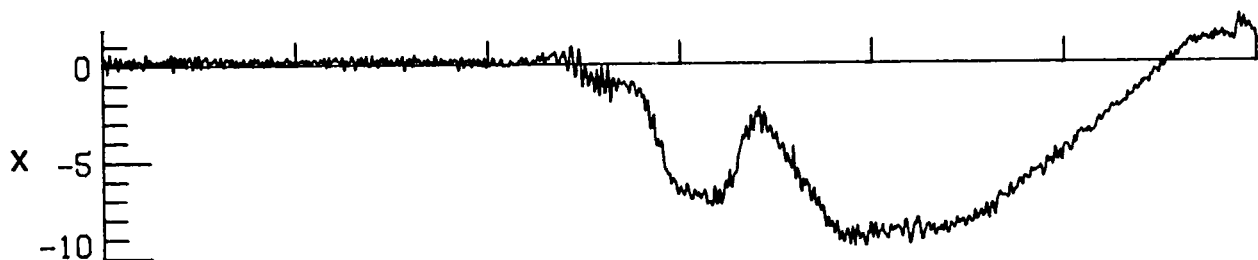
RIGHT DUMMY PELVIS  
NASA TEST NO. 8



LEFT DUMMY CHEST  
NASA TEST NO. 8



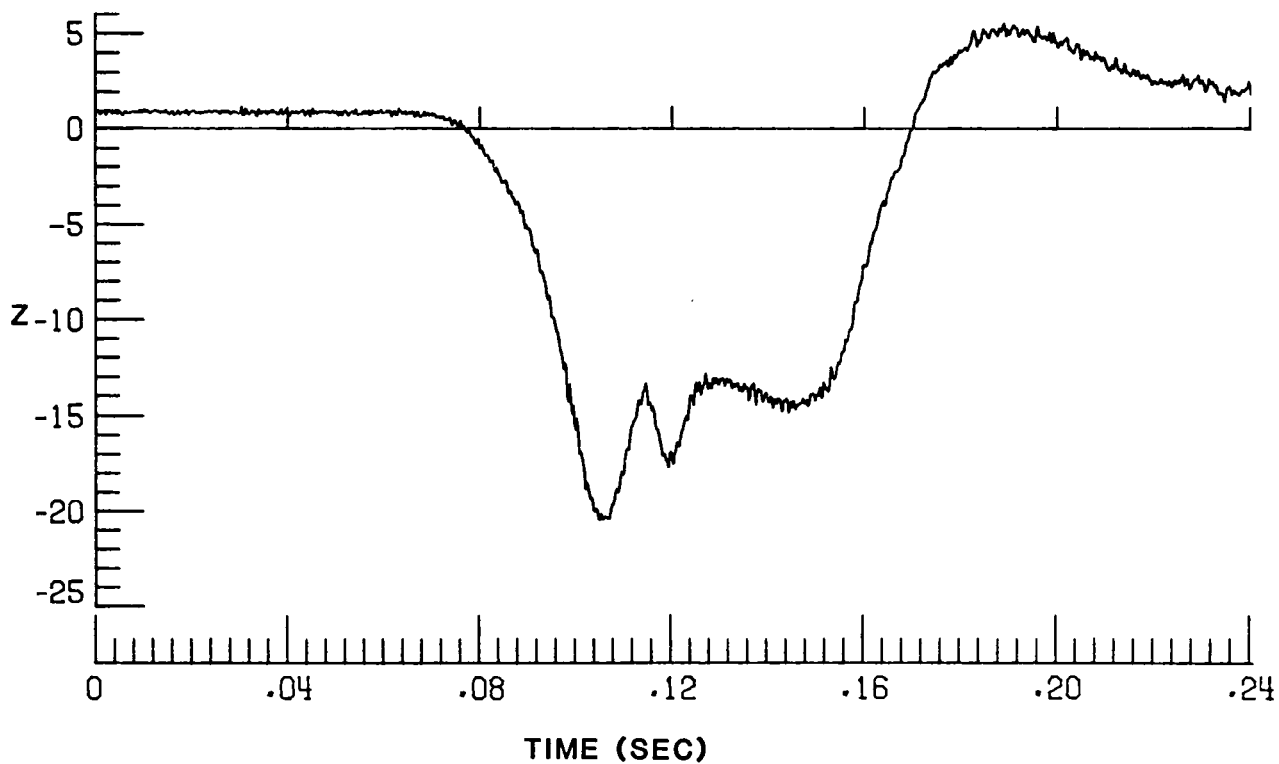
RIGHT DUMMY CHEST  
NASA TEST NO. 8



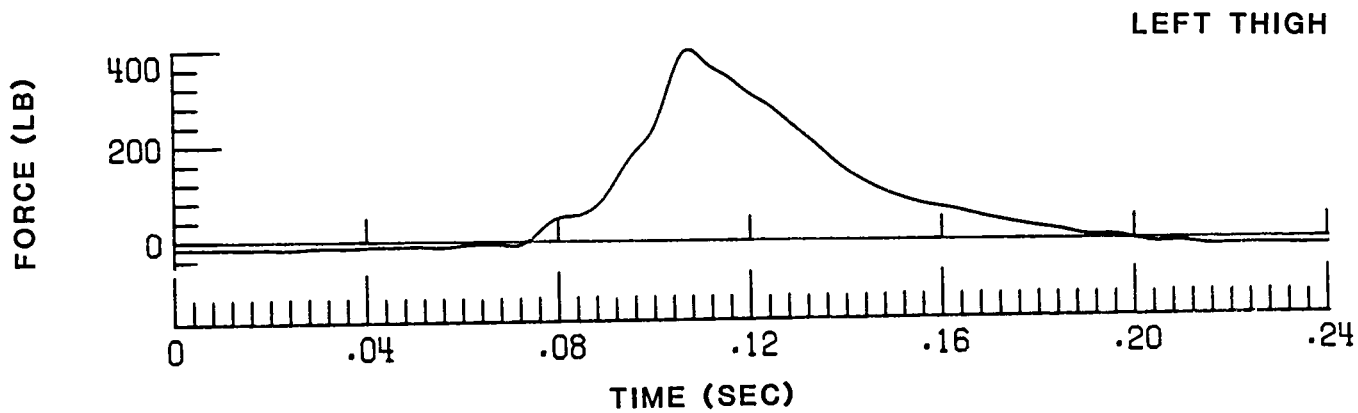
LEFT DUMMY HEAD  
NASA TEST NO. 8



ACCELERATION (G)

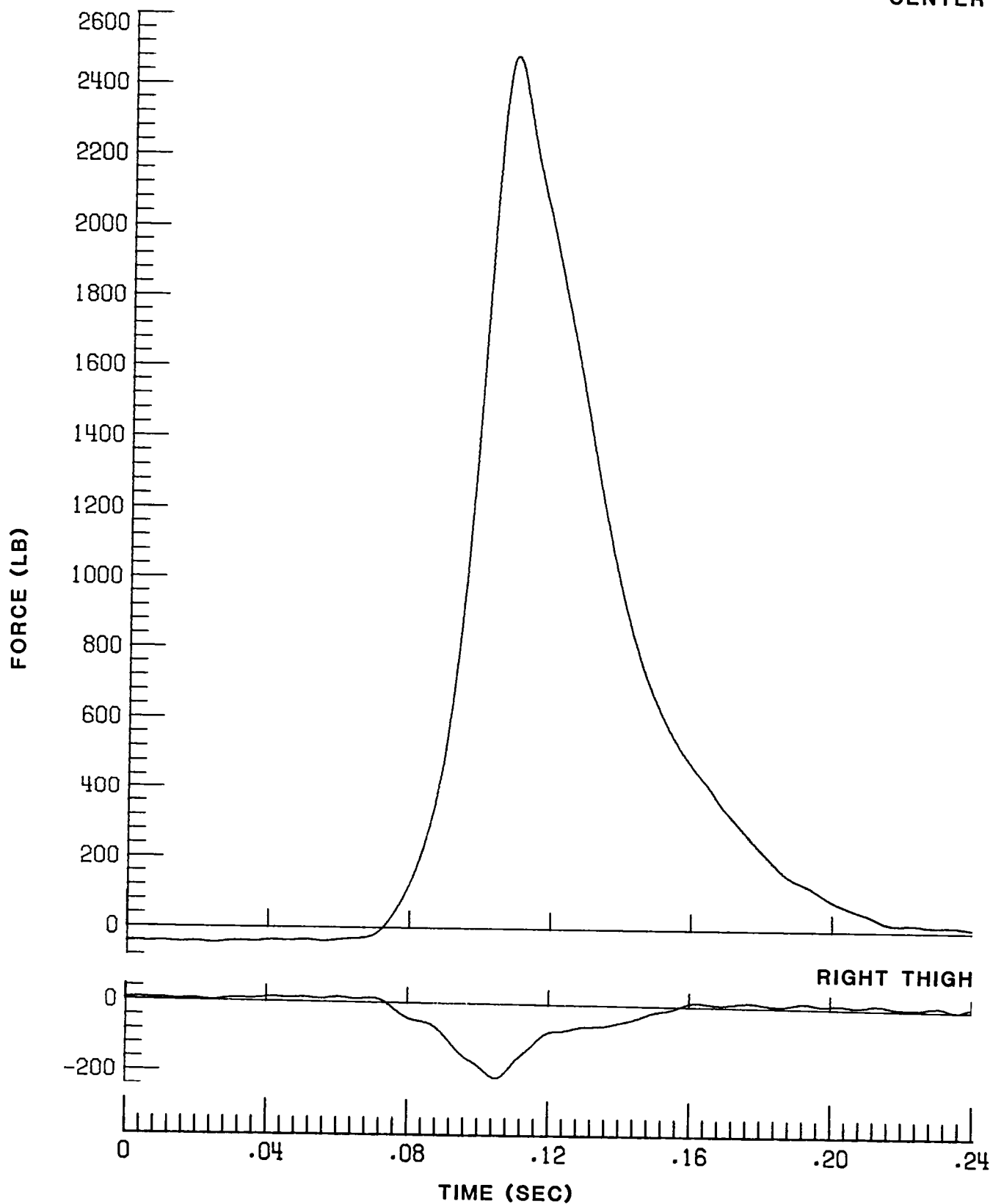


RIGHT DUMMY HEAD  
NASA TEST NO. 8



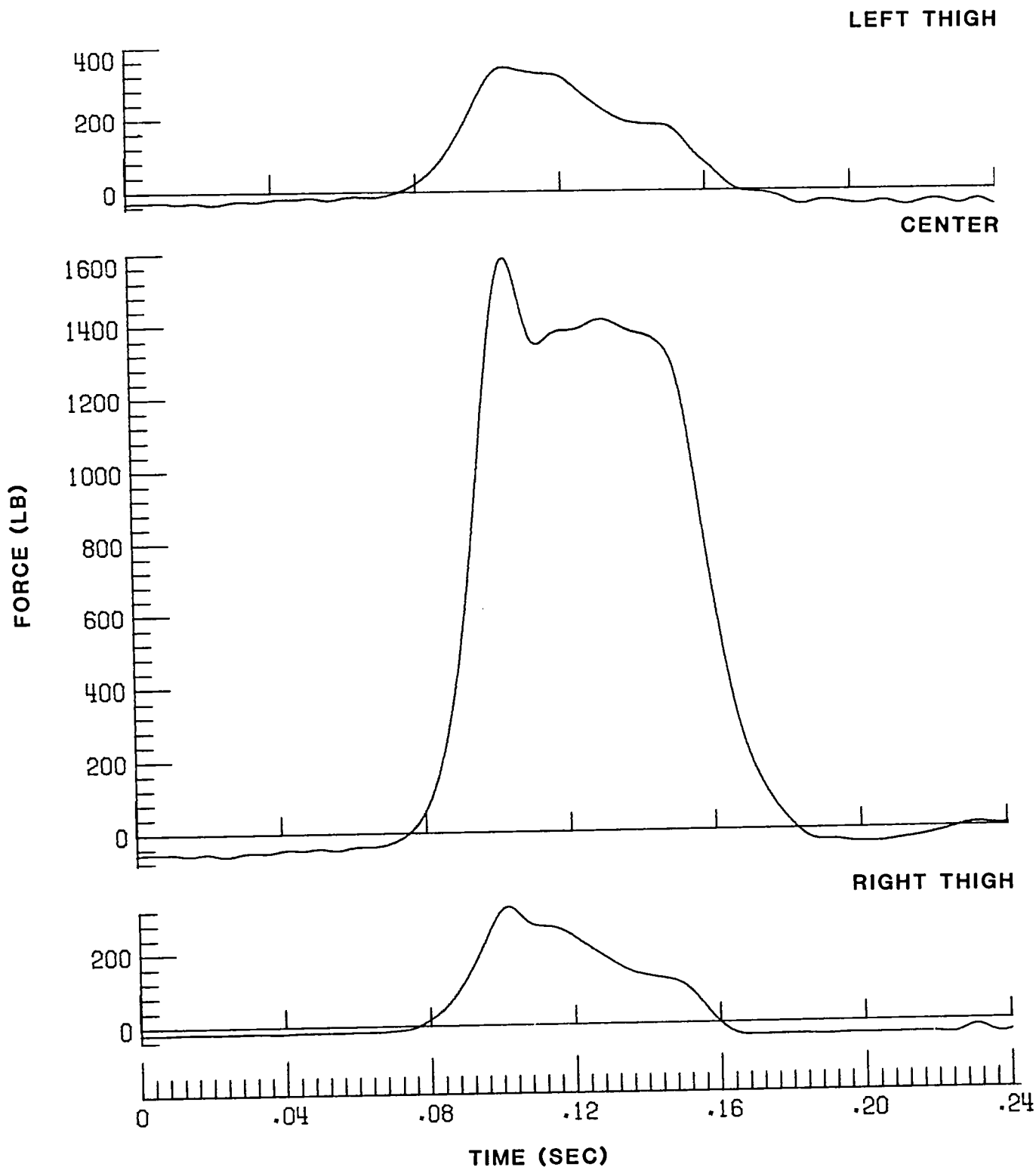
LEFT SEAT PAN LOADS  
NASA TEST NO. 8

CENTER

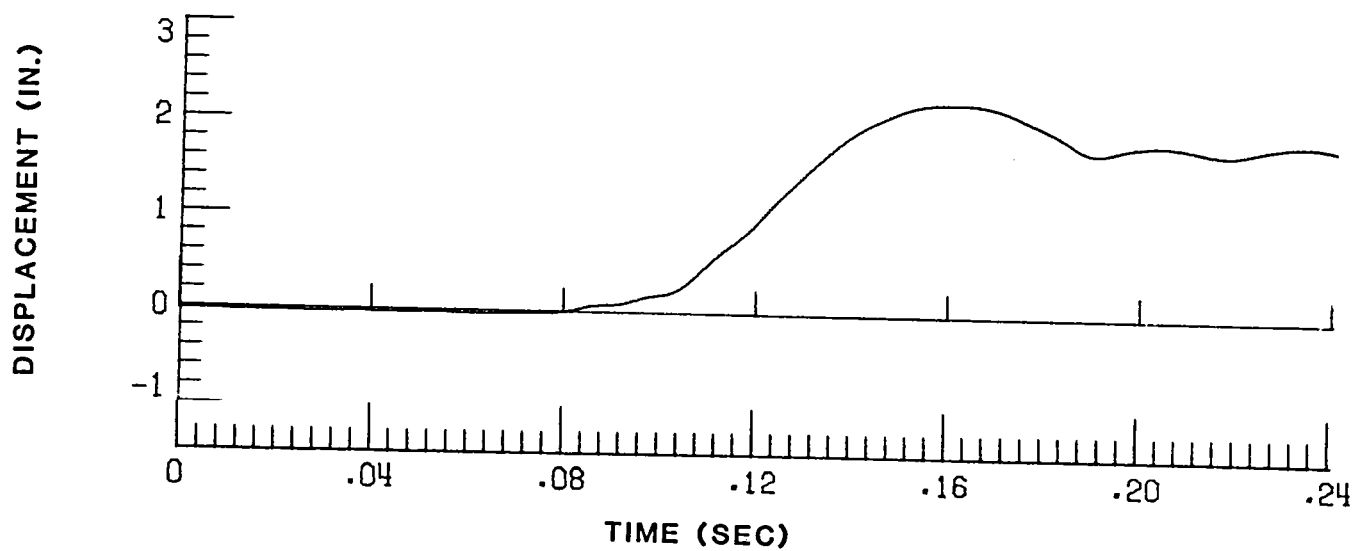


LEFT SEAT PAN LOADS  
NASA TEST NO. 8

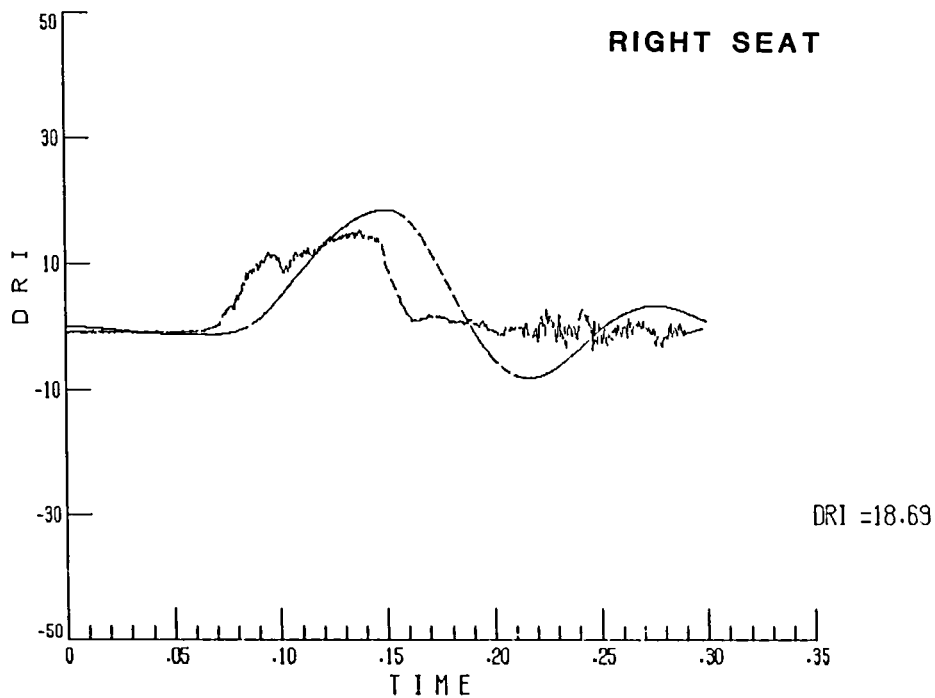
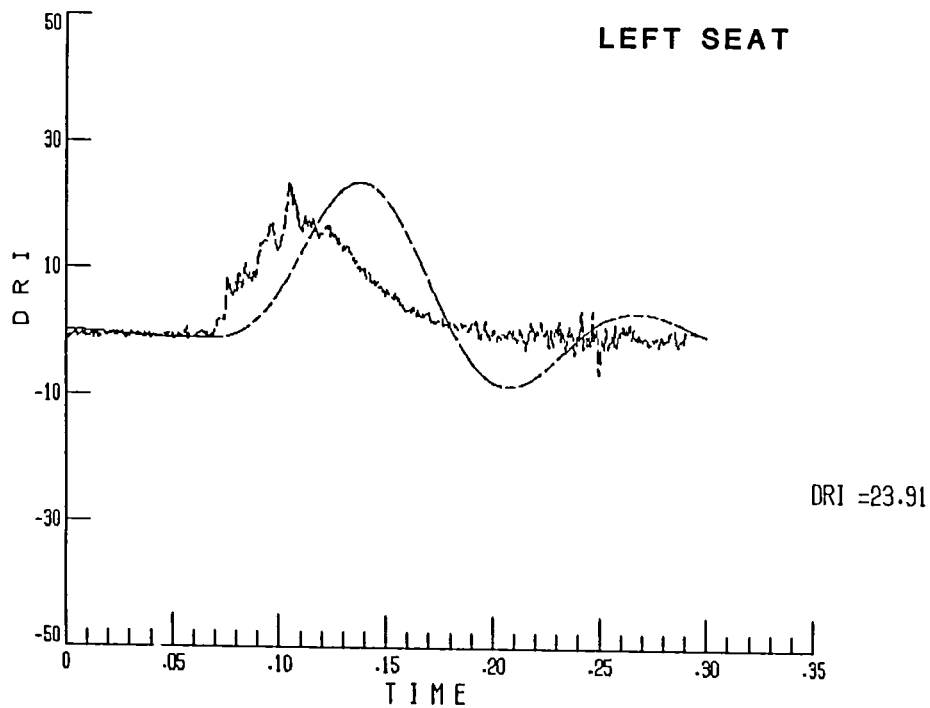




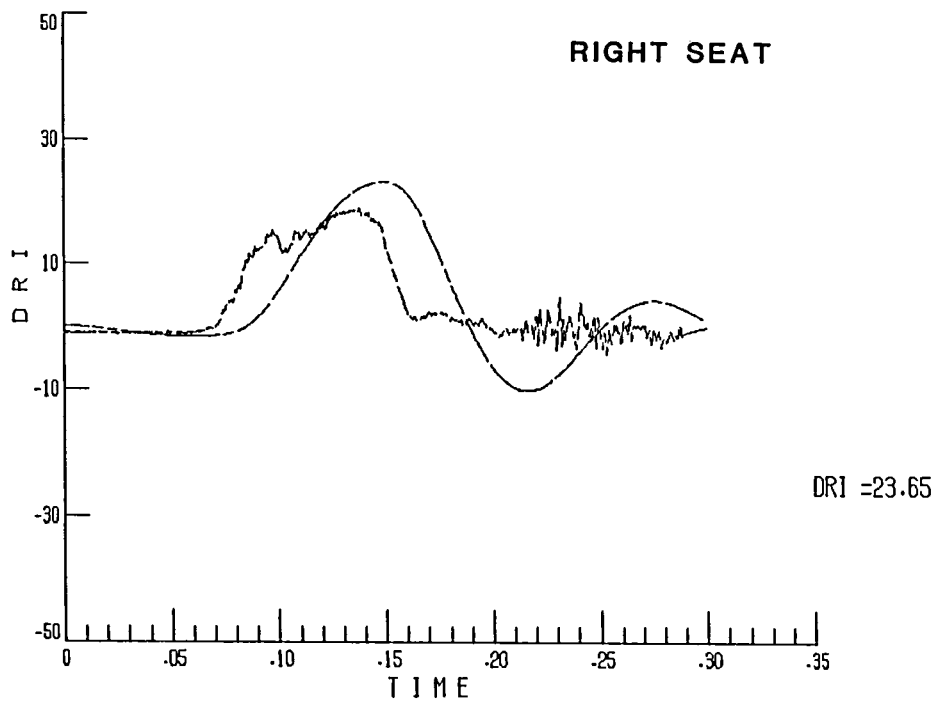
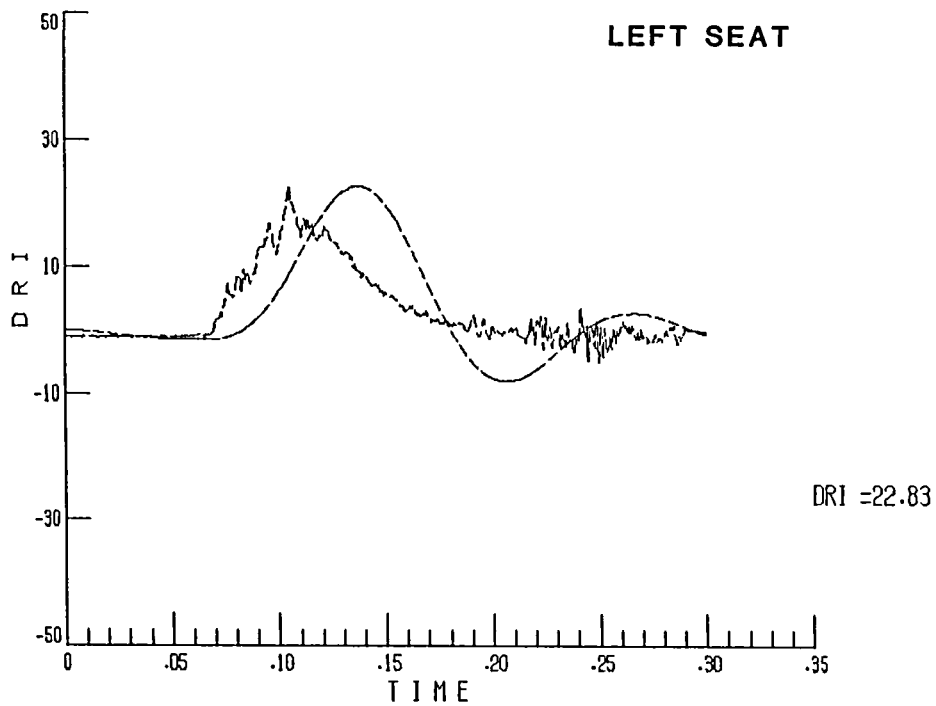
RIGHT SEAT PAN LOADS  
NASA TEST NO. 8



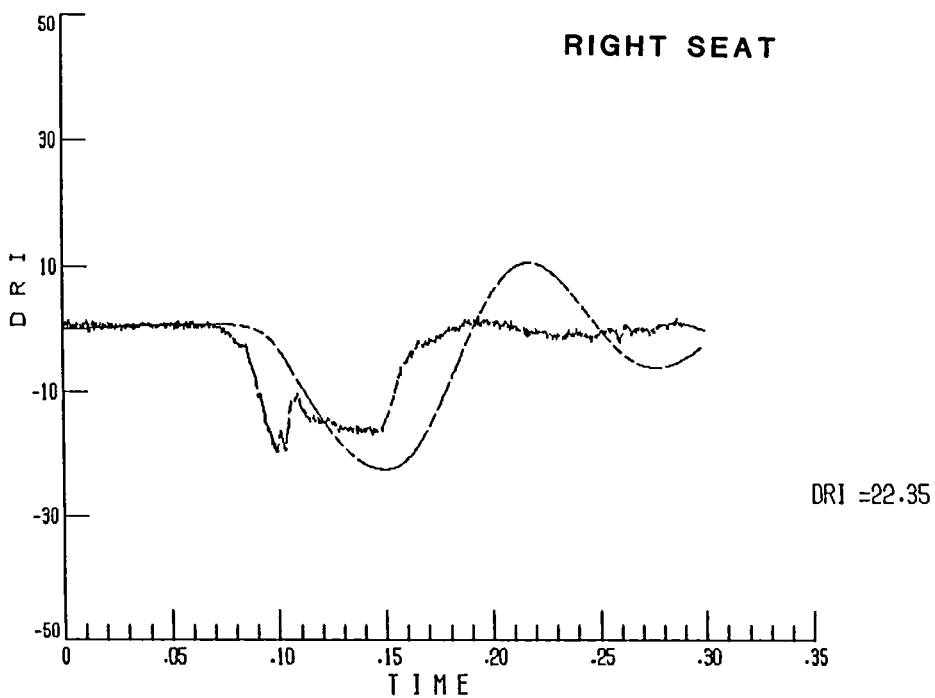
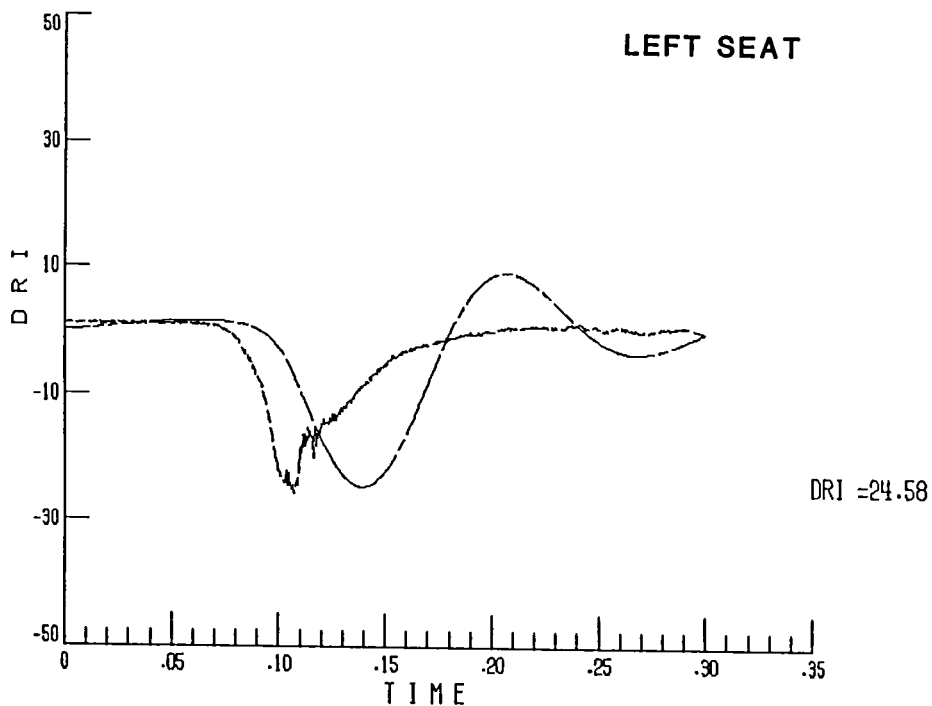
RIGHT SEAT STROKE  
NASA TEST NO. 8



**DYNAMIC RESPONSE INDEX  
SEAT PAN Z-AXIS  
NASA TEST NO. 8**



**DYNAMIC RESPONSE INDEX  
BACK TANGENT LINE  
NASA TEST NO. 8**



DYNAMIC RESPONSE INDEX  
DUMMY PELVIS Z-AXIS  
NASA TEST NO. 8  
151

10.0 NASA TEST NO. 9

HORIZONTAL VELOCITY: 0 ft/sec

VERTICAL VELOCITY: 25 ft/sec

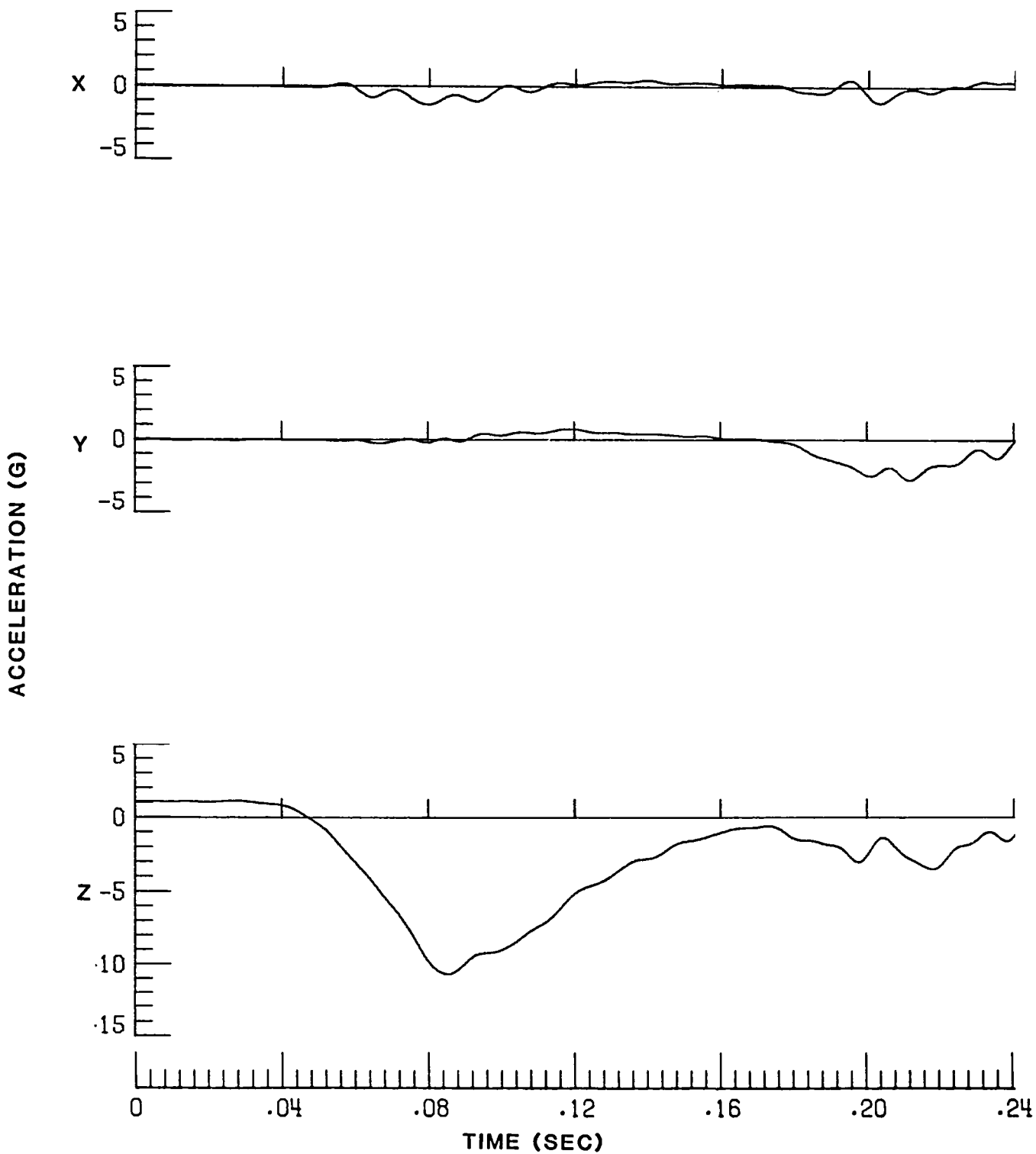
CREW MODULE PITCH:  $2^0$

CREW MODULE ROLL:  $0^0$

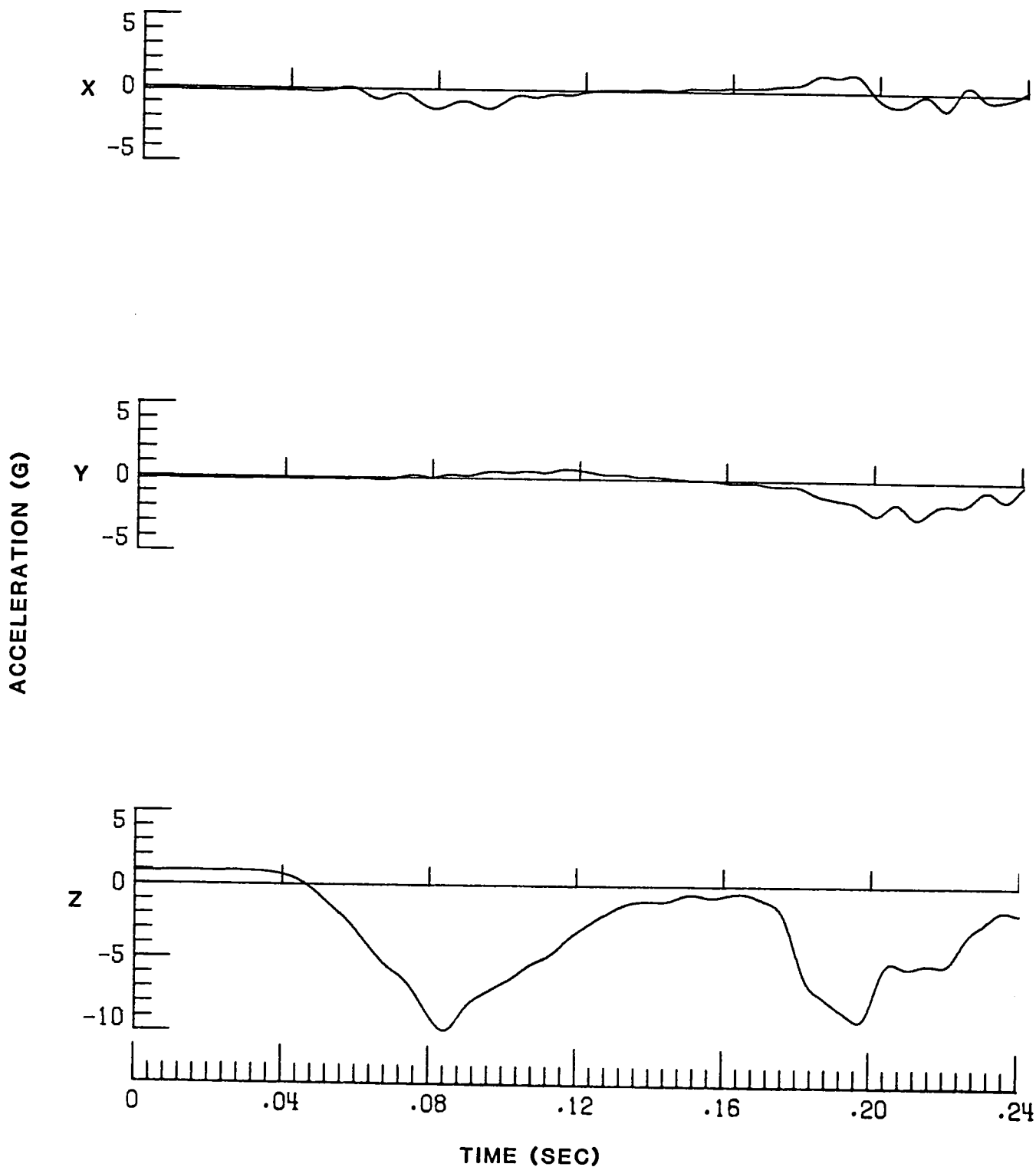
CREW MODULE YAW:  $0^0$

LEFT SEAT: F-111 Operational

RIGHT SEAT: Energy Absorbing

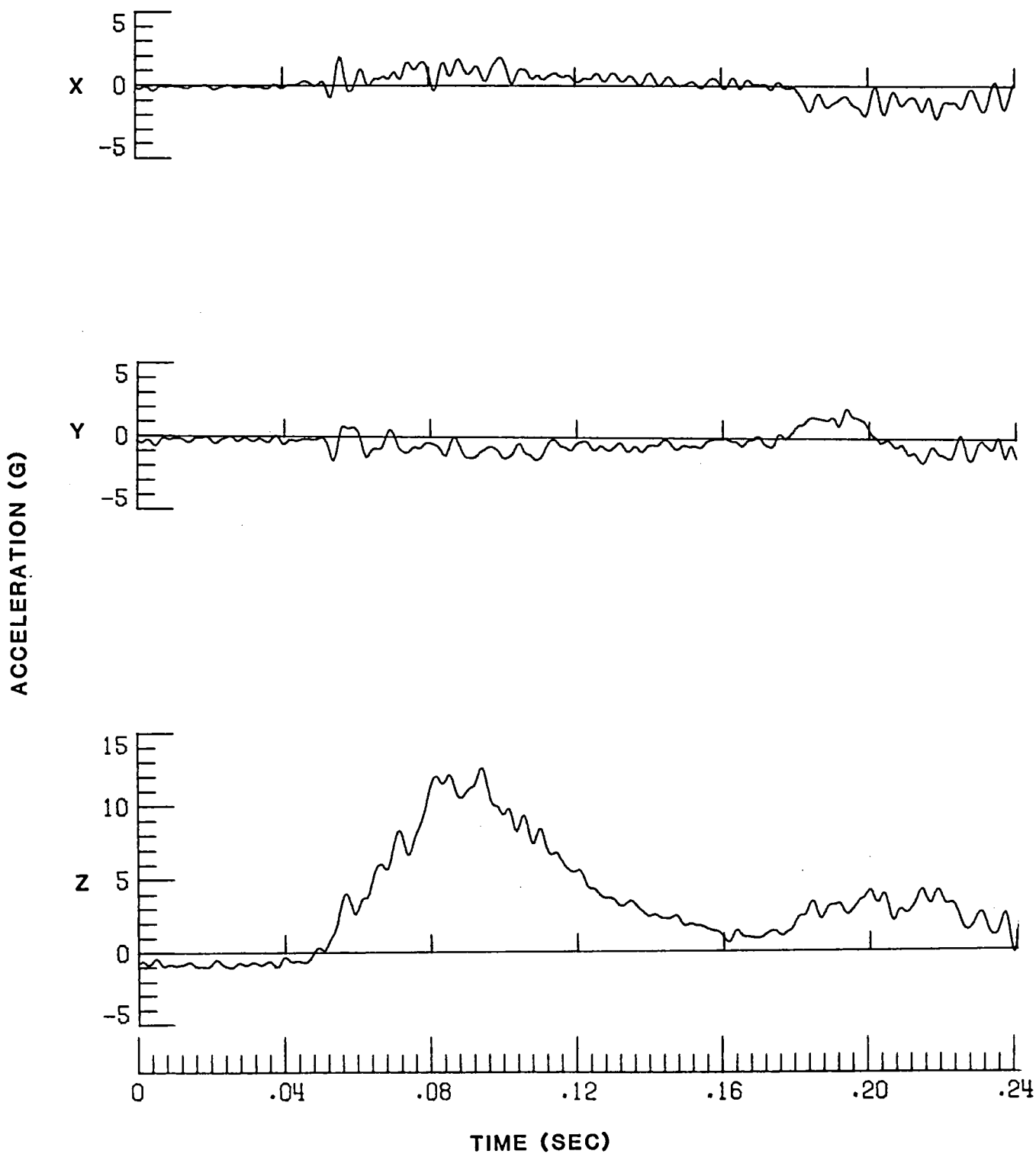


LEFT BULKHEAD  
NASA TEST NO. 9

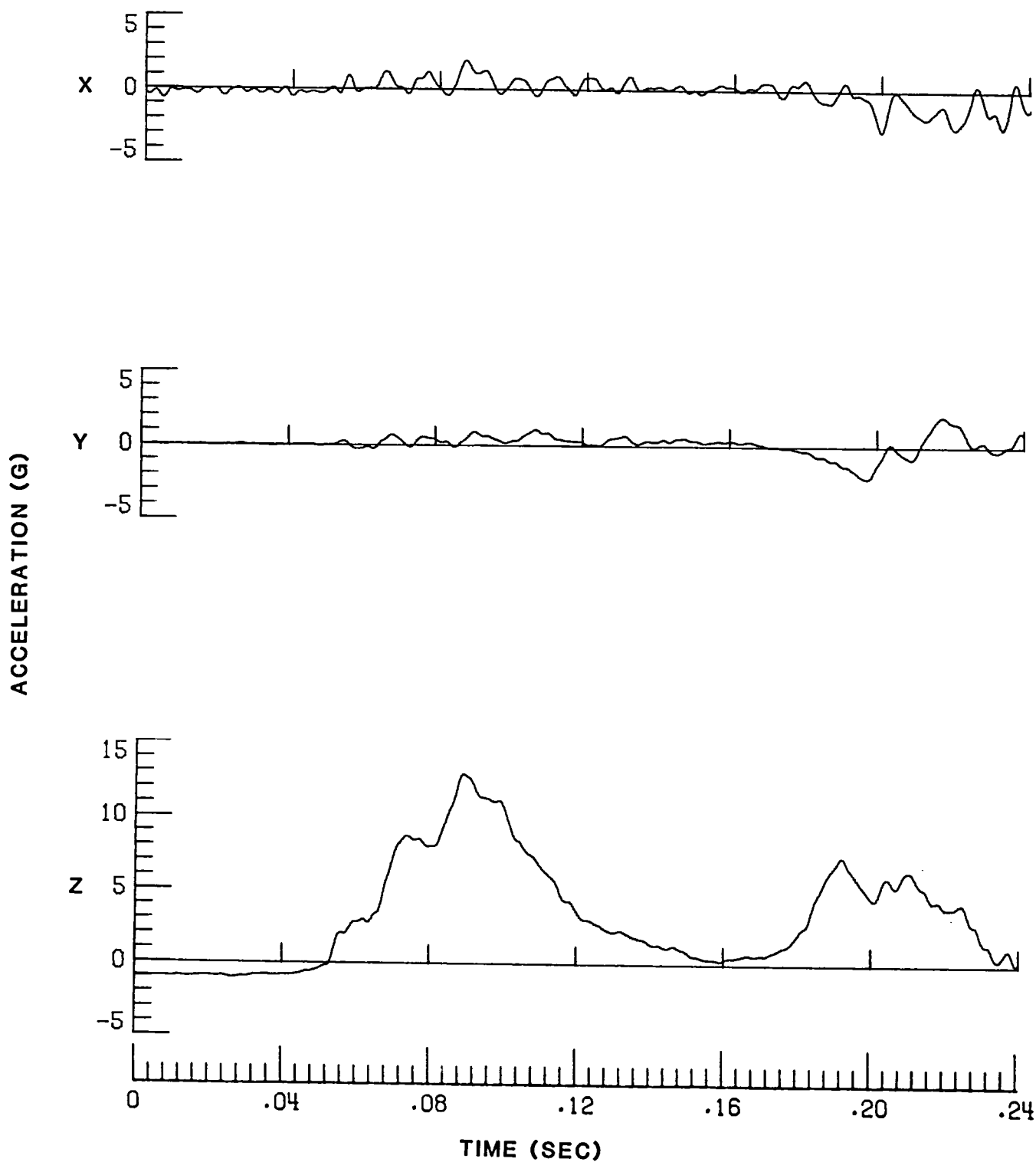


RIGHT BULKHEAD  
NASA TEST NO. 9

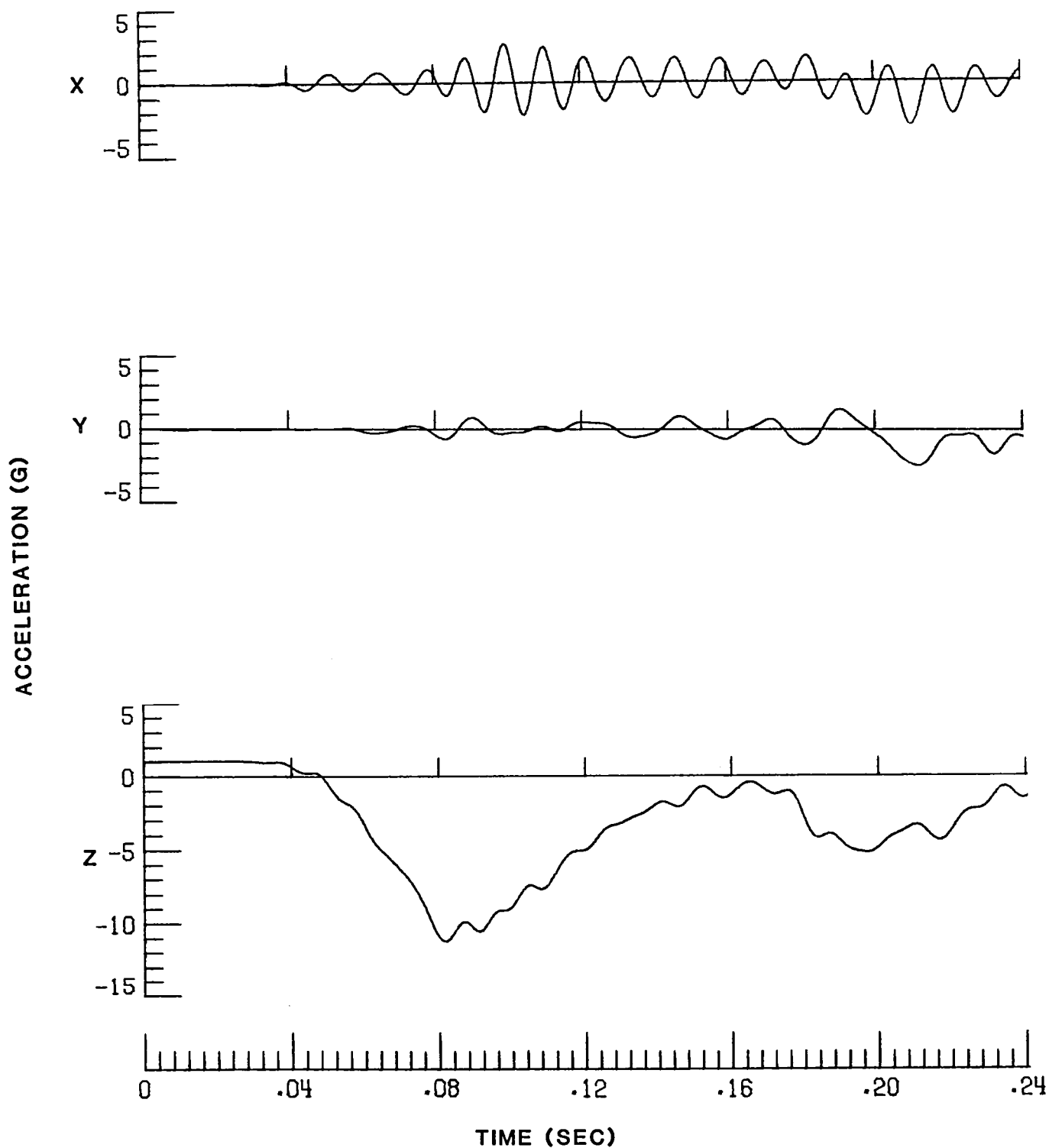




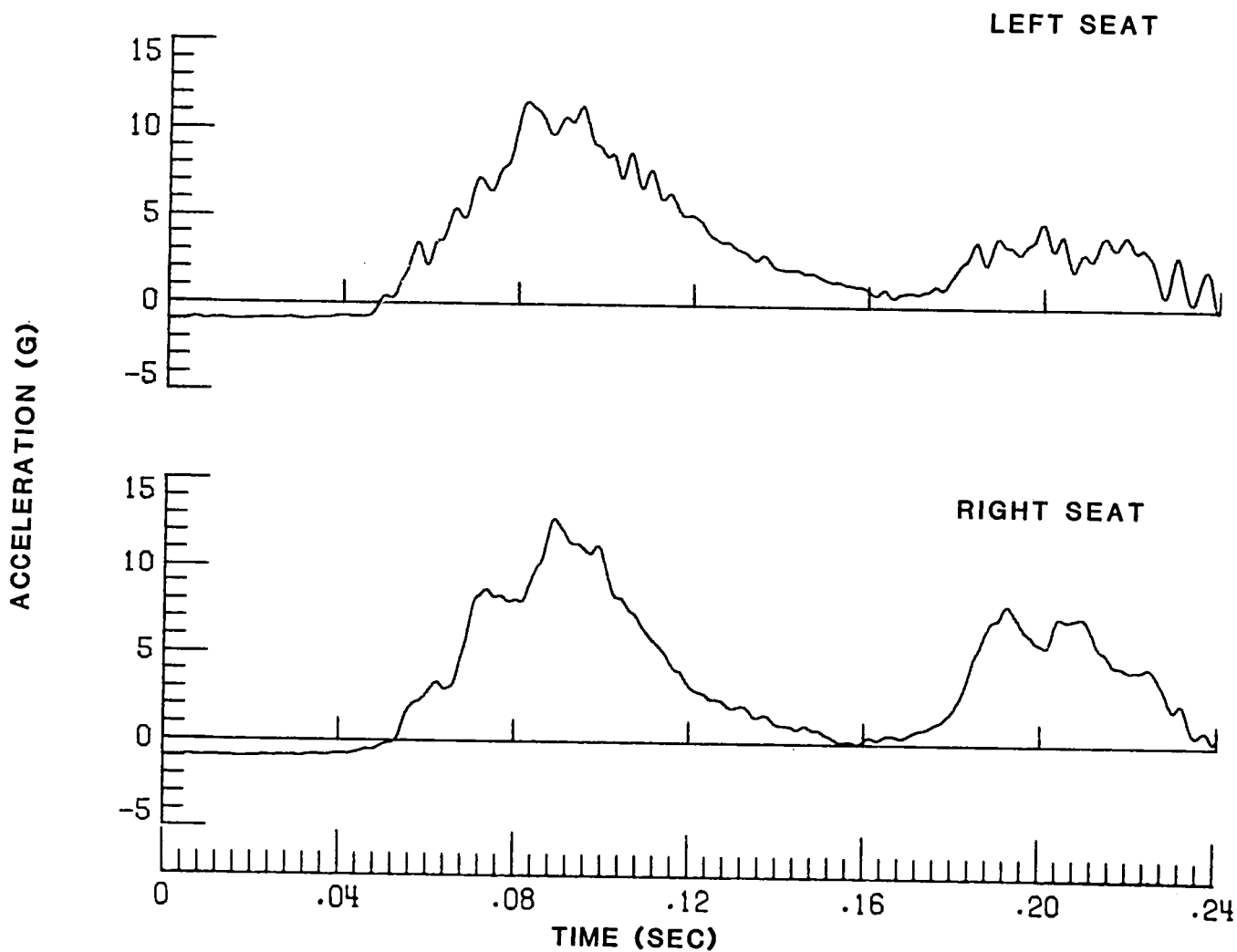
LEFT SEAT PAN  
NASA TEST NO. 9



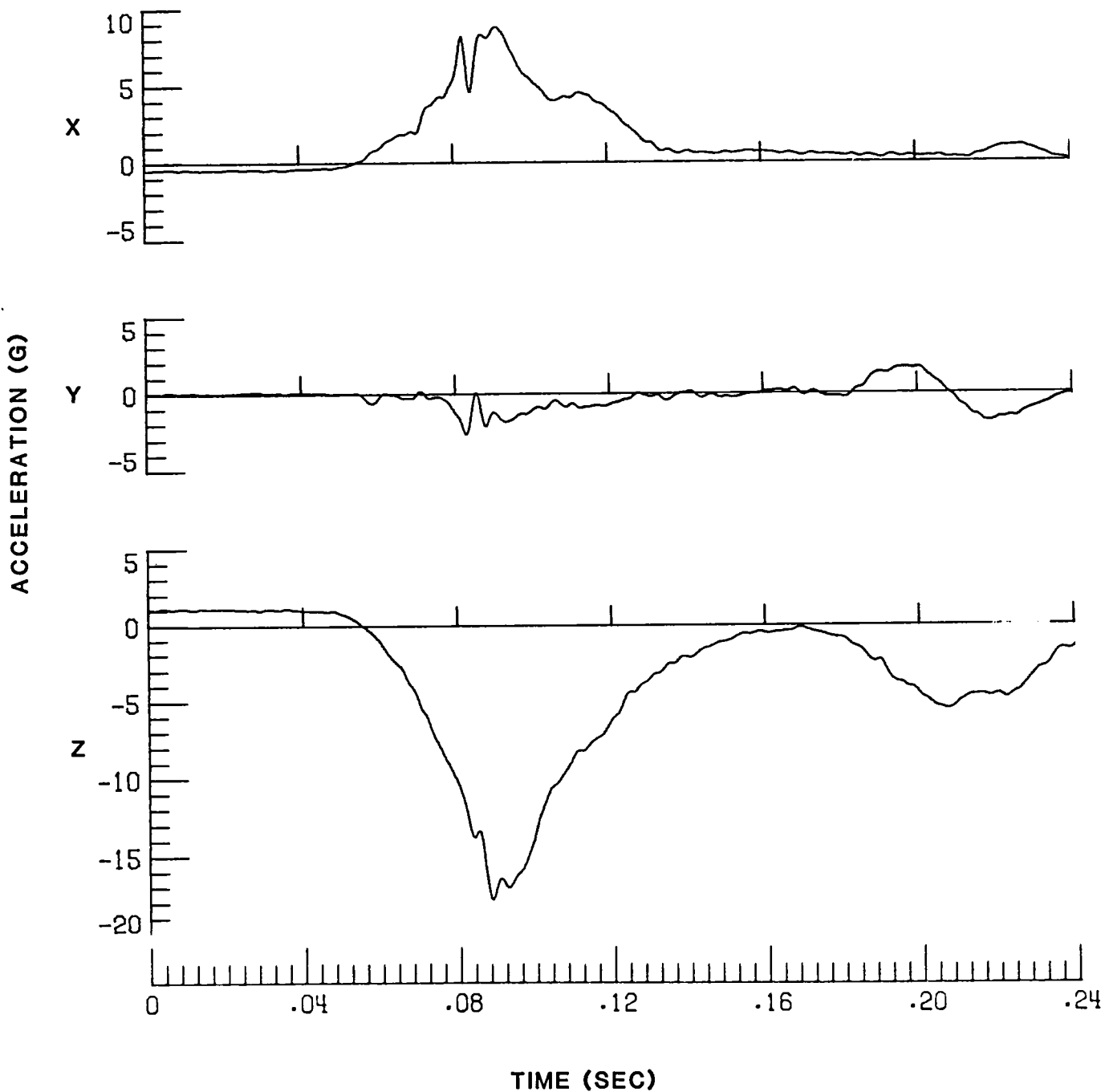
RIGHT SEAT PAN  
NASA TEST NO. 9



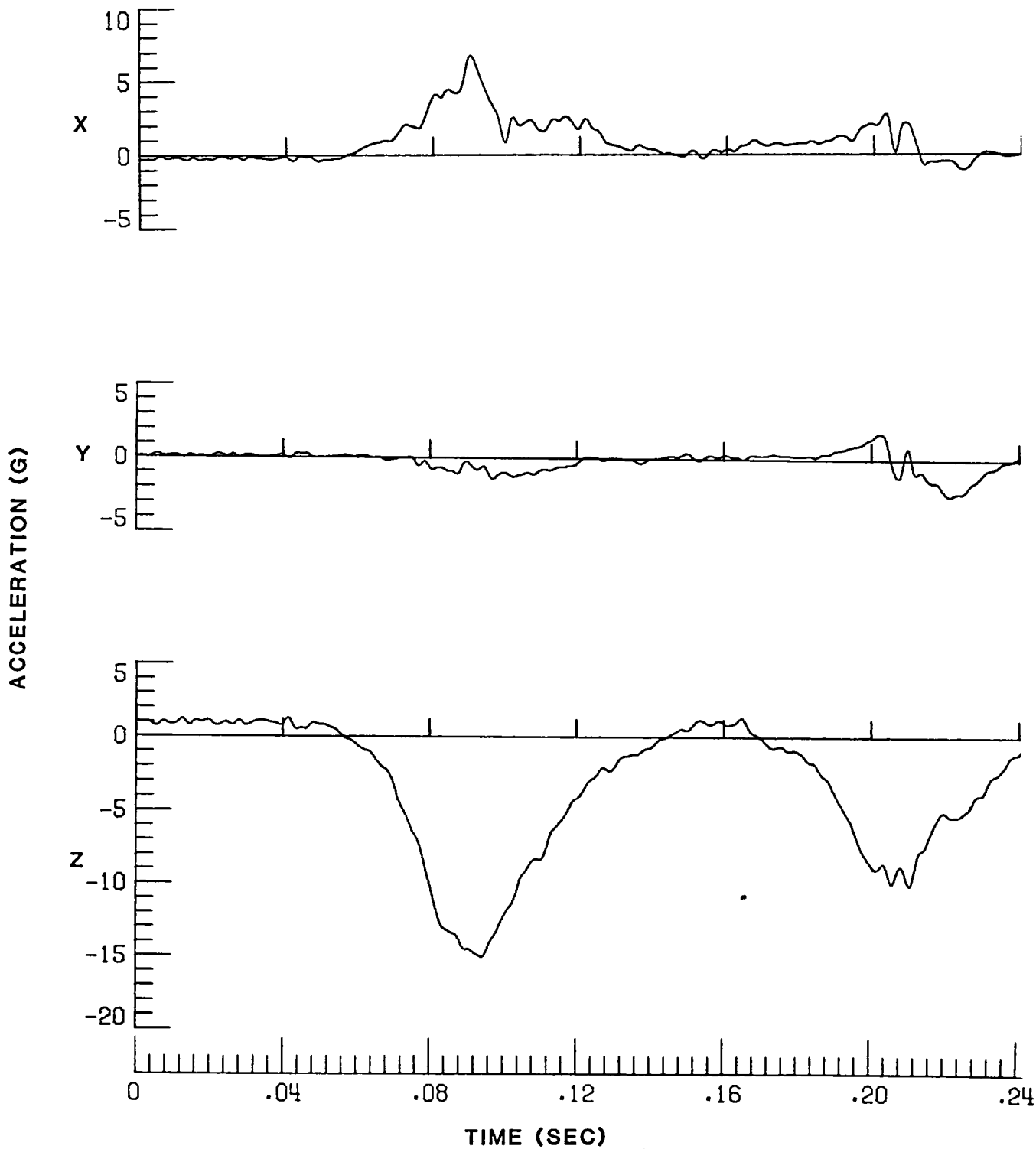
CAPSULE C.G.  
NASA TEST NO. 9



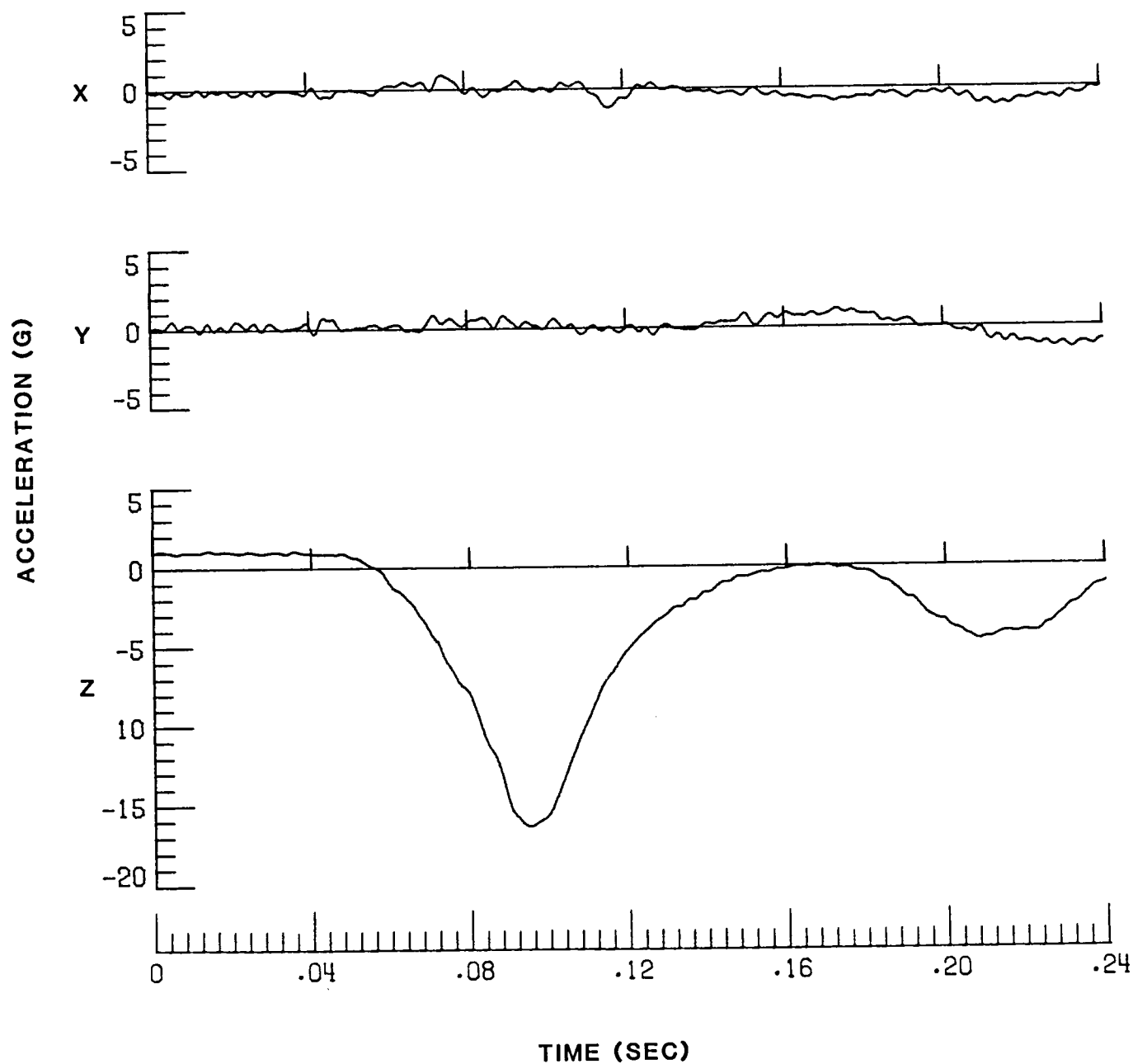
SEAT PAN ALONG BACK TANGENT LINE  
NASA TEST NO. 9



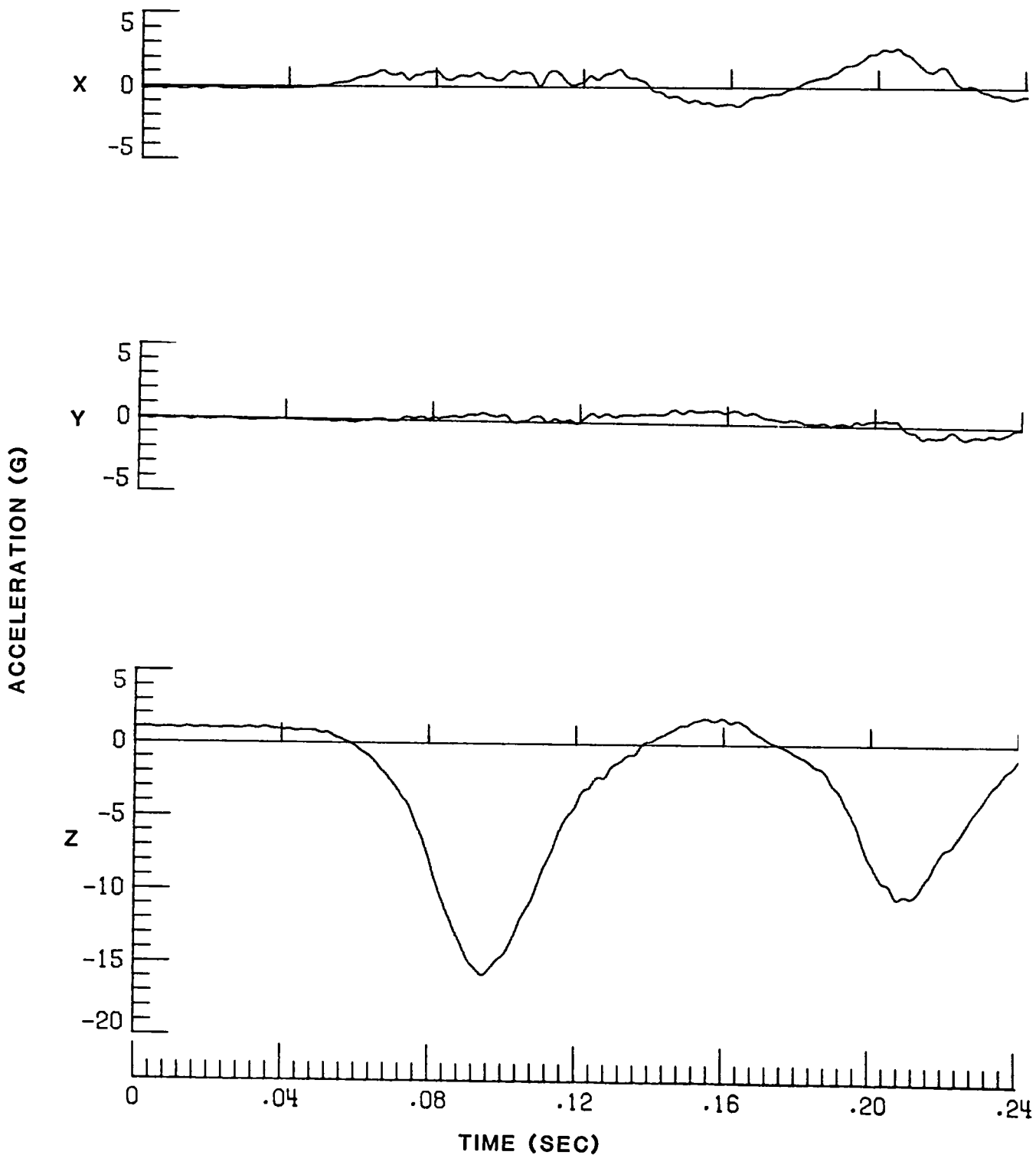
LEFT DUMMY PELVIS  
NASA TEST NO. 9



RIGHT DUMMY PELVIS  
NASA TEST NO. 9

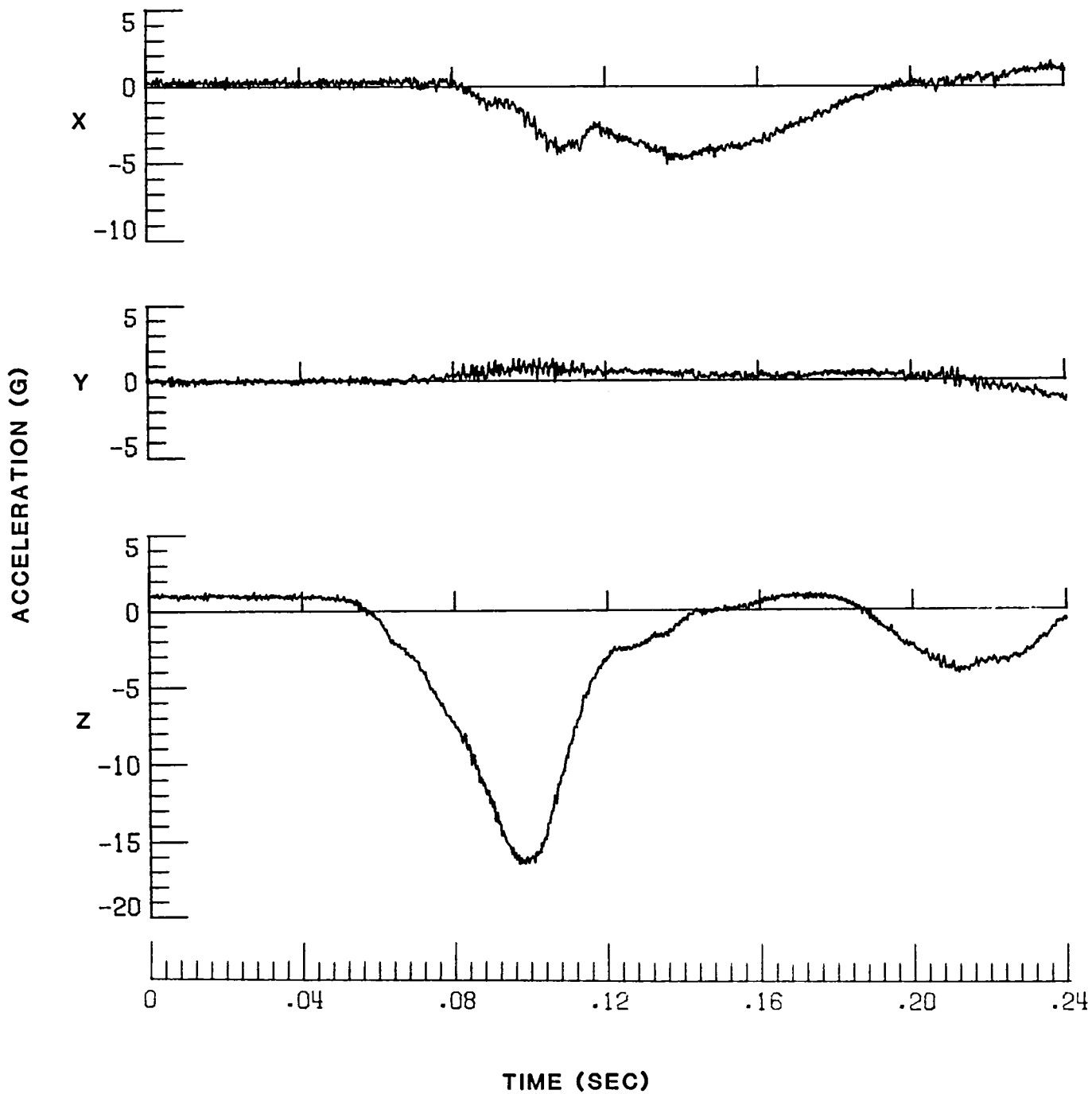


LEFT DUMMY CHEST  
NASA TEST NO. 9

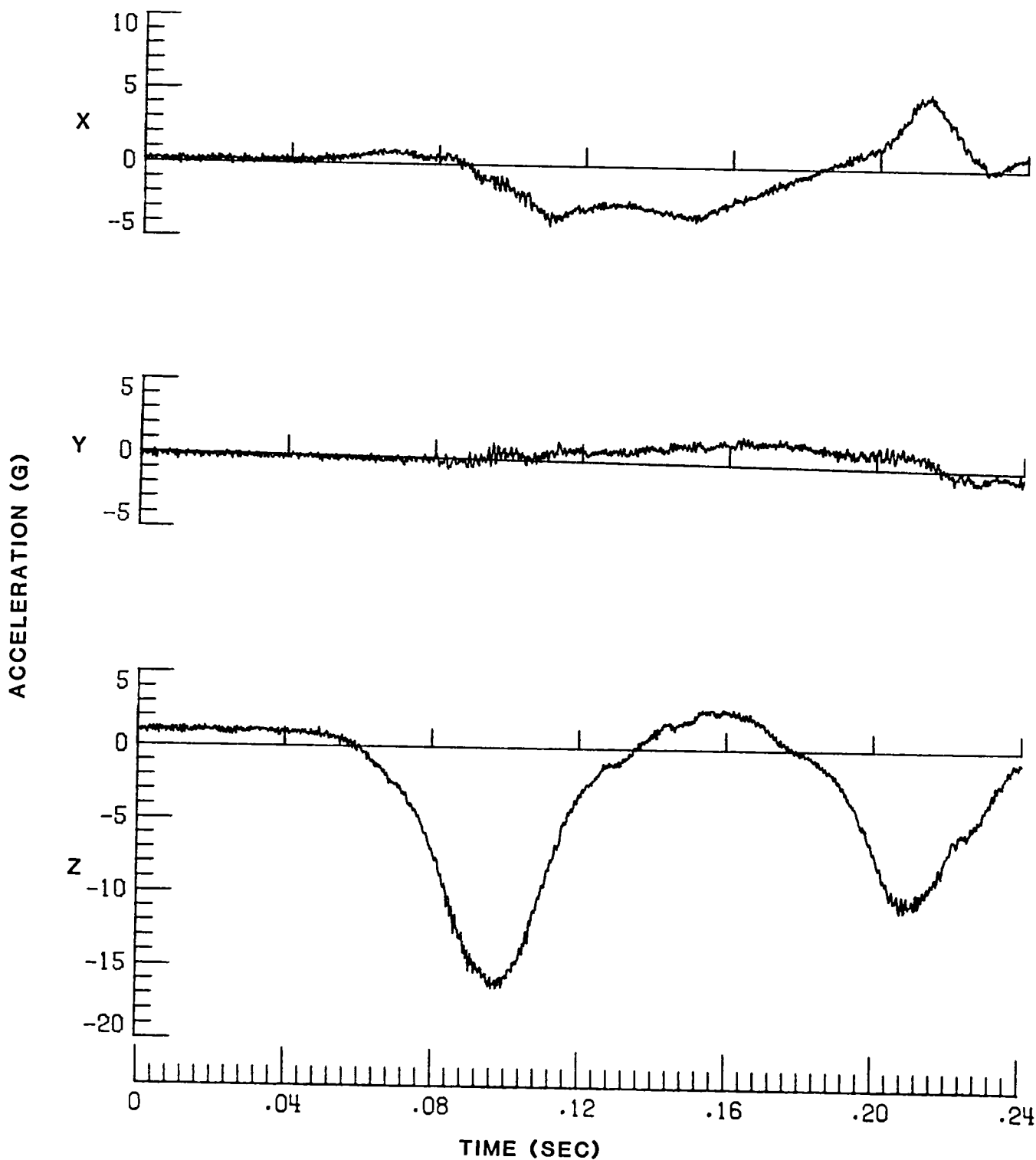


RIGHT DUMMY CHEST  
NASA TEST NO. 9

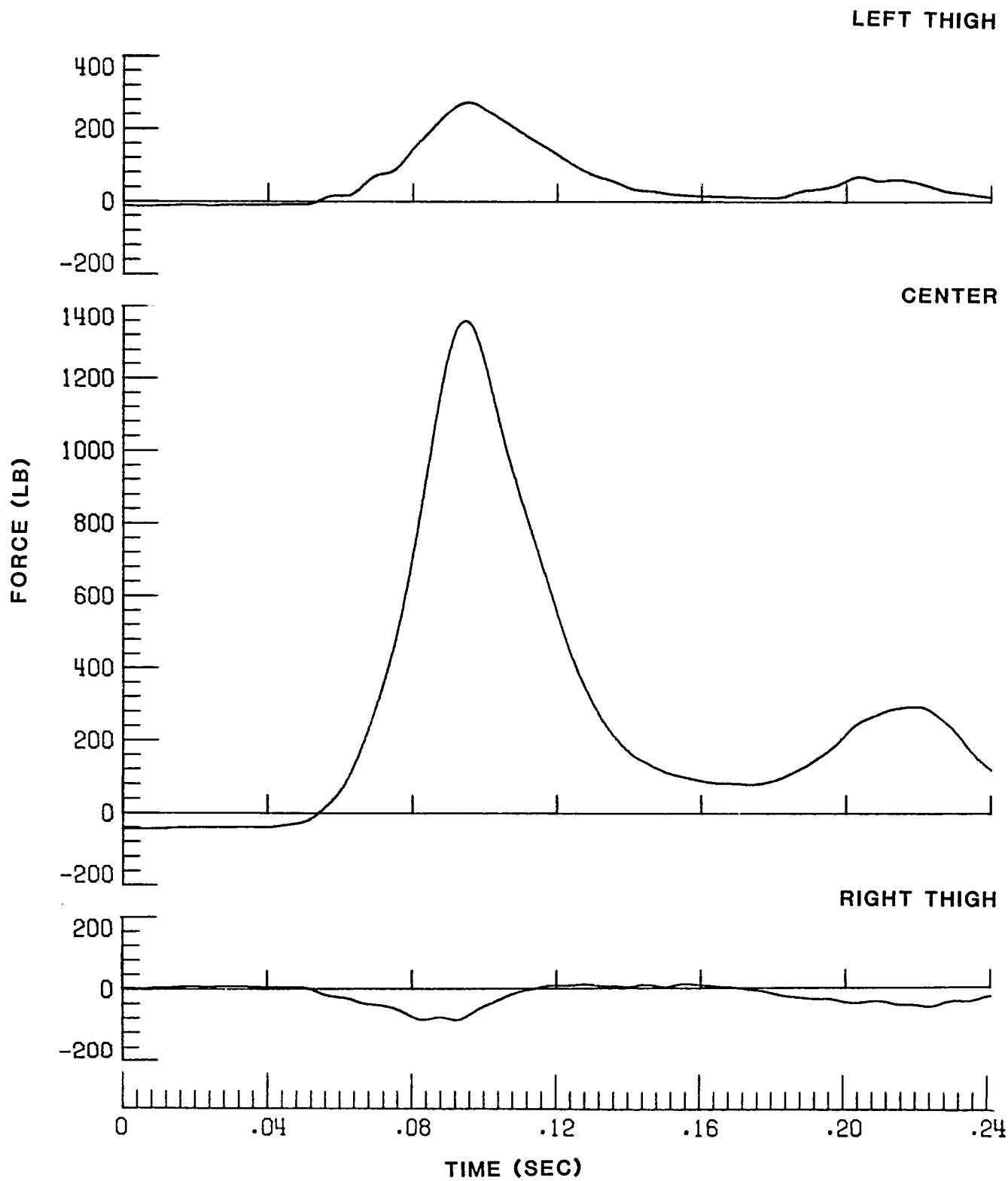




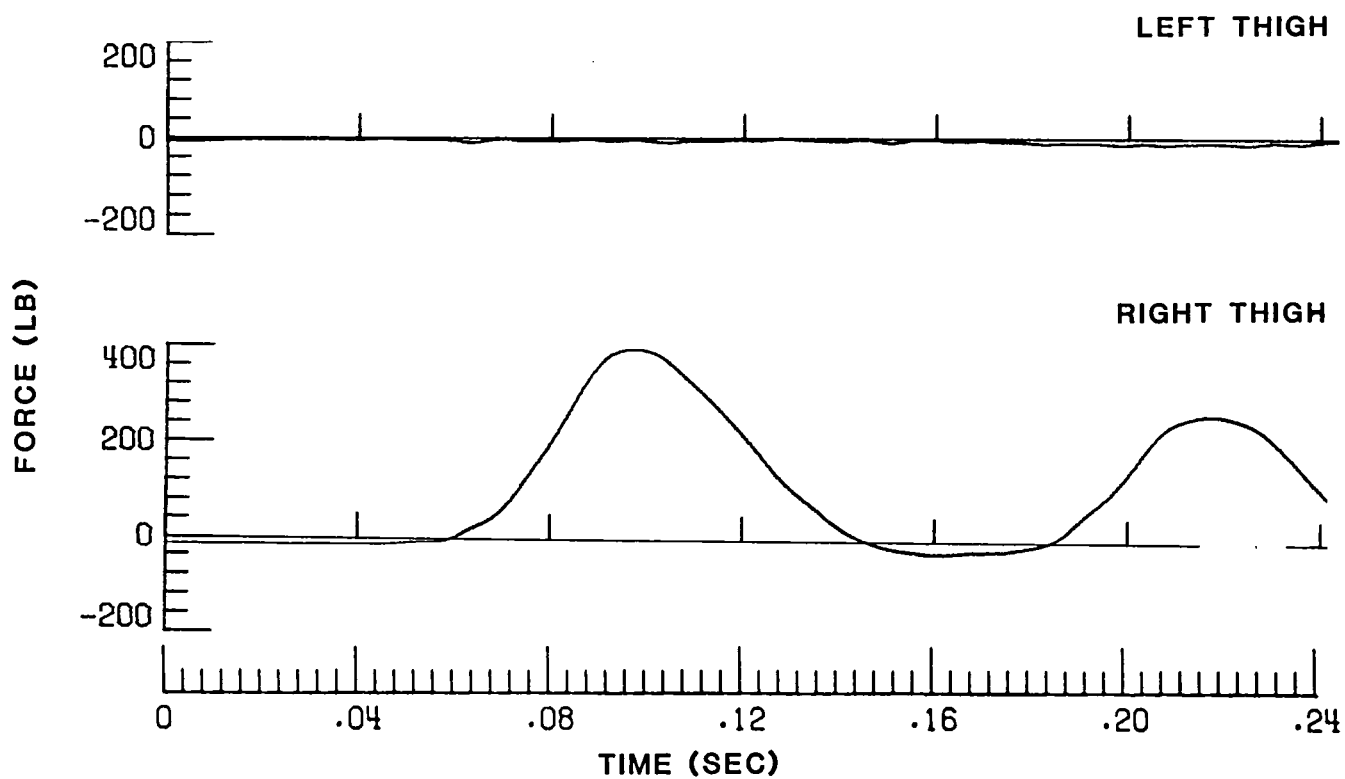
LEFT DUMMY HEAD  
NASA TEST NO. 9



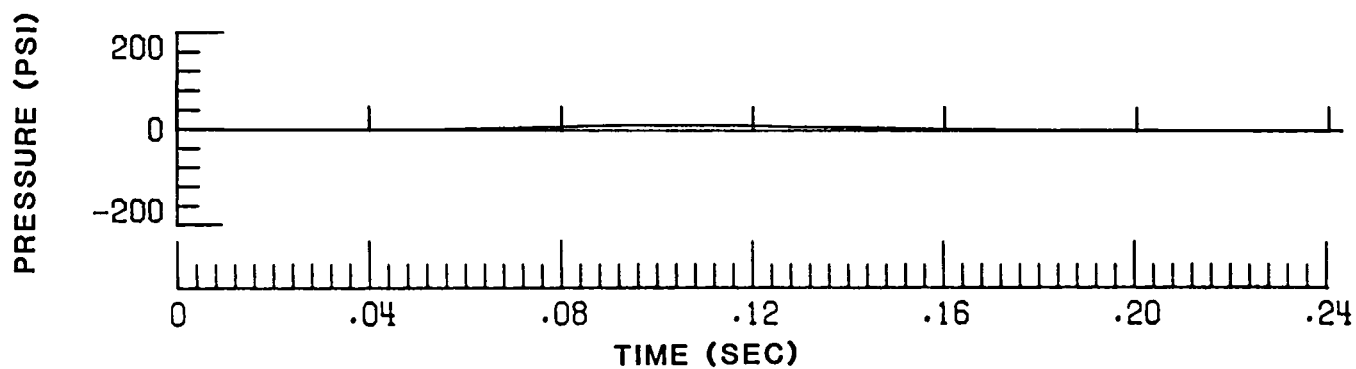
RIGHT DUMMY HEAD  
NASA TEST NO. 9



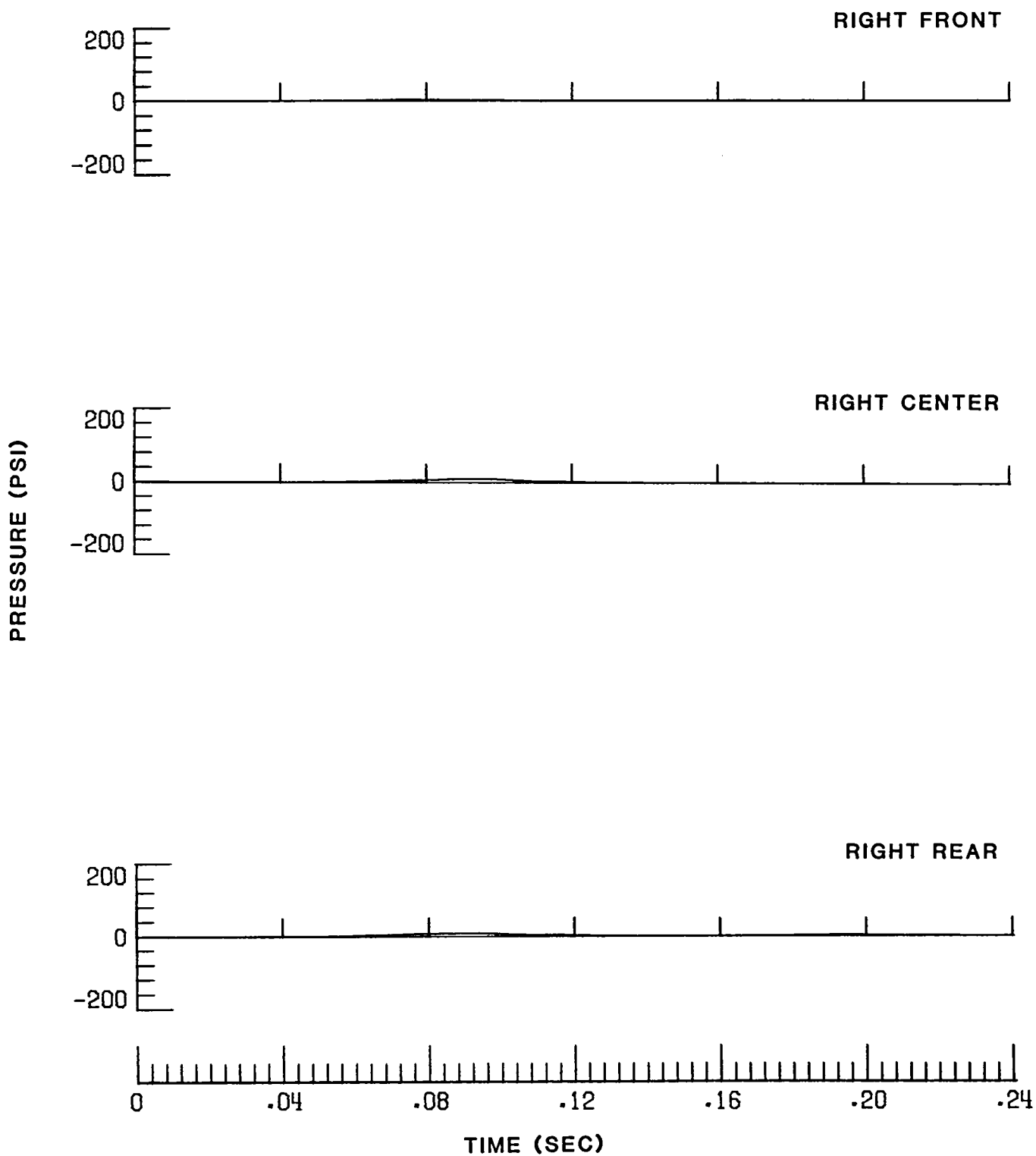
LEFT SEAT PAN LOADS  
NASA TEST NO. 9



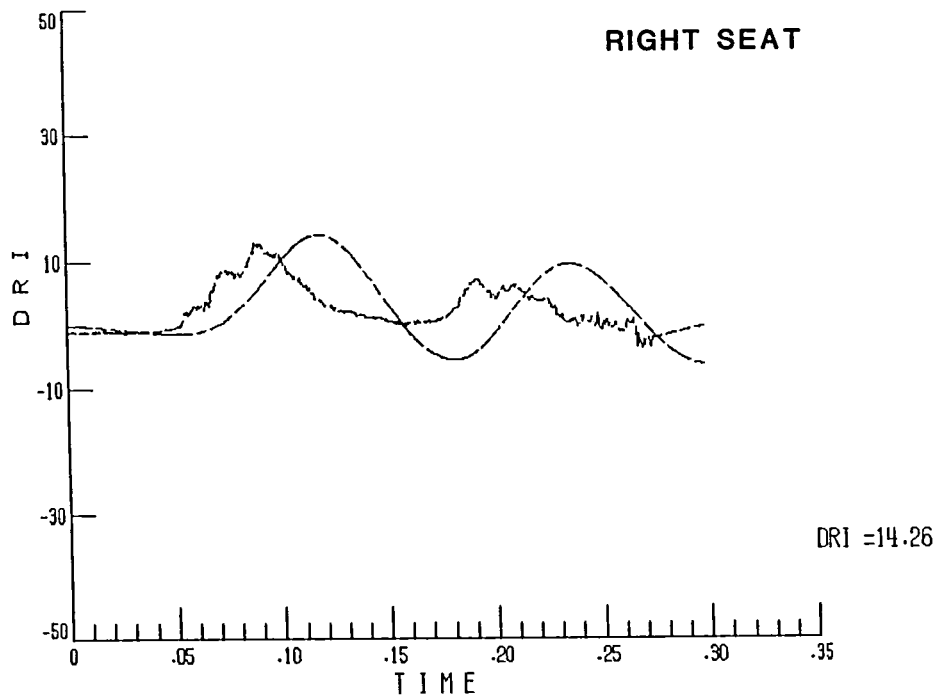
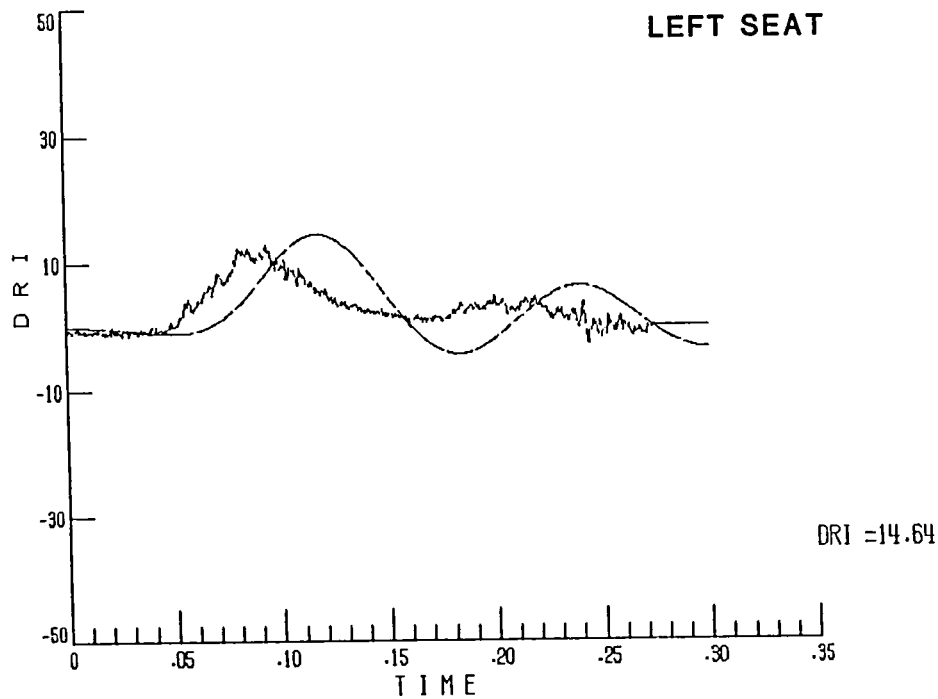
RIGHT SEAT PAN LOADS  
NASA TEST NO. 9



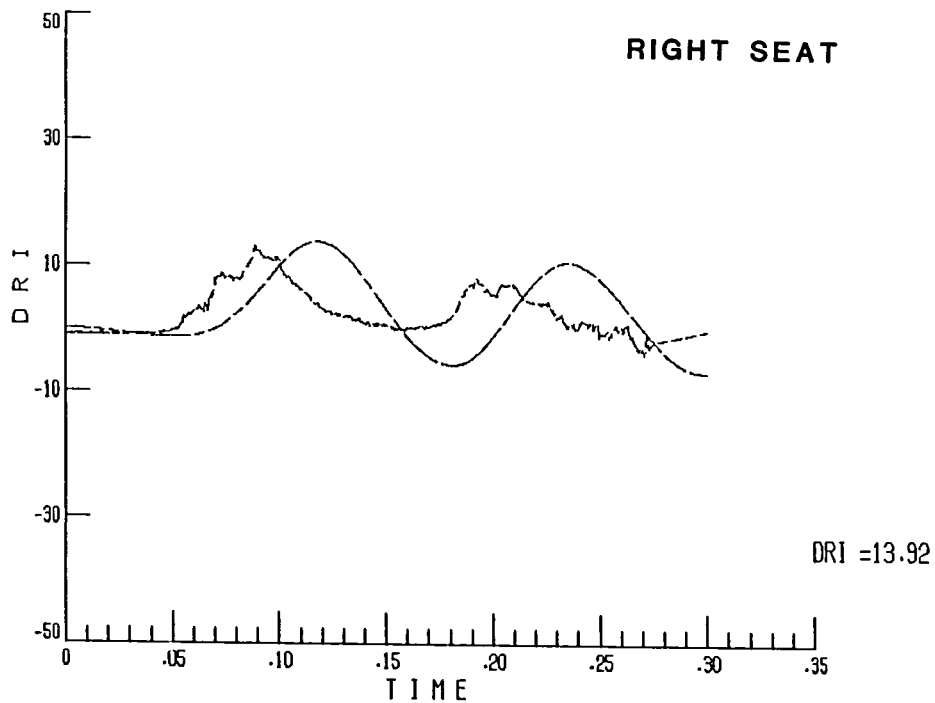
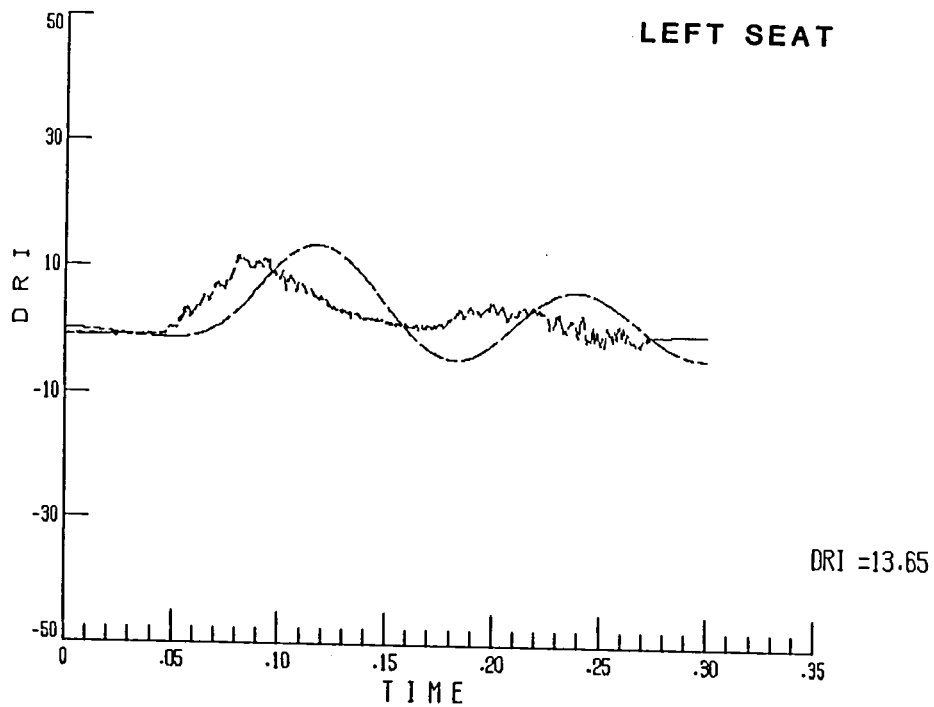
LEFT REAR EXIT PORT PRESSURE  
NASA TEST NO. 9



EXIT PORT PRESSURES  
NASA TEST NO. 9

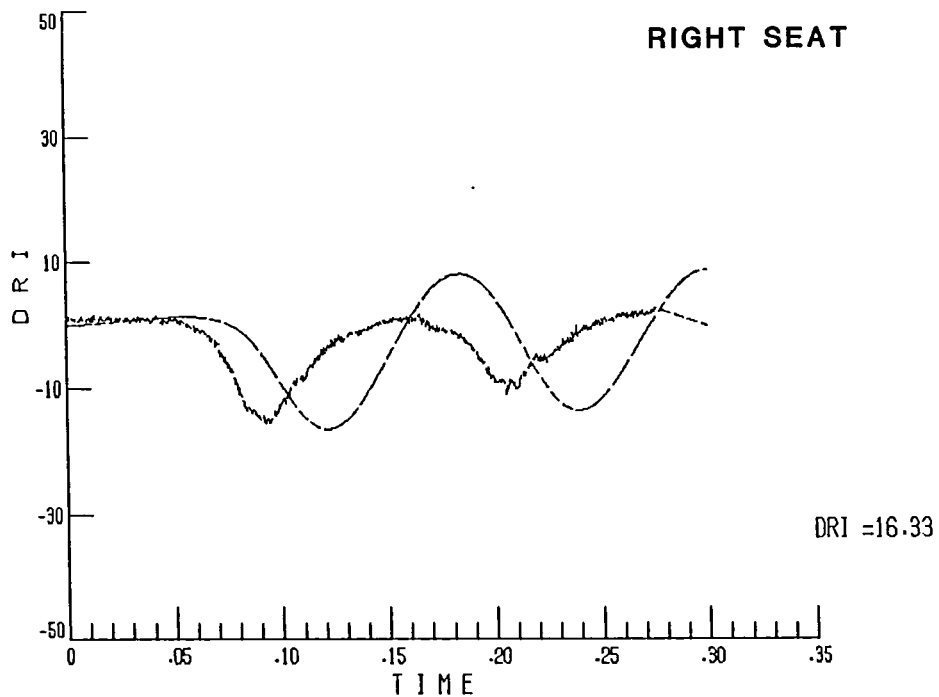
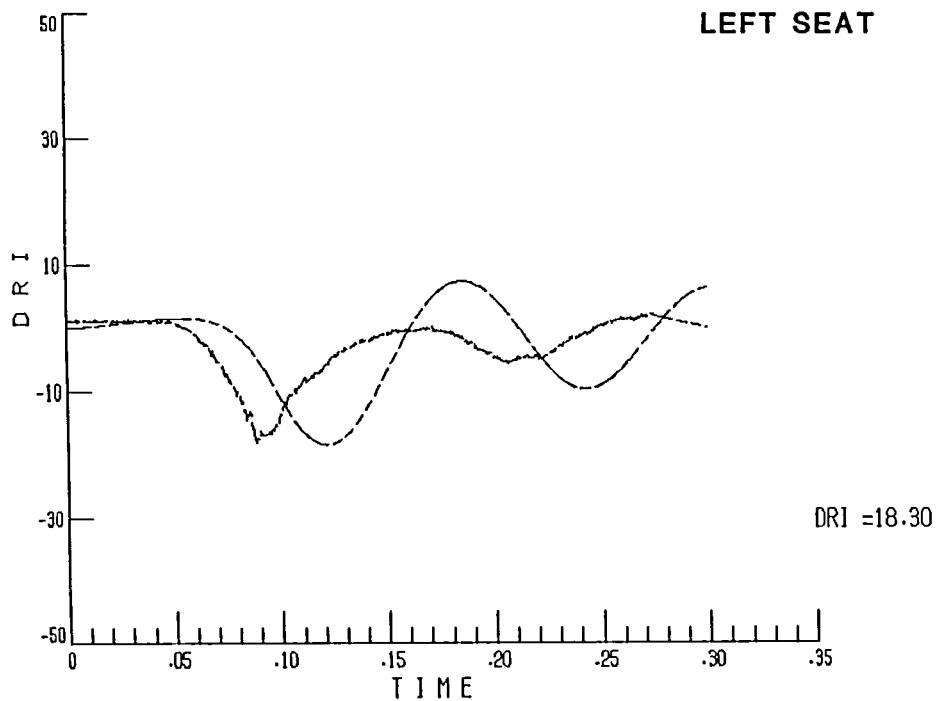


**DYNAMIC RESPONSE INDEX  
SEAT PAN Z-AXIS  
NASA TEST NO. 9**



**DYNAMIC RESPONSE INDEX  
BACK TANGENT LINE  
NASA TEST NO. 9**





**DYNAMIC RESPONSE INDEX  
DUMMY PELVIS Z-AXIS  
NASA TEST NO. 9**

11.0 NASA TEST NO. 10

HORIZONTAL VELOCITY: 0 ft/sec

VERTICAL VELOCITY: 25 ft/sec

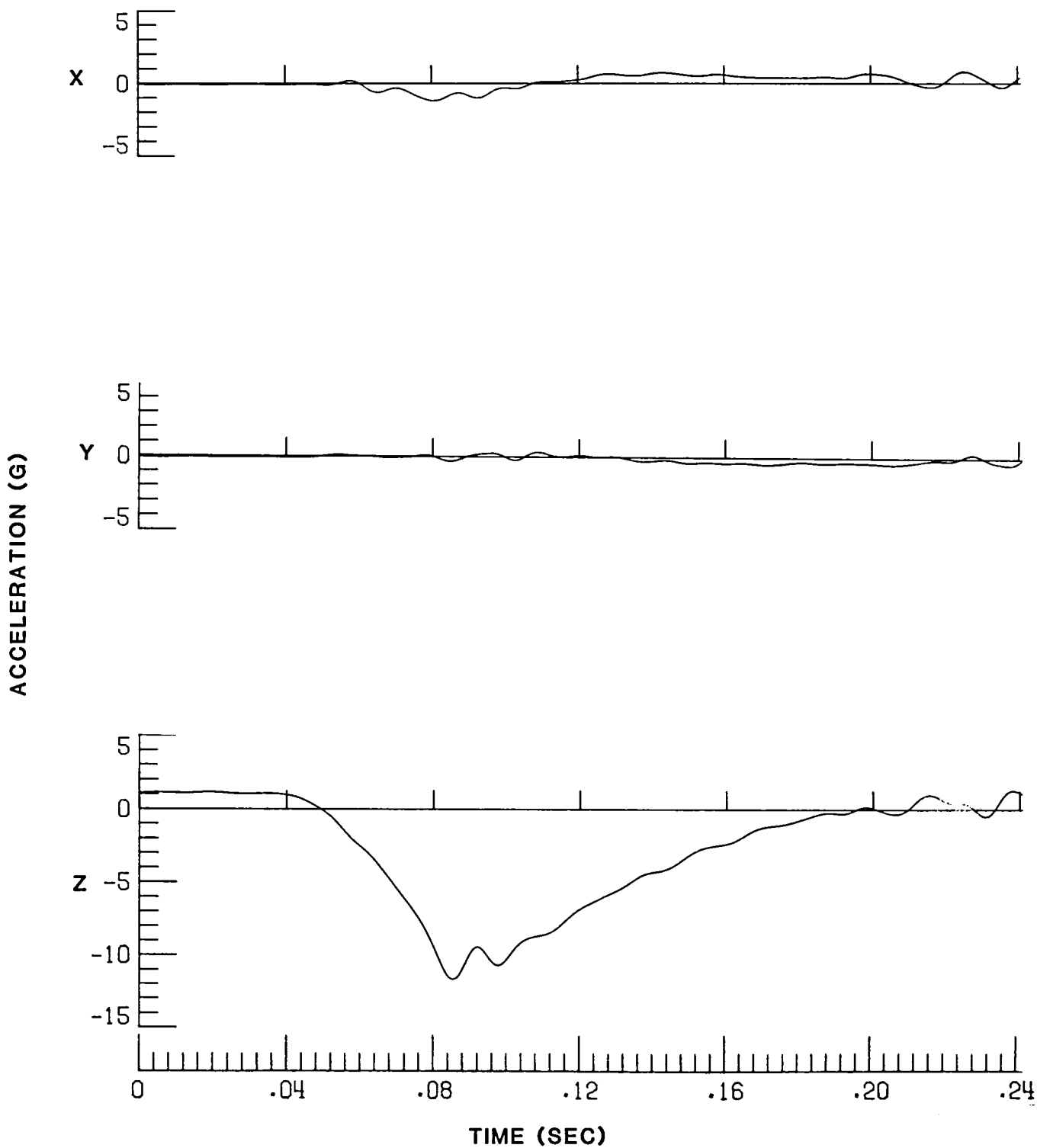
CREW MODULE PITCH:  $2^0$

CREW MODULE ROLL:  $0^0$

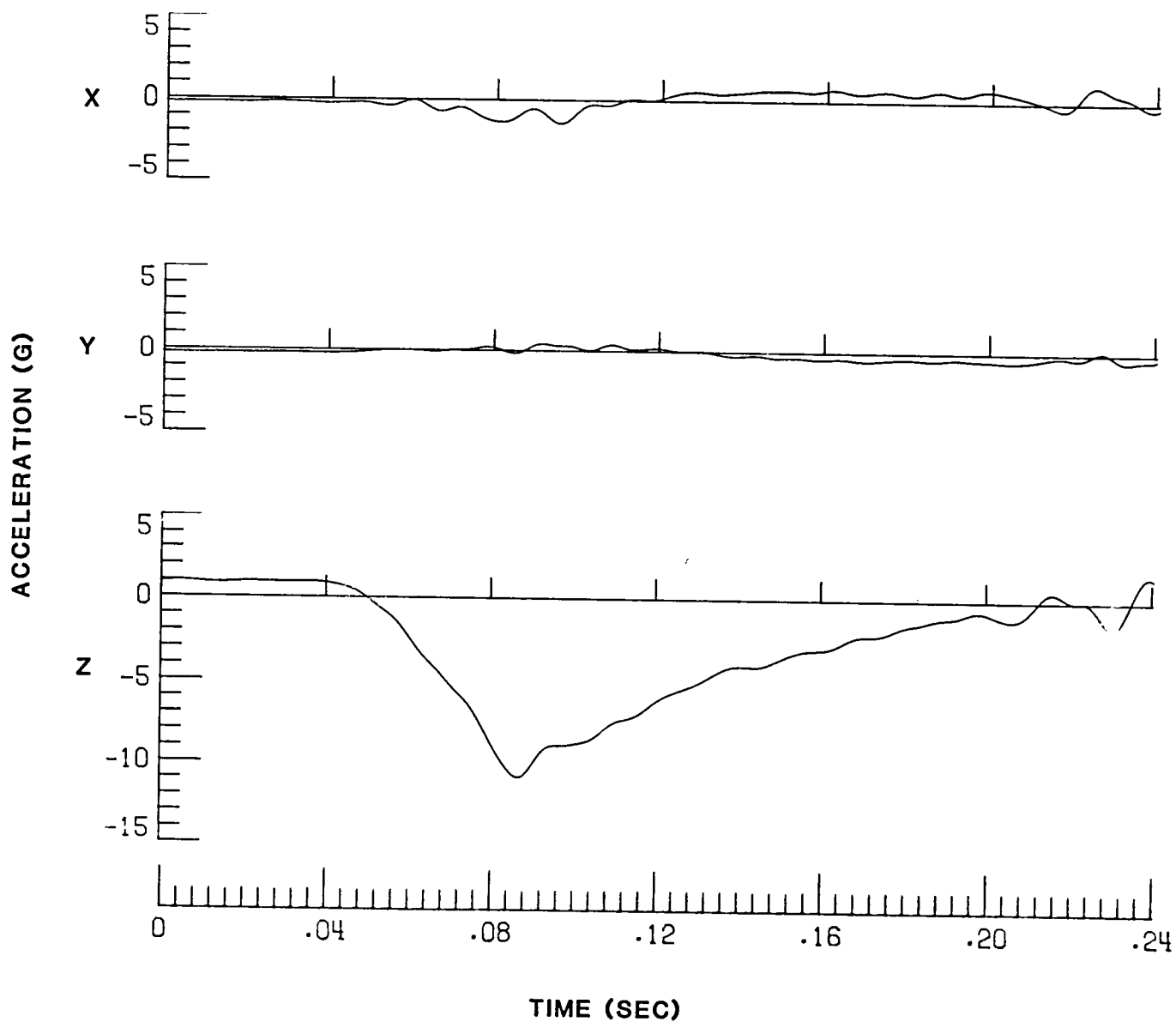
CREW MODULE YAW:  $0^0$

LEFT SEAT: F-111 Operational

RIGHT SEAT: Energy Absorbing

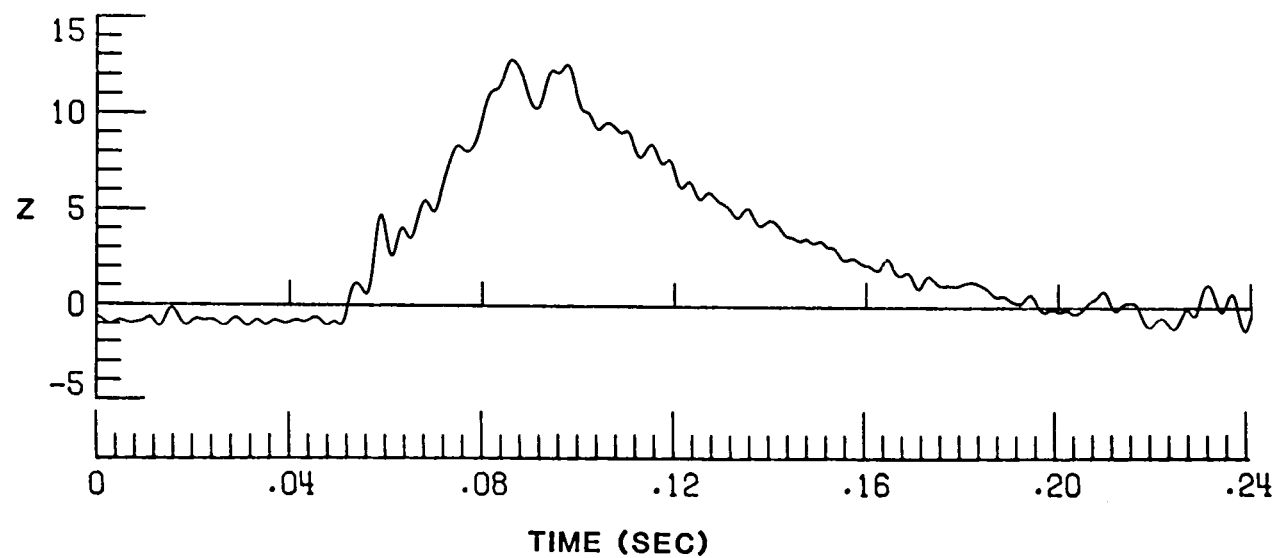
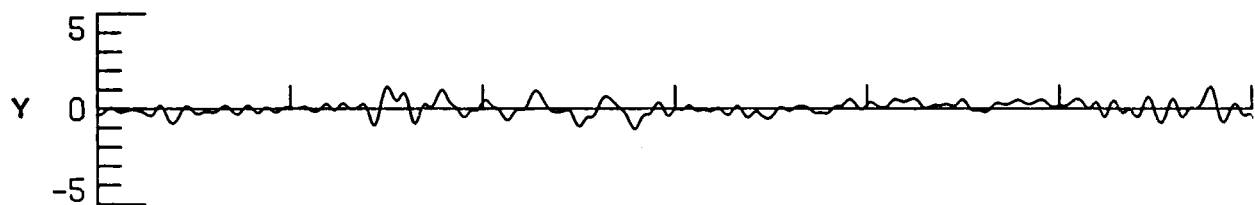
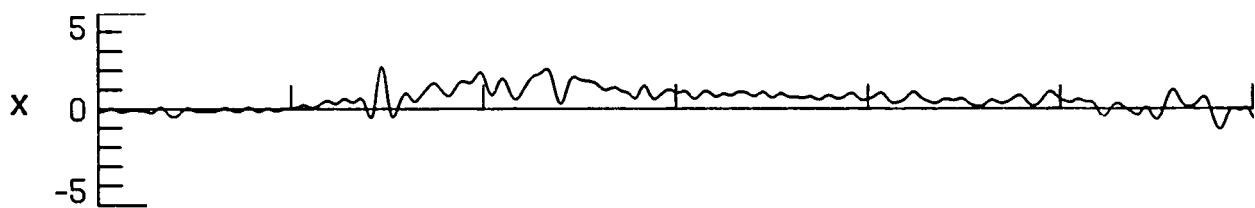


LEFT BULKHEAD  
NASA TEST NO. 10

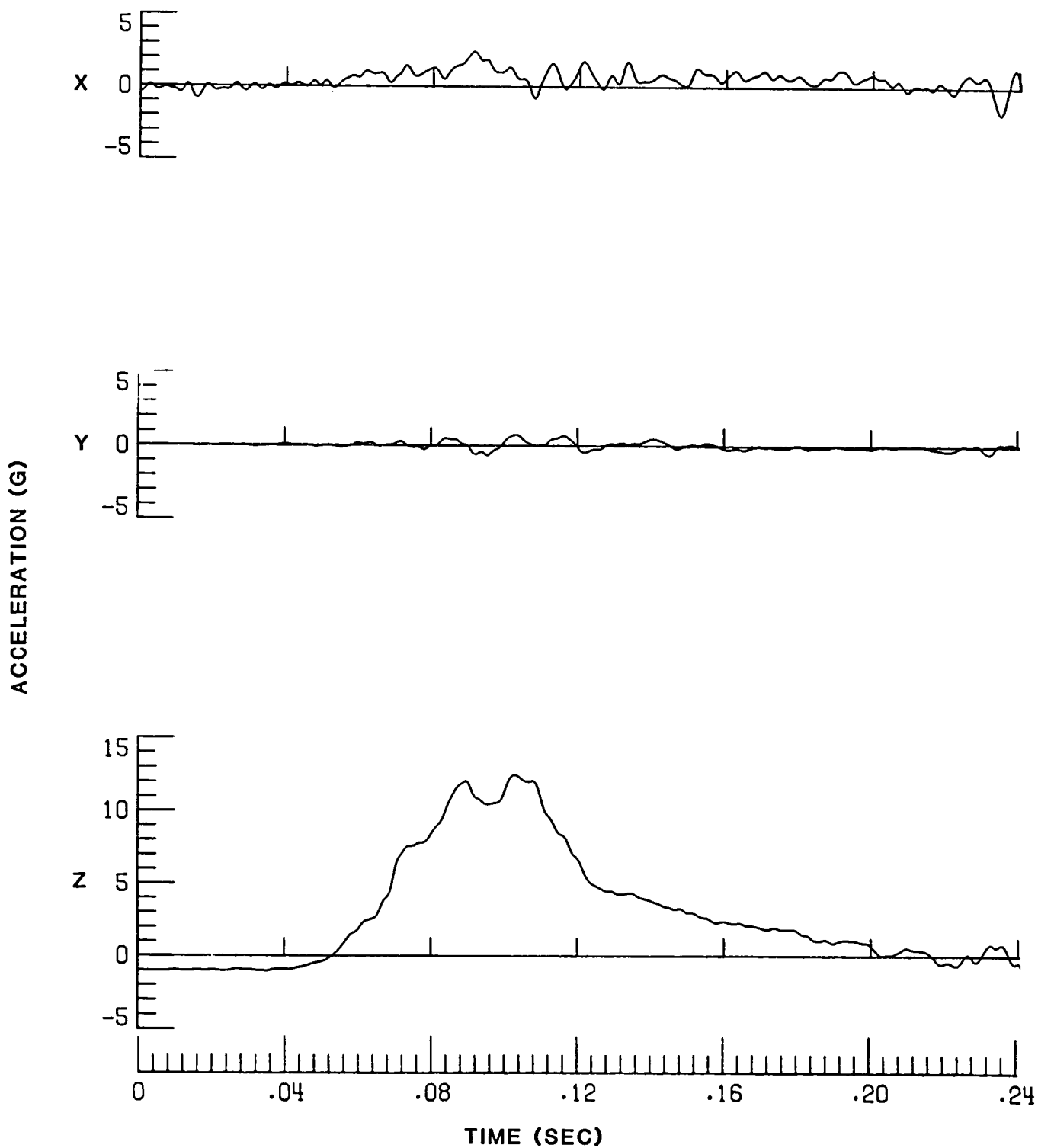


RIGHT BULKHEAD  
NASA TEST NO. 10

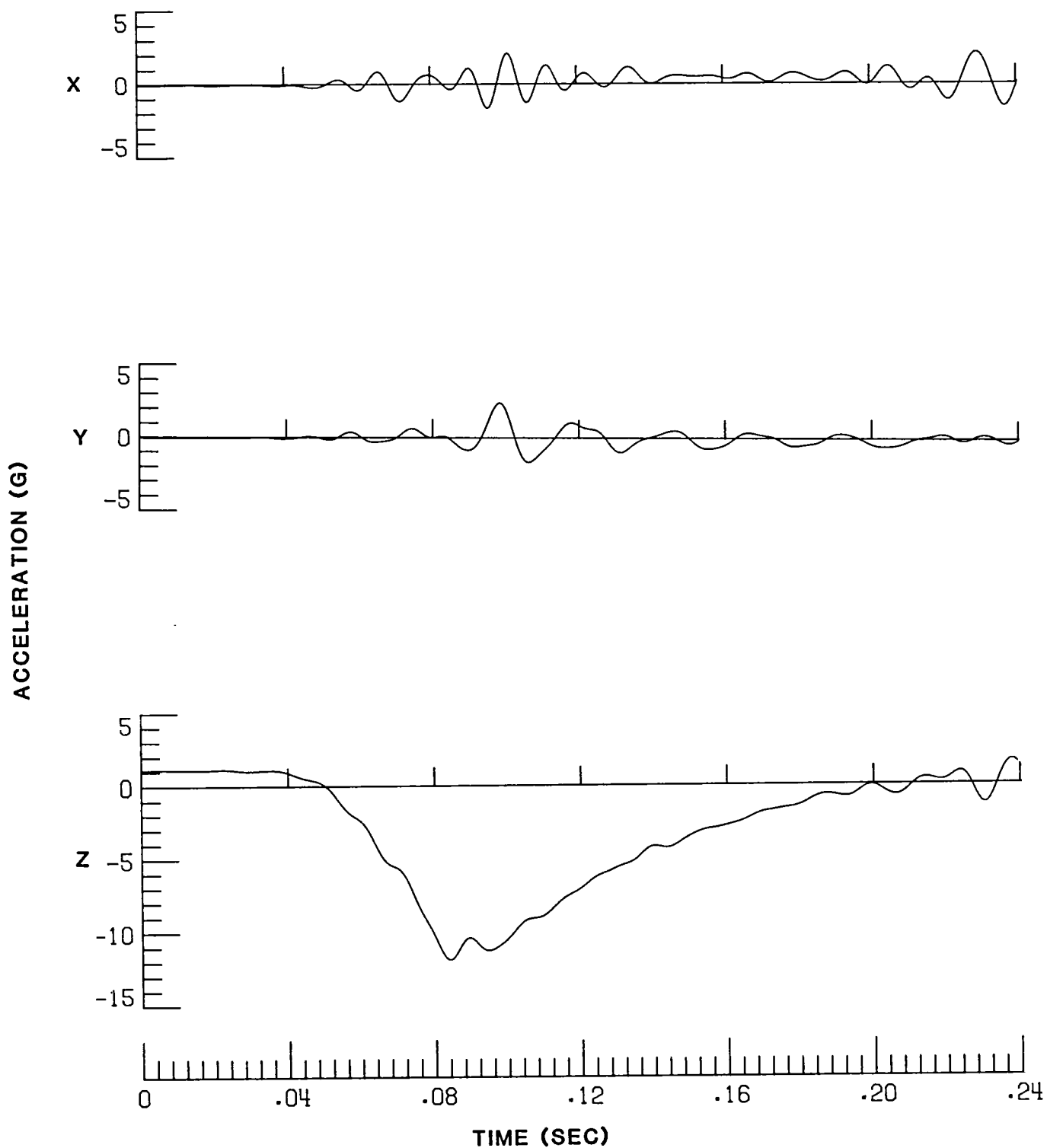
ACCELERATION (G)



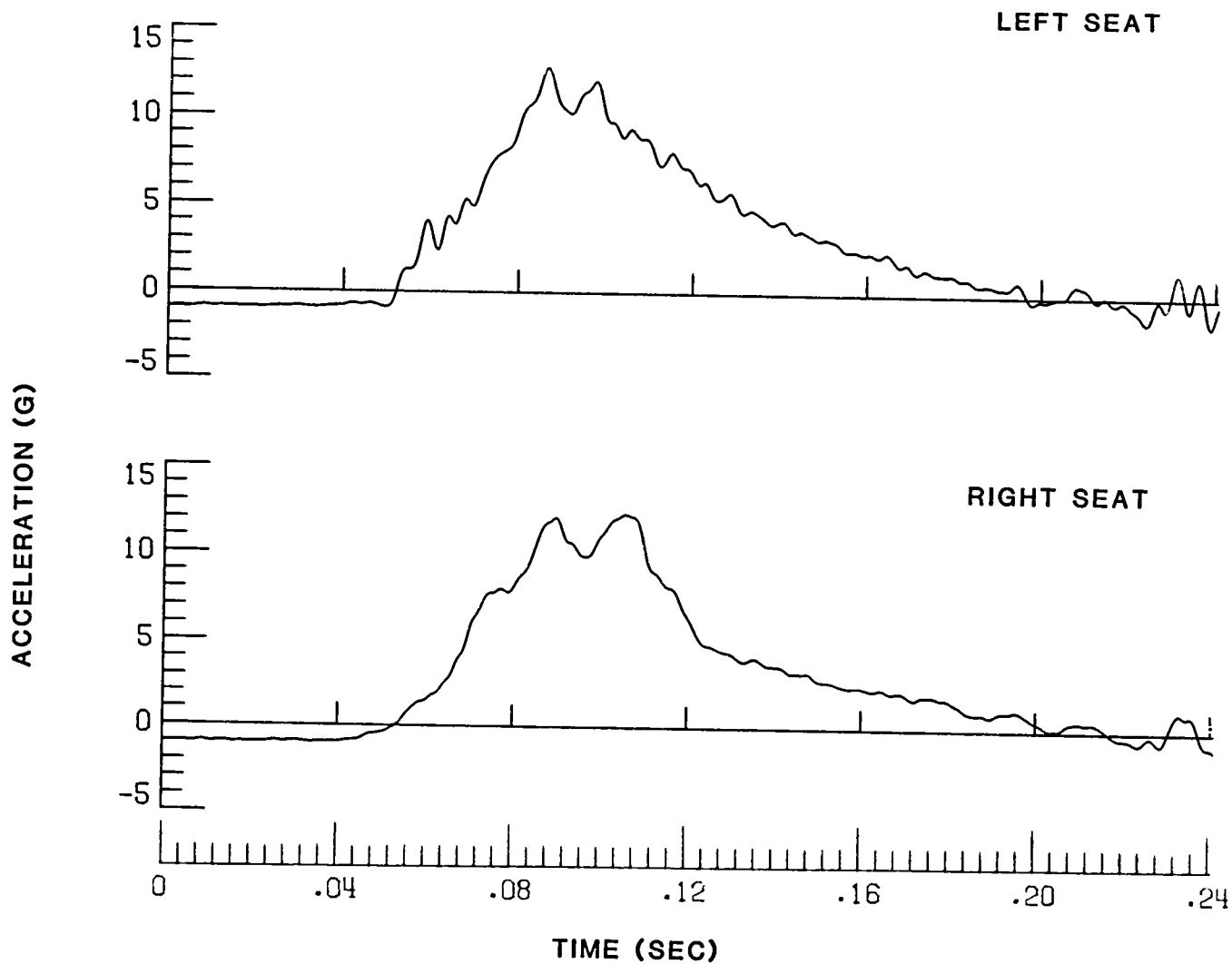
LEFT SEAT PAN  
NASA TEST NO. 10



RIGHT SEAT PAN  
NASA TEST NO. 10

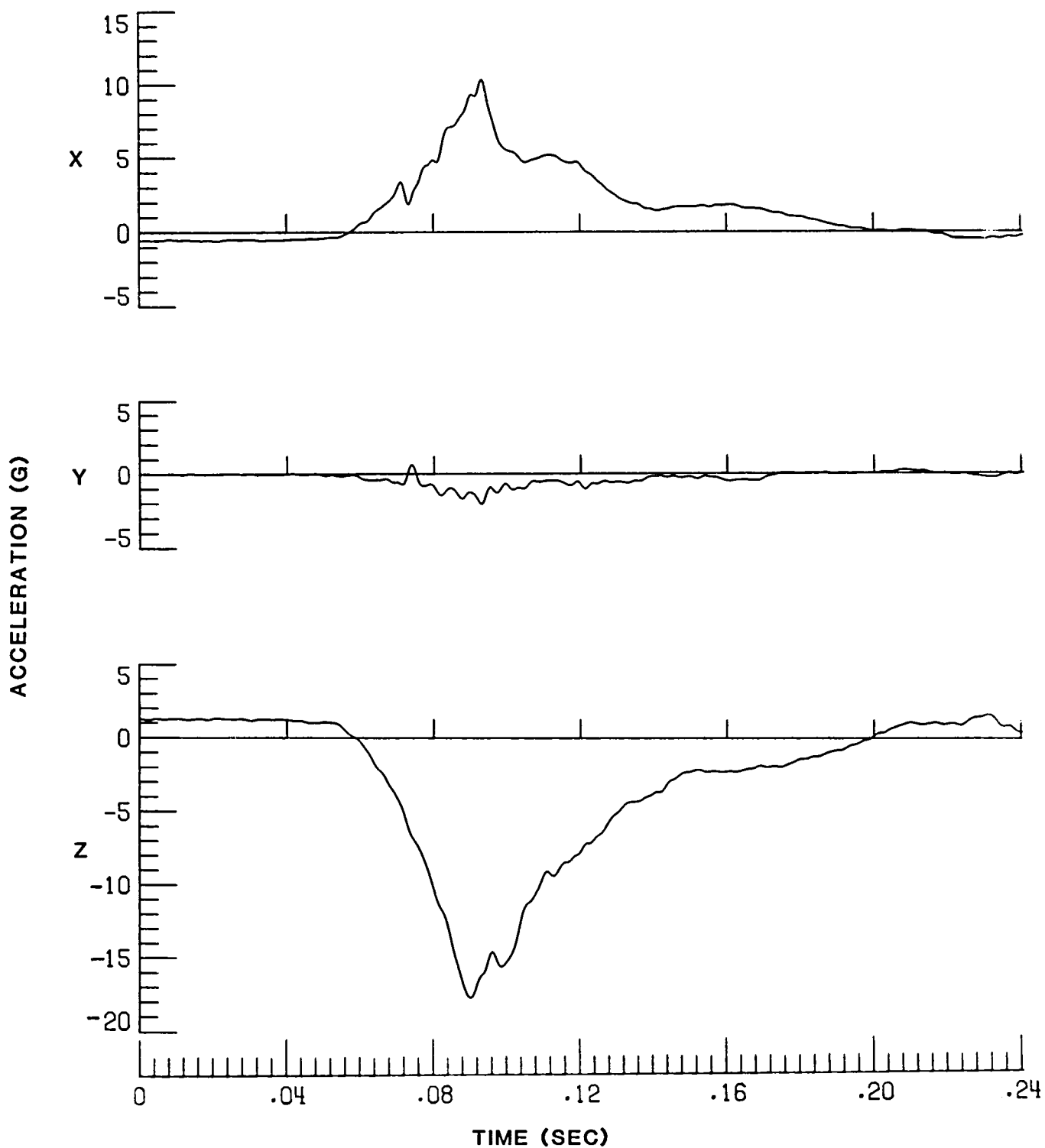


CAPSULE C.G.  
NASA TEST NO. 10

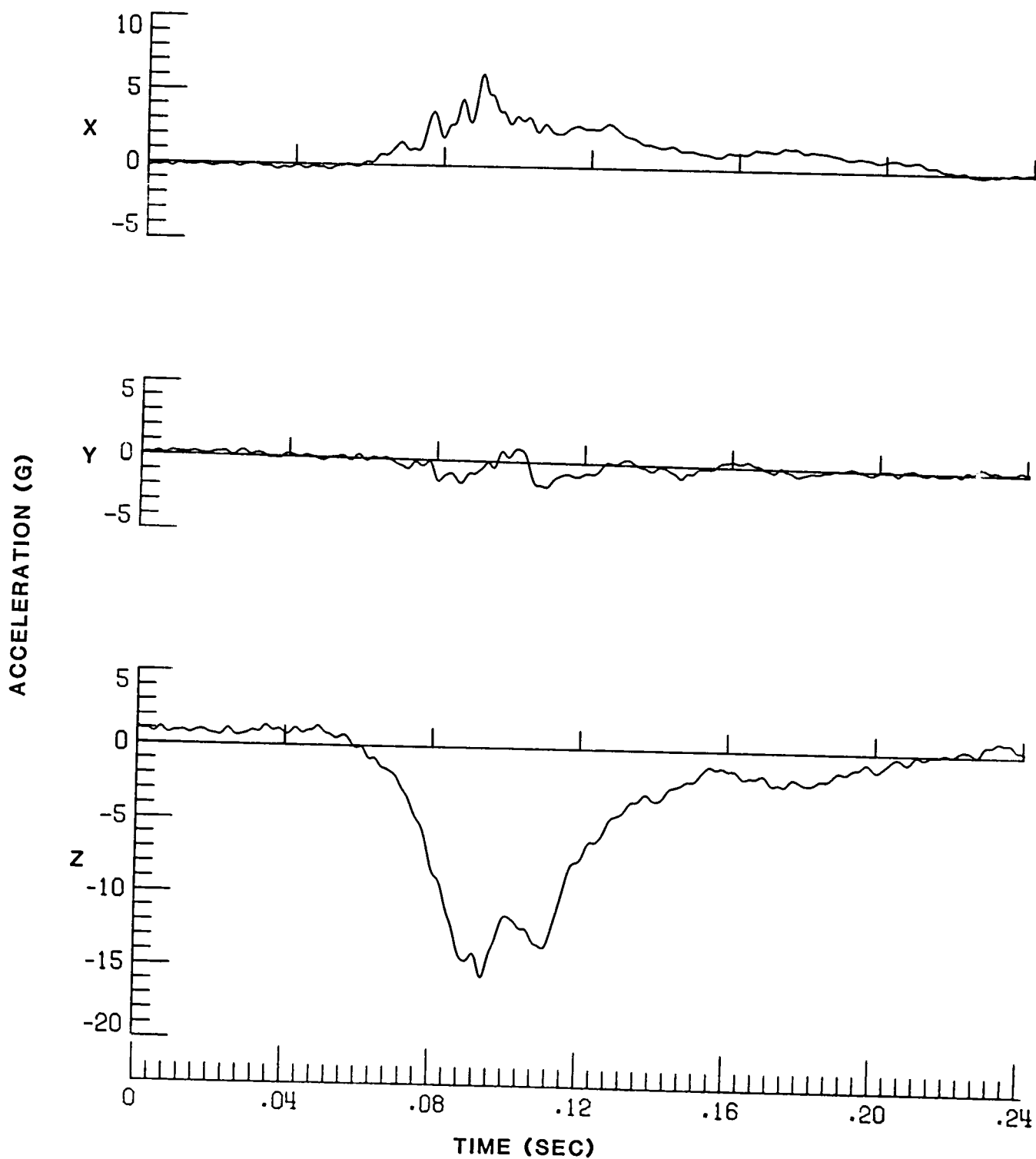


SEAT PAN ALONG BACK TANGENT LINE  
NASA TEST NO. 10





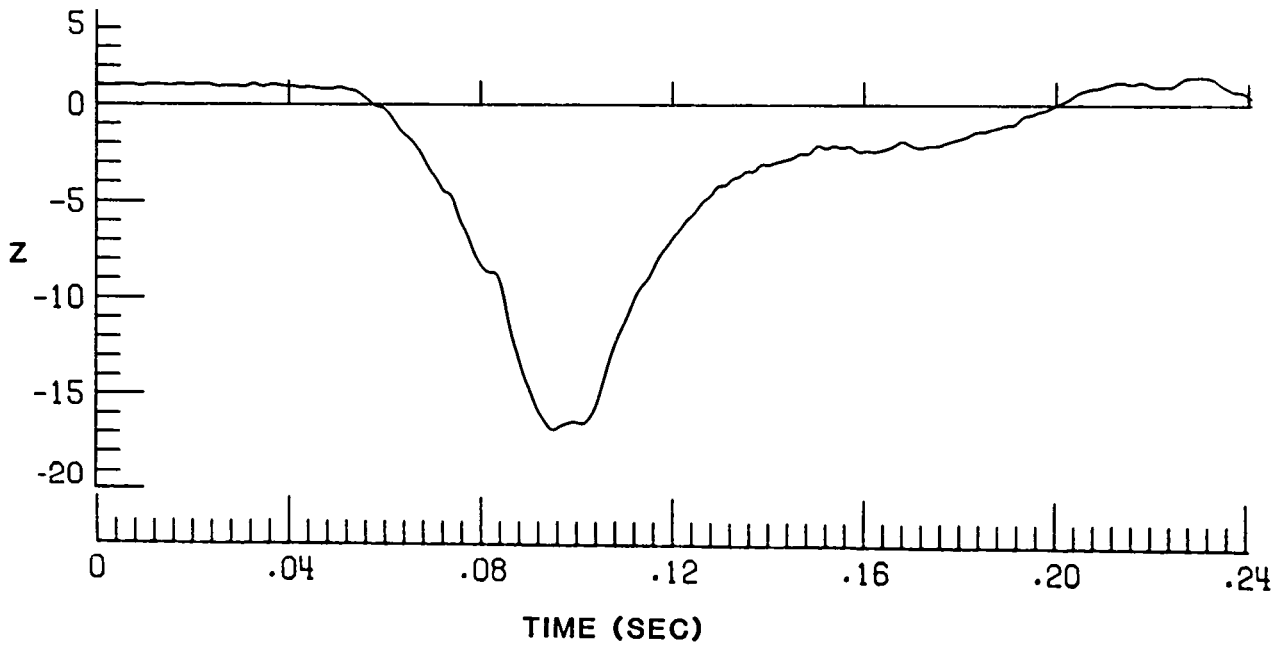
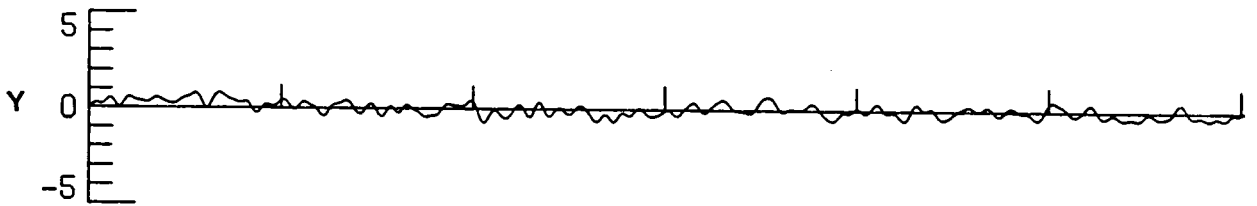
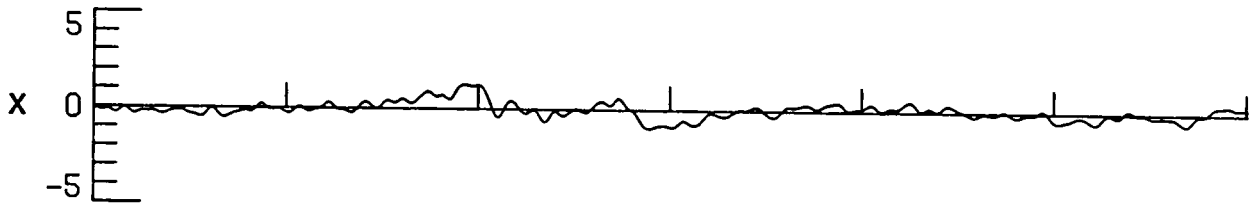
LEFT DUMMY PELVIS  
NASA TEST NO. 10



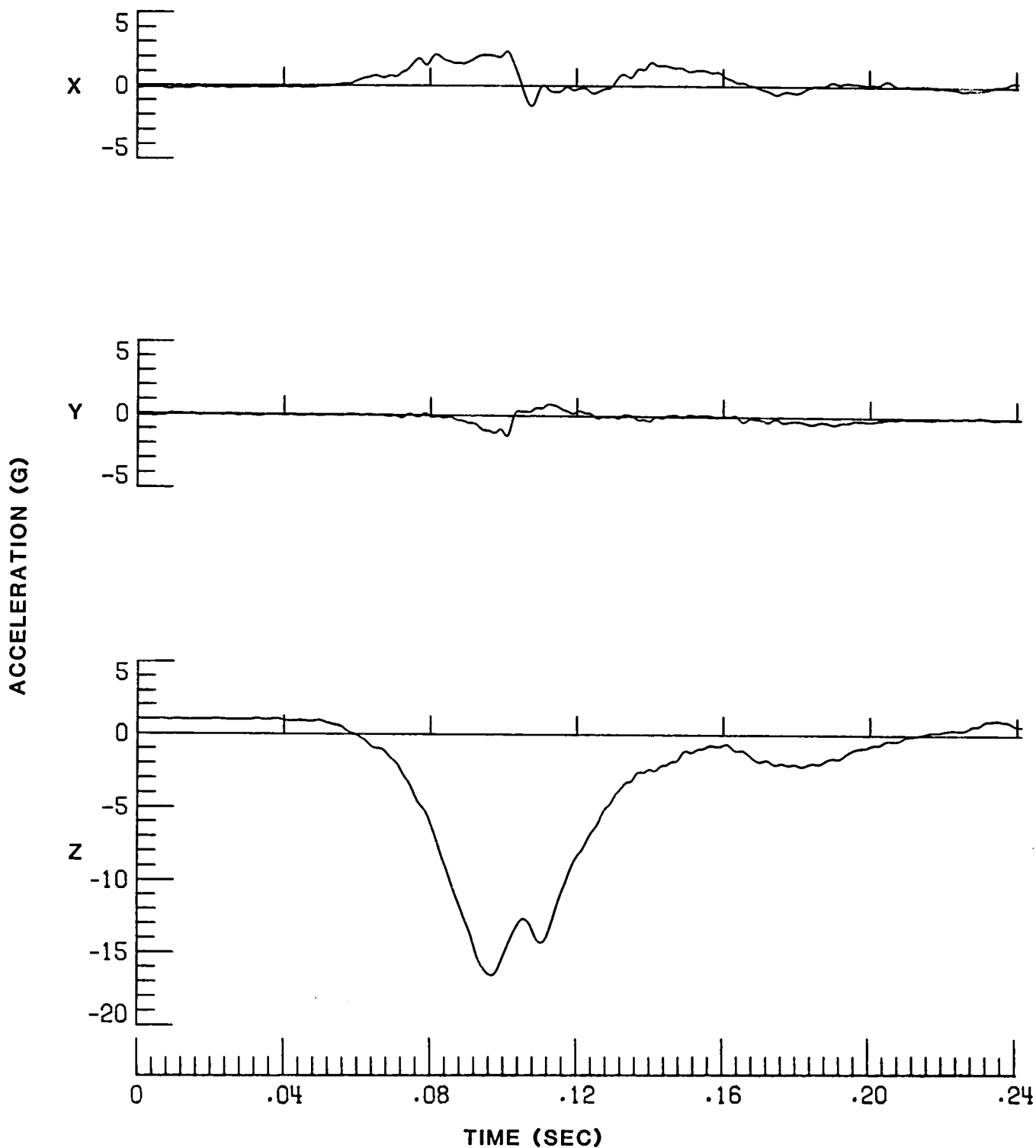
RIGHT DUMMY PELVIS

NASA TEST NO. 10

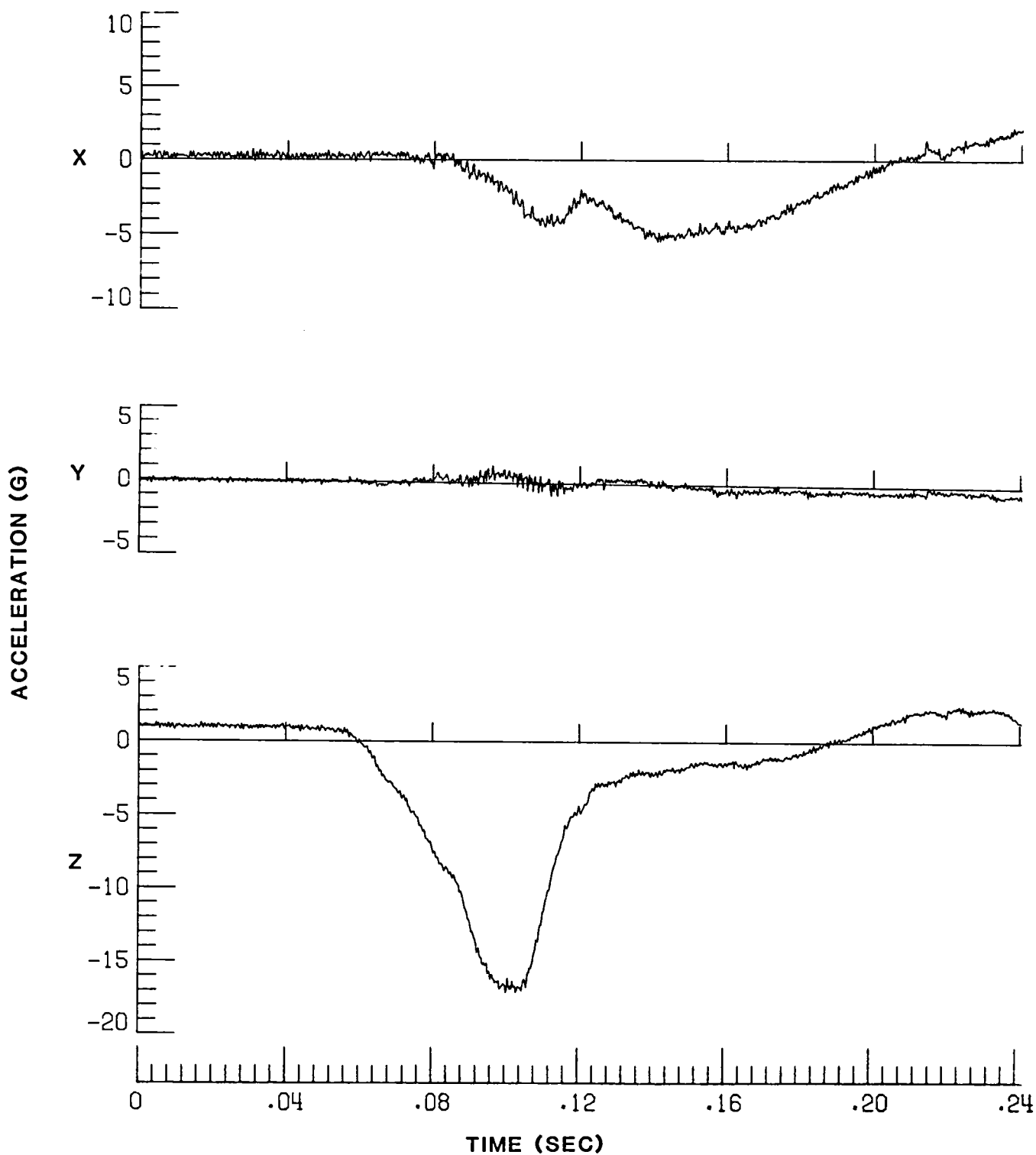
ACCELERATION (G)



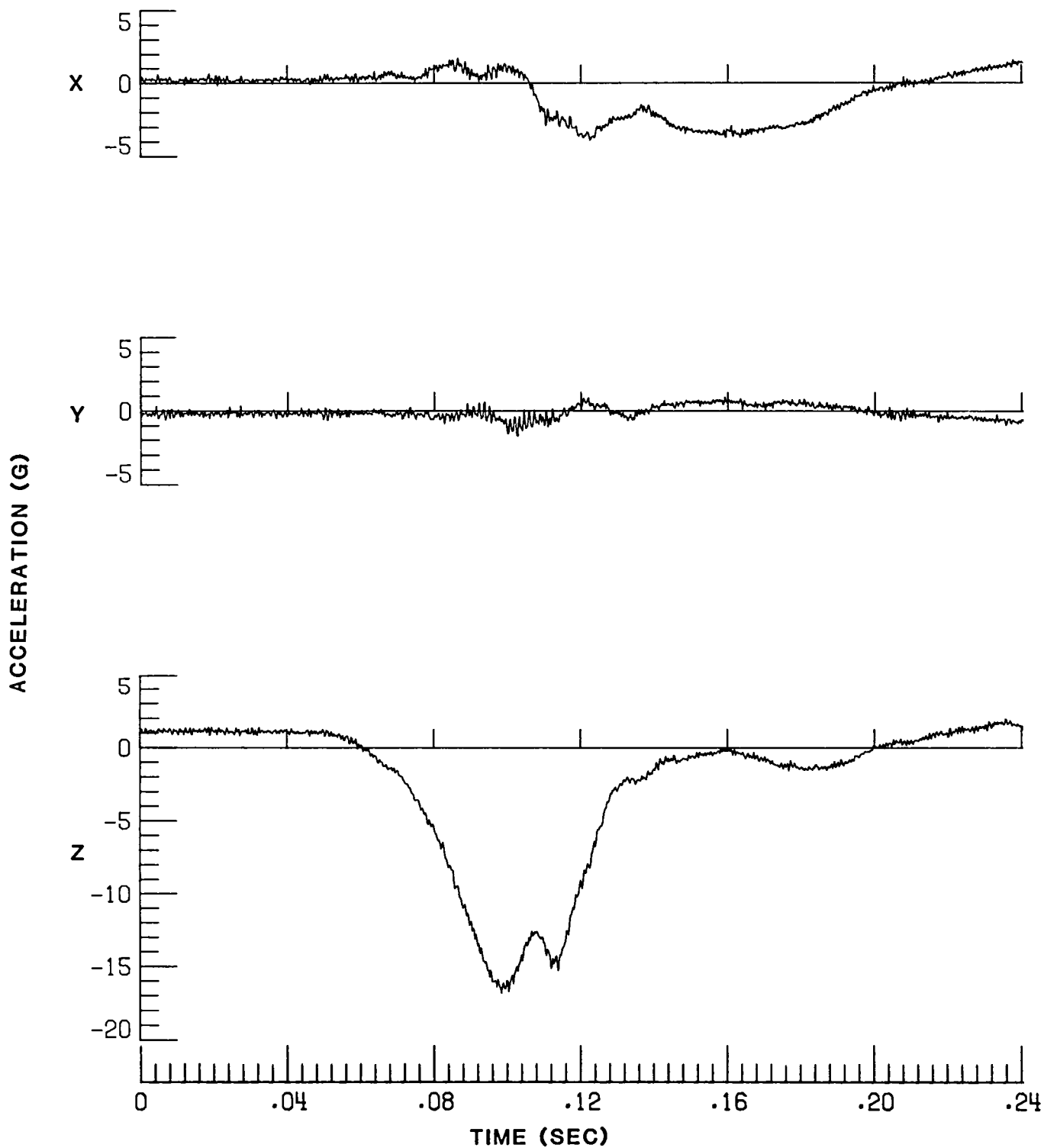
LEFT DUMMY CHEST  
NASA TEST NO. 10



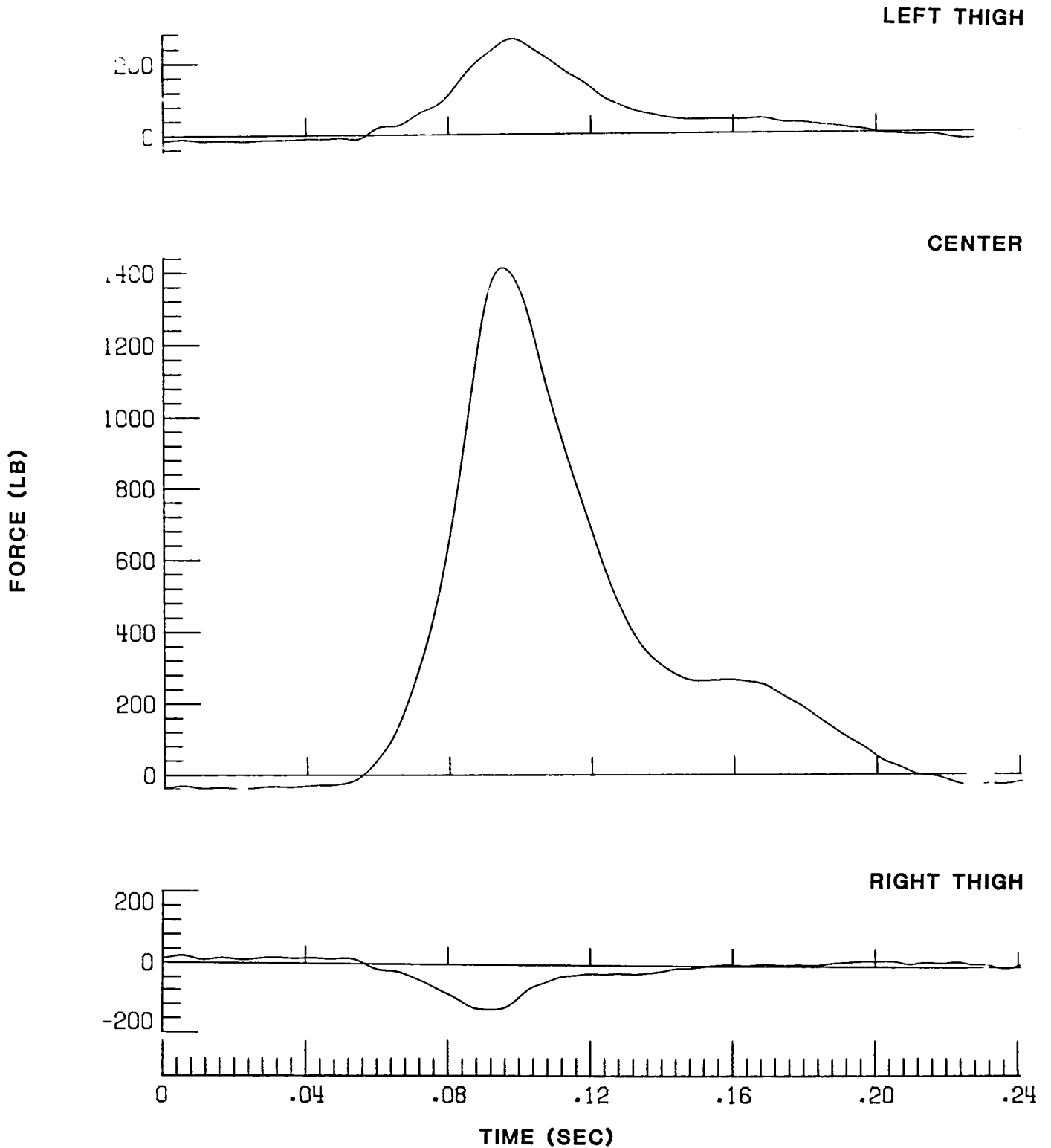
RIGHT DUMMY CHEST  
NASA TEST NO. 10



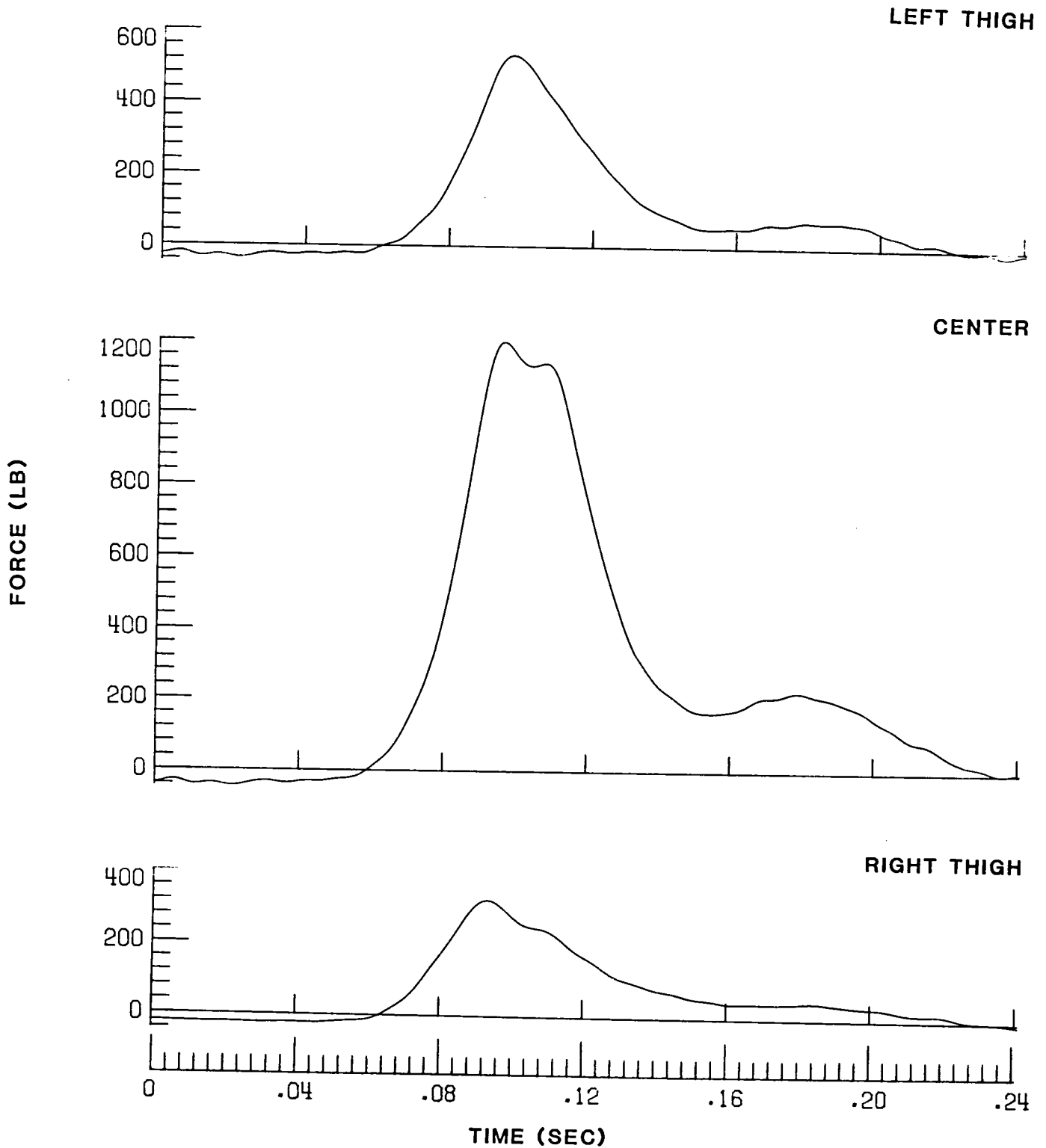
LEFT DUMMY HEAD  
NASA TEST NO. 10



RIGHT DUMMY HEAD  
NASA TEST NO. 10

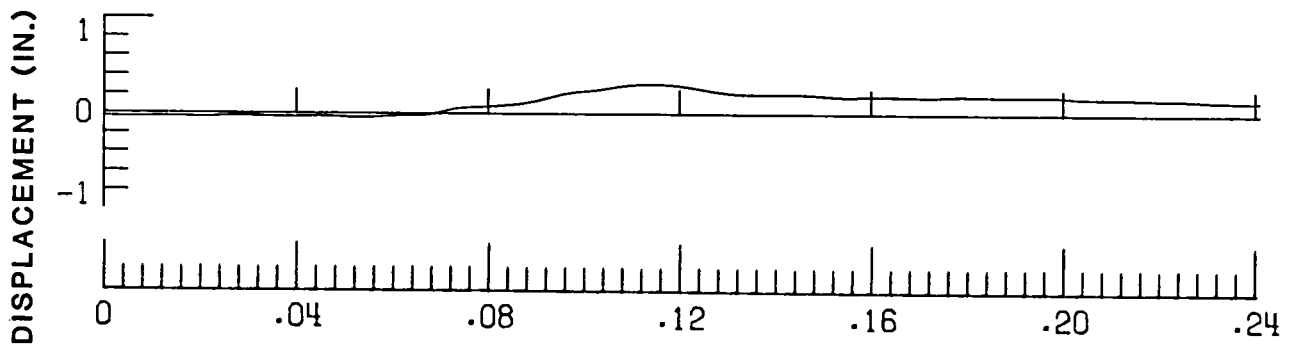


LEFT SEAT PAN LOADS  
NASA TEST NO. 10



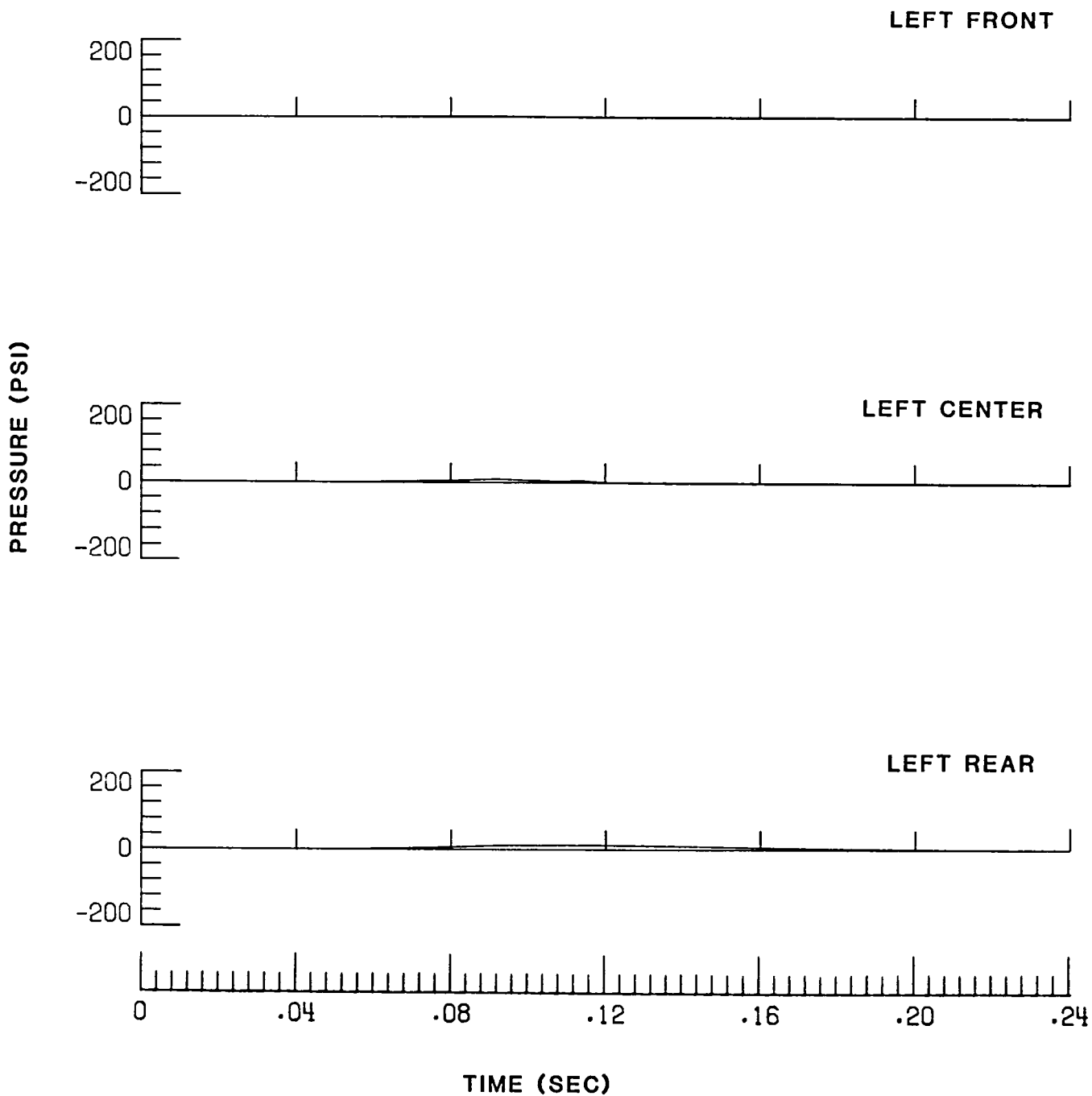
**RIGHT SEAT PAN LOADS  
NASA TEST NO. 10**



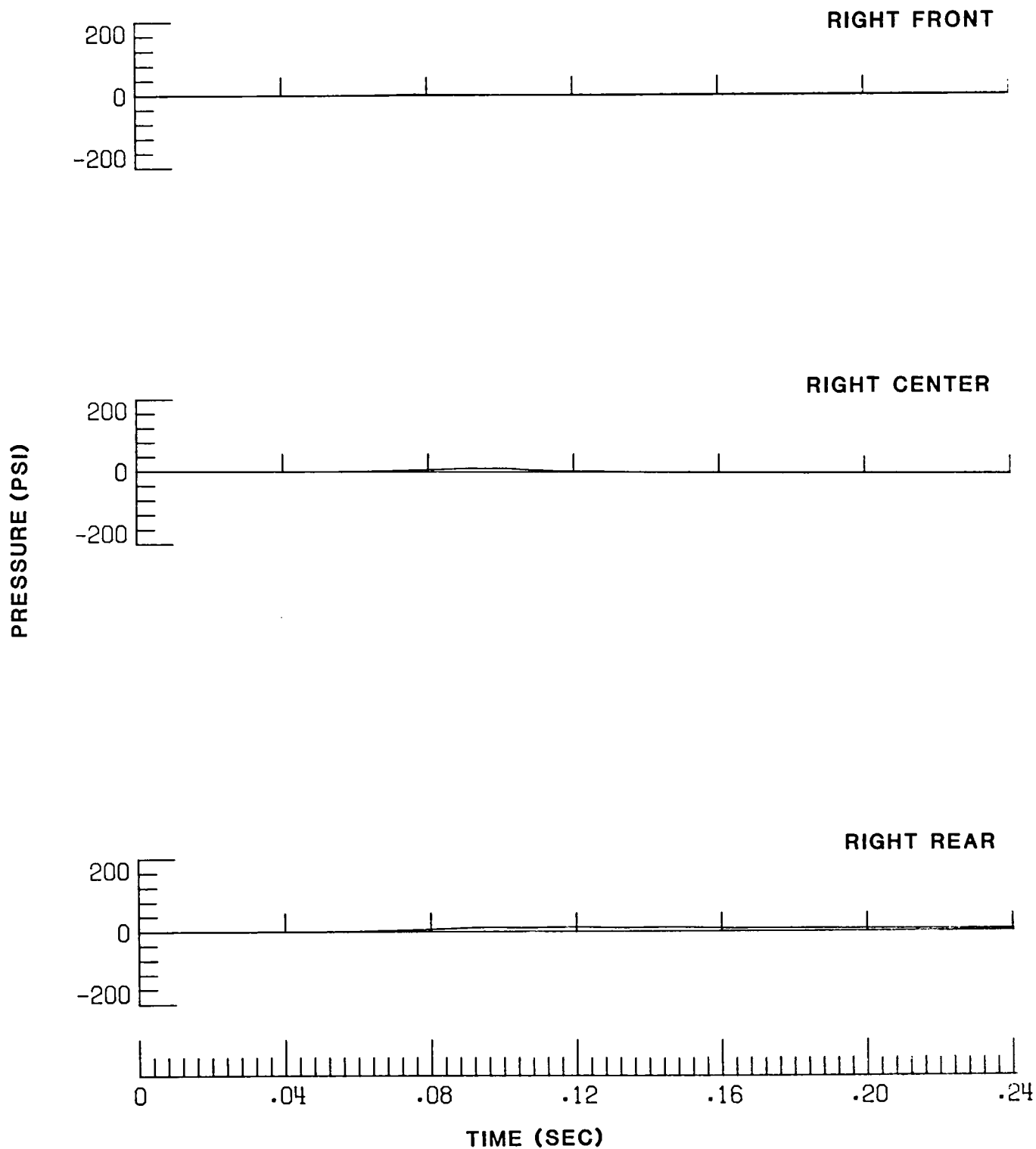


TIME (SEC)

RIGHT SEAT STROKE  
NASA TEST NO. 10

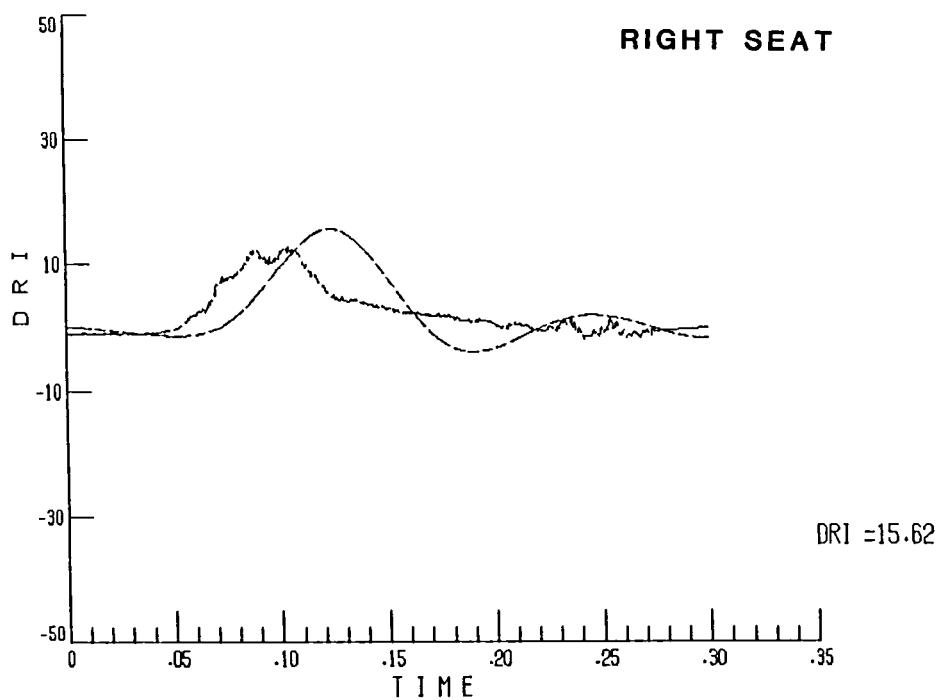
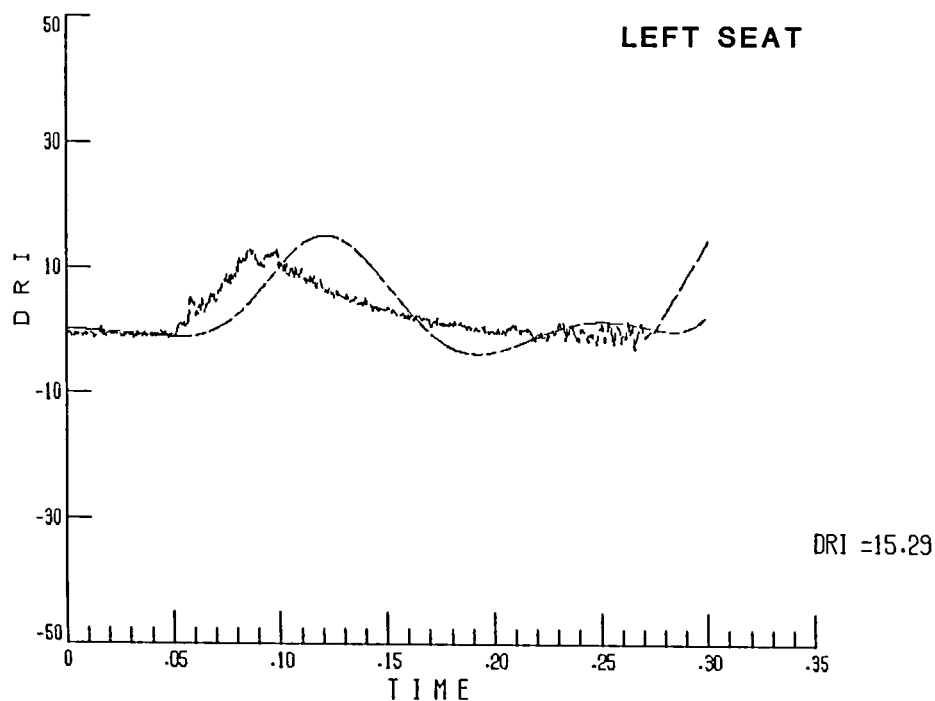


EXIT PORT PRESSURES  
NASA TEST NO. 10

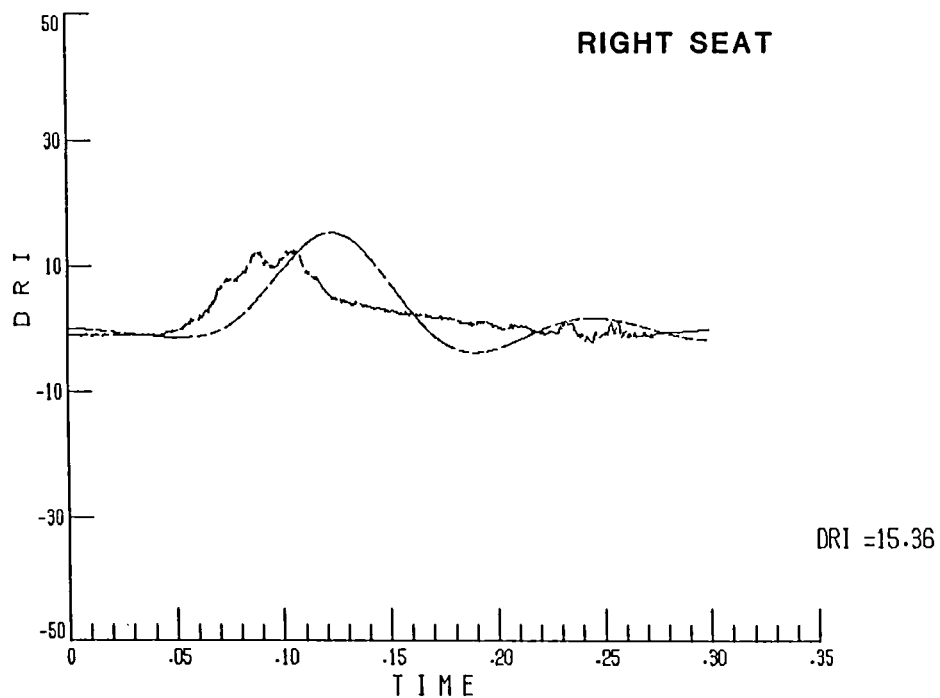
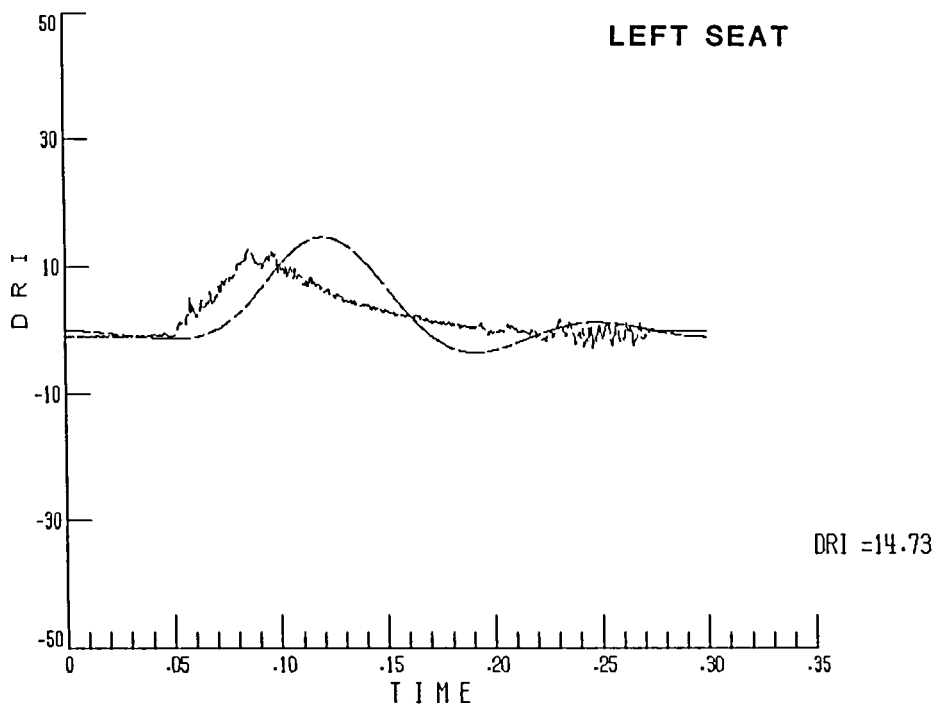


EXIT PORT PRESSURES

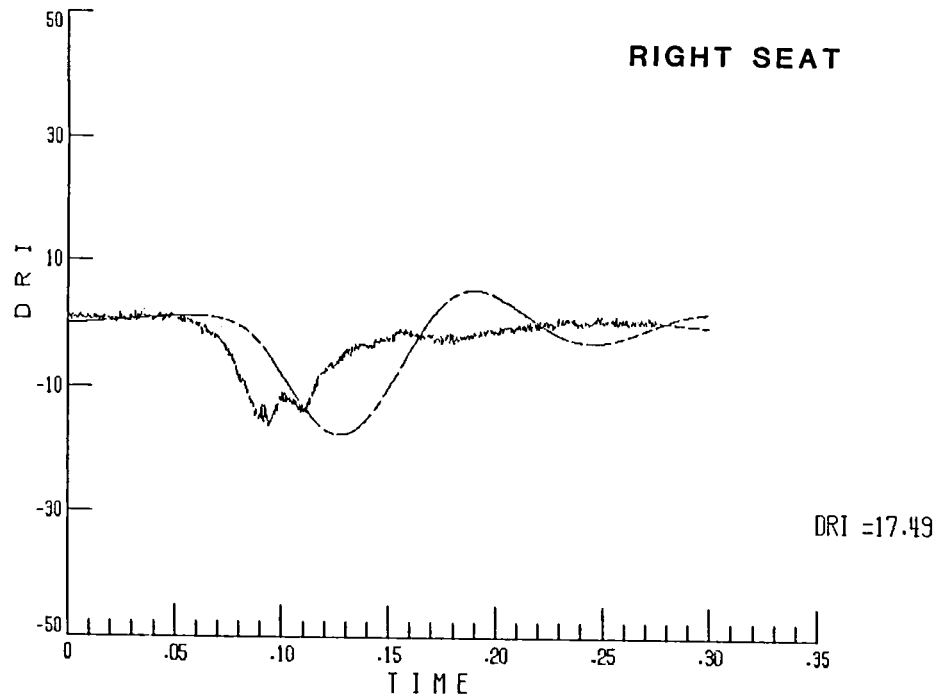
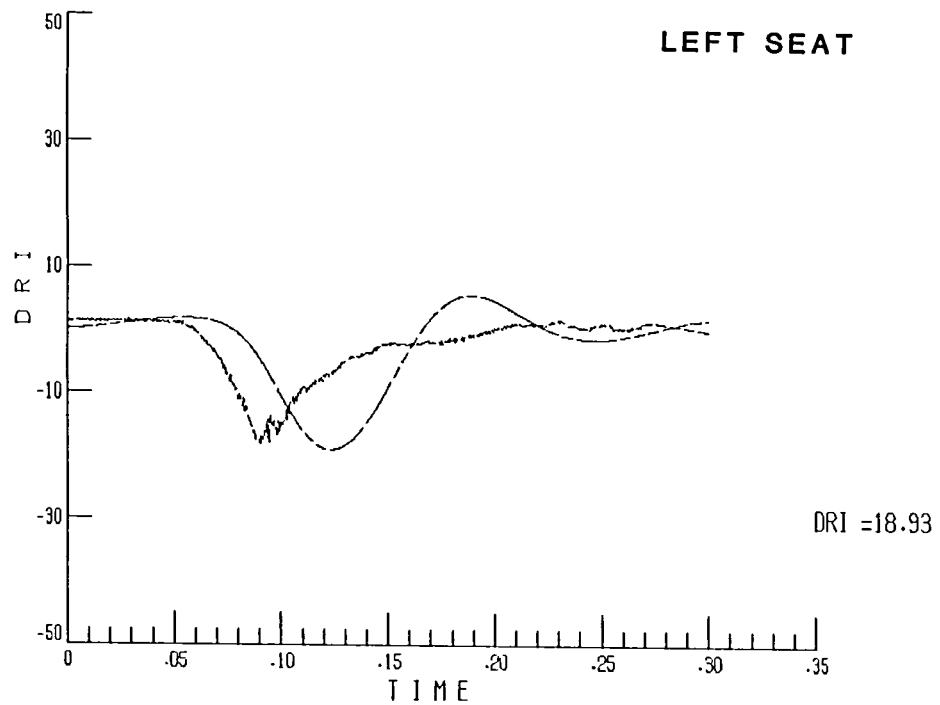
NASA TEST NO. 10



**DYNAMIC RESPONSE INDEX  
SEAT PAN Z-AXIS  
NASA TEST NO. 10**



**DYNAMIC RESPONSE INDEX  
BACK TANGENT LINE  
NASA TEST NO. 10**



**DYNAMIC RESPONSE INDEX  
DUMMY PELVIS Z-AXIS  
NASA TEST NO. 10**

## 12.0 NASA TEST NO. 11

HORIZONTAL VELOCITY: 0 ft/sec

VERTICAL VELOCITY: 25 ft/sec

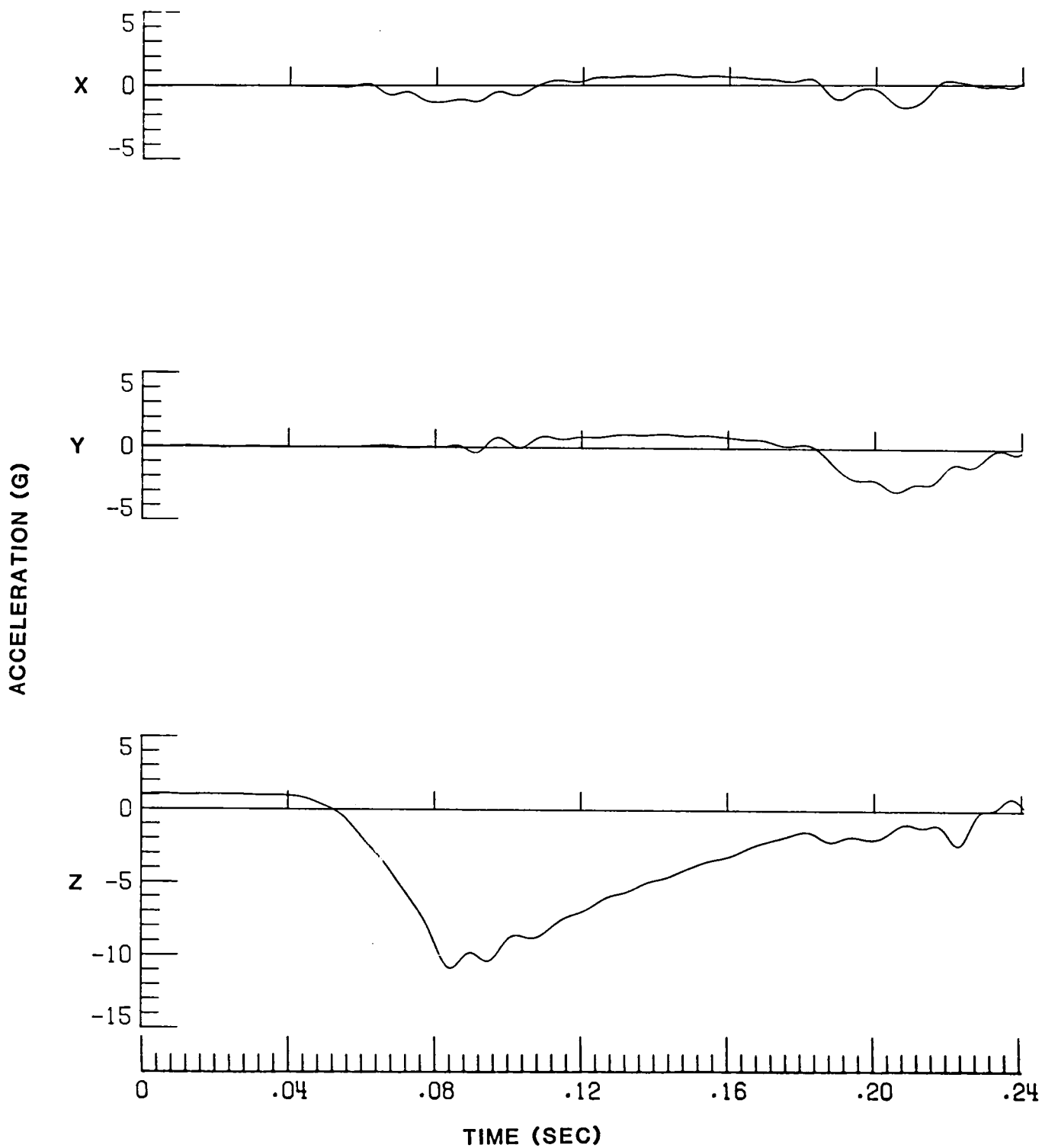
CREW MODULE PITCH:  $2^0$

CREW MODULE ROLL:  $0^0$

CREW MODULE YAW:  $0^0$

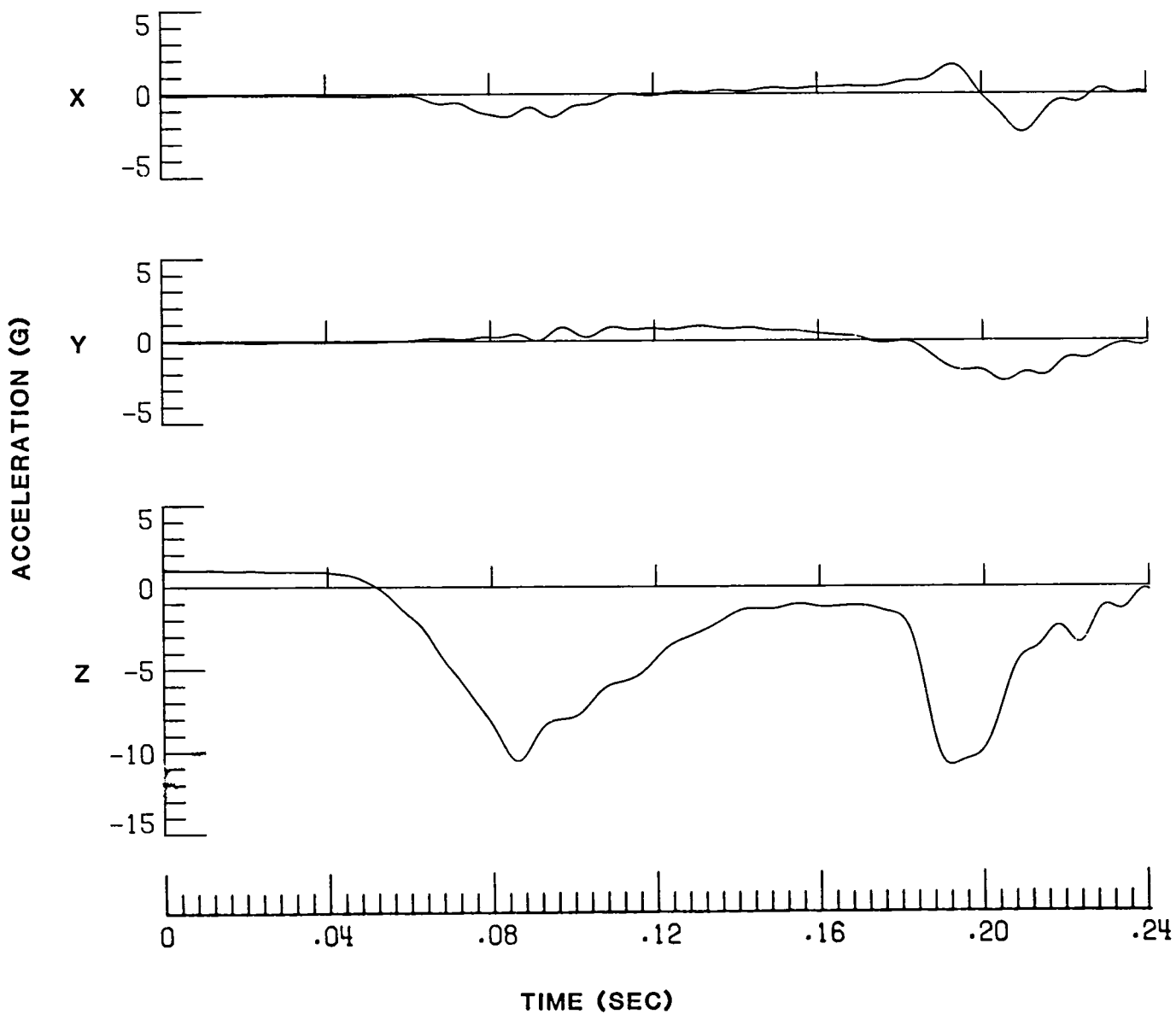
LEFT SEAT: F-111 Operational

RIGHT SEAT: Energy Absorbing

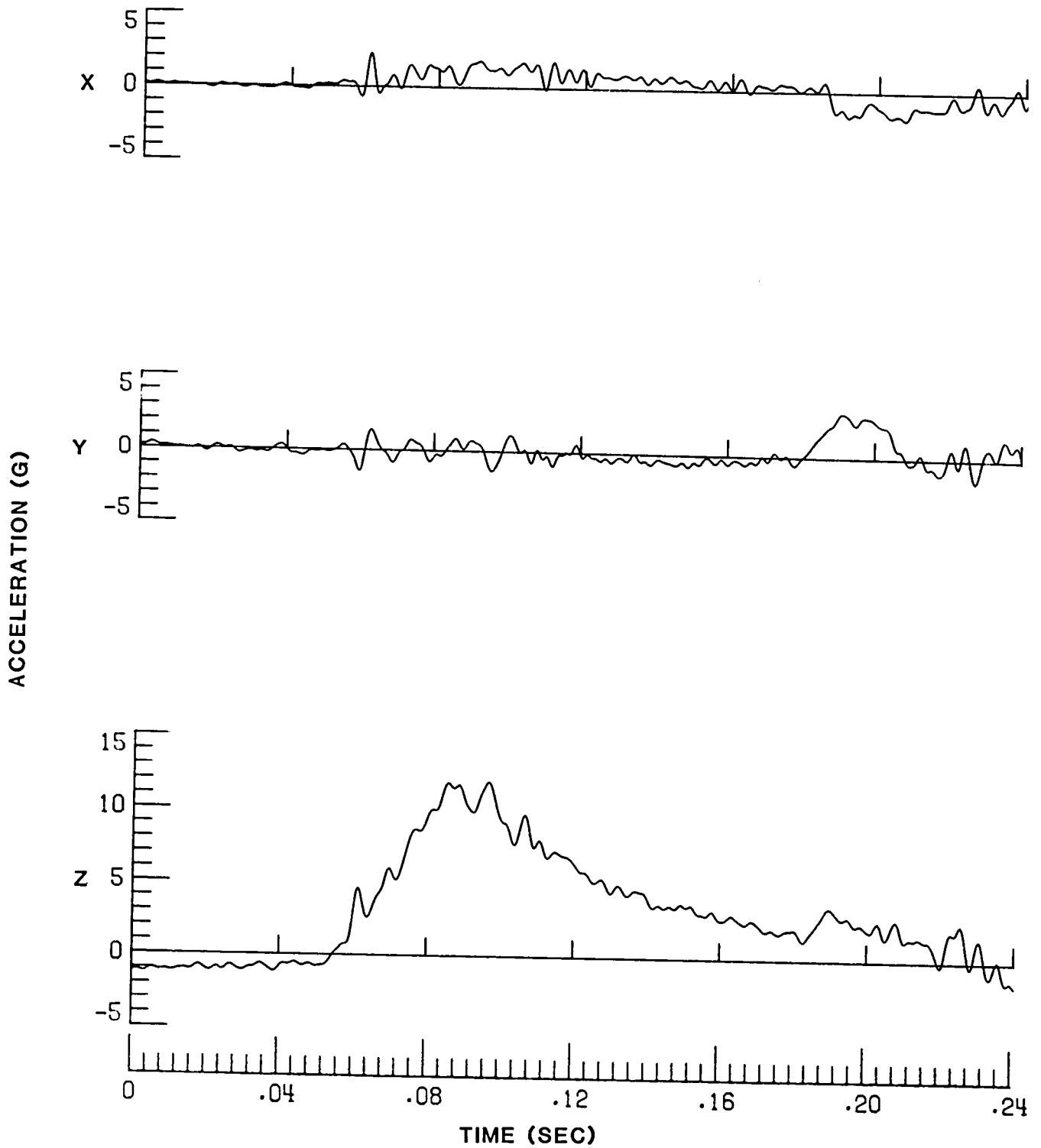


LEFT BULKHEAD  
NASA TEST NO. 11

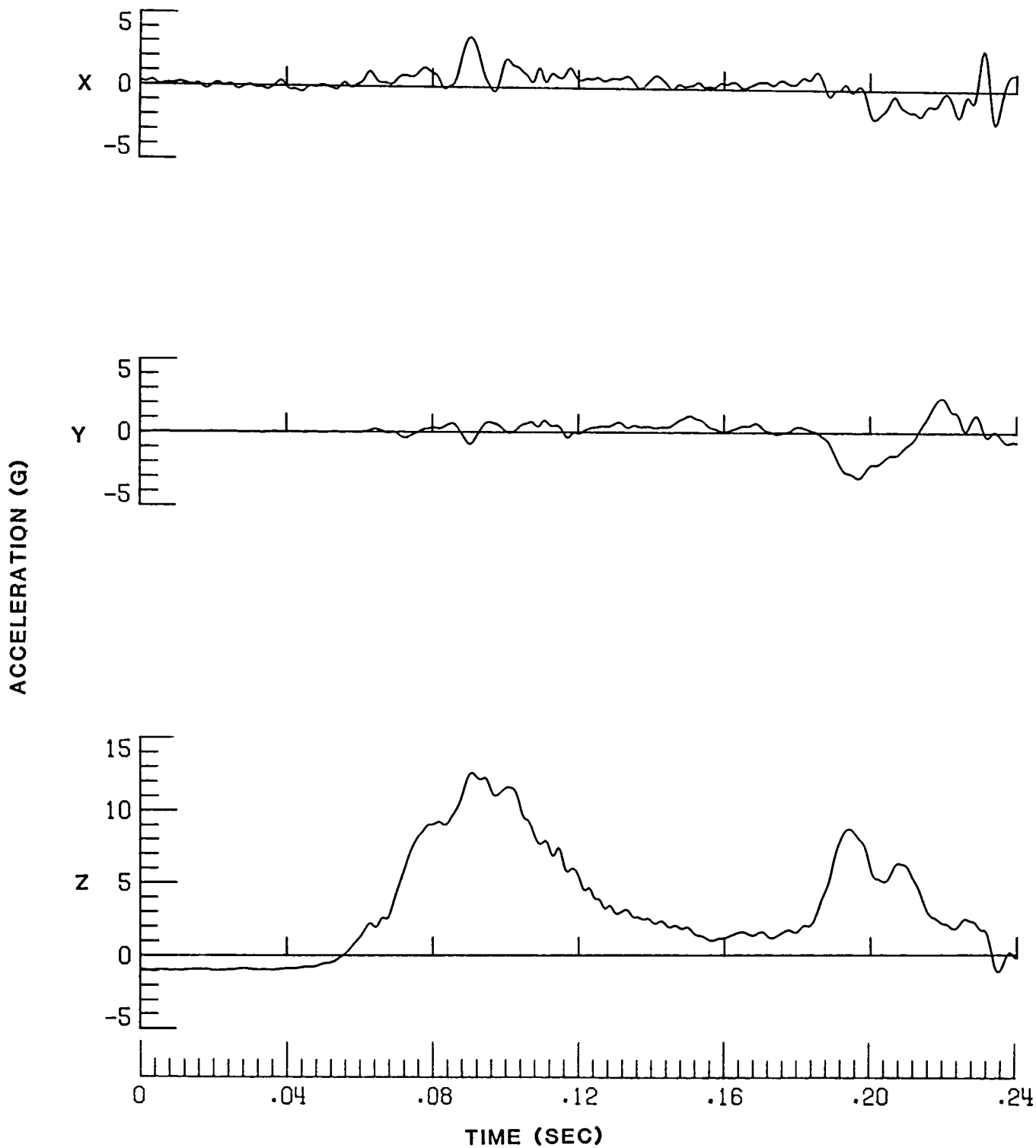




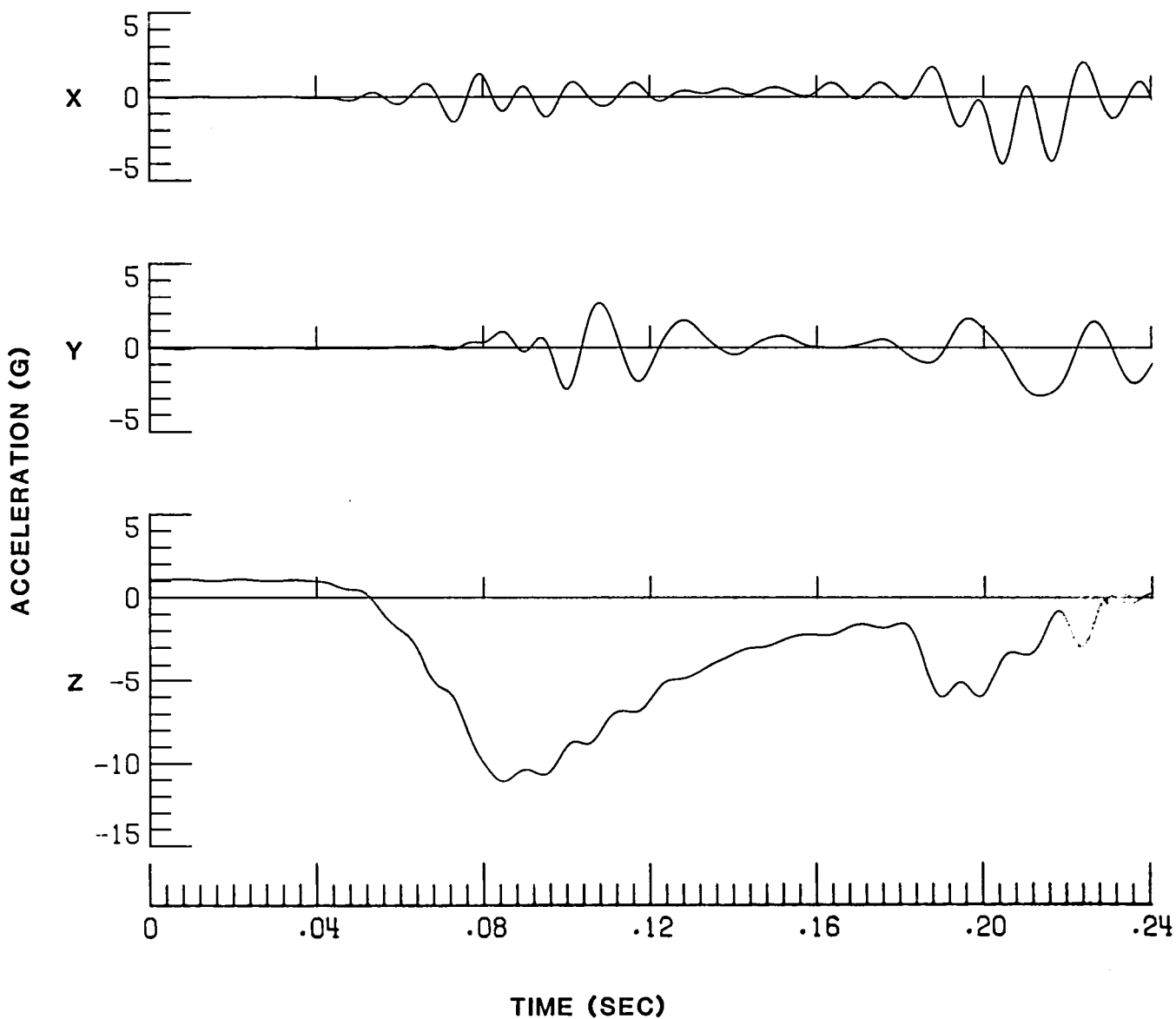
RIGHT BULKHEAD  
NASA TEST NO. 11



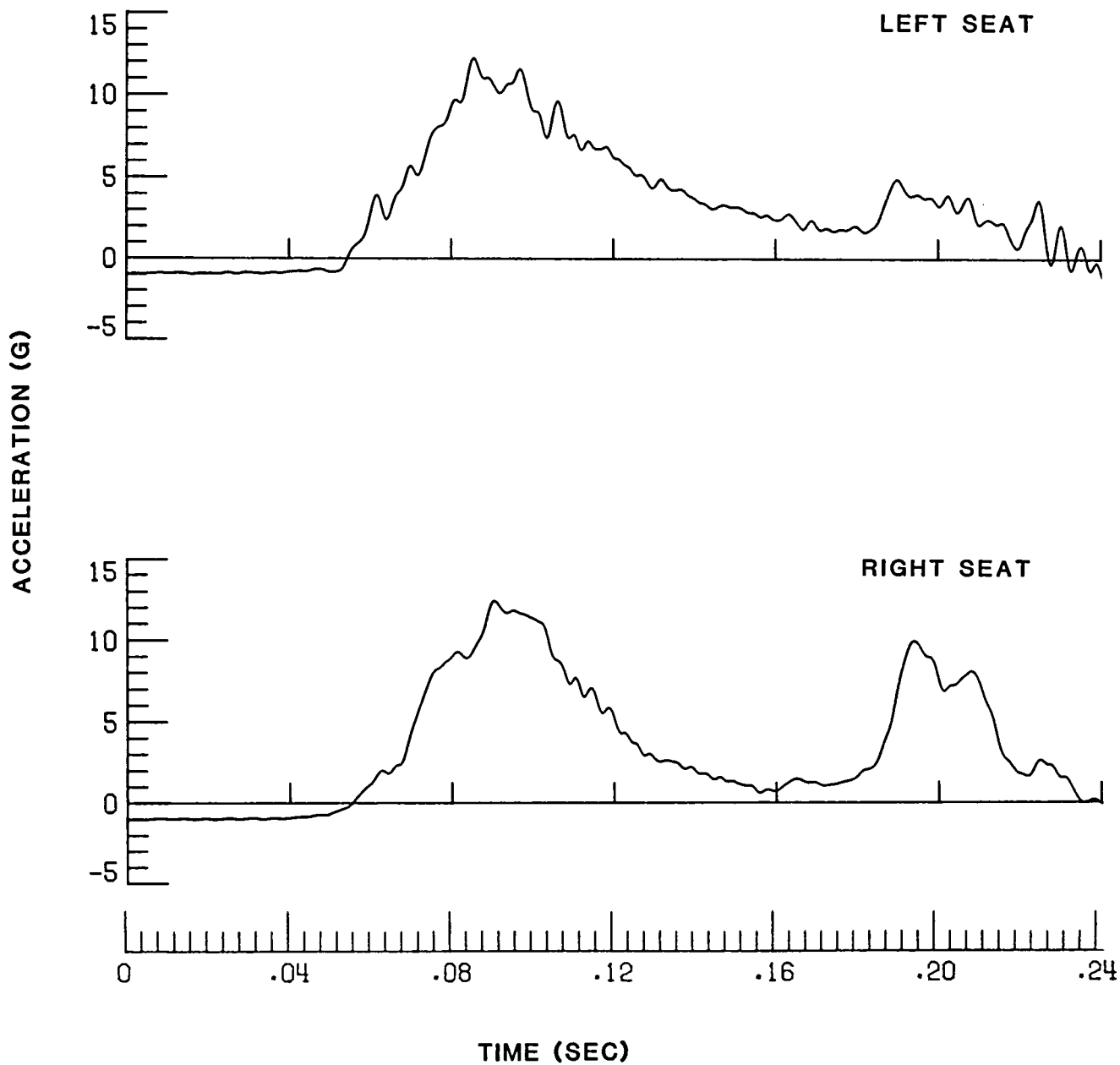
LEFT SEAT PAN  
NASA TEST NO. 11



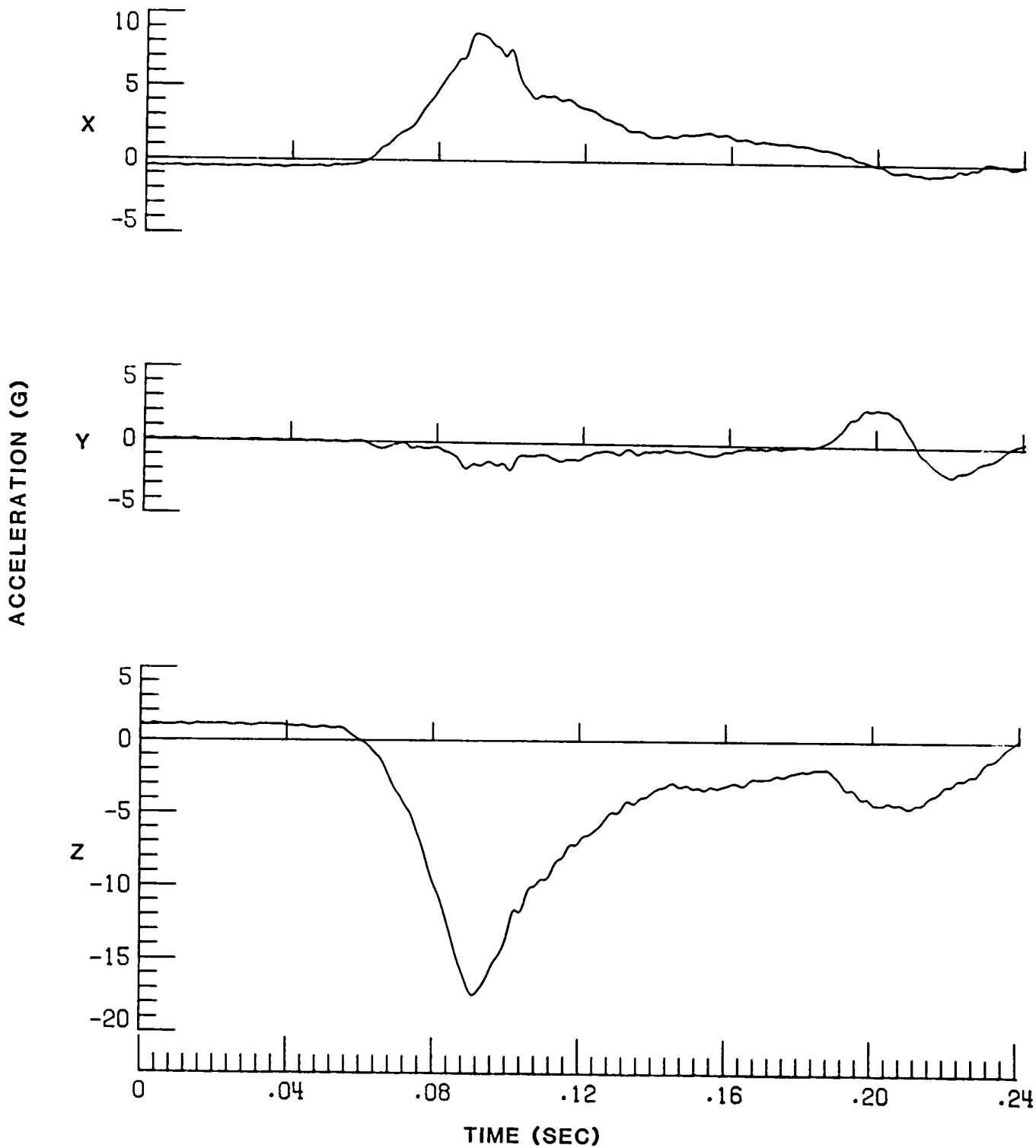
RIGHT SEAT PAN  
NASA TEST NO. 11



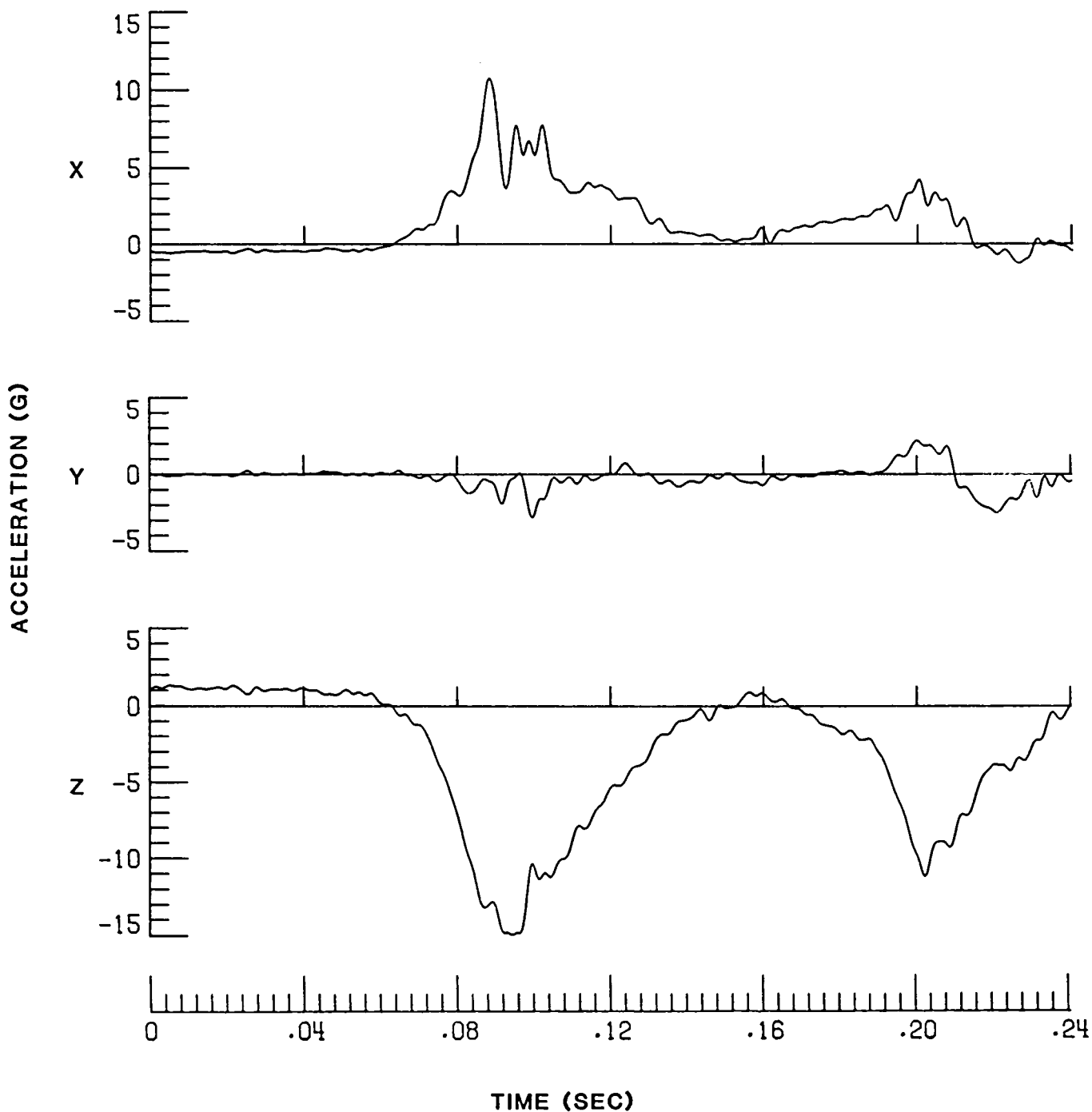
CAPSULE C.G.  
NASA TEST NO. 11



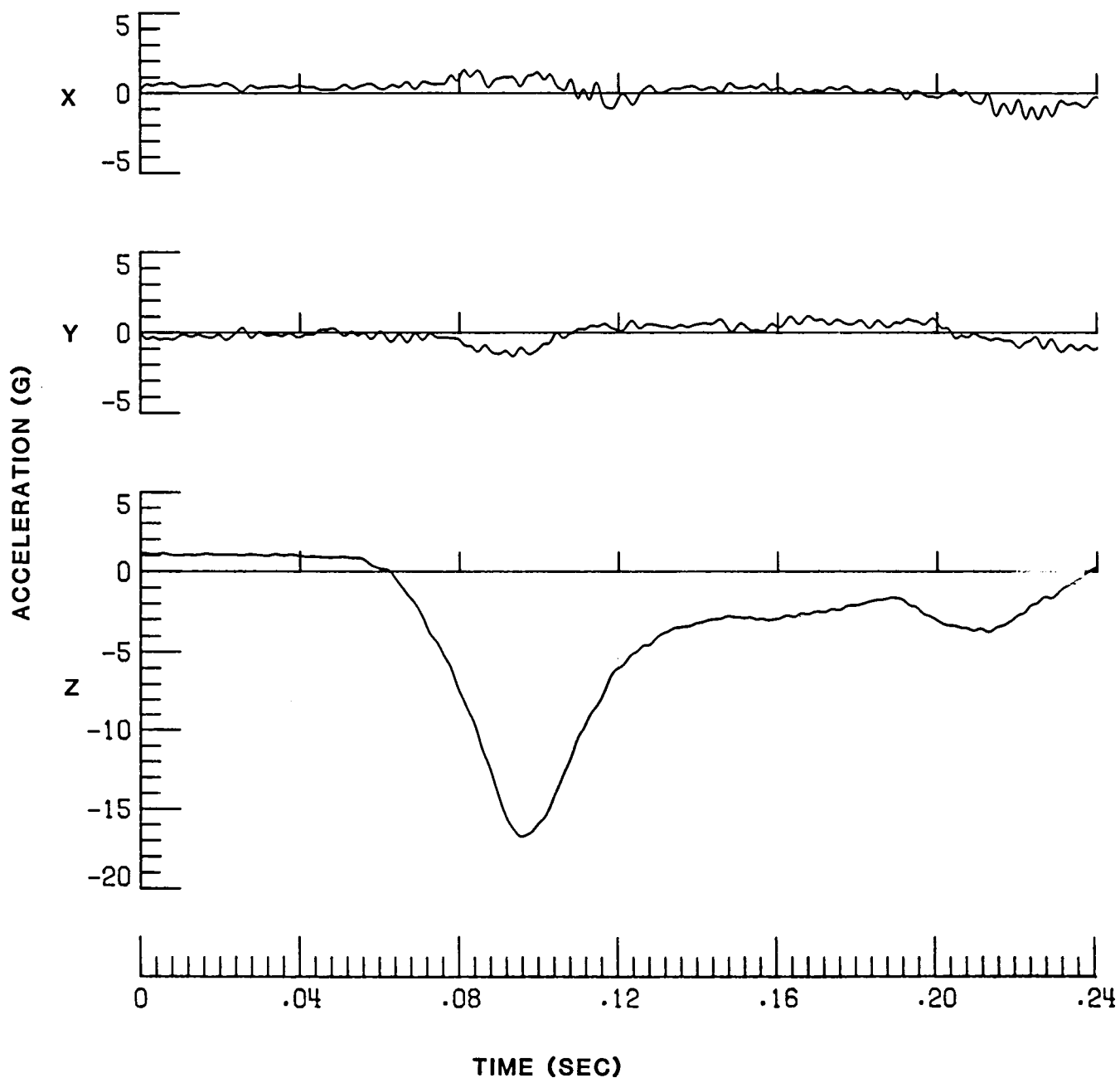
SEAT PAN ALONG BACK TANGENT LINE  
NASA TEST NO. 11



LEFT DUMMY PELVIS  
NASA TEST NO. 11

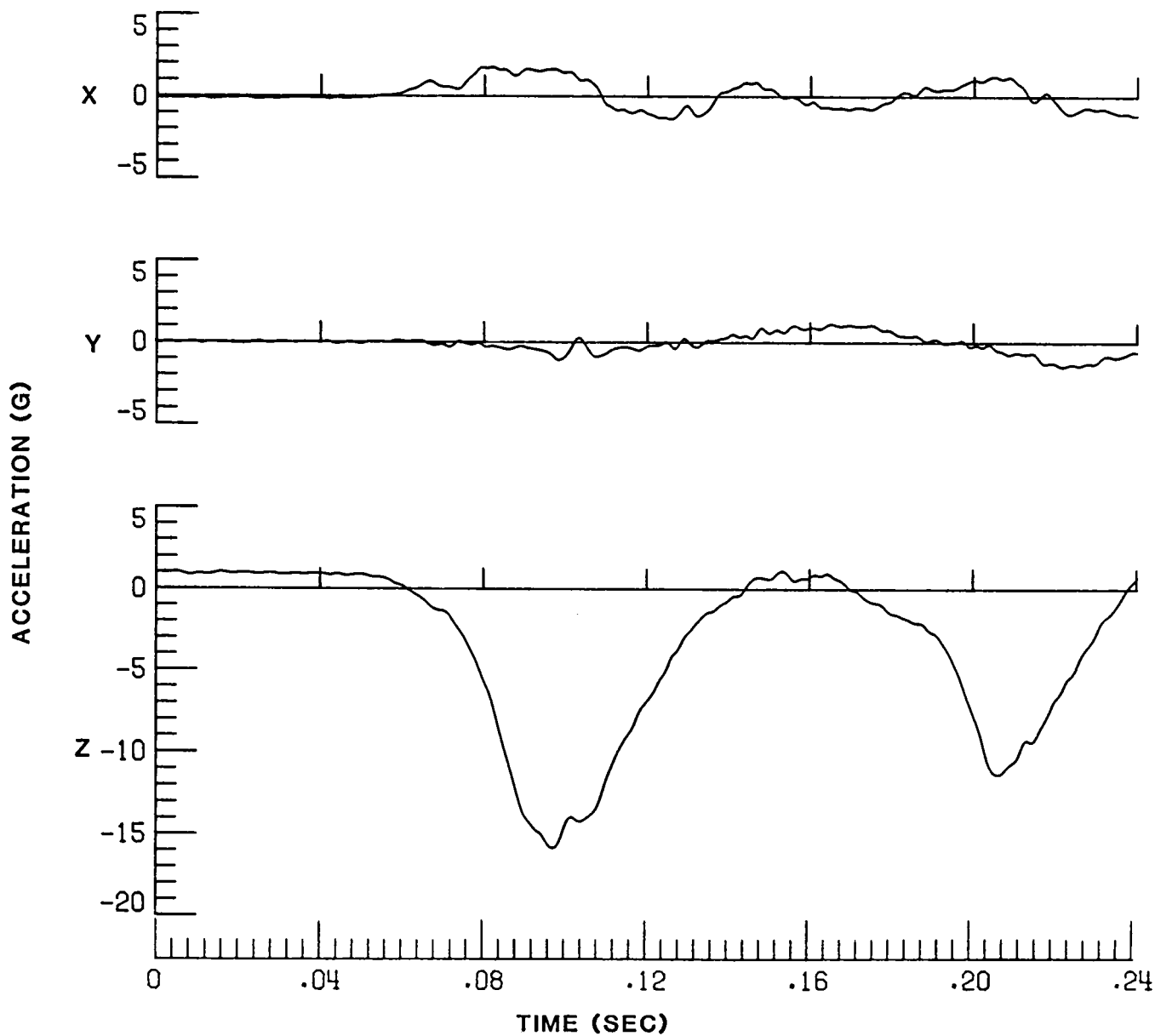


RIGHT DUMMY PELVIS  
NASA TEST NO. 11

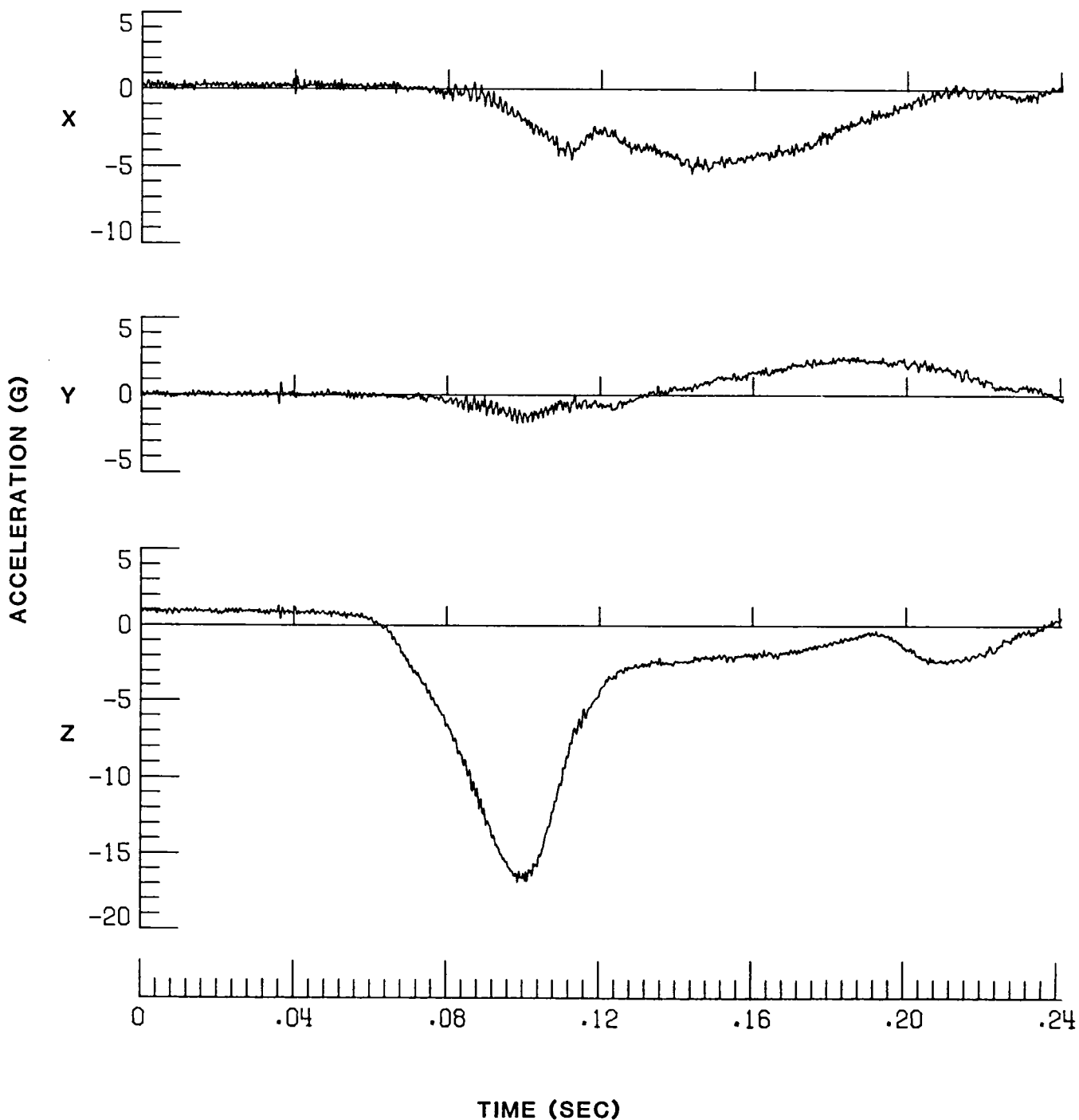


LEFT DUMMY CHEST  
NASA TEST NO. 11

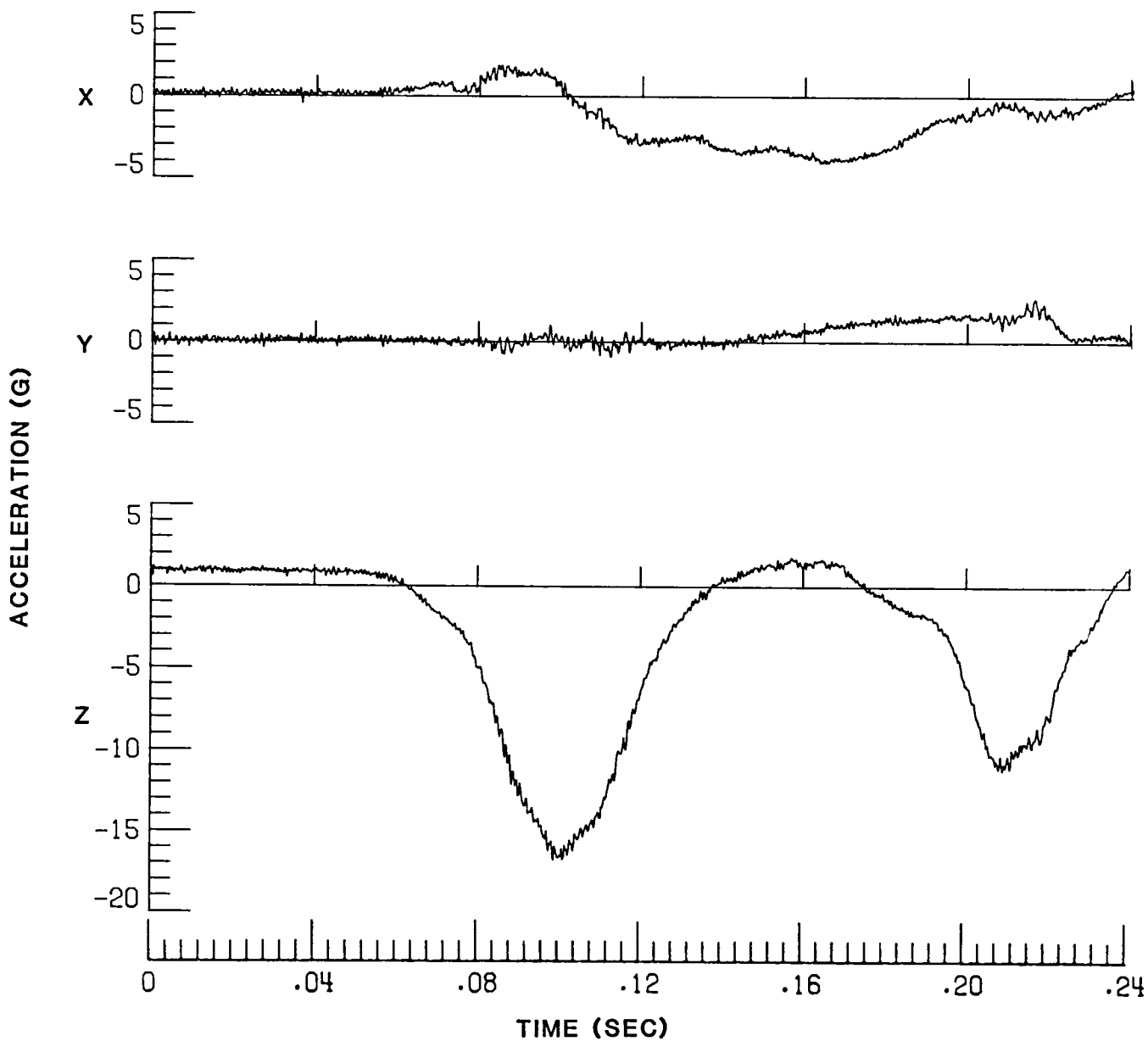




RIGHT DUMMY CHEST  
NASA TEST NO. 11

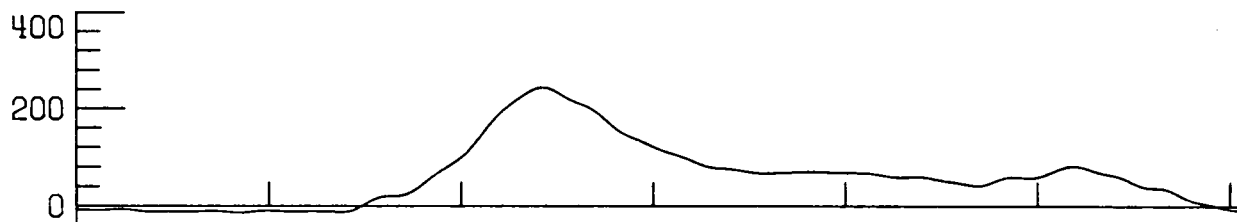


LEFT DUMMY HEAD  
NASA TEST NO. 11

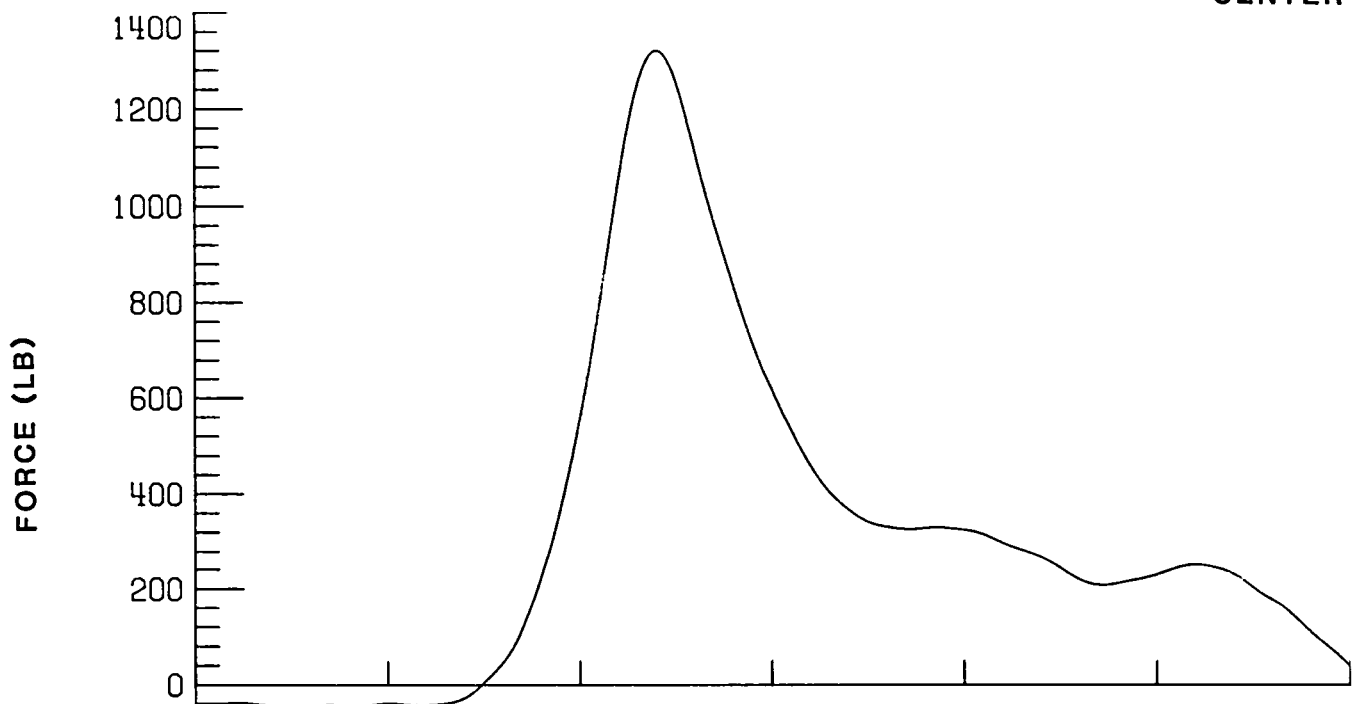


RIGHT DUMMY HEAD  
NASA TEST NO. 11

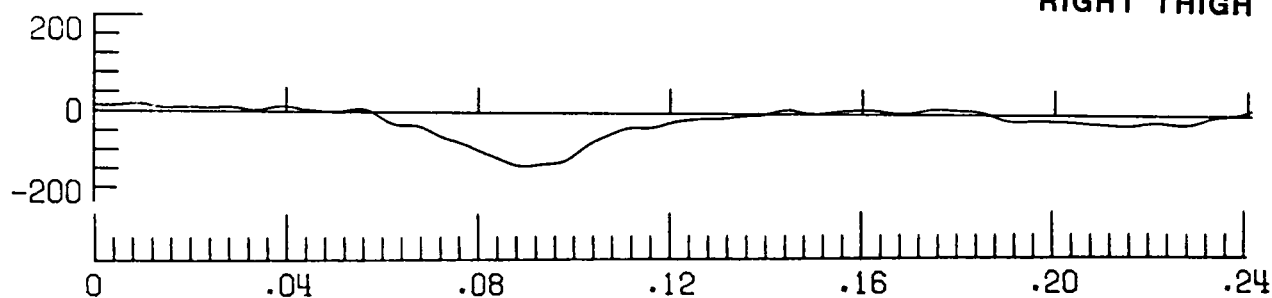
LEFT THIGH



CENTER

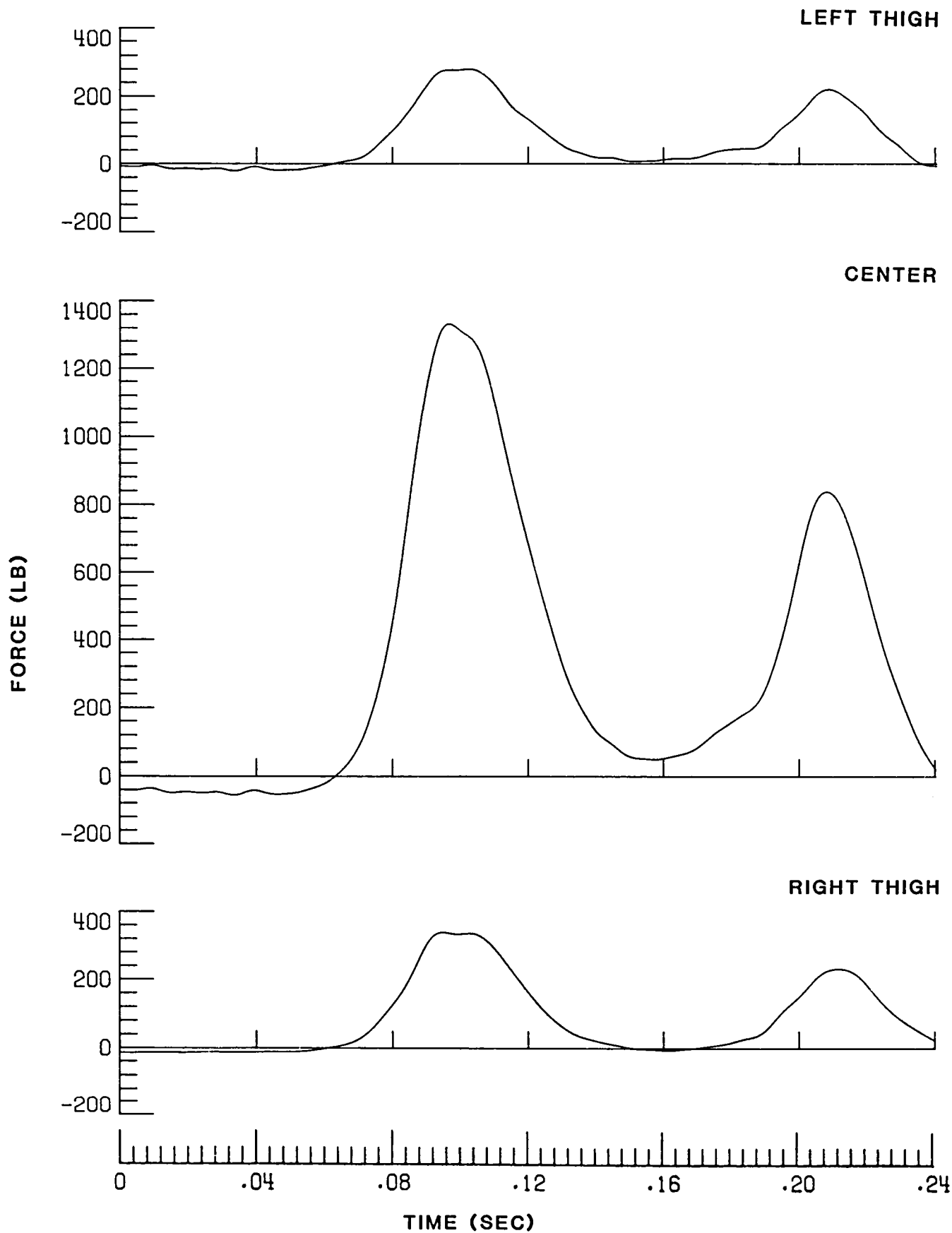


RIGHT THIGH



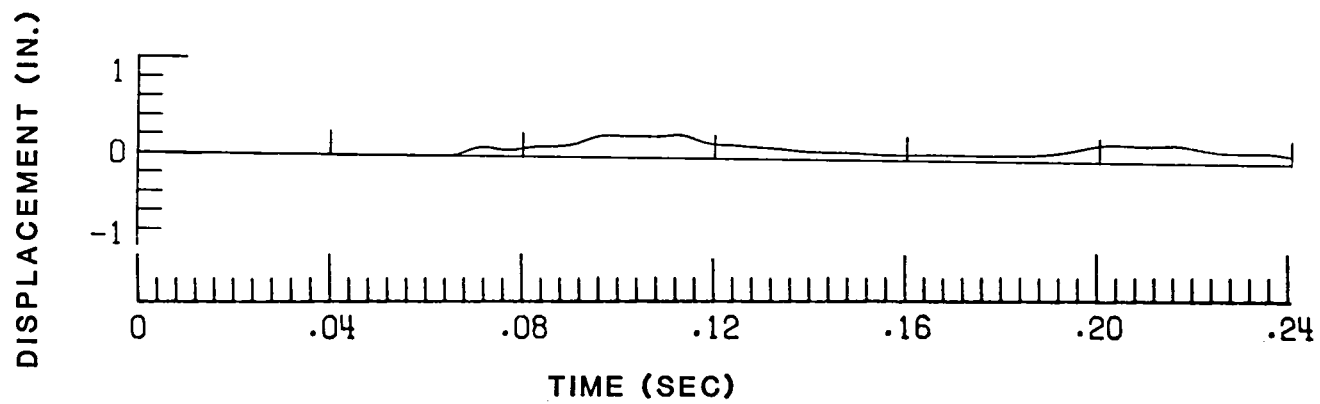
LEFT SEAT PAN LOADS

NASA TEST NO. 11

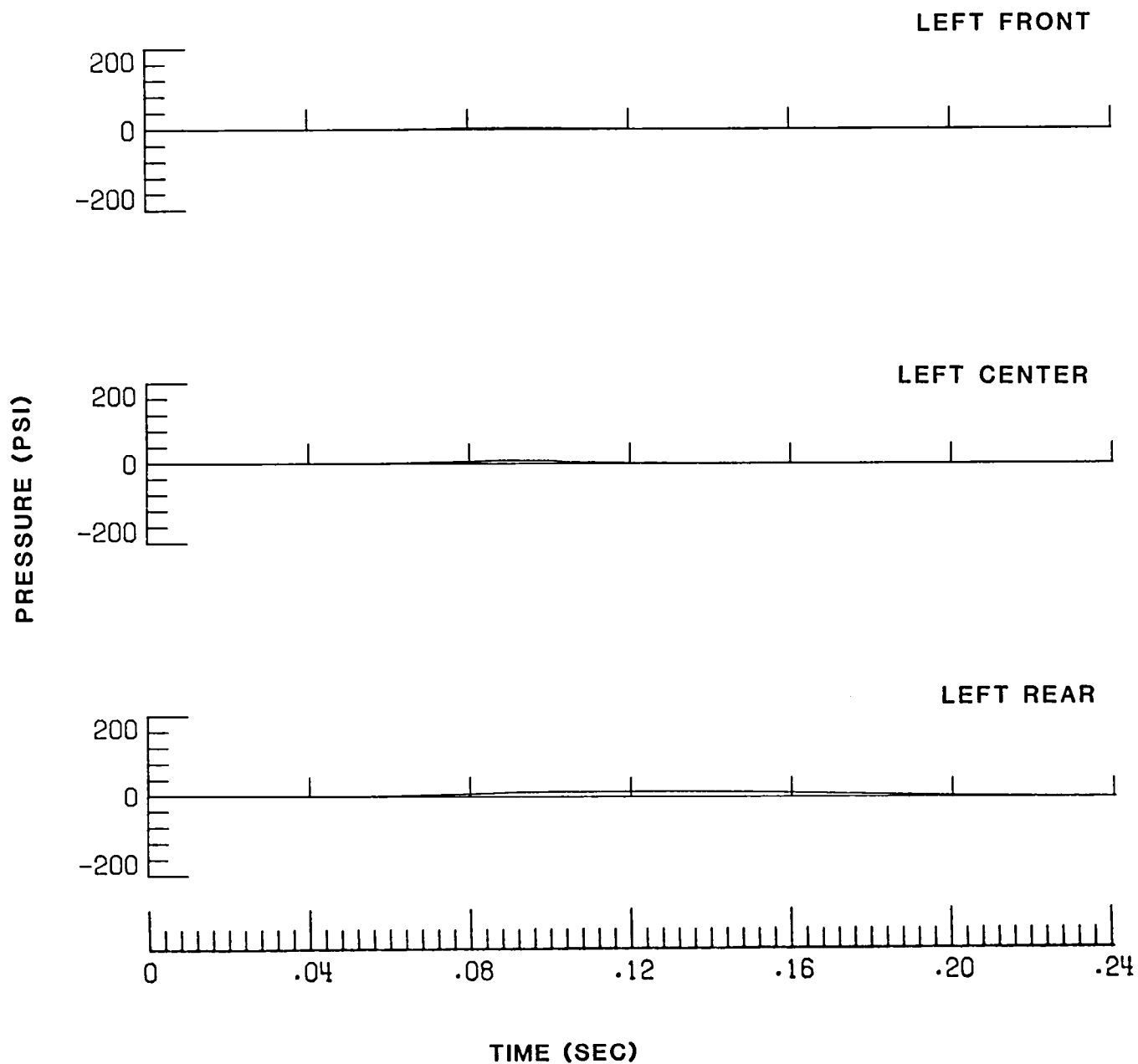


RIGHT SEAT PAN LOADS

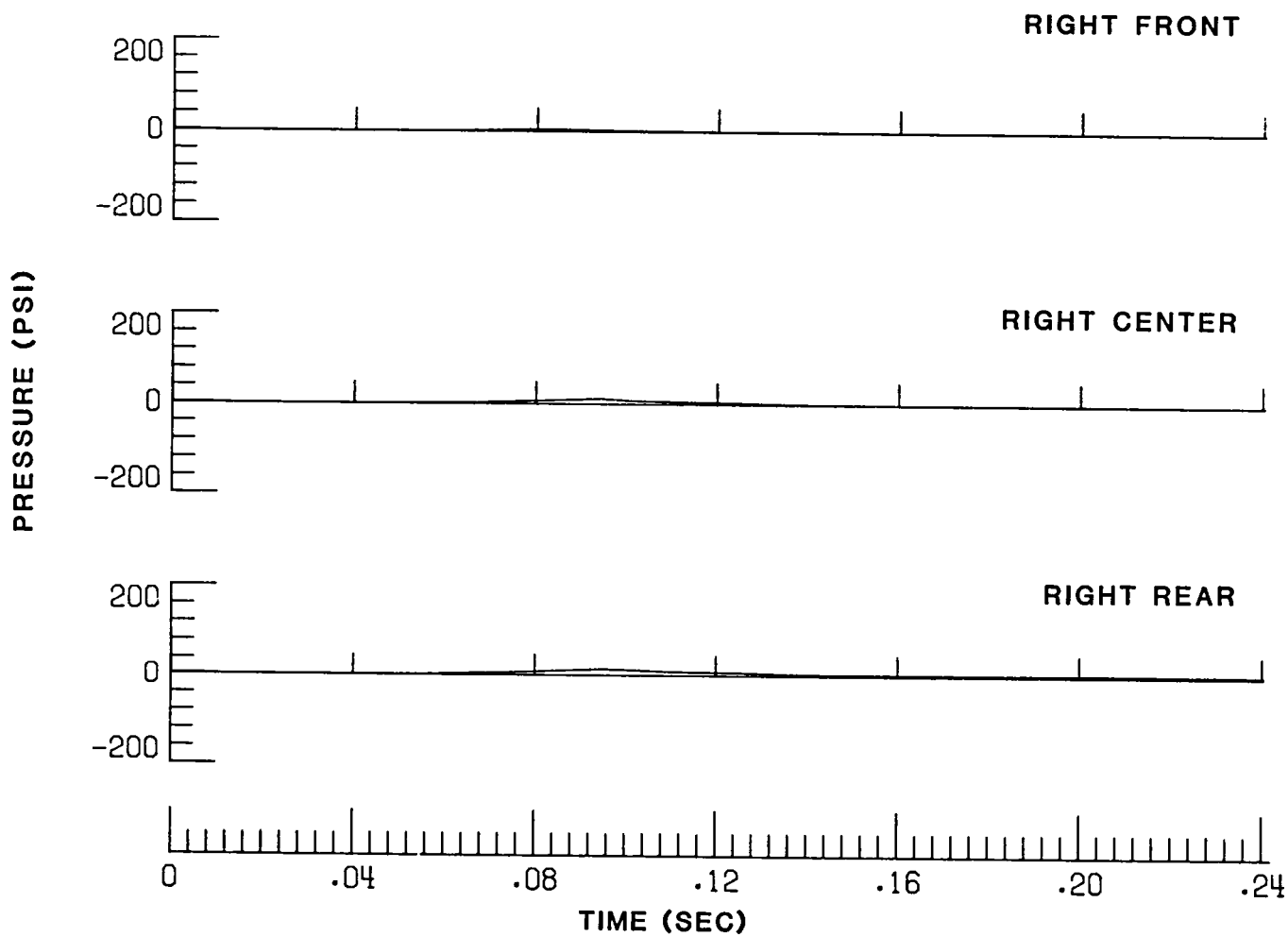
NASA TEST NO. 11



RIGHT SEAT STROKE  
NASA TEST NO. 11

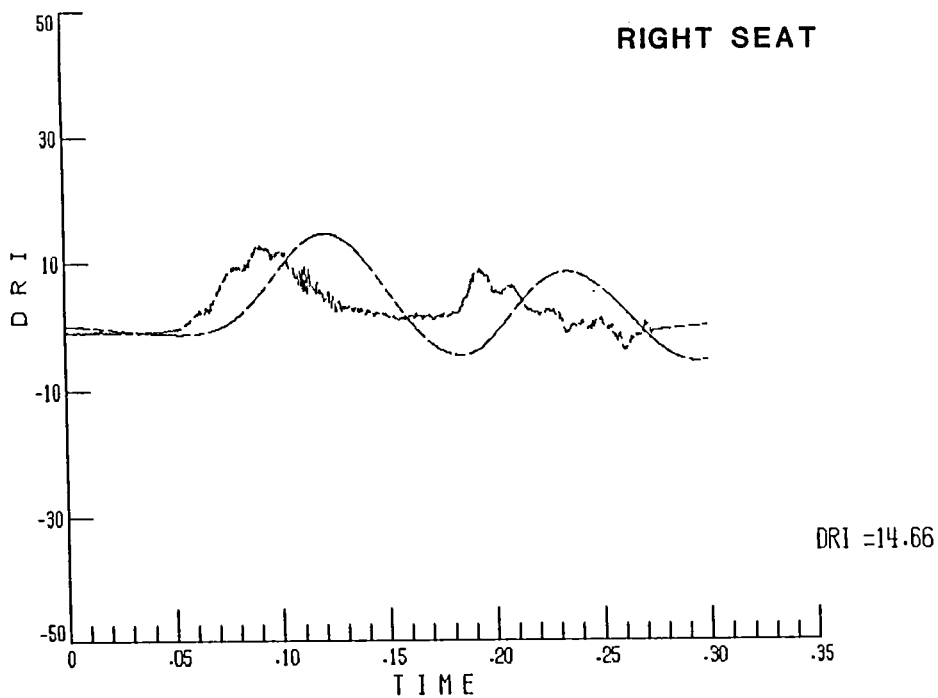
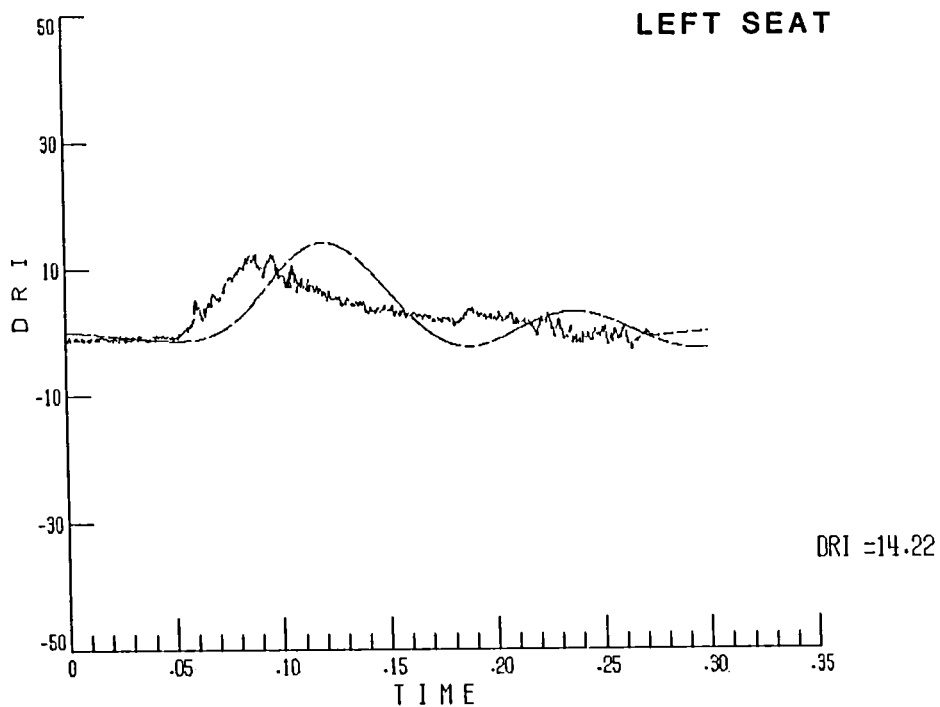


EXIT PORT PRESSURE  
NASA TEST NO.11

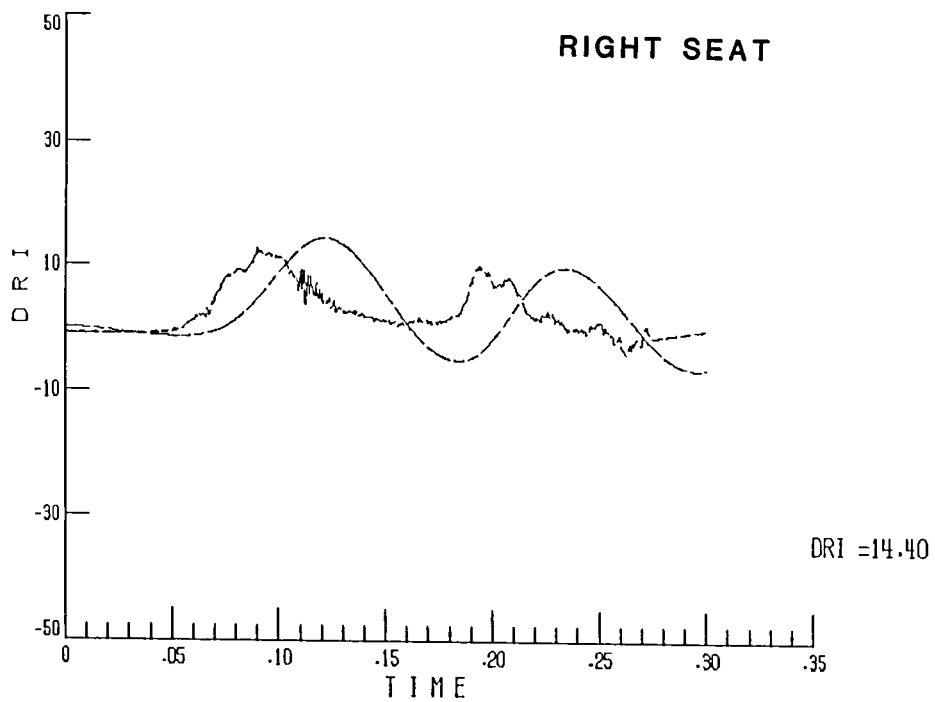
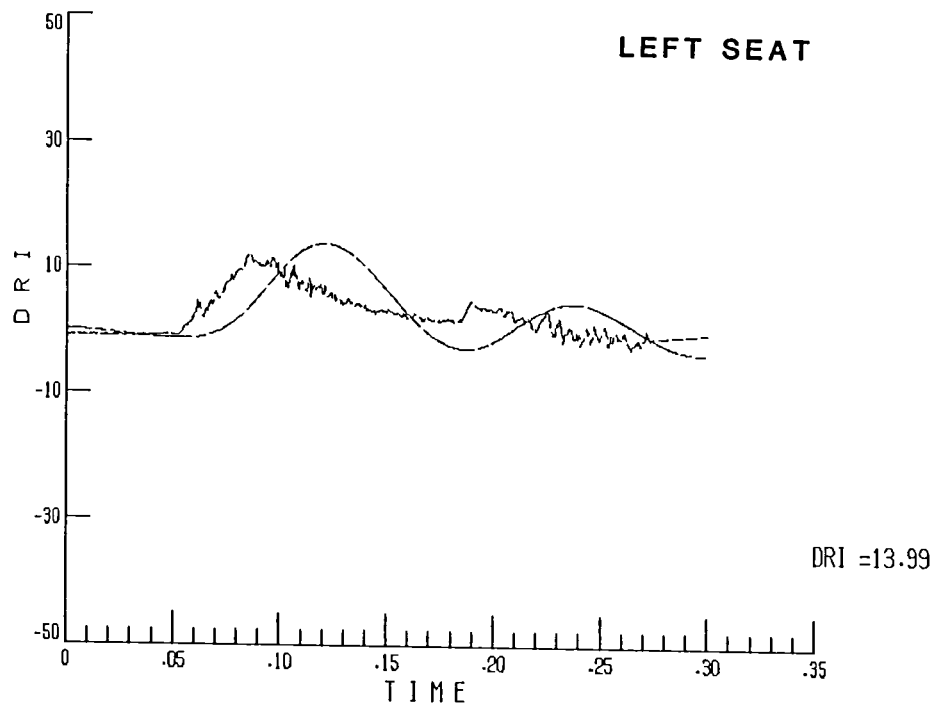


EXIT PORT PRESSURES  
NASA TEST NO. 11

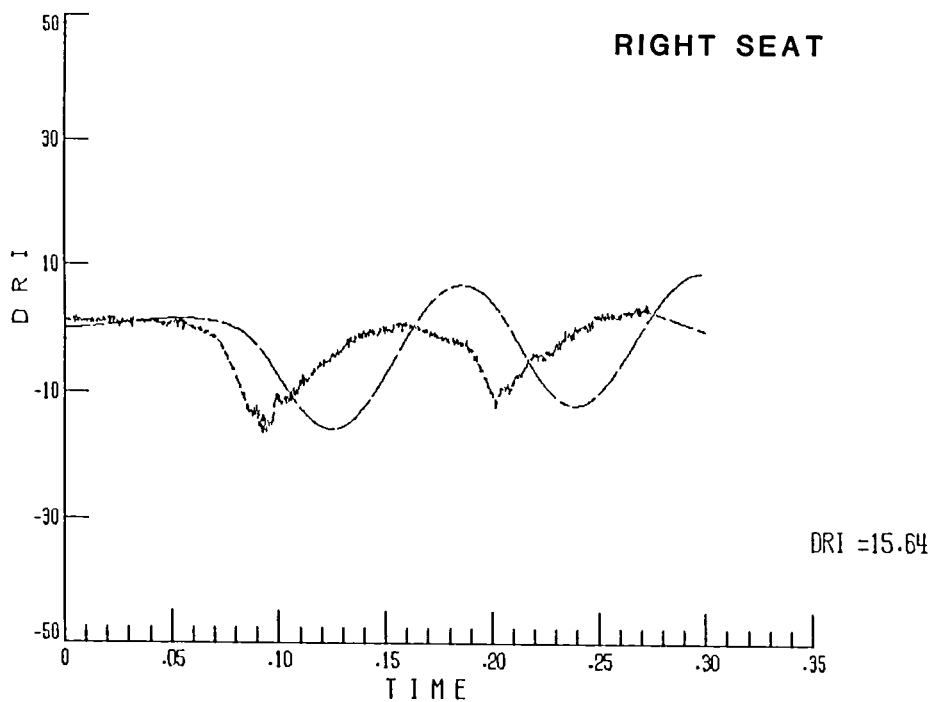
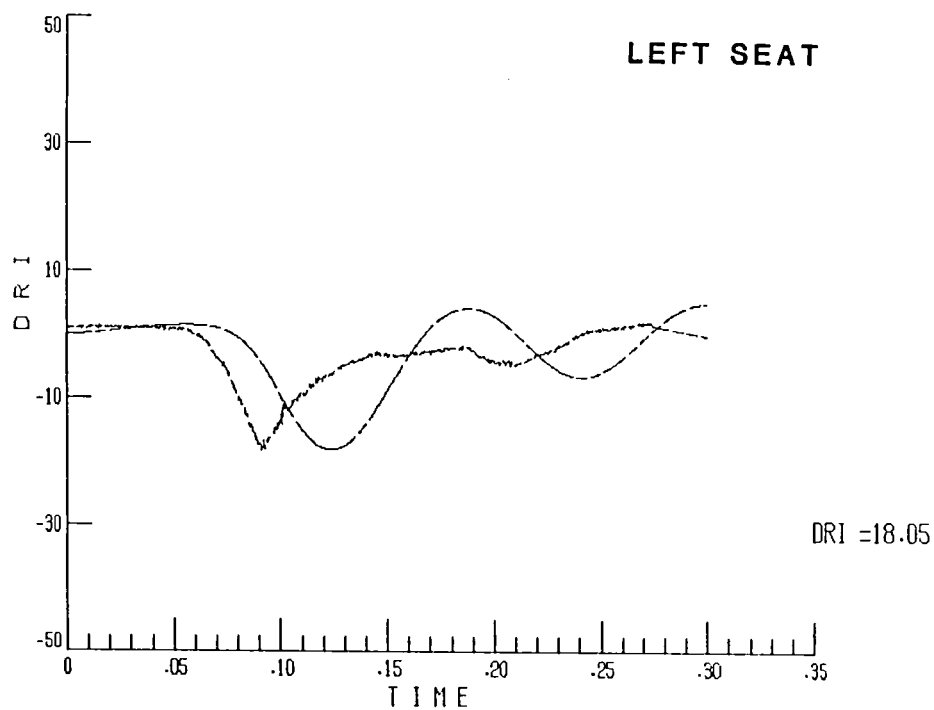




**DYNAMIC RESPONSE INDEX  
SEAT PAN Z-AXIS  
NASA TEST NO. 11**



**DYNAMIC RESPONSE INDEX  
BACK TANGENT LINE  
NASA TEST NO. 11  
212**



DYNAMIC RESPONSE INDEX  
DUMMY PELVIS Z-AXIS  
NASA TEST NO. 11

13.0 NASA TEST NO. 12

HORIZONTAL VELOCITY: 0 ft/sec

VERTICAL VELOCITY: 32 ft/sec

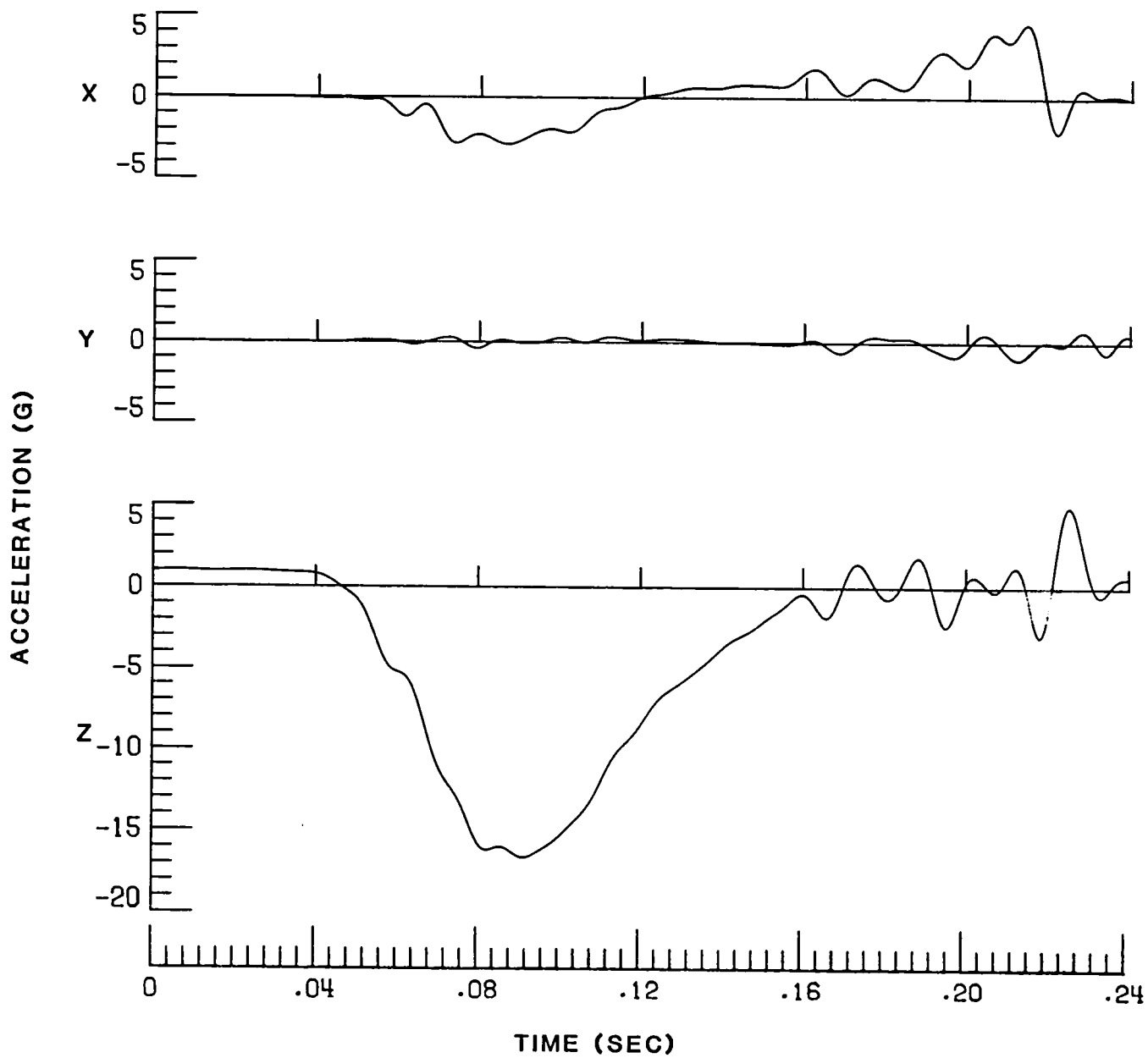
CREW MODULE PITCH:  $2^0$

CREW MODULE ROLL:  $0^0$

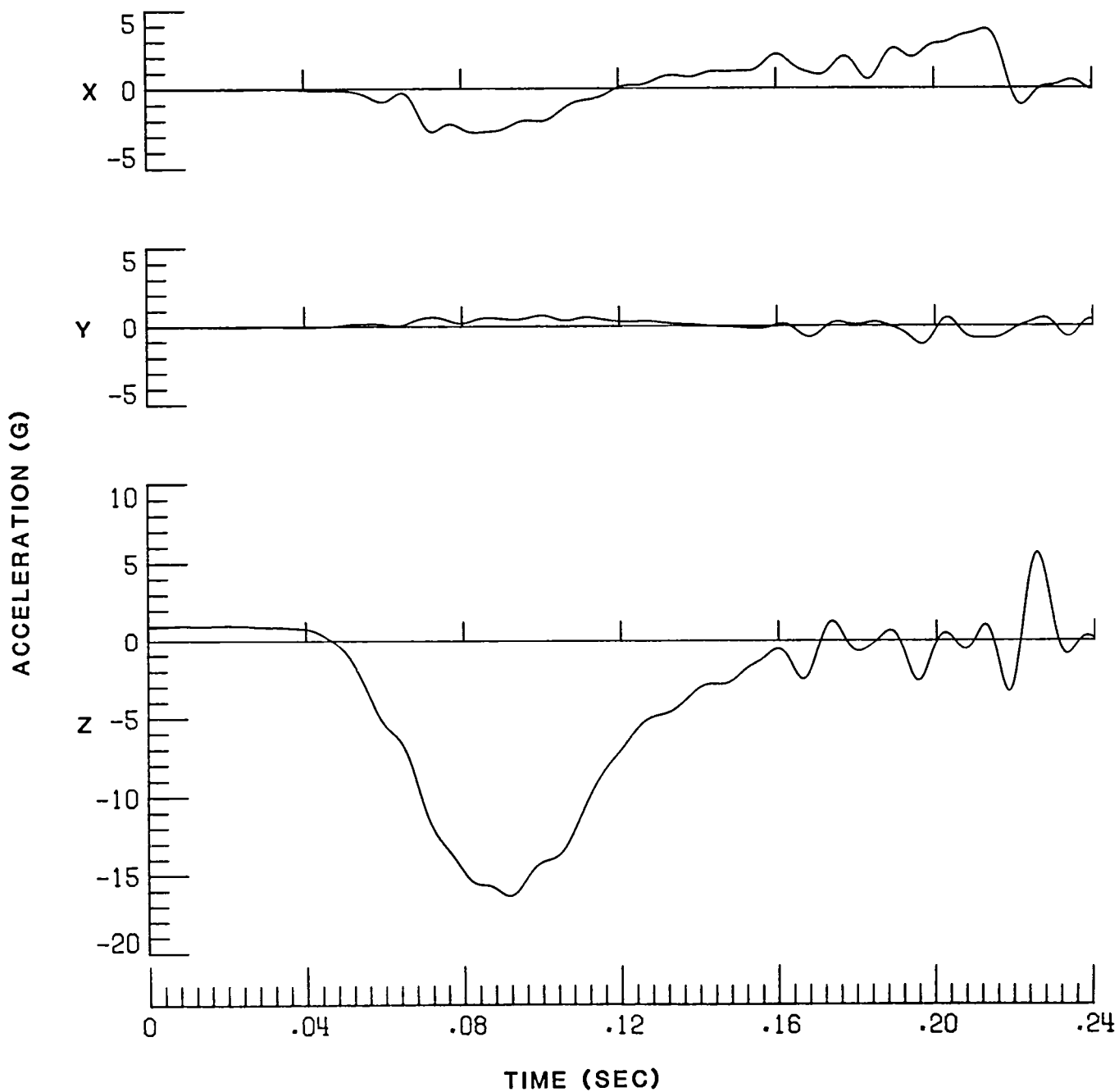
CREW MODULE YAW:  $0^0$

LEFT SEAT: F-111 Operational

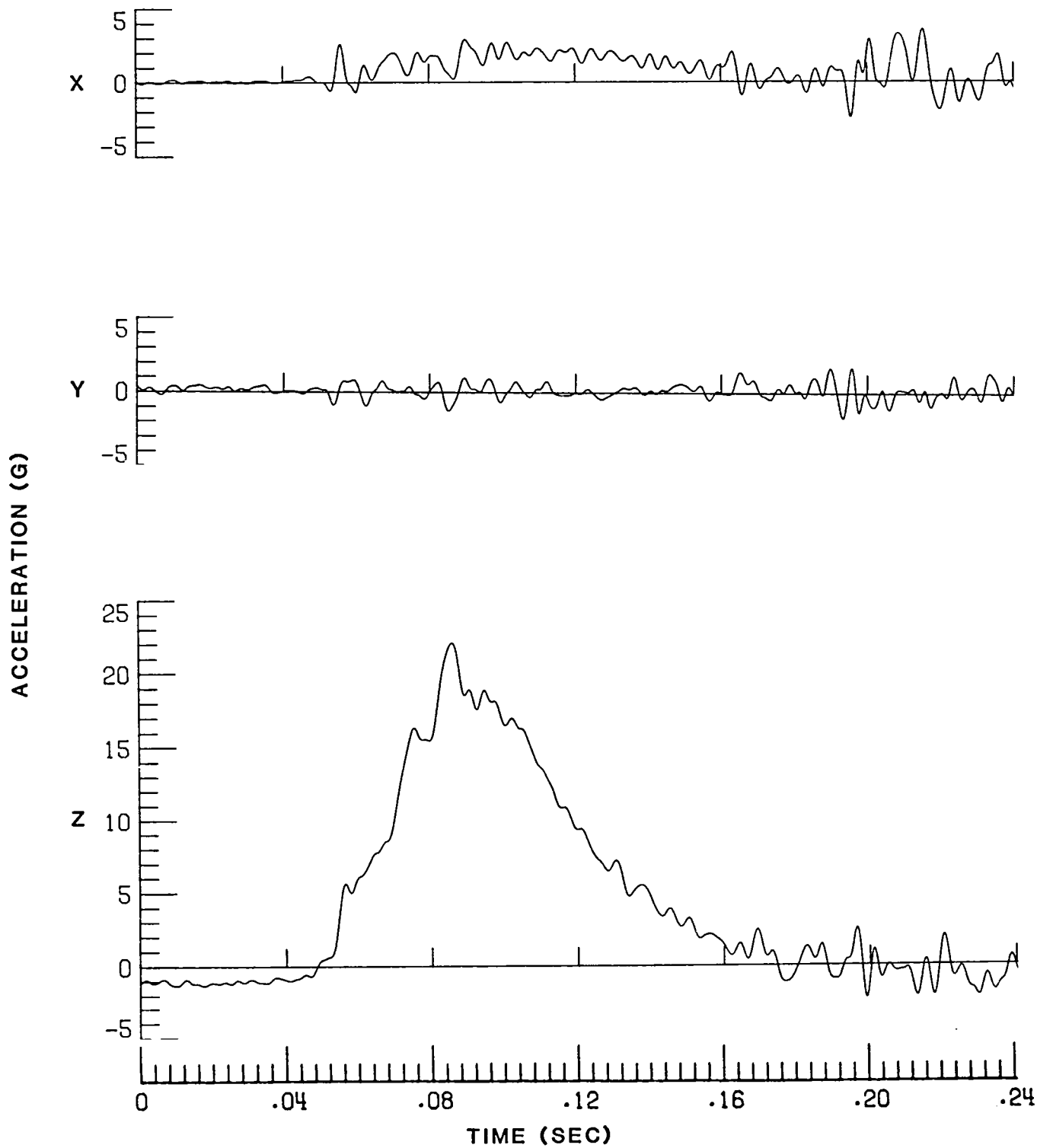
RIGHT SEAT: Energy Absorbing



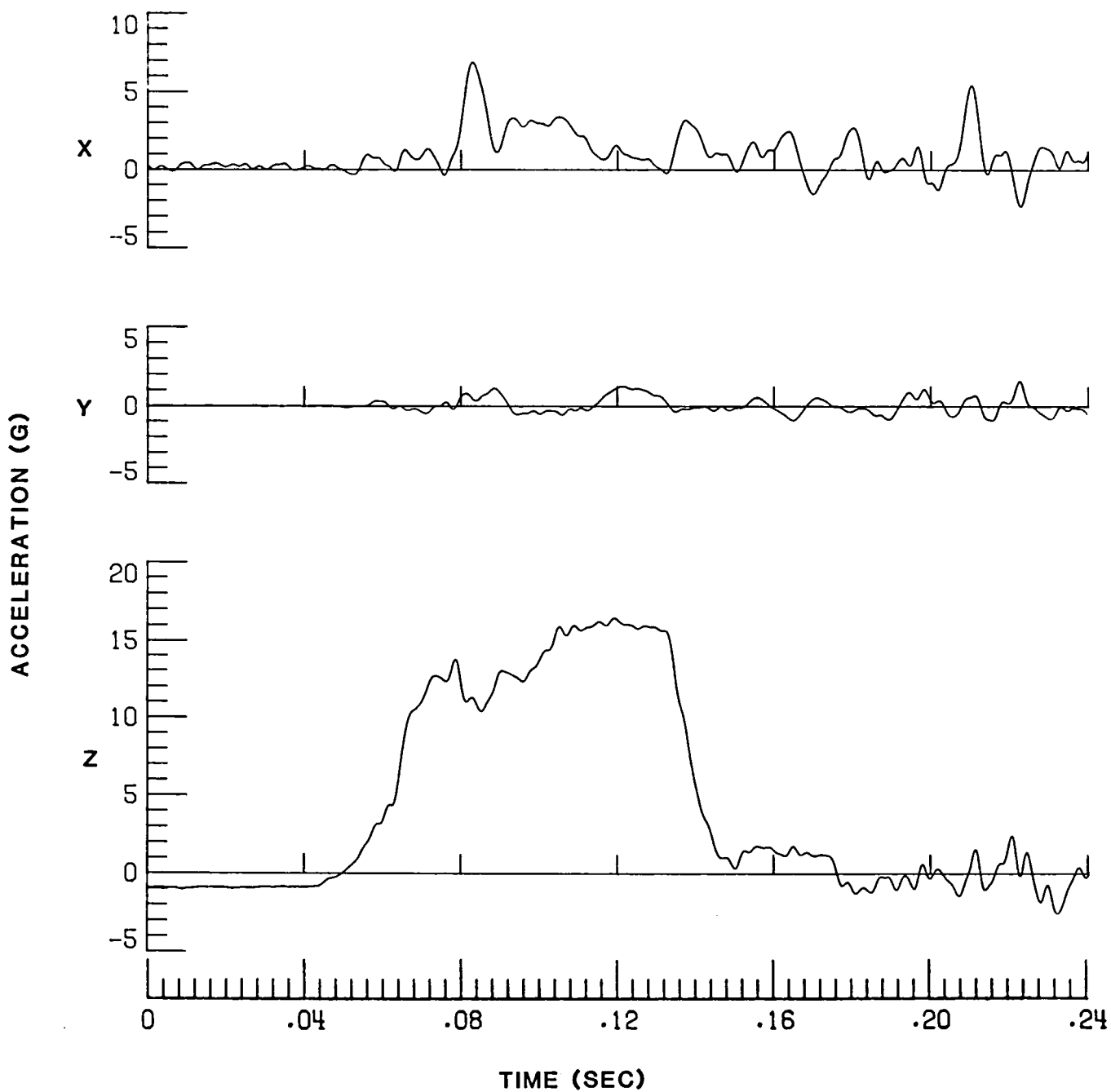
LEFT BULKHEAD  
NASA TEST NO. 12



RIGHT BULKHEAD  
NASA TEST NO. 12

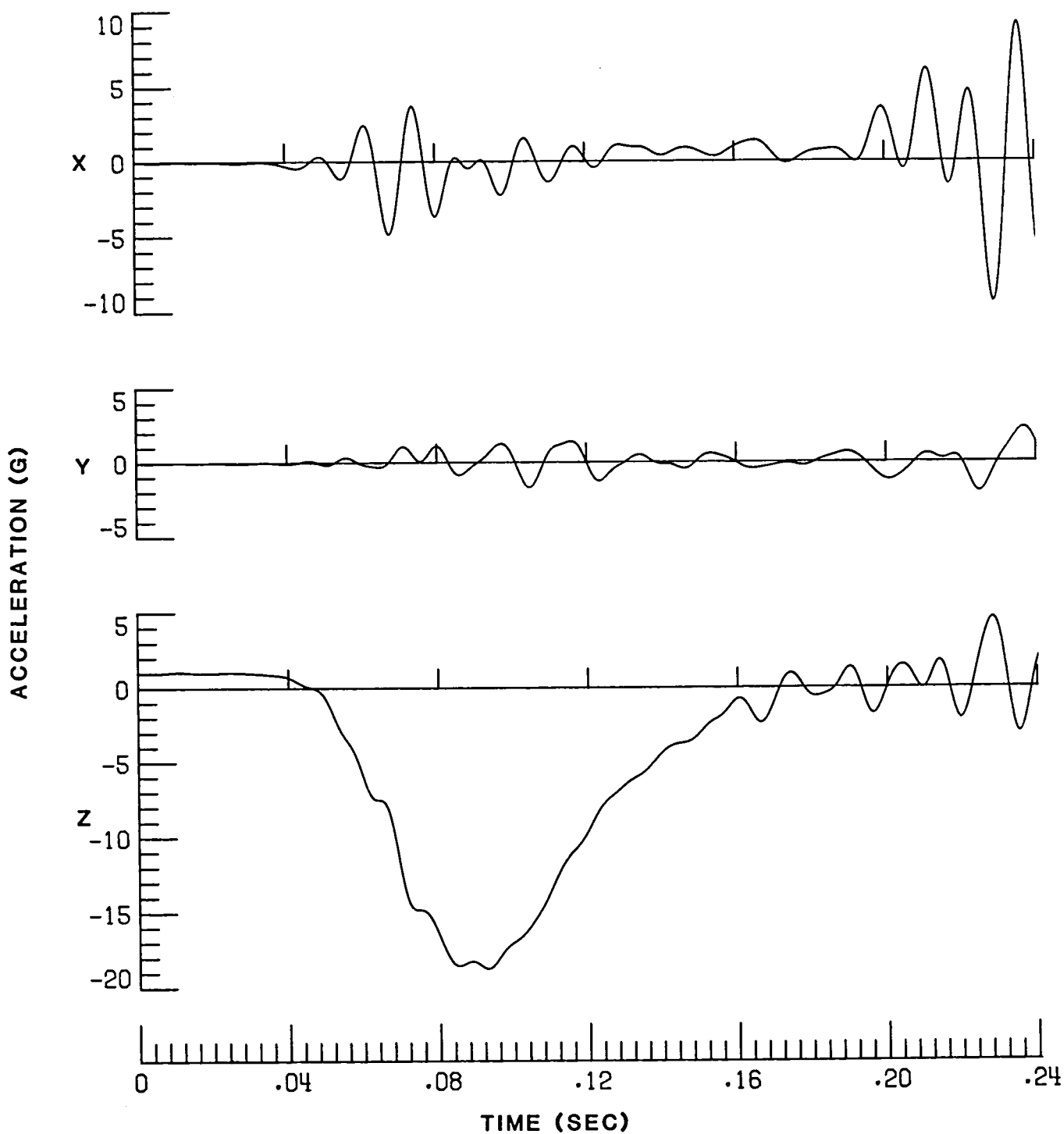


LEFT SEAT PAN  
NASA TEST NO. 12

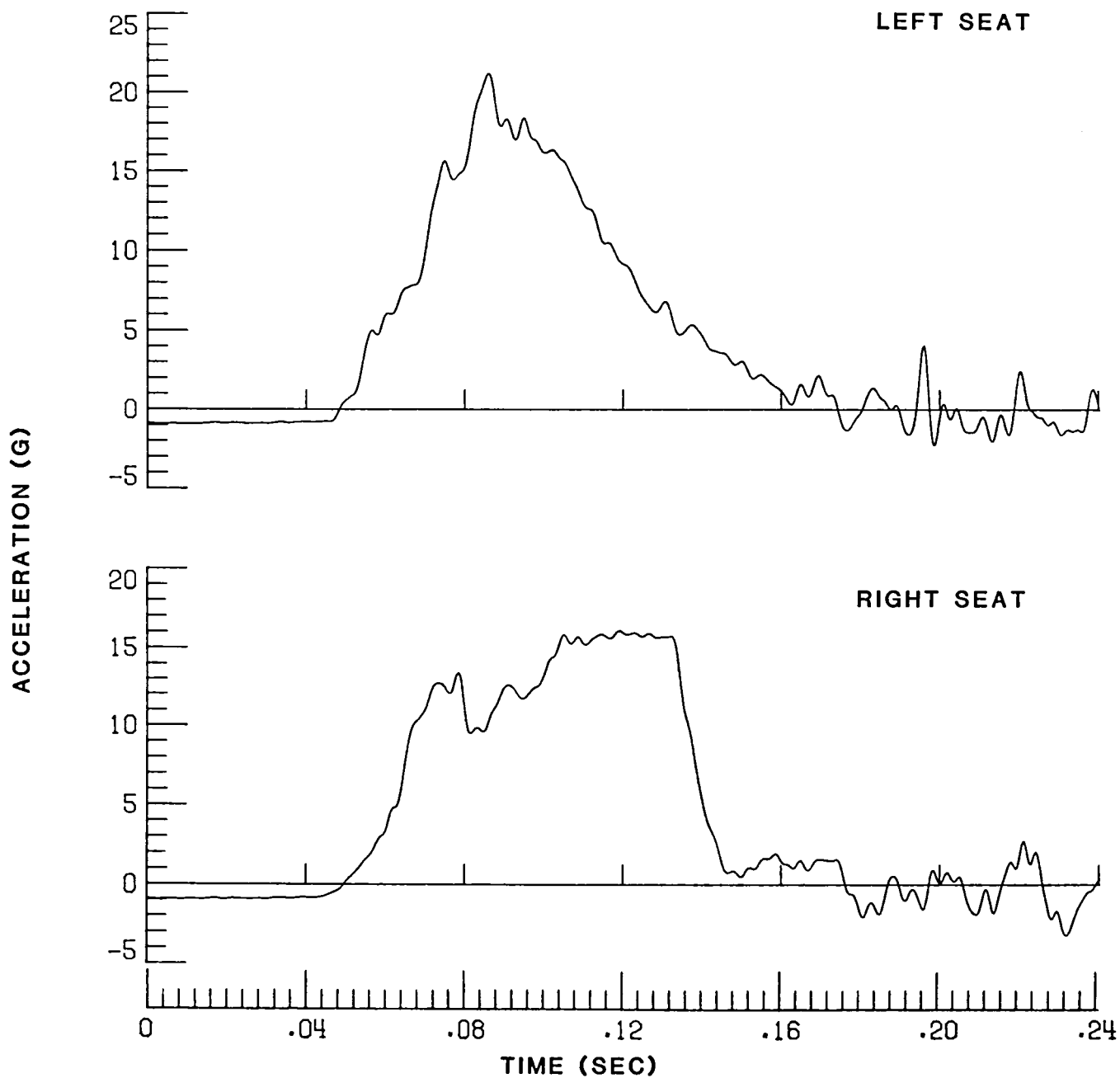


RIGHT SEAT PAN  
NASA TEST NO. 12

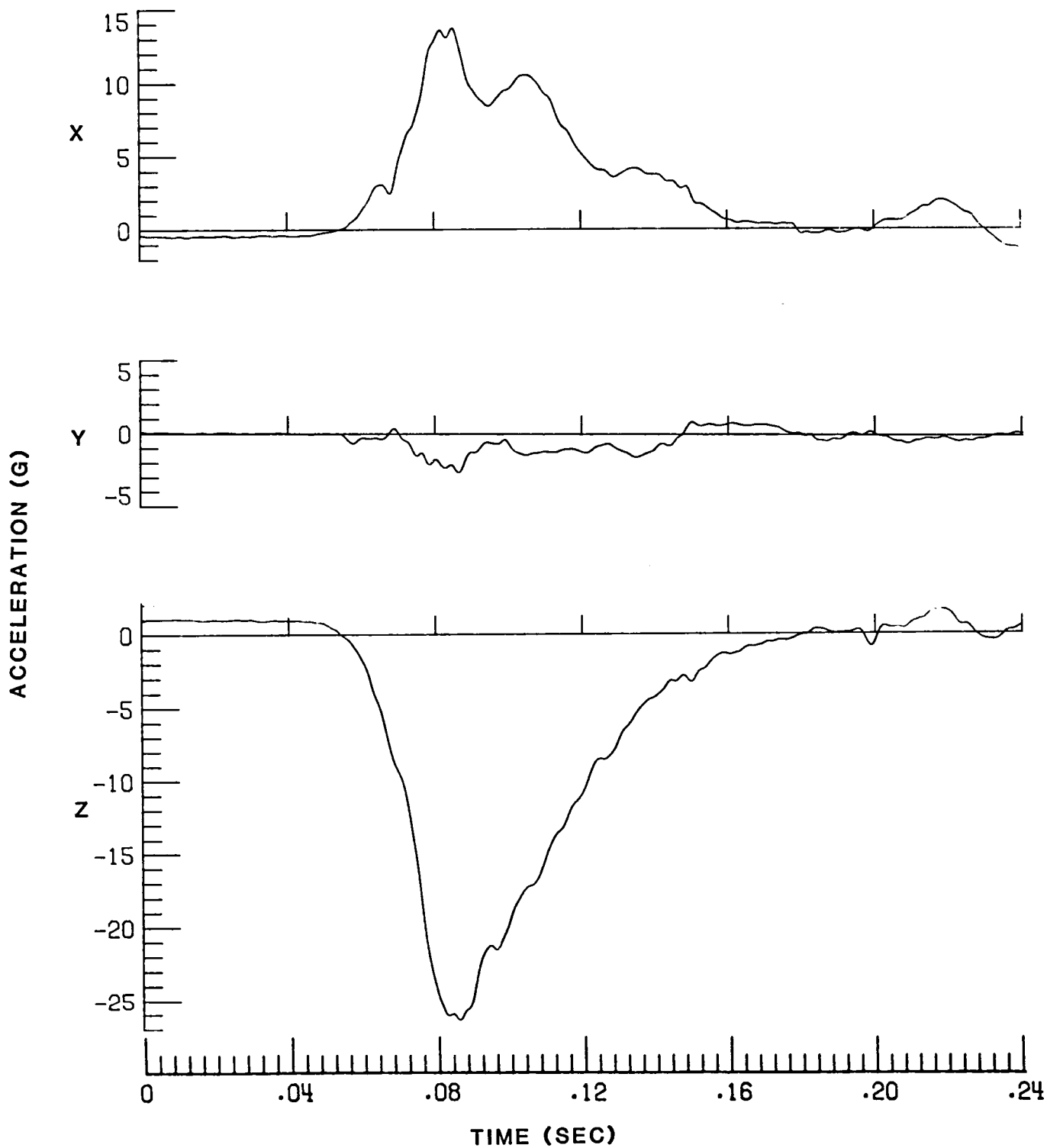




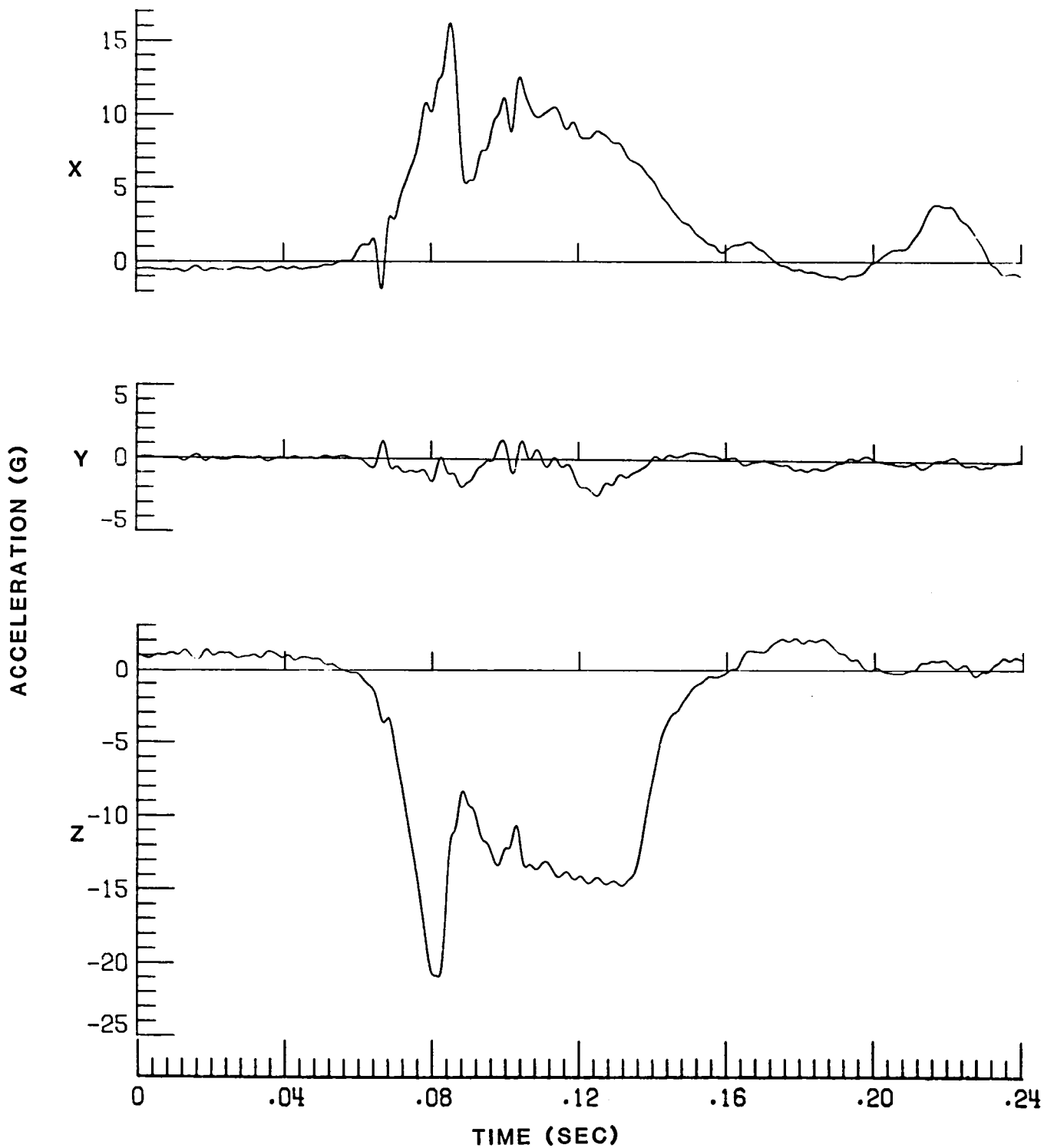
CAPSULE C.G.  
NASA TEST NO.12



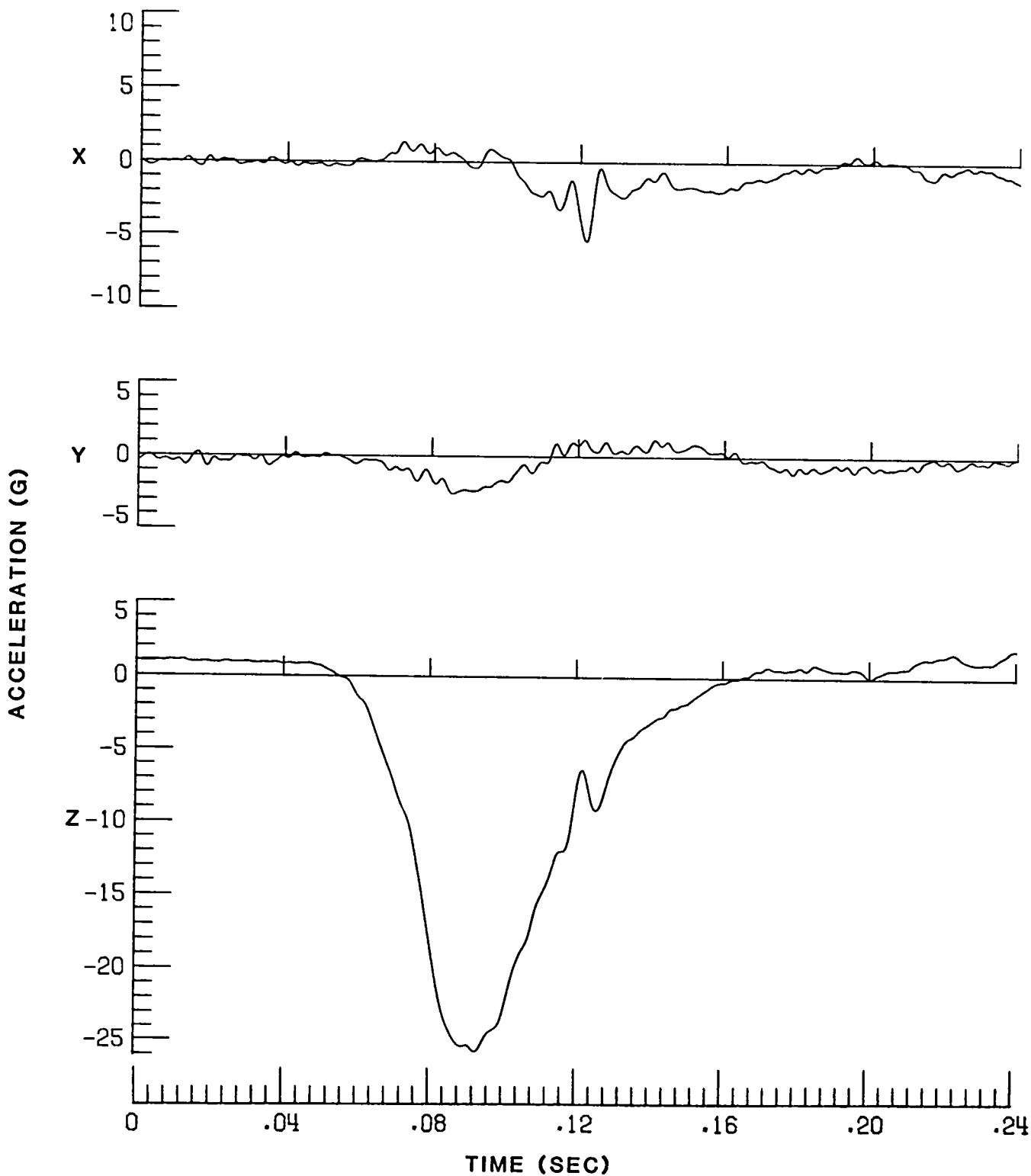
SEAT PAN ALONG BACK TANGENT LINE  
NASA TEST NO. 12



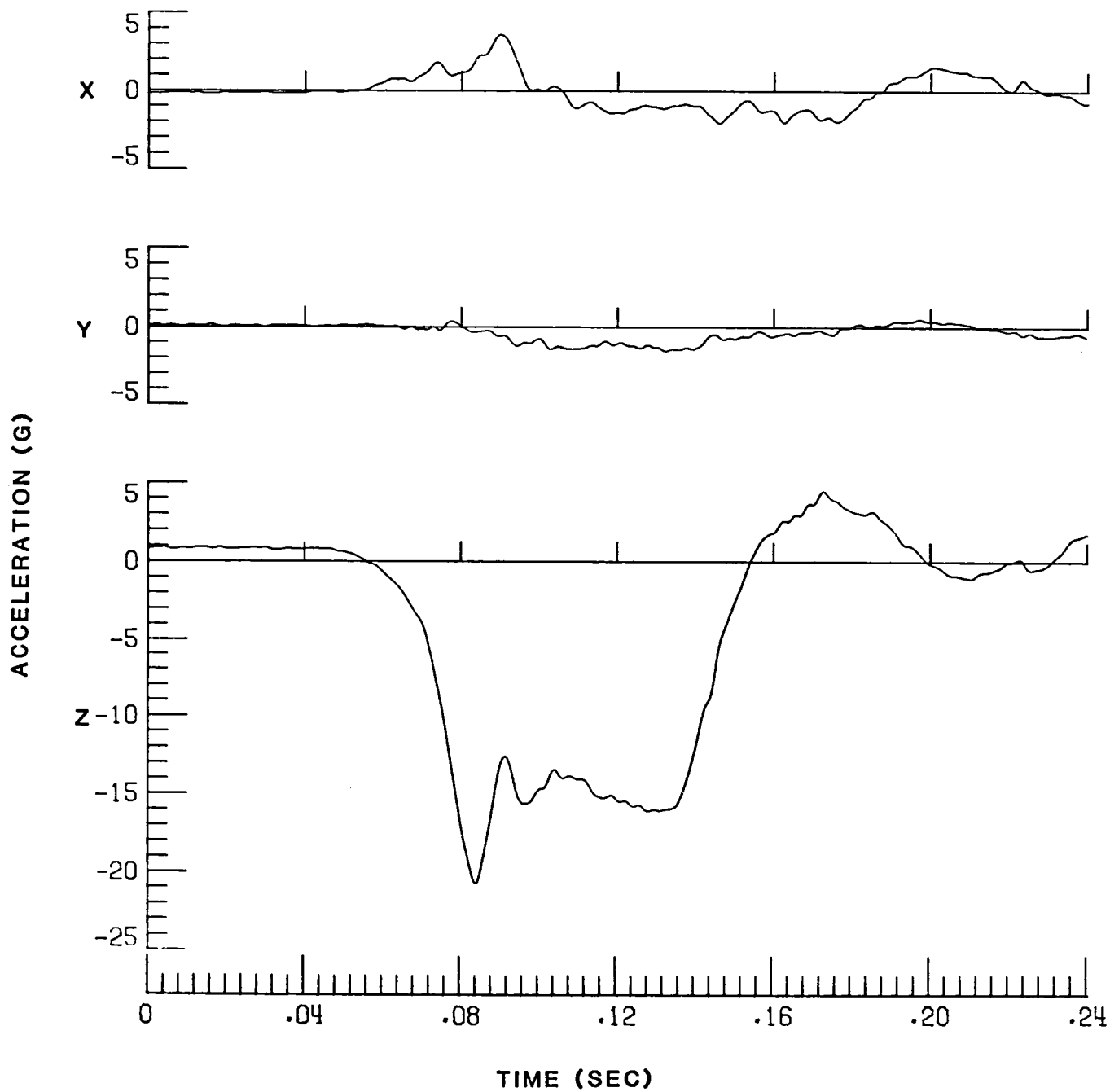
LEFT DUMMY PELVIS  
NASA TEST NO. 12



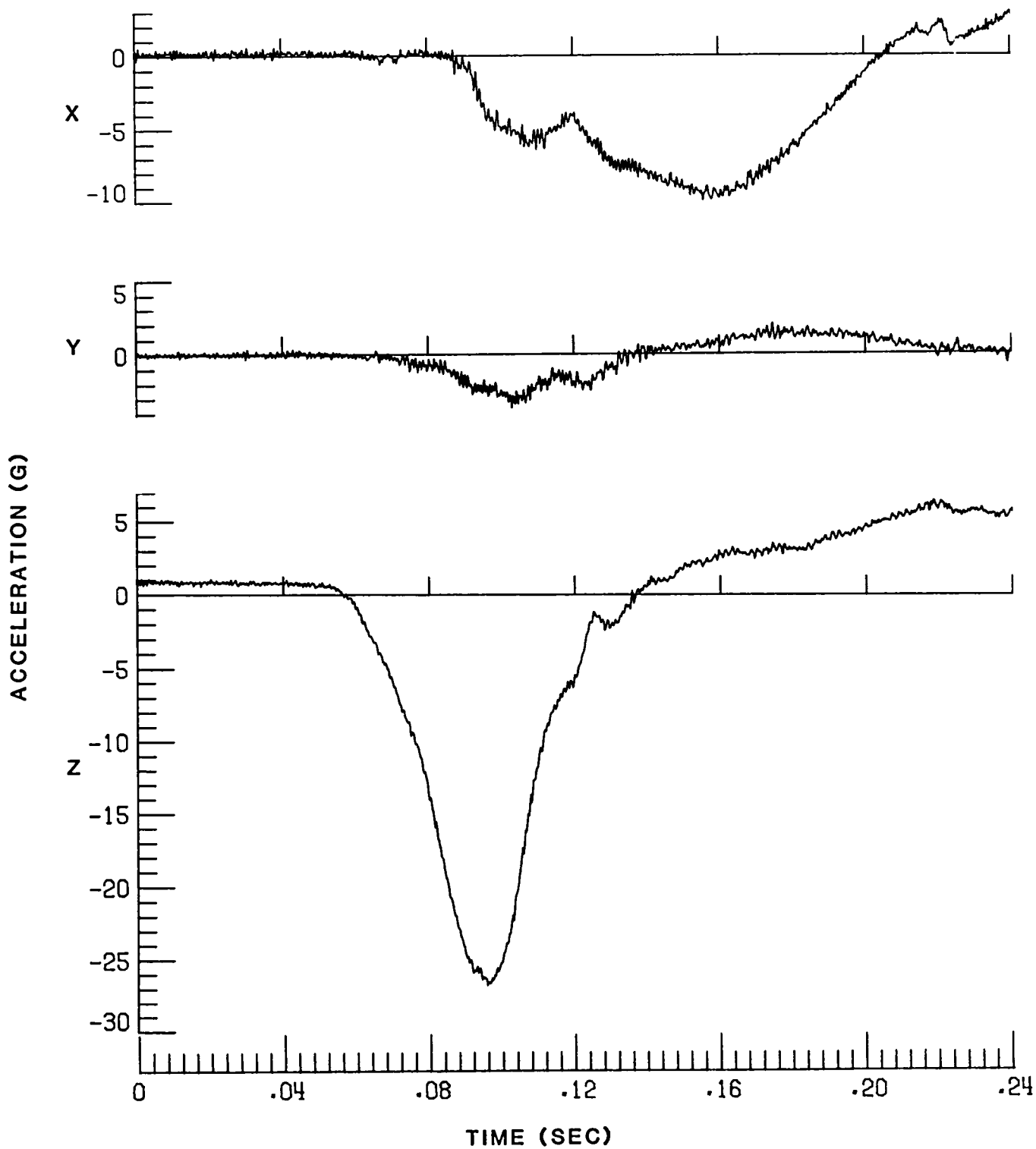
RIGHT DUMMY PELVIS  
NASA TEST NO. 12



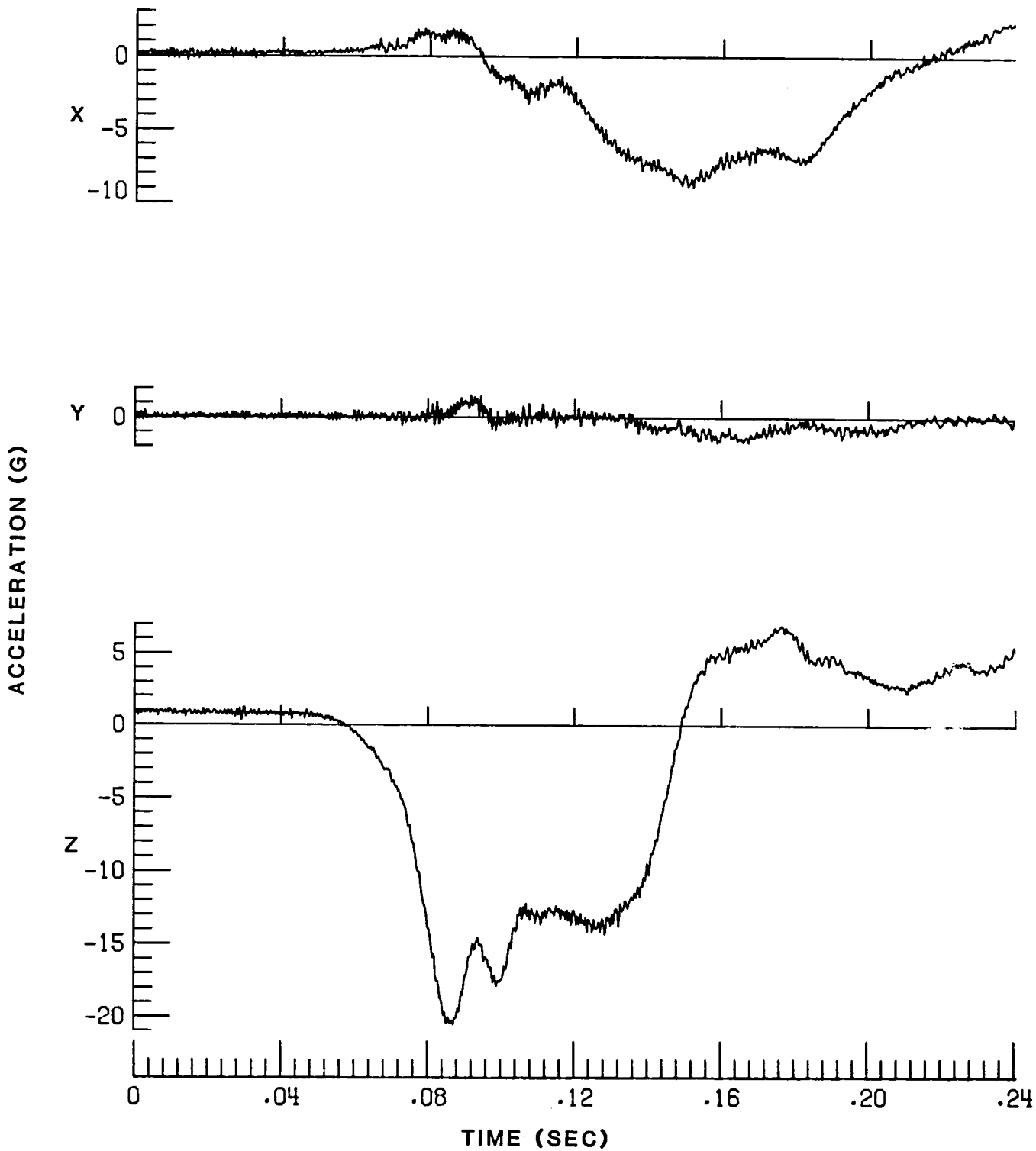
LEFT DUMMY CHEST  
NASA TEST NO. 12



RIGHT DUMMY CHEST  
NASA TEST NO. 12

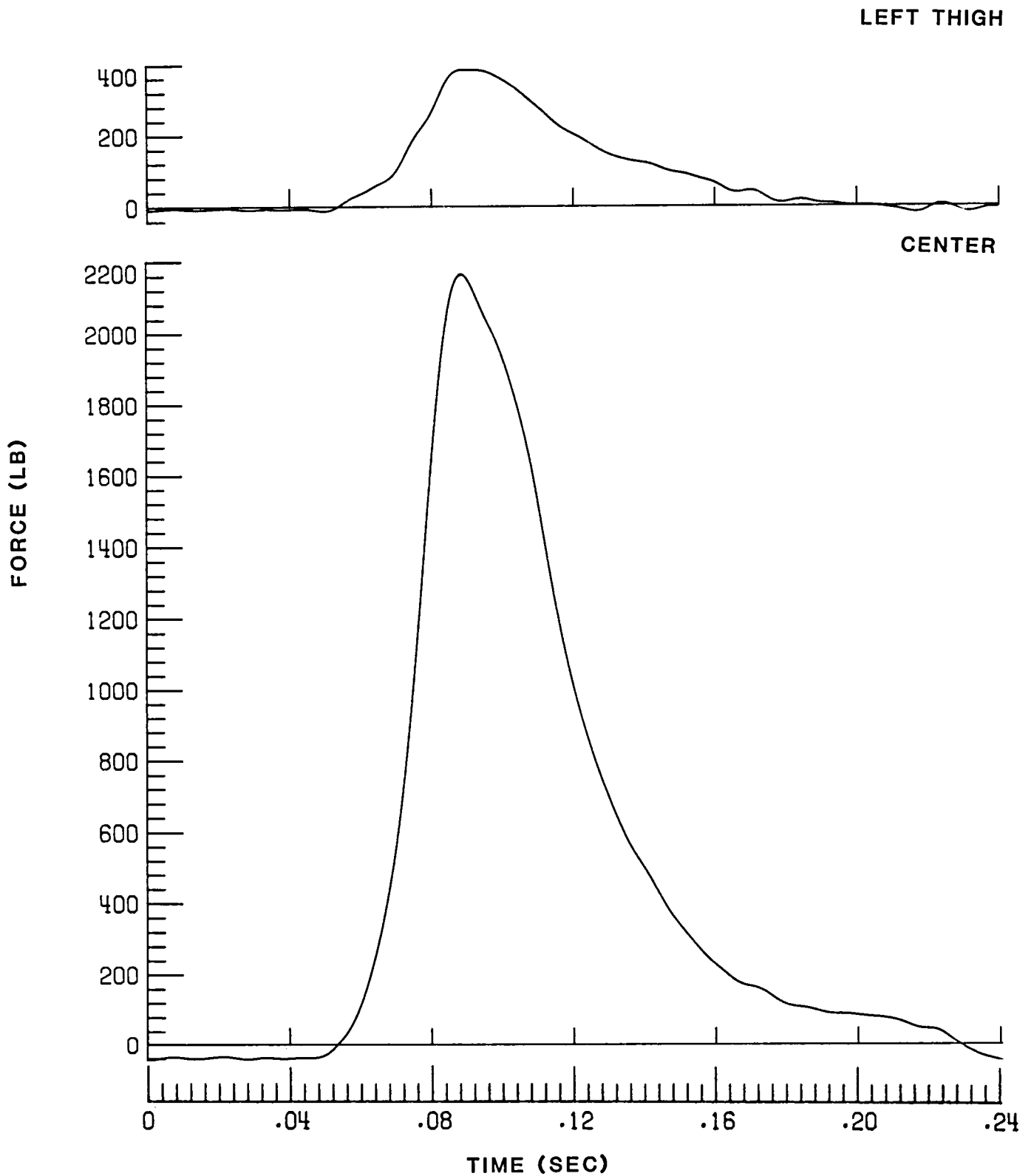


LEFT DUMMY HEAD  
NASA TEST NO. 12

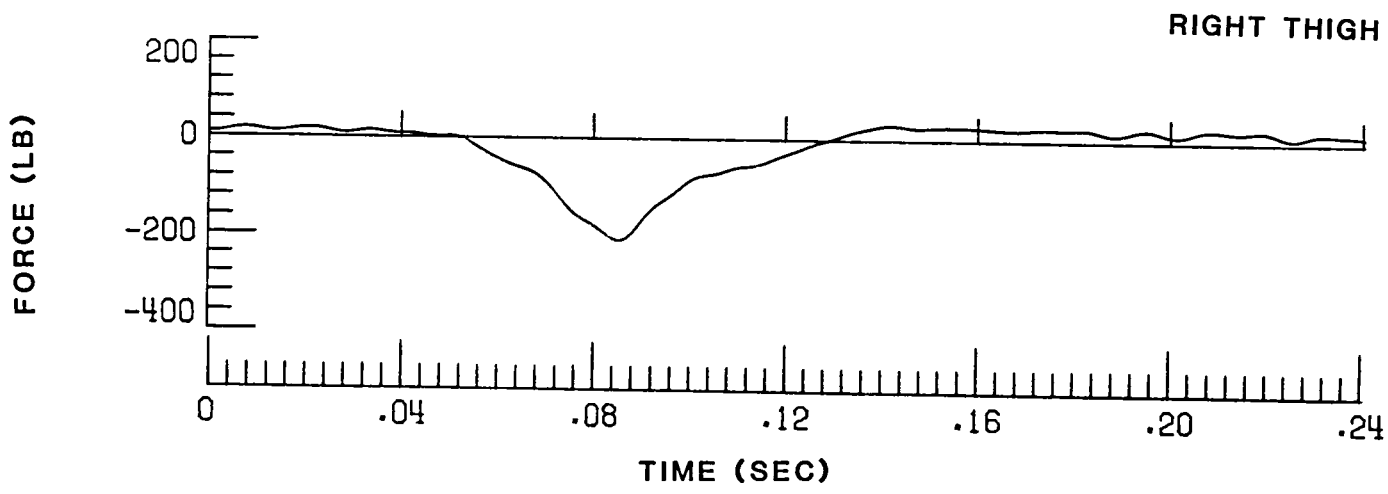


RIGHT DUMMY HEAD  
NASA TEST NO. 12

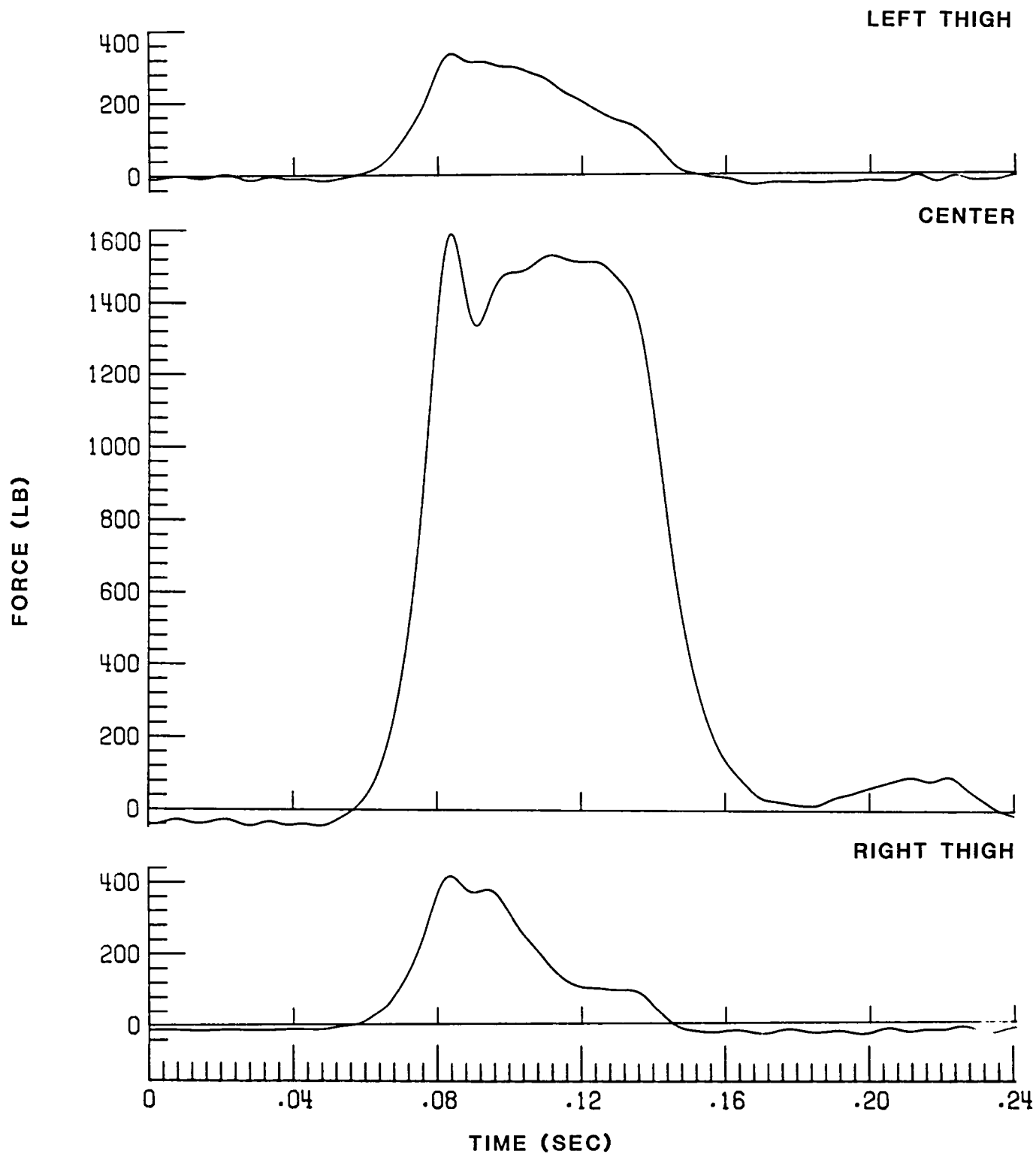




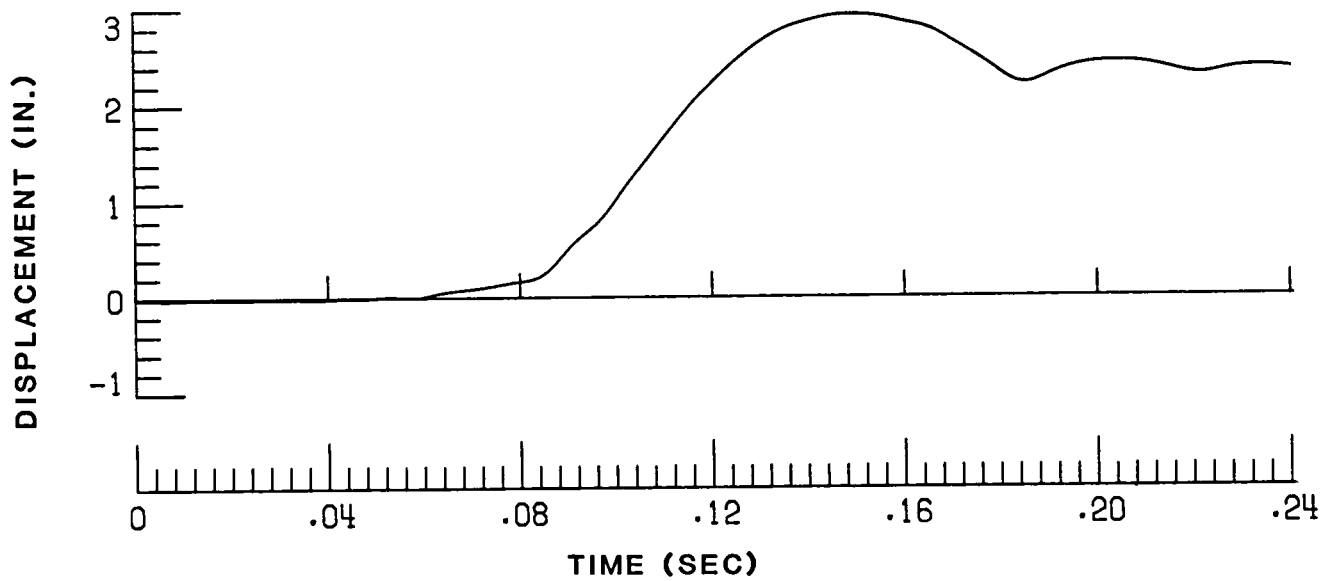
LEFT SEAT PAN LOADS  
NASA TEST NO. 12



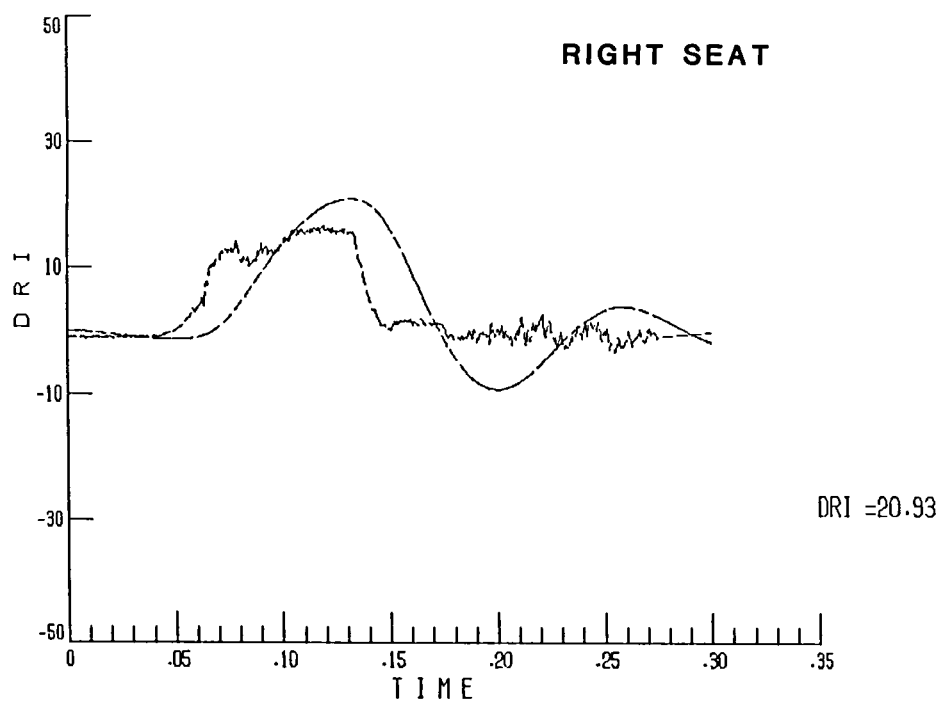
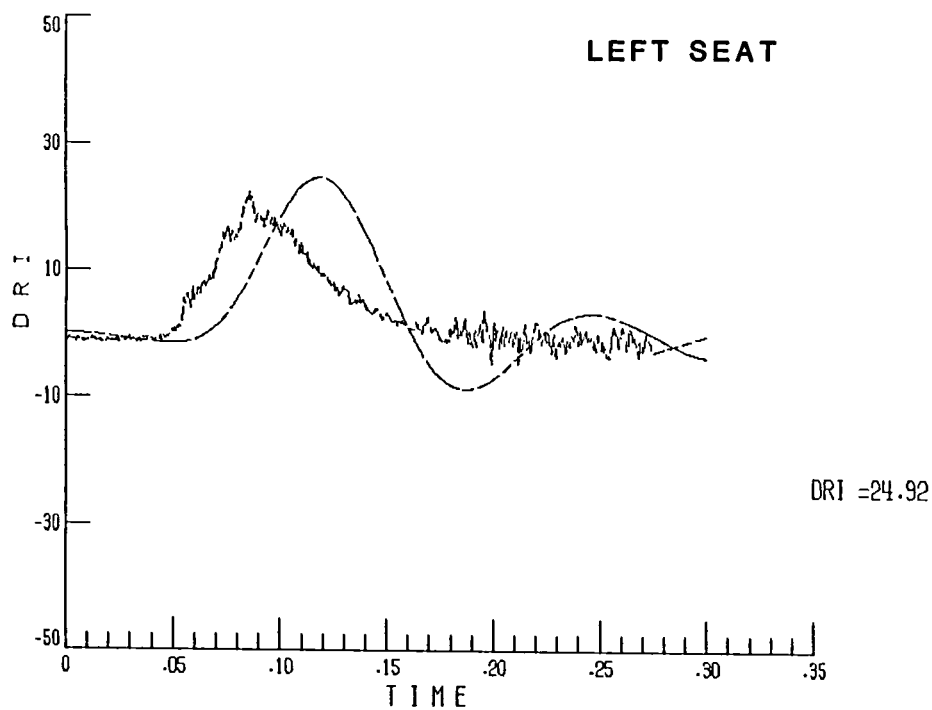
LEFT SEAT PAN LOAD  
NASA TEST NO. 12



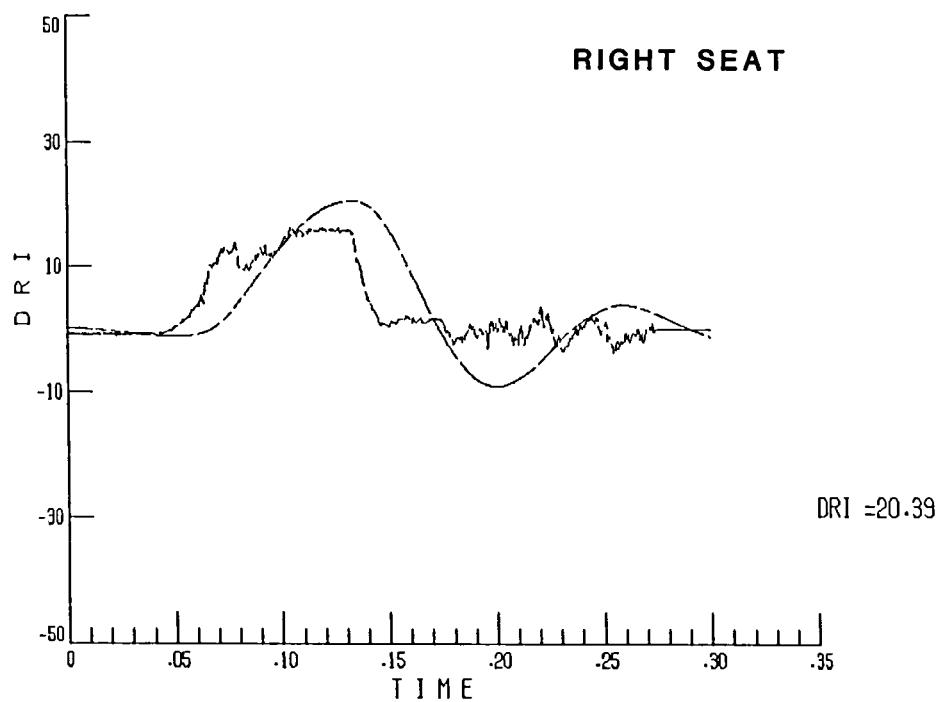
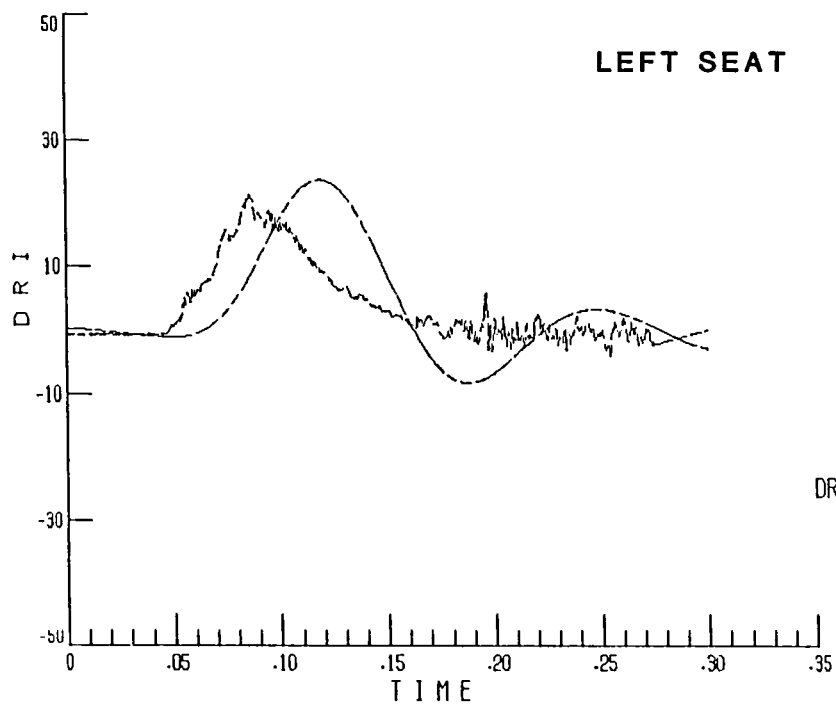
RIGHT SEAT PAN LOADS  
NASA TEST NO. 12



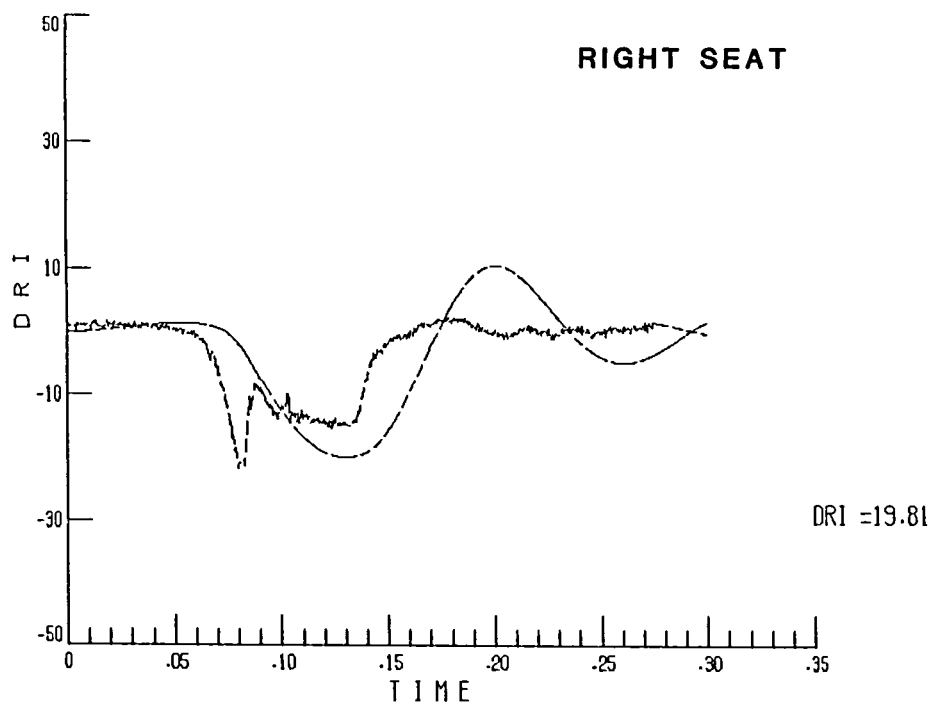
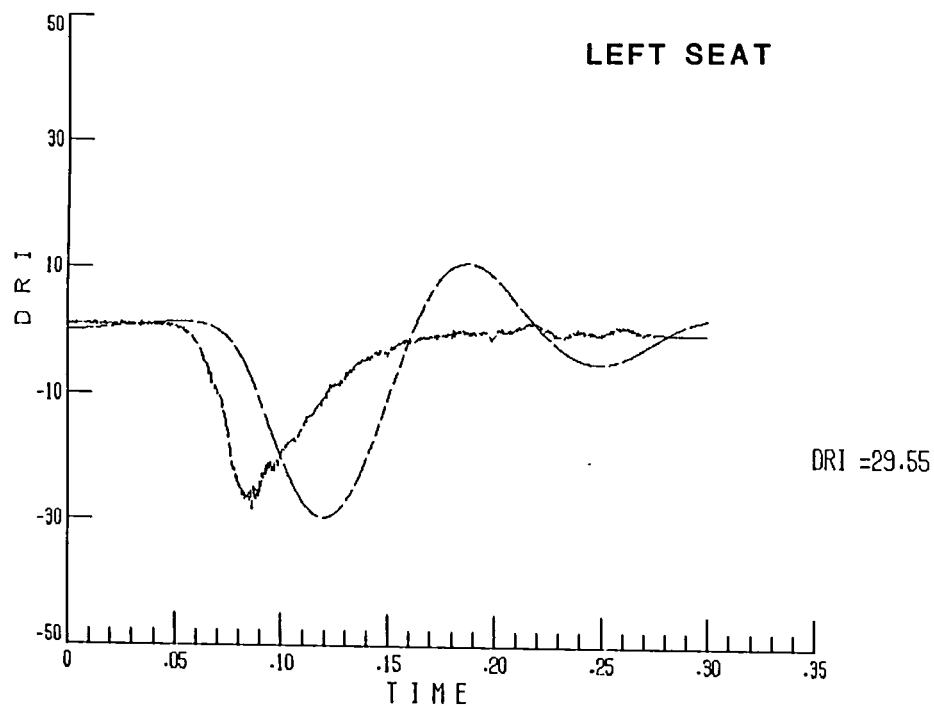
RIGHT SEAT STROKE  
NASA TEST NO. 12



**DYNAMIC RESPONSE INDEX  
SEAT PAN Z-AXIS  
NASA TEST NO. 12**



**DYNAMIC RESPONSE INDEX  
BACK TANGENT LINE  
NASA TEST NO. 12**



**DYNAMIC RESPONSE INDEX  
DUMMY PELVIS Z-AXIS  
NASA TEST NO. 12**

14.0 NASA (IBAHRS) NO. 17

HORIZONTAL VELOCITY: 34 ft/sec

VERTICAL VELOCITY: 32 ft/sec

CREW MODULE PITCH:  $2^0$

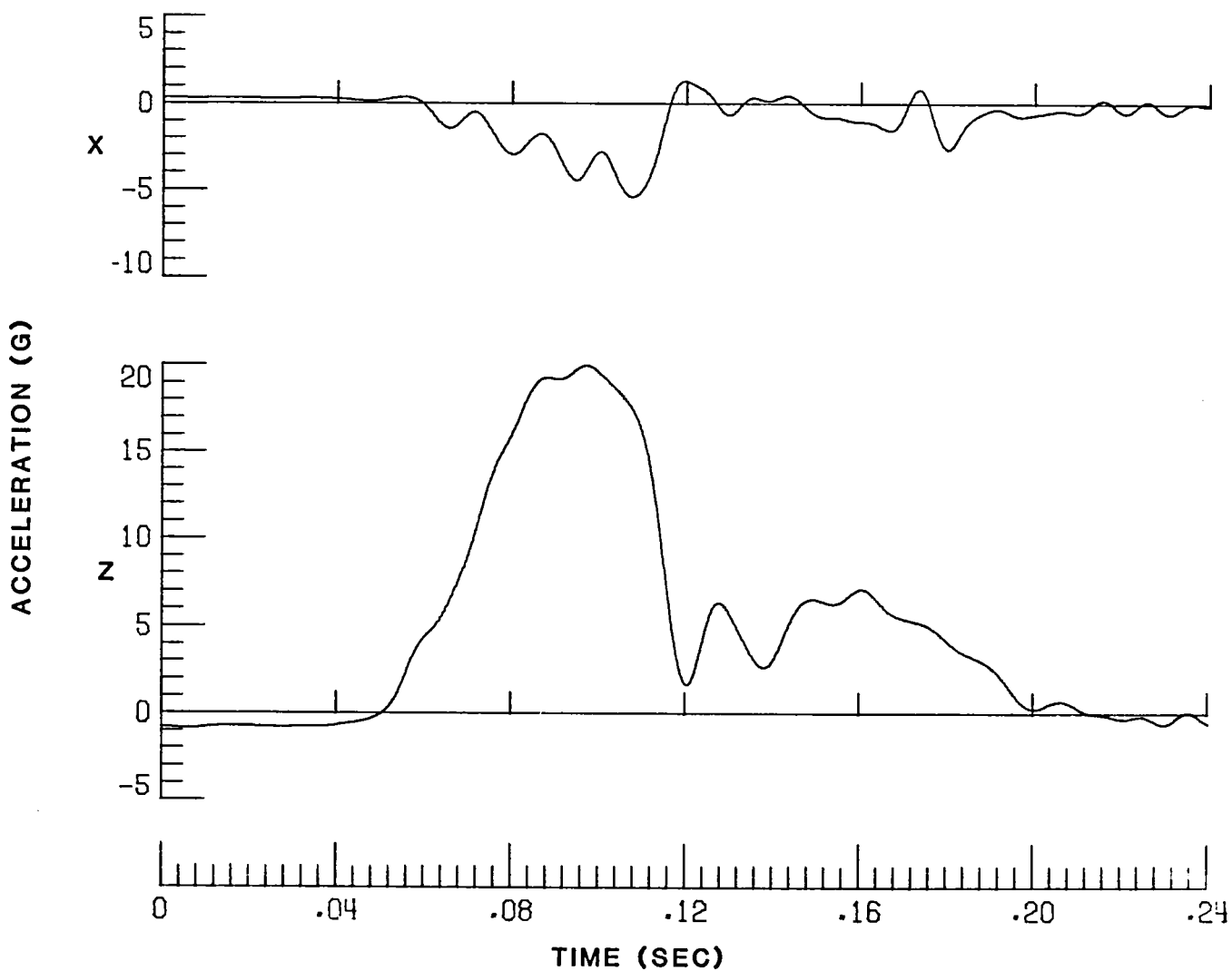
CREW MODULE ROLL:  $0^0$

CREW MODULE YAW:  $0^0$

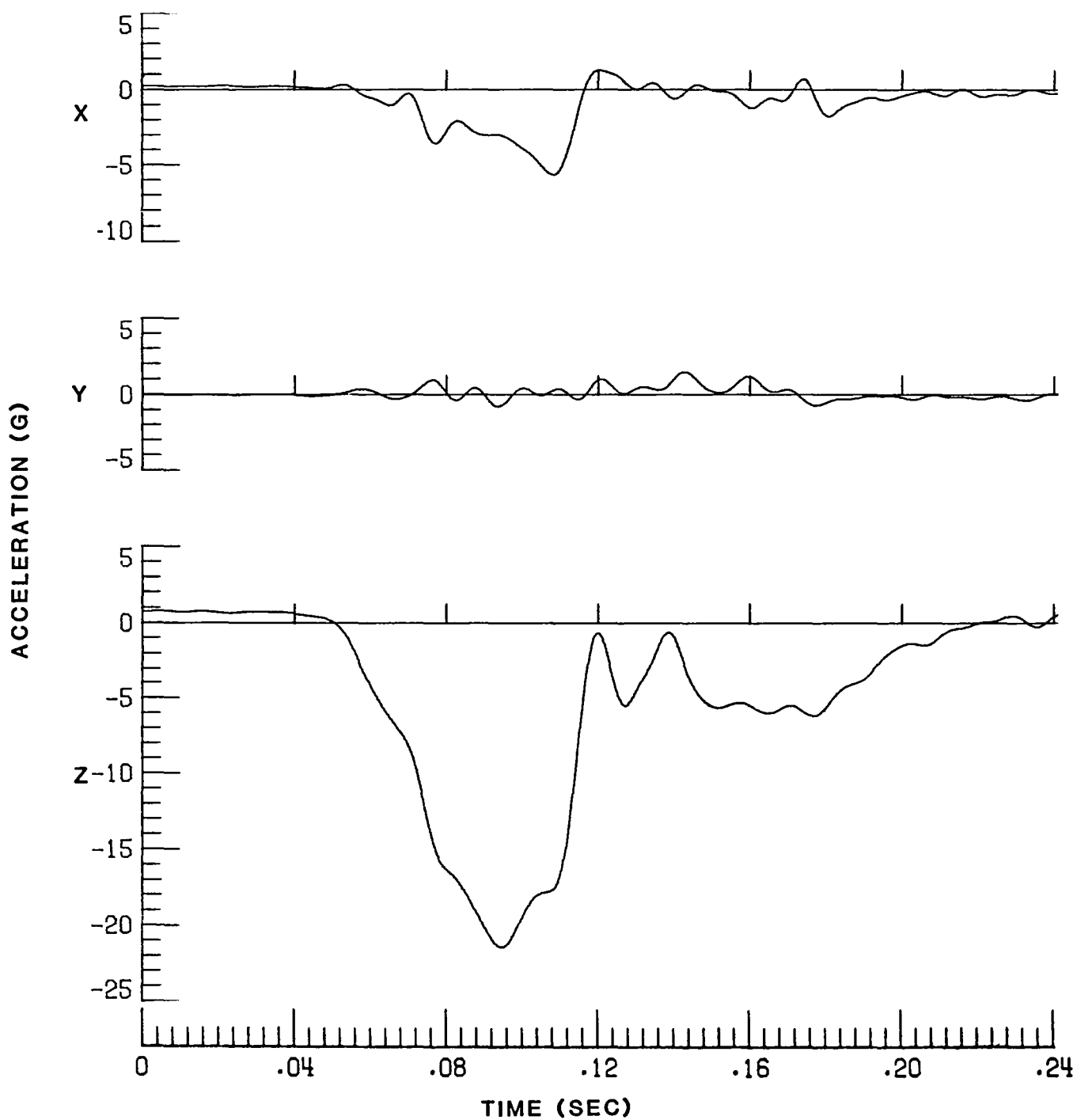
LEFT SEAT: F-111 Operational

RIGHT SEAT: Energy Absorbing

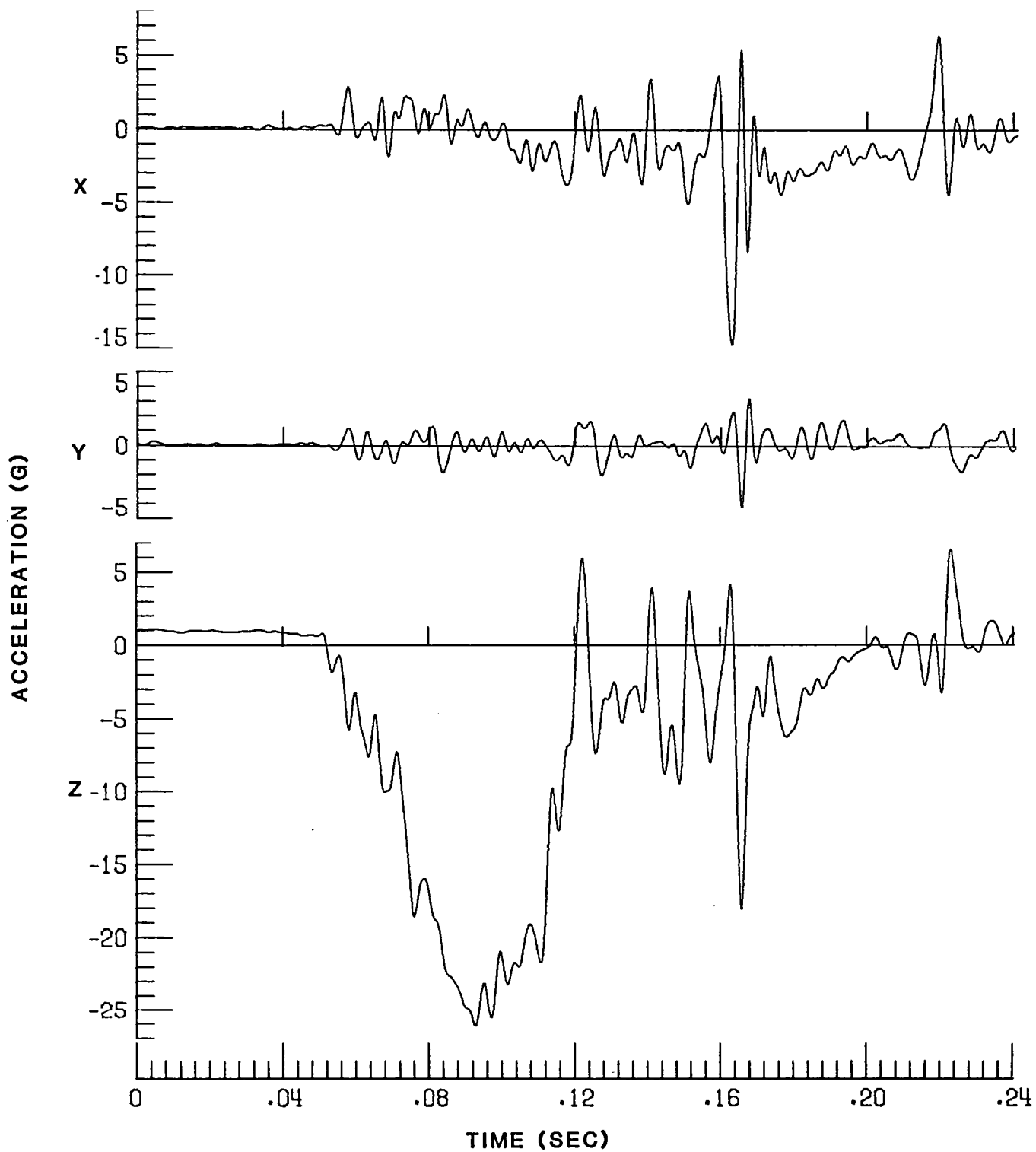




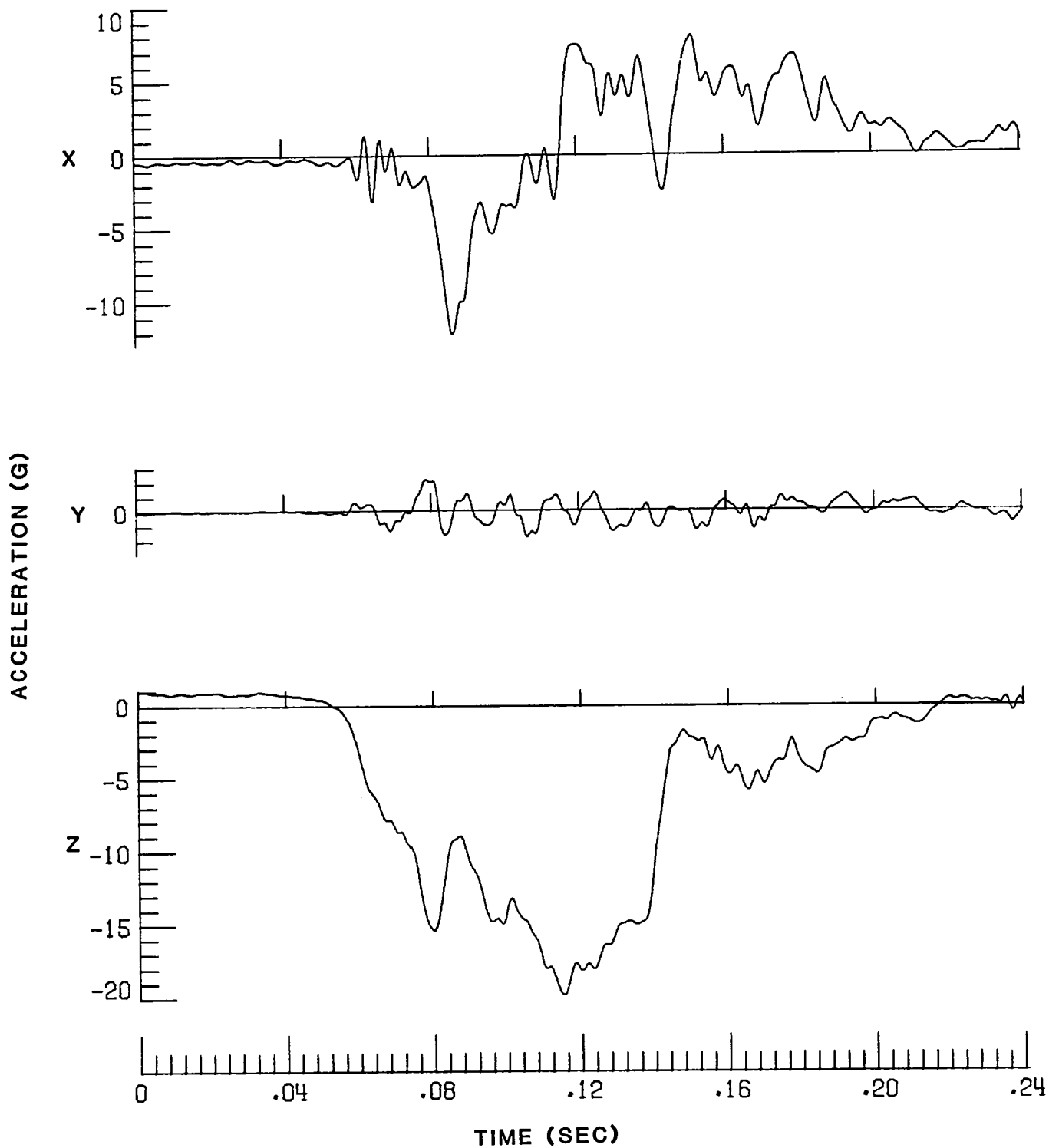
LEFT BULKHEAD  
NASA TEST NO. 17



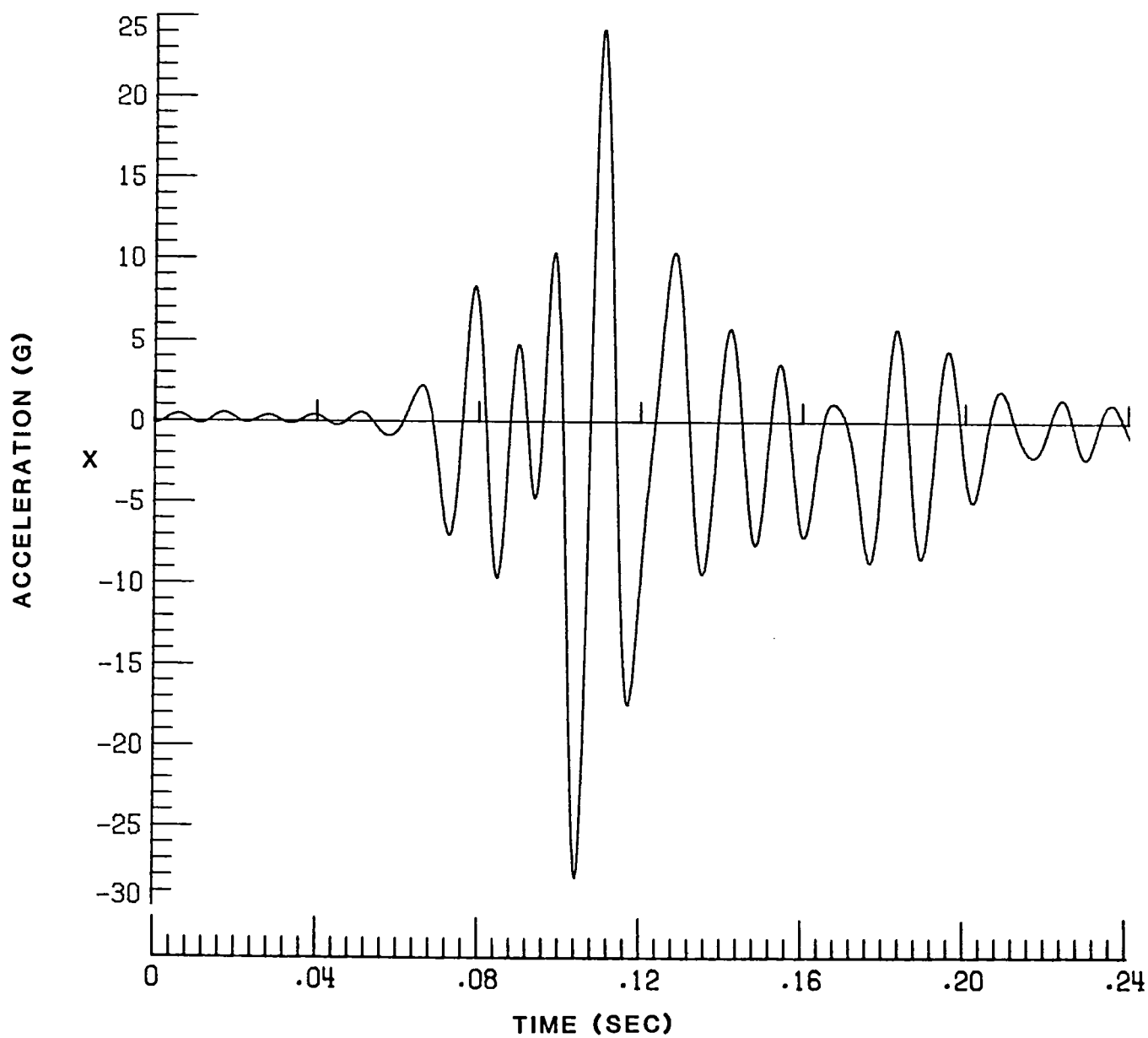
RIGHT BULKHEAD  
NASA TEST NO. 17



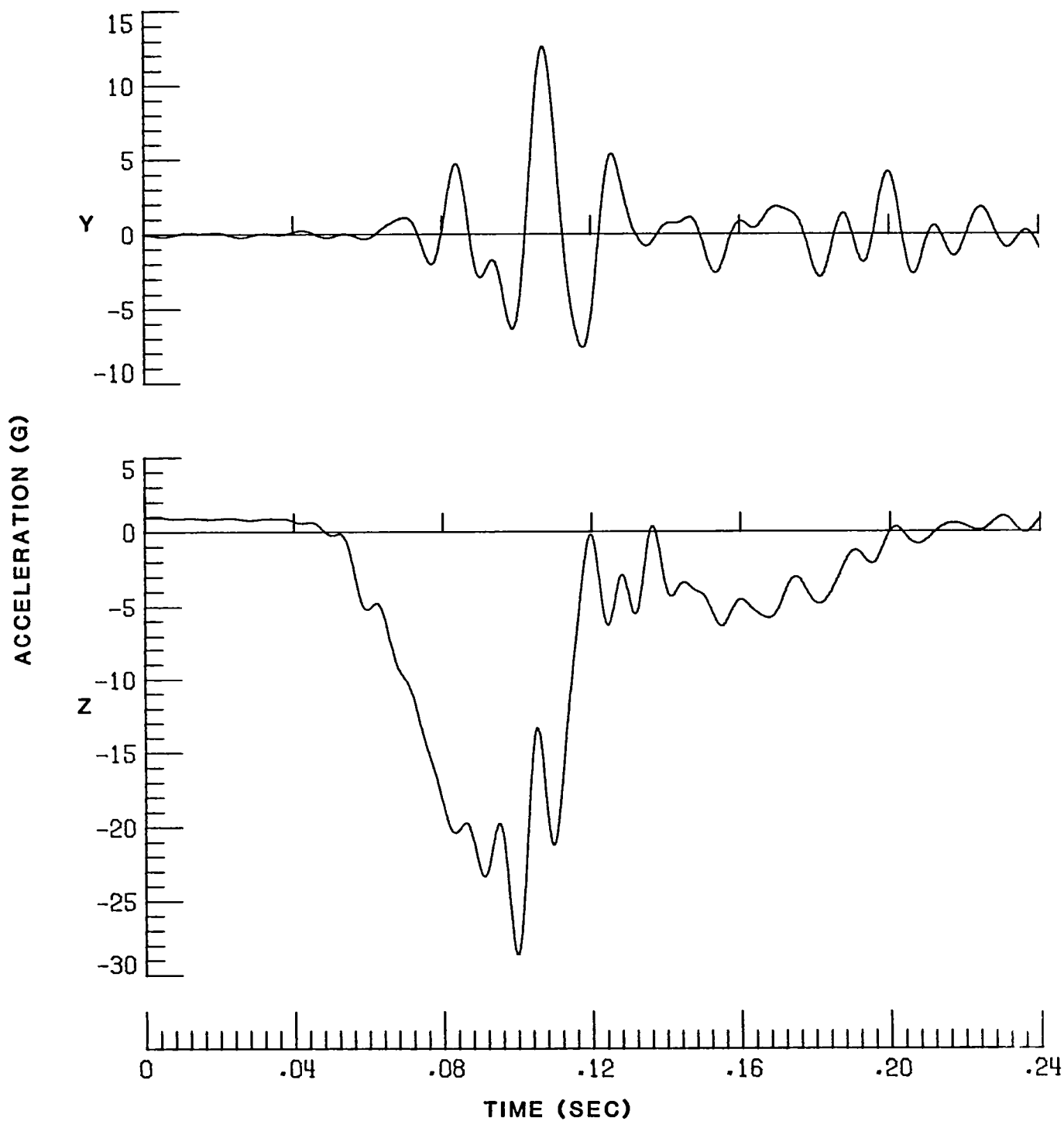
LEFT SEAT PAN  
NASA TEST NO. 17



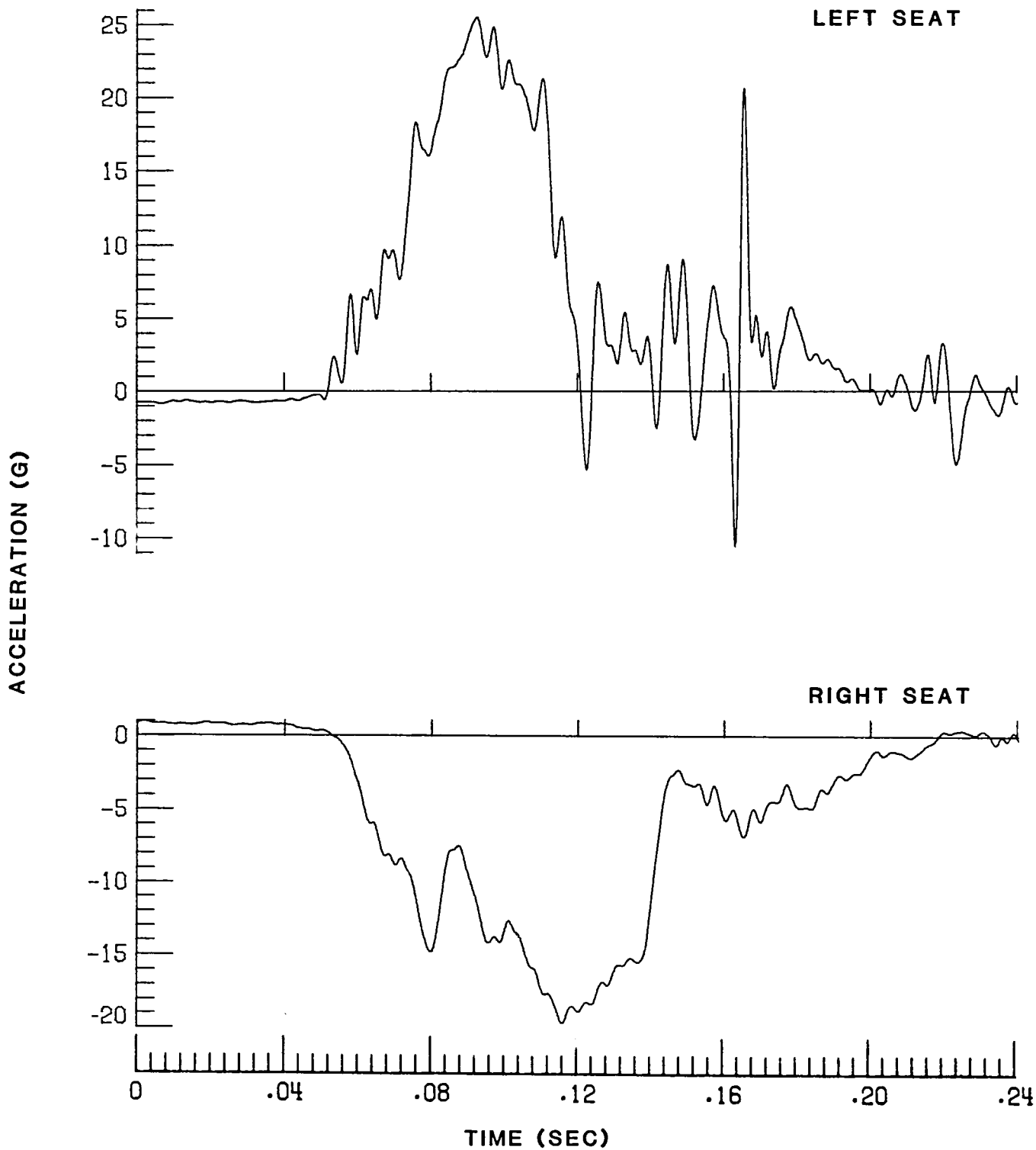
RIGHT SEAT PAN  
NASA TEST NO. 17



CAPSULE C.G.  
NASA TEST NO. 17

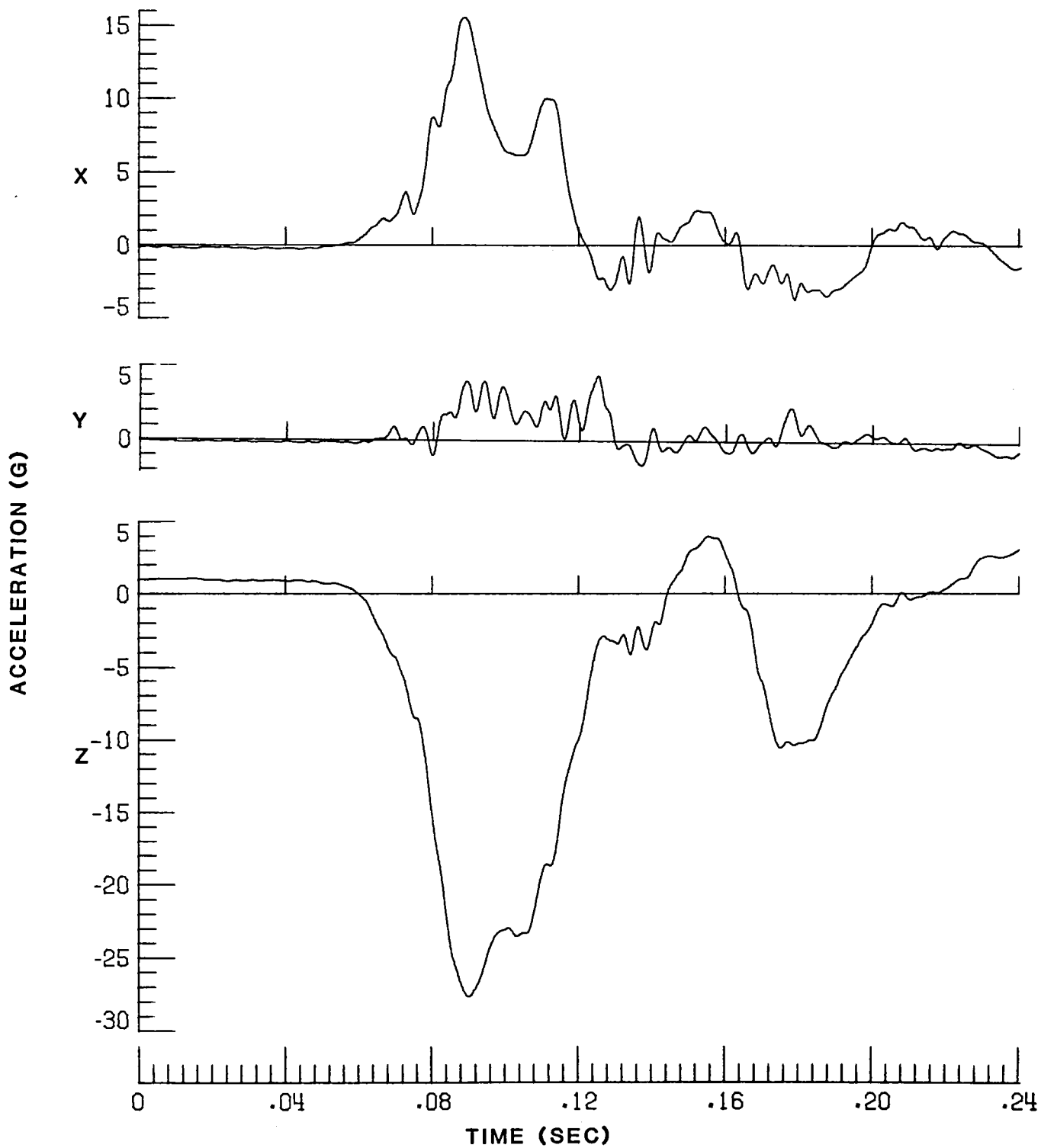


CAPSULE C.G.  
NASA TEST NO. 17



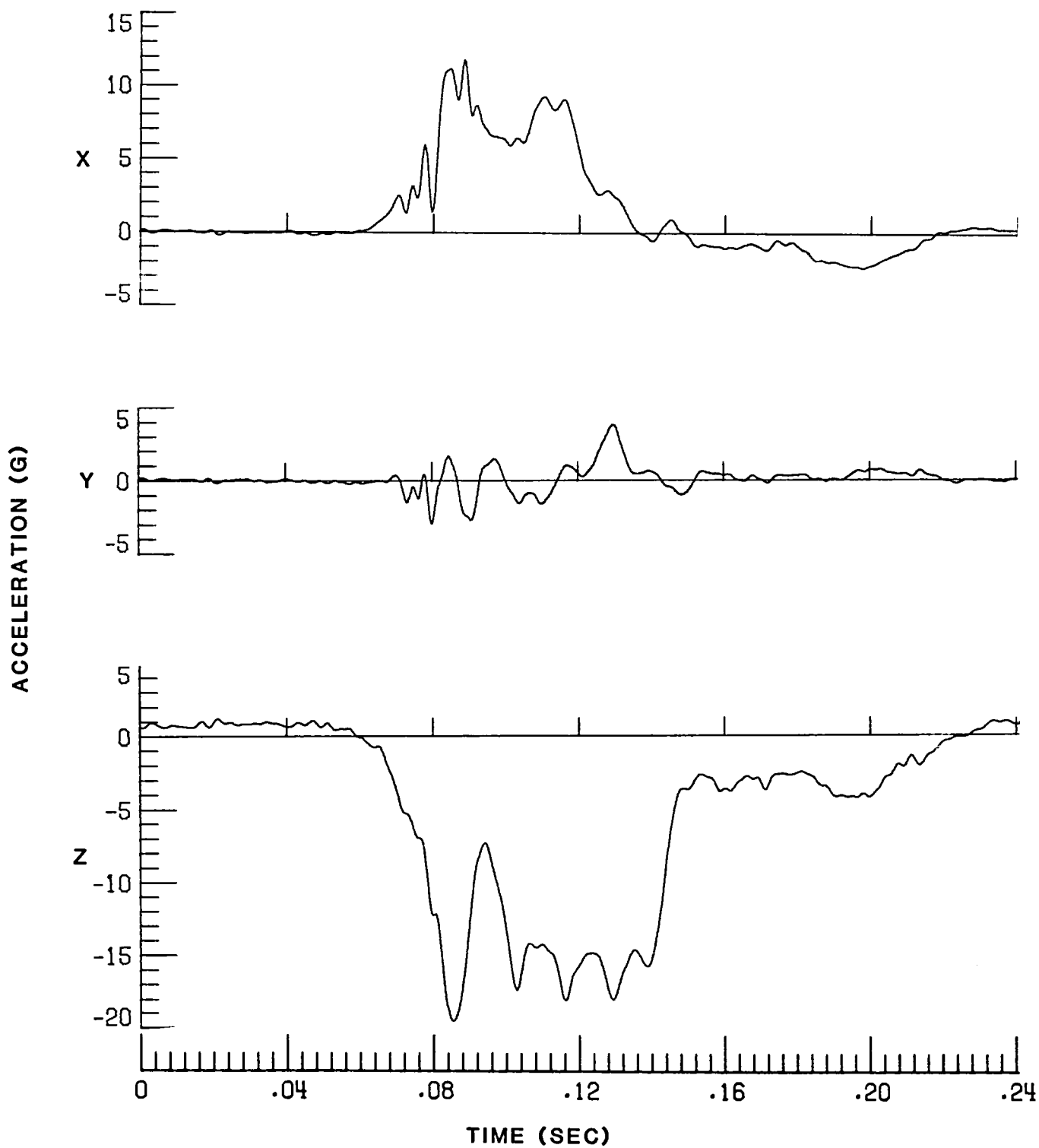
SEAT PAN ALONG BACK TANGENT LINE

NASA TEST NO. 17

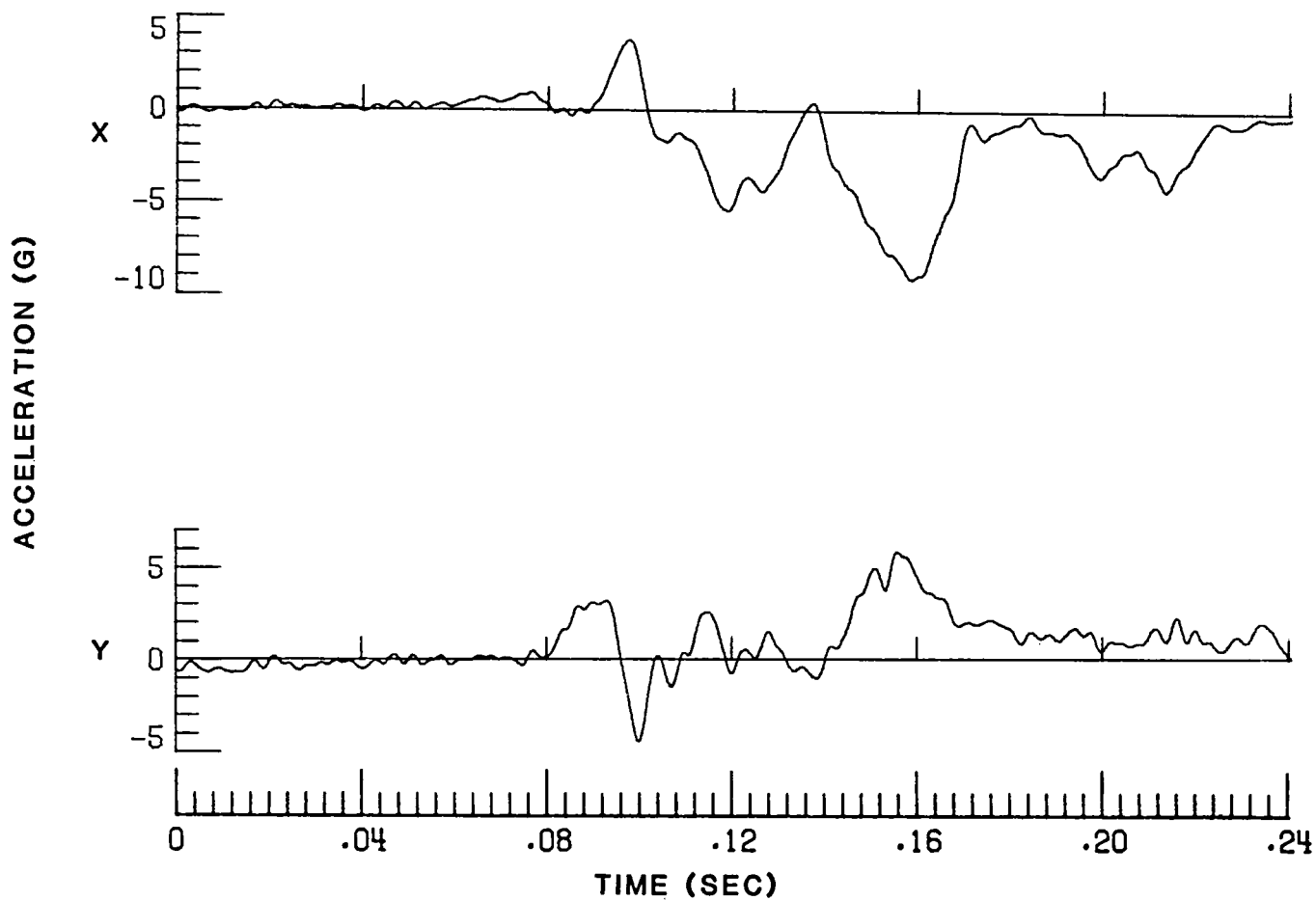


LEFT DUMMY PELVIS  
NASA TEST NO. 17

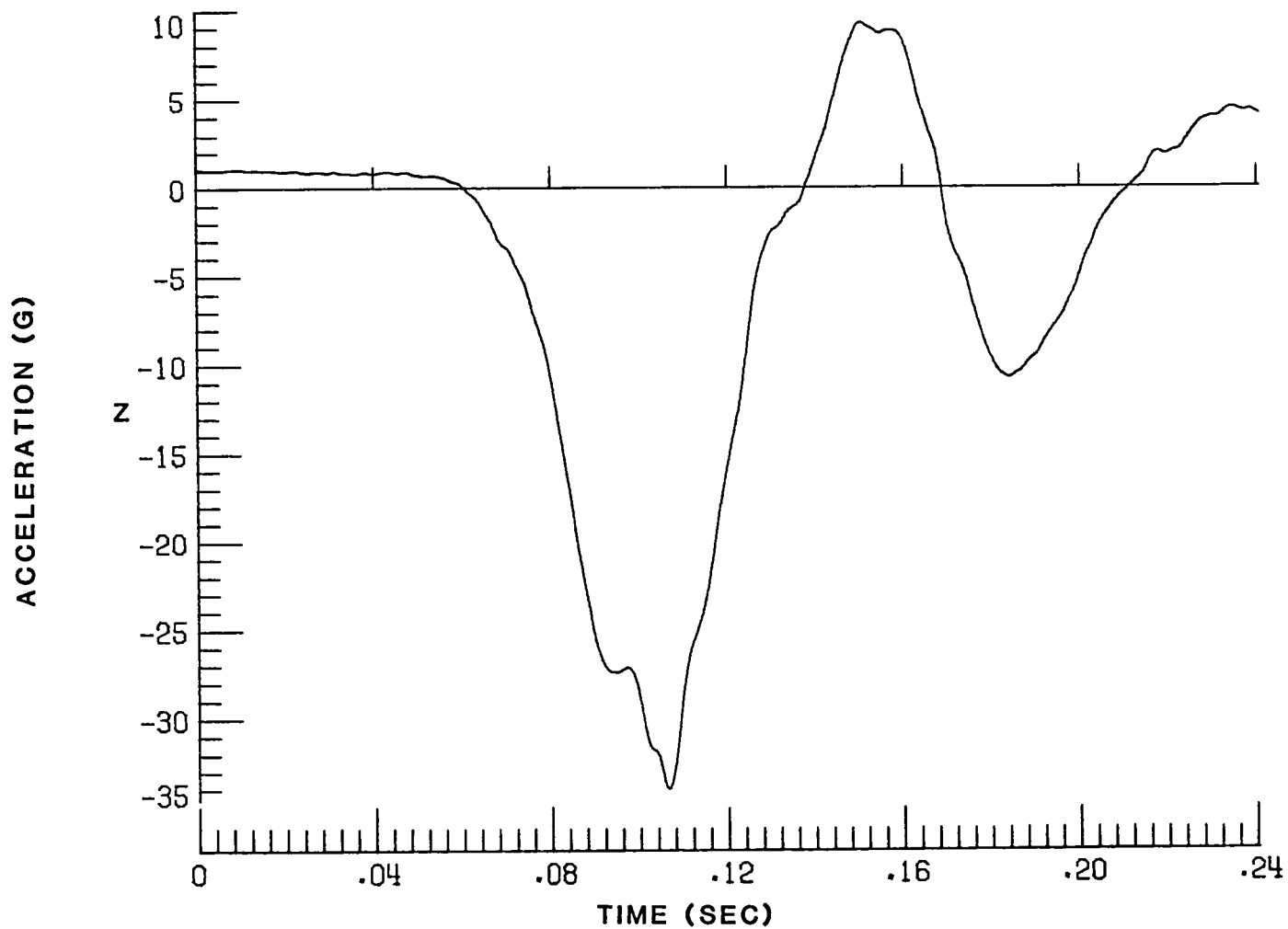




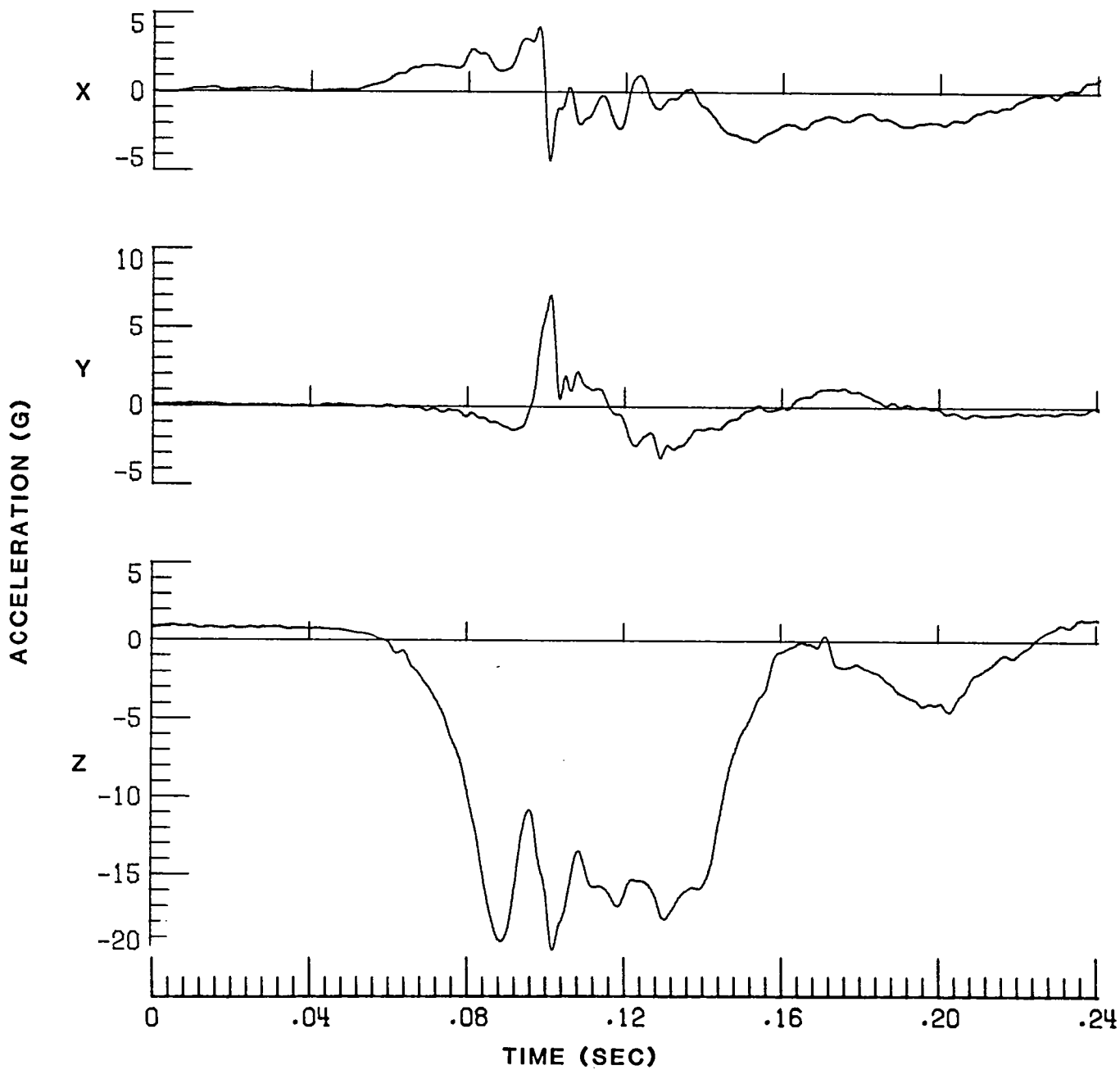
RIGHT DUMMY PELVIS  
NASA TEST NO. 17



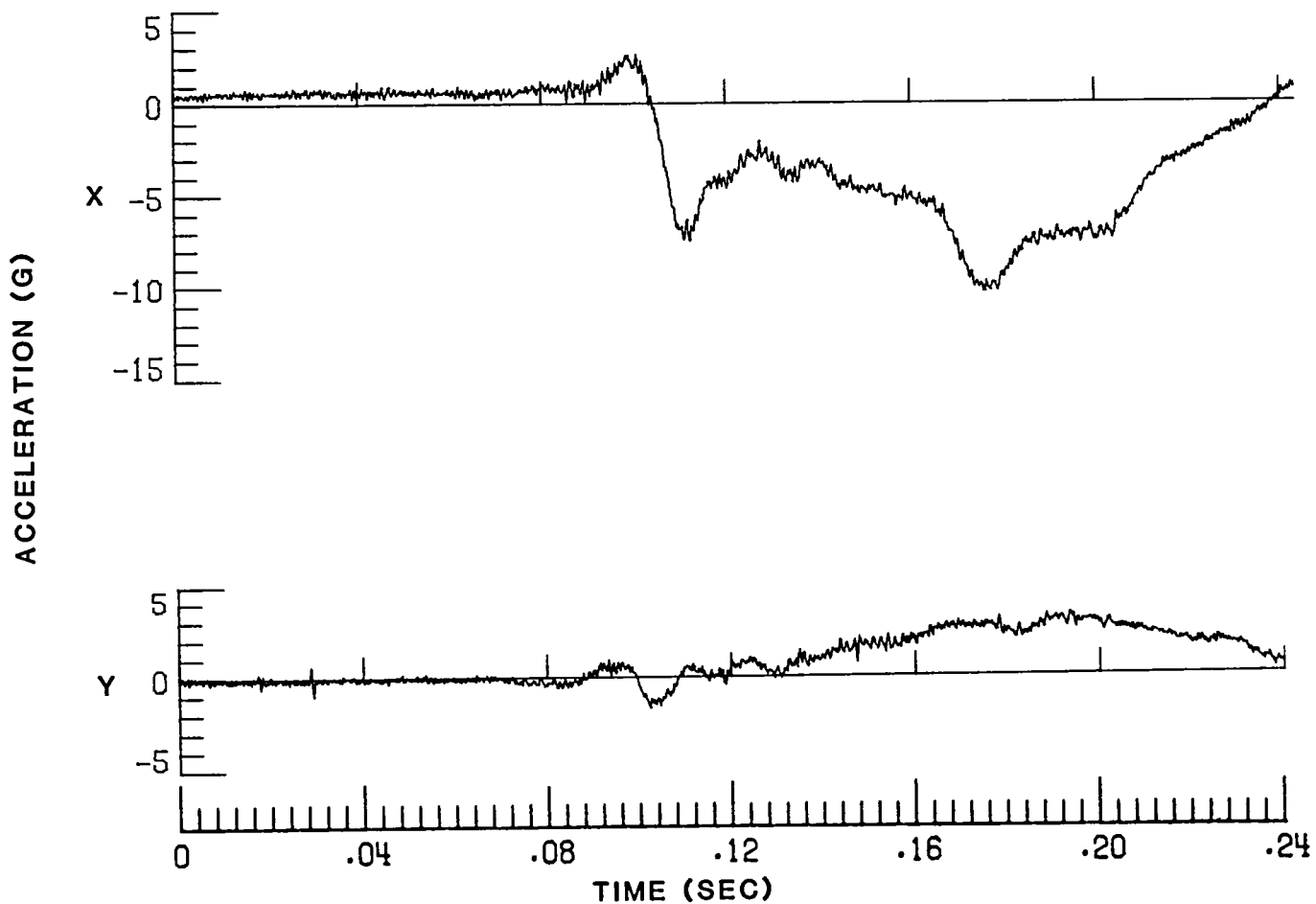
LEFT DUMMY CHEST  
NASA TEST NO. 17



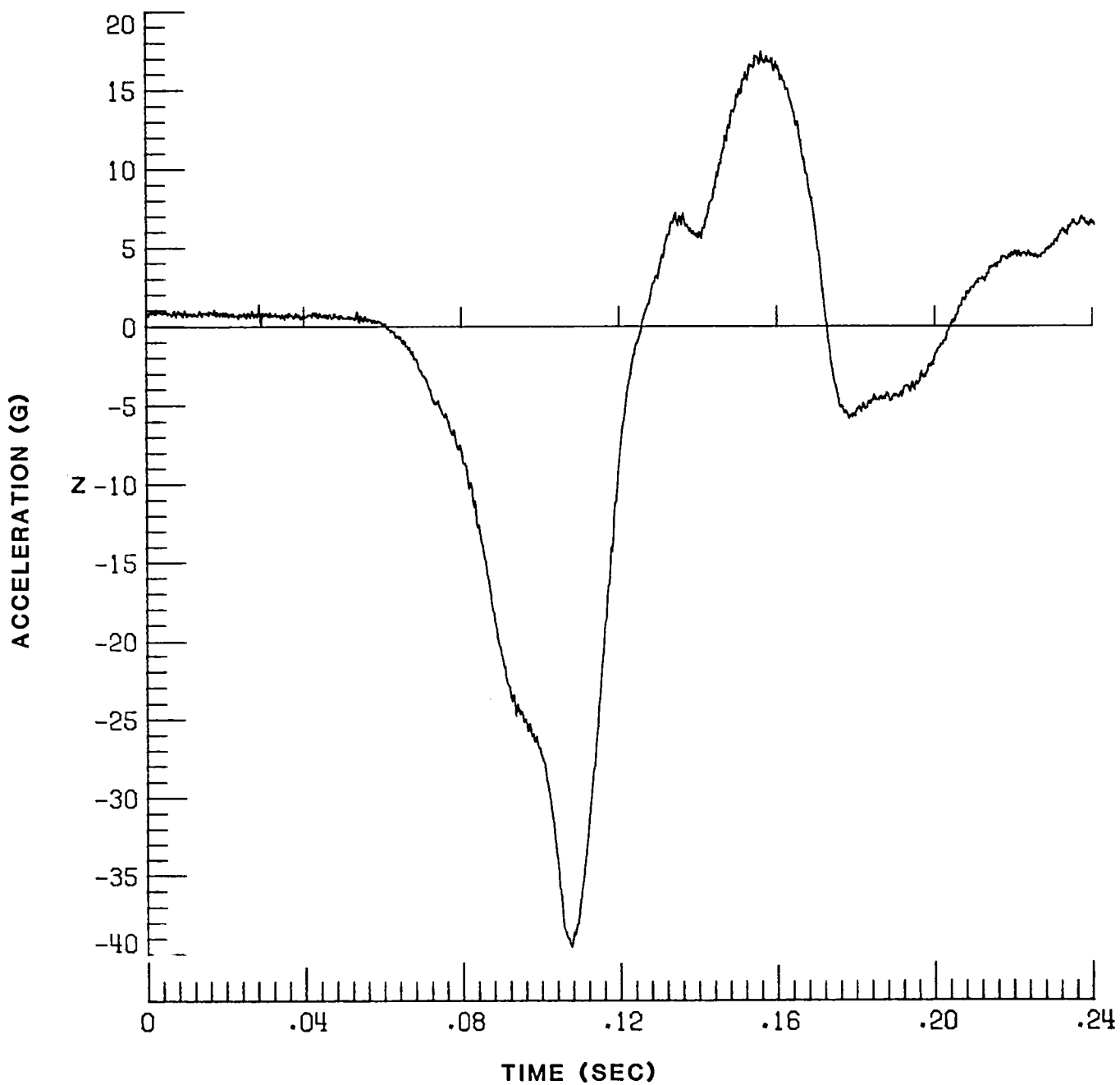
LEFT DUMMY CHEST  
NASA TEST NO. 17



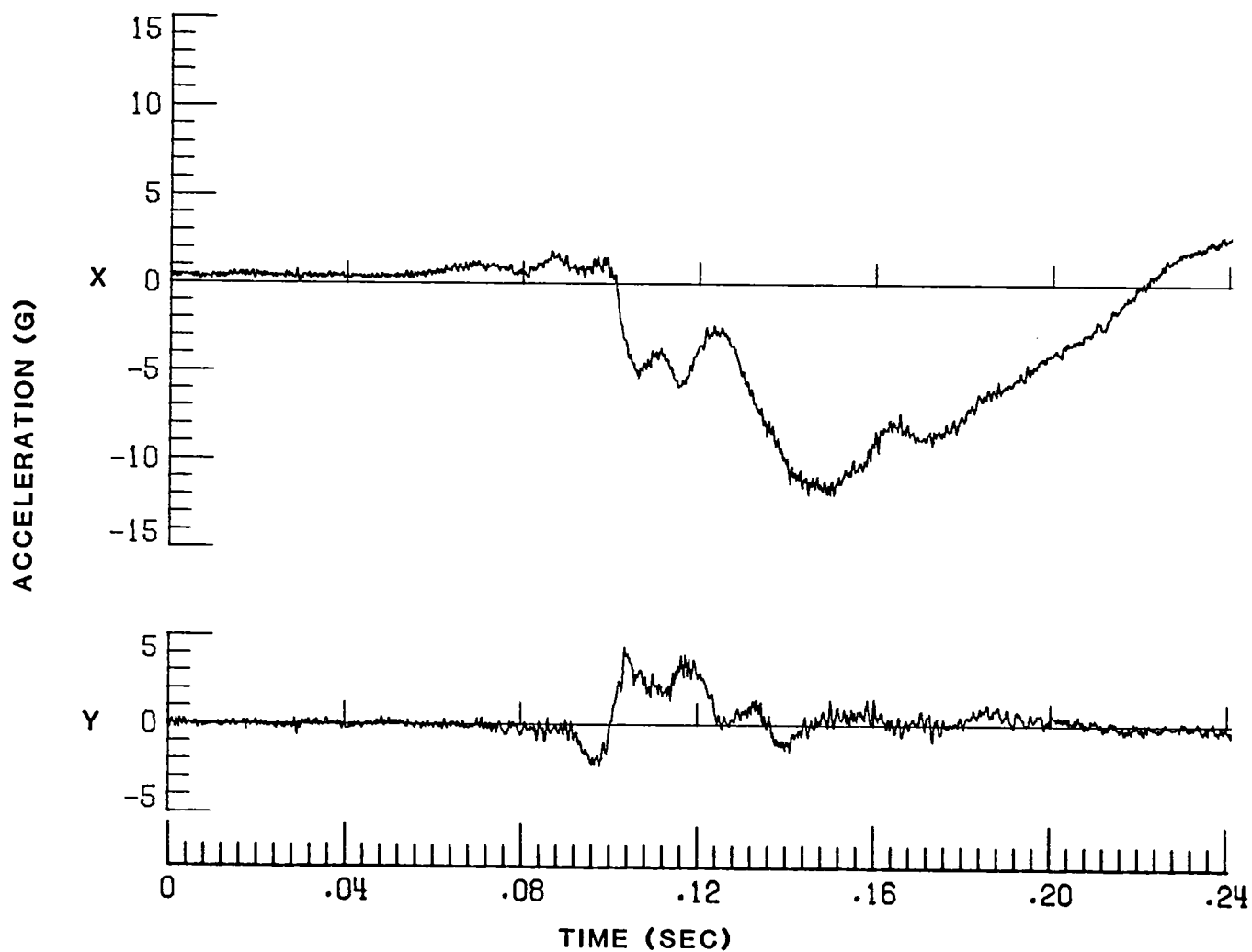
RIGHT DUMMY CHEST  
NASA TEST NO. 17



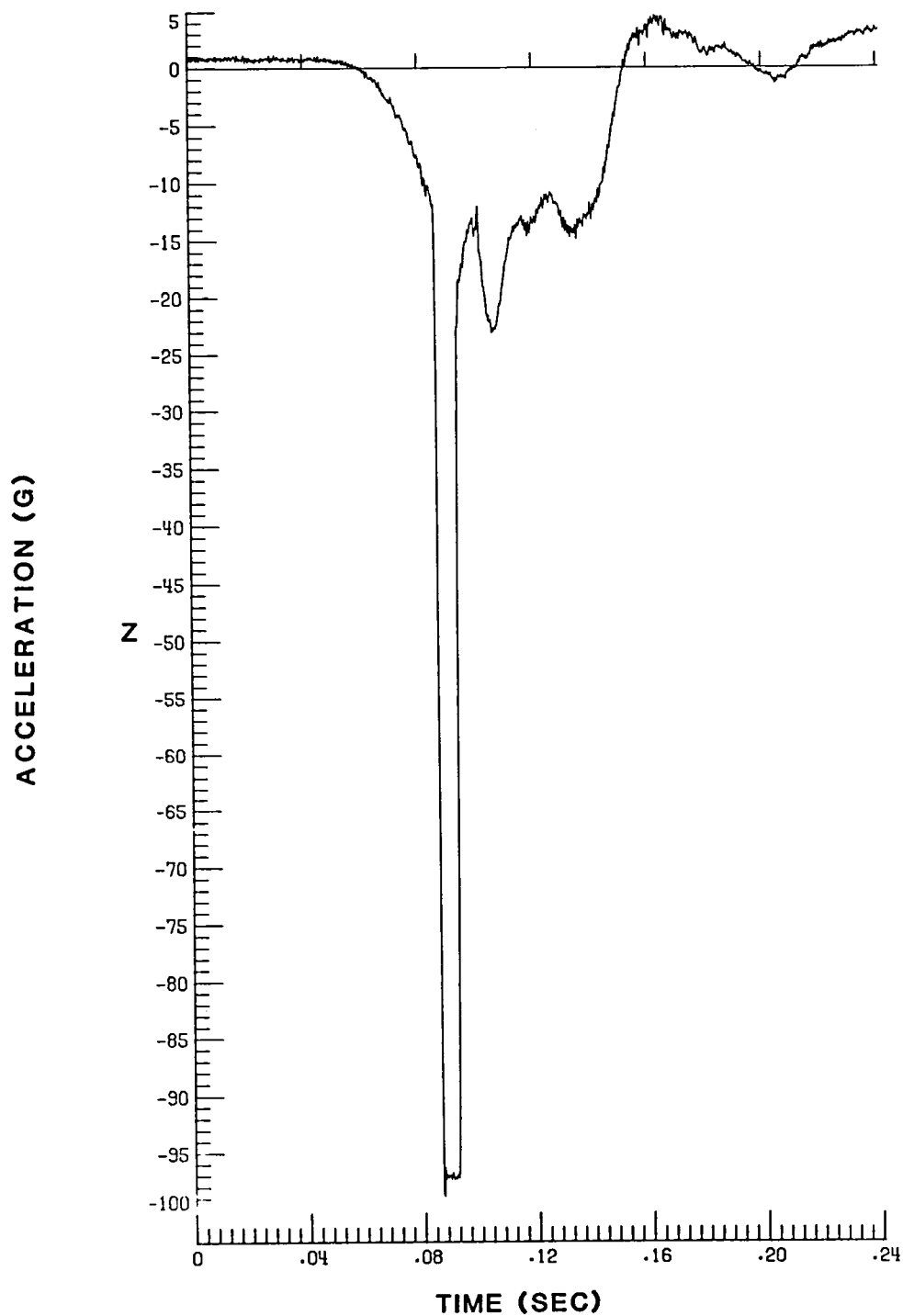
LEFT DUMMY HEAD  
NASA TEST NO. 17



LEFT DUMMY HEAD  
NASA TEST NO. 17

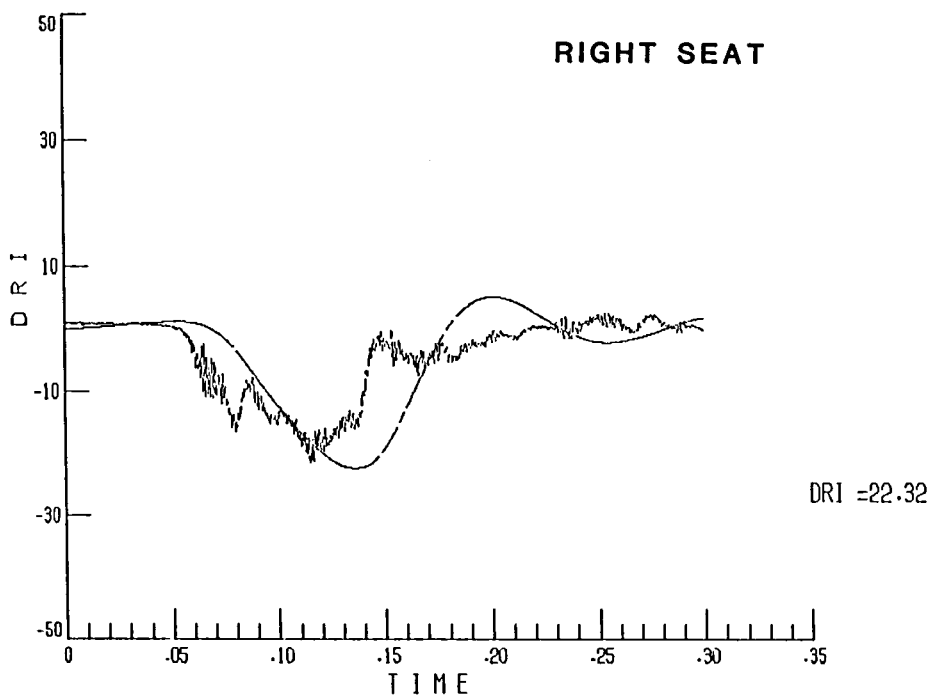
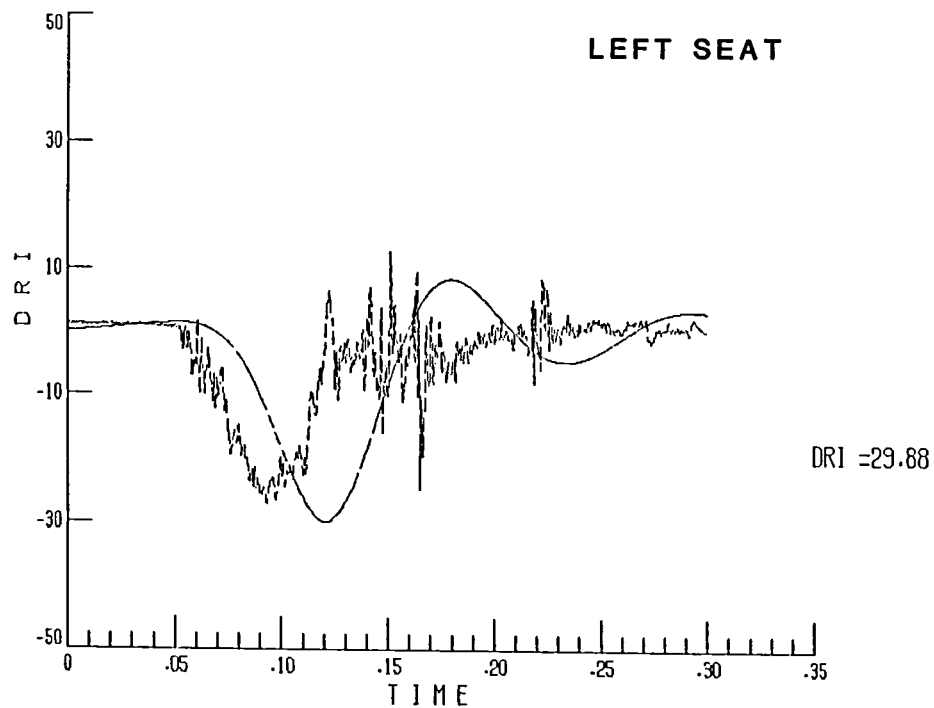


RIGHT DUMMY HEAD  
NASA TEST NO. 17

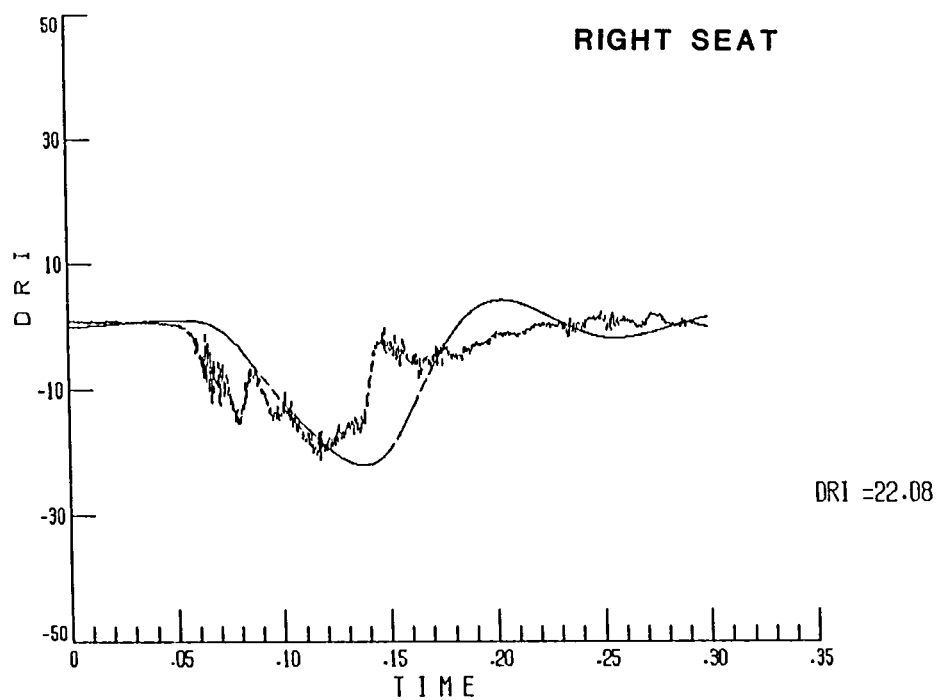
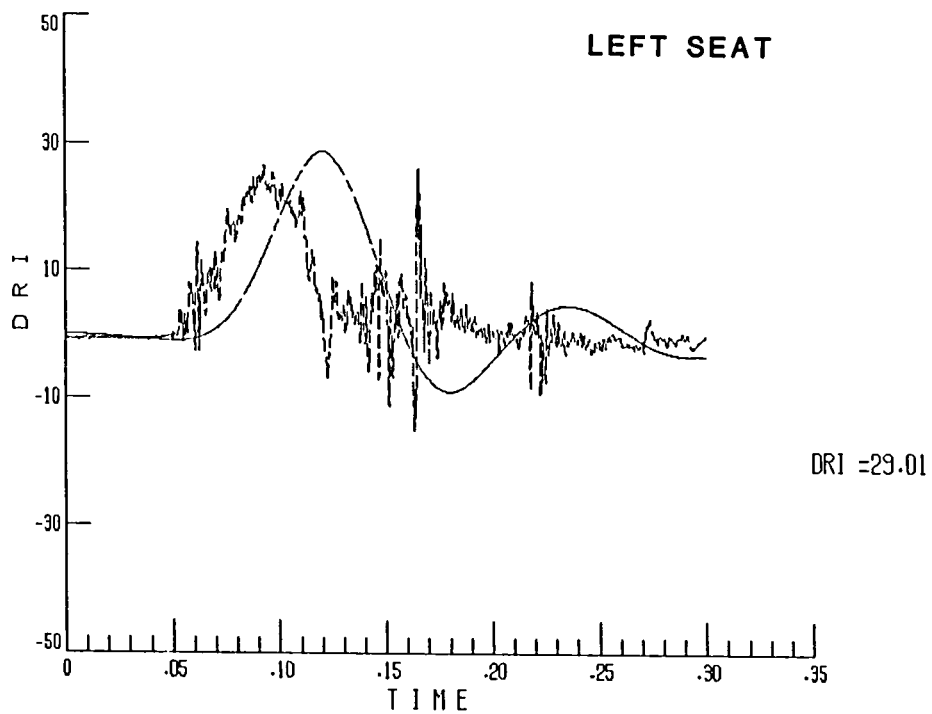


**RIGHT DUMMY HEAD**  
**NASA TEST NO. 17**





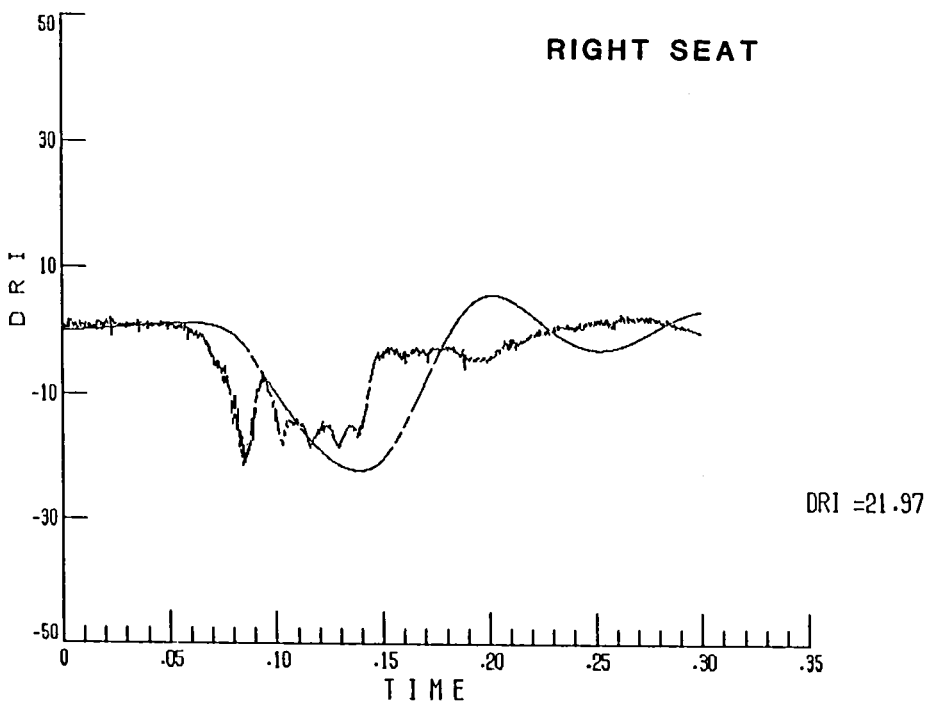
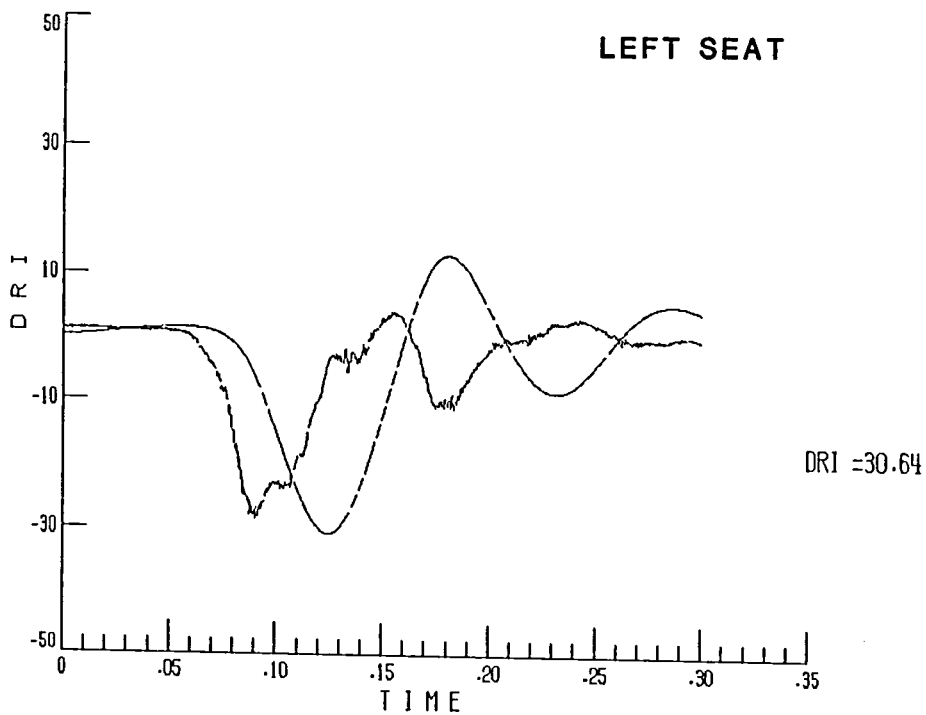
**DYNAMIC RESPONSE INDEX  
SEAT PAN Z-AXIS  
NASA TEST NO. 17**



**DYNAMIC RESPONSE INDEX**

**BACK TANGENT LINE**

**NASA TEST NO. 17**



**DYNAMIC RESPONSE INDEX  
DUMMY PELVIS Z-AXIS  
NASA TEST NO. 17**





1. Report No. NASA CR-3918		2. Government Accession No.		3. Recipient's Catalog No.	
4. Title and Subtitle Design and Testing of an Energy-Absorbing Crewseat for the F/FB-111 Aircraft. Volume III—Data From Crew Module Testing				5. Report Date August 1985	
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16. Abstract  Over the past years, several papers and reports have documented the unacceptably high injury rate during the escape sequence (including the ejection and ground impact) of the crew module for F/FB-111 aircraft. This report documents a program to determine if the injury potential could be reduced by replacing the existing crewseats with energy-absorbing crewseats. An energy-absorbing test seat was designed using much of the existing seat hardware. An extensive dynamic seat test series, designed to duplicate various crew module ground impact conditions, was conducted at a sled test facility. Comparative tests with operational F-111 crewseats were also conducted. After successful dynamic testing of the seat, more testing was conducted with the seats mounted in an F-111 crew module. Both swing tests and vertical drop tests were conducted. The vertical drop tests were used to obtain comparative data between the energy-absorbing and operational seats. Volume I describes the energy-absorbing test seat and testing conducted, and evaluates the data from both test series. Volume II presents the data obtained during the seat test series, while Volume III presents the data from the crew module test series.					
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